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IMPLEMENTATION OF AREA NAVIGATION IN THE NATIONAL AIRSPACE SYSTEM: an assessment of RNAV Task Force Concepts and Payoffs

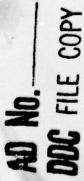


Final Report December 1976

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PREFACE

The Systems Research and Development Service of the Federal Aviation Administration has undertaken a program to assess the technical and economic impact of Area Navigation on the ATC System and the users of the National Airspace System. This work was performed under the RNAV Technical Support Contract to Systems Control, Inc. (Contract No. DOT-FA-72-WA-3098 Task Order No. 008). The work was performed by the Champlain Technology Industries Division (CTI) and the Aeronautical and Marine Systems Division (A&M) of Systems Control, Inc. (Vt), a subsidiary of Systems Control, Inc.

The FAA Technical Monitor for this work was D. M. Brandewie and the Technical Support Program Manager was D. W. Richardson of CTI. The Project Manager and principal author of this document was W. H. Clark of CTI.

This document is a final report containing the results of economic impact studies and the description of an RNAV implementation concept.

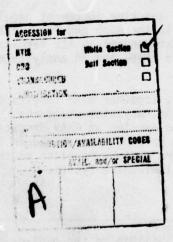
Particular acknowledgement to the members of the technical staffs of CTI and A&M is given to:

W. H. Clark (Project Management, Study Methodology, VNAV Analysis, User Benefit Analysis, Route Length Analysis, Operational Concept)

E. H. Bolz (Terminal Area Analysis, 4D Analysis, User Cost and Requirements Analysis, Slant Range Analysis, Terminal VORTAC Analysis, Automation Analysis)

H. L. Solomon (Low Altitude Route Length Analysis, Weather Route Simulation Analysis, High Altitude VORTAC Analysis)

A. R. Stephenson



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IMPLEMENTATION OF AREA NAVIGATION IN THE NATIONAL AIRSPACE SYSTEM:

An Assessment of RNAV Task Force Concepts and Payoffs

1.0 EXECUTIVE SUMMARY

1.1 PROGRAM SCOPE AND OBJECTIVES

This report contains the results of a three year study effort to define the cost and operational impact, and to assess the economic benefits which are expected to accrue to the ATC system and the various users of the National Airspace System as a result of the implementation of Area Navigation. The results of this study include benefits and costs for the complete spectrum of primary users of the National Airspace System and for the ATC System, and the definition of a system concept evolved from the FAA/Industry RNAV Task Force [1] concept which will allow a timely evolution to an all-RNAV environment. Although the ATC system and user costs and the ATC system savings derived in this report are based on an RNAV system which uses VORTAC as the position reference, RNAV is considered to be a generic descriptor of a random navigation system which can utilize navigation sensors other than VORTAC, and which will result in similar savings.

Costs and benefits are derived on a per aircraft basis as well as an annual aggregate basis. Annual benefits are shown for both an all-RNAV environment and for the transition period of mixed RNAV/VOR operations. The system concept which is defined includes a phased transition from VOR to RNAV, such as suggested by the Task Force, but the timing of the phases is dependent upon user demand rather than the fixed calendar periods recommended by the Task Force.

In assessing the impact of area navigation, the system design concept to be implemented will determine, to a great extent, the operational impact on both the system and the users and the degree of payoff for each. A series of related studies over the past three years, both completed and ongoing, have been aimed at the solution of the problem areas described in the Task Force Report [1] and the overall evaluation of Task Force RNAV concepts. These studies have examined terminal area design concepts [2], the development of waypoint designation standards [3], enroute design criteria [4], route width requirements [5], the application of 4D RNAV in the terminal area [6], the development of Avionics Standards [7], the quantifying of Flight Technical Error [8], the impact of mixed VOR/RNAV environments [9,10,11], various RNAV system features such as parallel offsets and paralleling in turns [7], and the application of VNAV in the terminal area [2,11]. Previous economic impact studies [12] have examined the terminal area and high altitude operations of trunk and local service carriers, the impact of those operations in a mixed VOR/RNAV environment, and the anticipated long term payoff of an all-RNAV environment. Other studies [1,13], made a preliminary assessment of Navaid requirements for the support of an RNAV and preplanned direct structure as well as terminal area RNAV designs. The impact on controller communications workload was also determined [9,11,12].

The major objectives of this report are as follows:

- 1) Expand preliminary analyses of ATC system impact to include the primary aspects of ATC system operation.
- Expand the preliminary user payoff analysis to include a broader spectrum of user groups, route structures, and types of operation, and to include latest available information.
- 3) Summarize the results of other RNAV system studies as they relate to the elements of system design which bear on the economic and operational impact on the users and the system.
- 4) Based on the results of related system studies and on a trade-off of impact vs system design elements, expand, clarify and modify as necessary the Task Force RNAV system operational concept, develop a total system design concept, and generate an implementation scenario which will support these concepts.

1.2 METHOD OF APPROACH

The final payoff analysis included in this report considered the three nominal RNAV implementation periods, both from the viewpoint of potential economic benefits to the user and the system and that of the ability of the system and user to function efficiently in the mixed VOR/RNAV environment. The payoff analysis effort concentrated on an expansion of the previous results relating to the ultimate costs and benefits associated with long term implementation rather than the transition phases, but also included the estimation of transition period benefits.

User Impact

The preliminary user benefit analyses [12] were limited to fuel and time effects of route length, altitude profiles, and conflict resolution as they applied to airlines jet aircraft operation. The current effort expanded that analysis in several ways.

A broader spectrum of users were considered, including business aircraft, general aviation, and military aircraft. The fuel and time savings achieved in enroute and terminal area operations by airline aircraft [12] were expanded to include the remainder of the primary users of the National Airspace System.

Preliminary terminal area designs were reviewed with users and were discussed with the respective FAA regions/facilities and were modified as appropriate [2]. The modifications also involved an iterative procedure designed to optimize the designs from a user economic point of view. A route structure based on one of the modified designs (New York) was utilized in a real time simulation at NAFEC [11] to further explore the benefits available to the user through the use of RNAV and to determine if fixed gradient VNAV routes provided any operational advantage to the user in the terminal area. Finally, annualized benefits due to altitude profile and route length effects were estimated by extrapolation for all CONUS terminal areas, and the incremental effects of 3D and 4D RNAV were determined. In the previous effort [12] route length savings were computed for both high altitude charted RNAV and

preplanned direct RNAV based on a nominal 150 airport pair route structure. This analysis has been updated and expanded in four ways: 1) an expanded high altitude structure containing 429 airport pairs was utilized; 2) a sample low altitude structure was developed, analyzed, and the results extrapolated to obtain an estimate of CONUS operation; 3) the no-wind route length analysis of the high altitude structure was calibrated by a simulation and analysis of RNAV weather routes compared with existing VOR weather routes; 4) the route length analysis of the high altitude structure was further calibrated by including the impact of restricted areas on the RNAV structure.

A series of interviews was conducted with various user groups to determine their respective views concerning RNAV implementation and to identify cost elements associated with that implementation as well as areas of potential benefit. Interviews were conducted with representatives of the National Business Aircraft Association, Aircraft Owners and Pilots Association, Helicopter Association of America, General Aviation Manufacturers Association, U.S. Army, U.S. Air Force, U.S. Navy, and U.S. Coast Guard. The information obtained was used to identify single event benefits, to calibrate the assumptions necessary in the analysis of enroute and terminal area fuel and time benefits, and to assess the overall impact of RNAV implementation on each of the user groups. The benefits available to six individual airlines, based on current schedule and frequency of operation, were also estimated.

ATC System Impact

Previous efforts [12,13] investigated the ability of the controller to operate in a mixed VOR/RNAV environment, estimated controller productivity increases due to RNAV, studied the effects of RNAV on enroute and terminal area capacity, and made preliminary estimates of terminal area and high altitude enroute requirements for VORTAC coverage to support an RNAV structure compared with that to support a VOR structure expanded to accommodate an equal amount of traffic.

In the current effort, the enroute high altitude VORTAC requirements analysis was expanded to include upgrading of VORTAC stations as well as the addition of new stations. The preliminary analysis of terminal area VORTAC requirements [13] was expanded to include a more detailed assessment of the relationship between traffic density, fix requirements, and VORTAC requirements. The analysis of the effects of slant range error which was initiated in Reference 7 was expanded through flight tests, additional simulation and additional analysis and the results applied to the assessment of low altitude and terminal area route design.

An assessment was made of RNAV route development requirements, including the requirement for a coordinated effort to produce optimum high altitude, low altitude, terminal and transition designs, from which individual routes can be implemented during the transition periods. This analysis included consideration of charting, ATC automation, video map requirements, VORTAC coverage and flight checking, and airspace capacity.

The impact of charted routes and preplanned direct RNAV operations on both enroute and terminal area automation requirements was investigated. Potential impact was considered with respect to storage and computational requirements for metering and sequencing, route definition, flow control, and automatic conflict prediction and resolution. Both hardware and software impact was assessed and an overview of potential problem areas and elements of probable impact was prepared.

A preliminary assessment of the impact of RNAV implementation on controller training requirements was made, including recognition of various levels of RNAV equipment capability, and the utilization of RNAV procedures and phraseology.

Finally, an additional real time simulation was conducted at NAFEC [11] to evaluate the impact of various levels of RNAV, VNAV, and radar vectored traffic on system performance and capacity to determine the effect of fixed gradient VNAV routes or controller workload and system performance.

RNAV Operational Concepts

The RNAV concept as envisioned by the Task Force [1] involved several elements which could have significant economic and operational impact (penalty or benefit) on the user and/or the AIC system. The operational aspects of these elements have been analyzed in this and other studies [12,13]. The results of these studies were utilized to develop an overall operational concept for area navigation which is based on the Task Force concept, but which utilizes the results of subsequent analyses, as were recommended by the Task Force. The resultant operational concept is intended to form a baseline for the implementation of RNAV in a manner most beneficial to the ATC system and a broad spectrum of users of the National Airspace System.

An analysis was conducted of cost versus capability of existing area navigation systems, with capability delineated in a manner which relates to operational utility, pilot workload, and economic benefit. The Task Force proposed that RNAV be the "price of entry" to high and medium density terminal areas in the post-1982 time period. An analysis was made of the relative numbers of aircraft, by type and use, which would require RNAV equipment as a result of each of several candidate definitions of a "terminal area" for this purpose.

A detailed analysis was made of various VNAV concepts, from the viewpoints of airspace capacity, user economics, user operational utility, ATC system considerations, and controller workload. The technical and operational aspects of turn anticipation, paralleling through turns, waypoint designation and storage, parallel offsets, and slant range correction were considered. Terminal area and enroute airspace design concepts were developed, based on results of terminal area design studies, enroute studies, slant range considerations, VNAV analyses, and route width requirements studies.

The intent was not to find a "compromise" solution which would not fully satisfy any individual user, but rather to describe a concept which recognizes several levels of RNAV/VNAV capability, allows their use without compromise to the system as a whole, and provides maximum benefits for all users.

1.3 USER RELATED RESULTS

The benefits available to the various users of the National Airspace System in an all RNAV environment accrue primarily from the economic and operational advantages of a more efficient and ordered route structure both enroute and in the terminal area. These benefits, which are heavily dependent upon the operational concept which is implemented, are derived through reduction in route lengths, improvement in vertical flight profiles, reduction in arrival holding delays, reduced pilot workload, and improvement in the availability and safety of instrument approaches.

The savings possible through use of a charted high altitude RNAV structure and 2D RNAV SIDs and STARs in six major terminal areas were derived in a previous study [12] for air carrier jet aircraft. The current study expanded the savings estimate to include a low altitude charted RNAV structure, the effects of weather routes in the high altitude structure, 2D and 3D RNAV benefits at sixty major airports, and the savings which could be realized with 4D time control navigation in twenty-five M & S terminals. Savings were estimated for air carrier, business, and other general aviation aircraft, although 3D and 4D savings were estimated only for air carrier aircraft.

The average enroute route length savings over VOR were estimated to be 2.36% for a charted low altitude structure and 1.61% for a charted high altitude structure. The high altitude structure savings include the effect of weather routes and assumes that no restricted areas will be violated. Enroute benefits are those that may be expected by an aircraft at random in a fully developed charted RNAV structure, but they are also indicative of the savings that may be expected on individual RNAV routes as they are implemented in the transition phases. The results are somewhat conservative in that the NAFEC structure upon which they are based is not an optimum coordinated structure. Terminal area 2D RNAV savings are a result of shorter route lengths for piston aircraft, and of both shorter route lengths and better altitude profiles for jet and turbo-prop aircraft. The 2D RNAV benefits include the savings due to improved climb profiles in vertical departure envelopes which generally allow each aircraft to select an optimum climb schedule. High performance climb envelopes are utilized where the limiting of the use of the route to aircraft of a certain minimum performance capability will result in shorter departure routes for such aircraft.

3D benefits are realized through pilot selection of 3D descent profiles as compared with procedural or "rule of the thumb" descent procedures. It was determined through real time simulation [11] that pilot selection of 3D descent profiles has no impact on controller workload or other ATC system benefits. Additional 4D savings in M & S terminals were derived, based on a capacity and delay analysis for 8 major airports which was extrapolated to 25 M & S airports.

Terminal area benefits, demonstrated by the results of real time simulations, also include a significant decrease in arrival holding delays, a decrease in time and distance flown, and in pilot communications workload.

Annual savings for calendar year 1984 are summarized in Table 1.1. The annual savings were computed on the basis of fuel cost plus the flight time sensitive portion of direct operating cost and did not consider the worth of

executive time in business aircraft operations. This factor can be particularly significant for business aircraft operations in which the "Value Added" concept of the worth of executive time is used and executive time savings may be valued as high as \$40 per minute.

The savings summarized in Table 1.1 are applicable to an all-RNAV environment in 1984. The realization of benefits by individual users, however, is not dependent upon an all-RNAV environment. Based on an assumed implementation scenario, which is described in Section 5.1.5, it was estimated that aggregate user savings would grow almost linearly over a seven year implementation period in reaching the levels indicated in Table 1.1. In a related study [14], where costs and benefits were computed for this implementation scenario at their present (discounted) value, it was determined that the user benefit to cost ratio for the period of 1982 to 2000 would be in a range of 5.5 to 7.1.

TABLE 1.1 Summary of 1984 Annual RNAV Savings over VOR

	Total	1984 Annua	1 Saving	s in Millio	ns of 19	75 Doll	ars	
	Ai	r Carrier		Business			Other General	Total
	Aircraft	Passenger Time	Total	Aircraft	Pass Time	Total	Aviation	
2D-60 Major Airports 3D Descents - 60 Major	91.1	93.5	184.6	.6	.9	1.5	1.2	187.3
Airports 4D at 25 M & S Terminals	49.1 110.4	45.8 142.1	94.9 252.5	:	:	:	:	94.9 252.5
Terminal Area Total	250.6	281.4	532.0	.6	.9	1.5	1.2	534.7
Charted Enroute	98.3	95.9	194.2	1.9	2.2	4.1	3.1	201.4
Direct Enroute	119.1	116.2	235.3	2.3	2.7	5.0	5.5	245.8
Total Charted and Terminal	348.9	377.3	726.2	2.5	3.1	5.6	4.3	736.1
Total Direct and Terminal	369.7	397.6	767.3	2.9	3.6	6.5	6.7	780.5

^{*} Not estimated

Of at least equal importance to the dollar savings with RNAV are the fuel savings which are contained therein. A summary of annual fuel savings at 1984 traffic levels is given in Table 1.2 for an all-RNAV environment.

Table 1.2 Summary of 1984 Annual RNAV Fuel Savings Over VOR

	Fuel Savings in Millions of Gallons							
	Air Carrier	Business Aircraft	Other General Aviation	Total				
Terminal Area-2D 3D Descents 4D in M & S Terminals	156 76 153	1 *	1 * *	158 76 153				
Terminal area total	385	1	1	387				
Enroute charted	171	2	4	177				
Enroute Direct	208	3	7	218				
Total with charted	556	3	5	564				
Total with direct	593	4	8	605				

^{*}not estimated

Analysis of existing RNAV approaches indicates that a reduction in minimum descent altitude may be possible in some cases for an RNAV approach compared with a VOR approach, but that the primary advantages of RNAV (during approach) lie in the elimination of circling approaches, and in providing approaches to non-instrumented runways. In addition to the increased safety element in the elimination of circling approaches, considerable distance savings are also possible. In non-radar terminals an RNAV approach may save as much as 30 miles over a VOR approach, depending upon location of the initial approach fix with respect to the arrival waypoint. RNAV also provides the potential for application of noise abatement procedures under IFR conditions.

Projected payback for a user is dependent upon both the cost of RNAV installation and the specific savings unique to his type of operation. The benefits available to the approximately 3100 air carrier jet aircraft in 1984 would support an average amortization or payback on the order of \$113,000 per year per aircraft in a total RNAV environment. This average payback is derived by dividing the total annual airline savings by the number of airline aircraft. Individual per aircraft payback, which is not dependent upon a total RNAV environment, will vary over an approximate range of \$20,000 to \$150,000 per aircraft per year depending upon type of aircraft and schedule. The individual aircraft benefits were derived from an analysis of six airlines' current operations.

It was determined that, if the RNAV TCA concept described in this report were chosen, 11,000 of the projected 67,000 business aircraft in 1984 and only 7,500 of the projected 145,000 non-business general aviation aircraft in 1984 would be required to have RNAV. Business and other general aviation jet and turboprop aircraft would realize an annual savings of \$5850 and \$2070 per aircraft respectively, based on fuel and aircraft time alone, which would yield a reasonable payback period for the cost of airborne equipment. A reasonable payback period cannot be projected for small piston aircraft on the same basis. However, as evidenced by the current popularity of RNAV equipment among general aviation users (over 4500 equipped), there are other benefits available, such as VFR pilotage aid, elimination of circling approaches, and access to airports which do not currently have a VOR approach.

The potential cost impact on military operations and equipment requirements which would be caused by the implementation of an all-RNAV environment varies widely among the services. The impact on the Coast Guard would be nominal since all USCG aircraft have relatively sophisticated multi-sensor navigation systems. The impact on the Army would depend greatly upon the way in which terminal areas are defined for purposes of requiring RNAV capability. If the RNAV TCA concept were chosen, whereby RNAV were required in airspace defined in a manner similar to current terminal control areas, only 3-5% of Army aircraft would be required to have RNAV.

The impact on the Navy and Air Force could be considerably greater. Only a small percentage of their aircraft have tactical navigation systems which could be adapted for RNAV use and little spare space exists to accommodate the installation of an RNAV computer. Many tactical aircraft are only occasional users of the National Airspace System and therefore would accrue the benefits of RNAV at a slower rate than the constant users. New aircraft system designs could probably accommodate an RNAV capability with minimal impact, but a large scale retrofit requirement could probably be fulfilled only through new and innovative equipment design.

1.4 SYSTEM RELATED RESULTS

The cost impact of RNAV on the ATC system and the benefits derived therefrom are as closely related to the operational concept which is implemented as are the corresponding costs and benefits for the users. The system impact which is presented in this report, as is the user impact, is based on the implementation of the system design concept described herein. This impact is specifically related to an RNAV environment which is achieved through an orderly evolution, based on user demand, throughout the transition phases of mixed VOR/RNAV operations.

The most important step in this evolutionary process is the development of optimum, coordinated high altitude, low altitude, terminal, and transition structures from which individual routes can be implemented throughout the transition periods. An analysis was made of route development requirements

and their impact on various elements of the ATC system as discussed below. Close coordination will be required between various services of the FAA and user groups in both the initial development of the optimum route structures and in the continuing process of implementation.

In a previous analysis [12] it was determined that the pre-1982 VORTAC coverage for the charted high altitude structure could be attained with the addition of one station, and that the post-1982 total CONUS direct coverage could be attained for \$14 million plus approximately \$1 million per year maintenance costs - - these estimates were based on the assumption that all VORTACs for which Task Force postulated accuracies were required had already been upgraded to achieve that accuracy, and that the Task Force recommended route width of ±2.5 nm would be required. Subsequent analysis has resulted in the conclusion that constant ±4 nm route widths will adequately accommodate expected traffic. An analysis of VORTAC coverage cost based on a \pm 4 nm route width and an optimum mix of upgrading existing stations and adding new stations resulted in initial costs of \$600,000 for a 1977 charted high altitude structure and approximately \$6 million for the post-1982 complete CONUS coverage. These costs are nominal when compared with the annual F & E budget plan for Navigation Aids upgrading and expansion for the VOR airways system. Also, these costs are based on an overall estimate and do not reflect the requirements for the specific route structure to be implemented, nor do they consider annual maintenance savings achievable through removal of redundant stations. It can be expected that a number of stations will no longer be required, and that the annual maintenance savings would provide an early offset for any upgrading costs for RNAV coverage requirements.

The effects of slant range error were found to be negligible, from an airspace planning viewpoint, below 12,500 ft AGL in the terminal area and low altitude enroute structure. It was determined that the number of VORTACs required to support RNAV terminal areas over the CONUS was 48 less than the number required to support VOR terminal areas at the projected 1982 traffic levels. The resultant savings were estimated as \$2.3 million per year due to maintenance costs associated with the VORTACs which were removed.

Preliminary estimates of the impact of RNAV on ATC automation were derived. A total of six areas in ATC automation which were determined to exhibit potential for impact were analyzed: enroute and terminal area route definition, enroute and terminal area conflict prediction, terminal metering and spacing, and enroute flow control. For the most part the impact was found to be minimal or even favorable. It was determined that the cost of enroute and terminal computer system core requirements due to the implementation of RNAV is \$258,000 to support a mixed VOR/RNAV environment. However, core requirements for an all-RNAV environment would be \$174,000 less than the current requirement.

A major concern of the RNAV Task Force was the possibility of penalties to the ATC system resulting during the transition period when a mix of VOR and RNAV traffic would be required. The results of real time simulations [11,12] indicated that controllers are capable of operating in a mixed VOR/RNAV environment with reduced workload and increased productivity. An increase in operation rates and decreased time and distance flown by RNAV aircraft in the terminal

area was also demonstrated. Furthermore, an increase in system capacity was evident which resulted in a significant reduction in arrival holding delays. A continuing controller training program will be necessary to insure a level of familiarity with RNAV procedures and capabilities that will result in a timely and orderly evolution to an all-RNAV environment.

RNAV will increase airspace capacity by providing a more ordered route structure, which reduces the potential for conflicts in the enroute environment. The use of 4D in M & S terminals will also provide a significant increase in capacity. A summary of the impact of RNAV on the ATC system over the period of 1982-2000 is given in Table 1.3.

Table 1.3 ATC System Impact Summary	1982-2000			
	Savings		Costs	
	Total	Present Value	Total	Present Value
VORTAC REQUIREMENTS Terminal area (maintenance) Enroute (implementation and maintenance)	\$36.1M	\$8.0M	\$2.4M	\$0.8M
CONTROLLER PRODUCTIVITY Terminal Area Enroute	\$26.4M \$422.0M	\$8.0M \$120.7M		
IMPLEMENTATION (Training, automation, charting and flight inspection, route development)			\$19.8M	\$13.0M
TOTAL	\$484.5M	\$136.7M	\$22.2M	\$13.8M

1.5 RNAV OPERATIONAL CONCEPT

Certain modifications, expansions and clarifications of the Task Force operational concepts, which are based on the results of studies to date are summarized below.

Terminal Area Concept

2D SIDs and STARs with a route width of \pm 2nm or \pm 4 nm depending upon distance from the VORTAC [15] should be established at all radar terminals which overlie, and are compatible with, optimum radar vector paths in order to allow controllers and users to accrue immediate benefits and to gain experience with RNAV procedures in a manner which is compatible with radar vector procedures during

the transition period to an all-RNAV environment. The all-RNAV terminal area designs for the 1982 period should be based on a modified Task Force Design with the terminal maneuvering area aligned according to runway usage and the terminal transition area aligned according to the traffic flow. The octant concept should be relaxed as appropriate to accommodate traffic flow and to optimize the interface with the enroute structure. Slant range correction will not be required below 12,500 feet.

3D capability should not be required in the terminal area, but terminal route design should accommodate 3D separation requirements in order that 3D equipped aircraft could utilize individually selected 3D profiles within the vertical envelope provided in the 2D route design.

VNAV

The ability to utilize pilot-defined vertical routes on an ad hoc basis can provide significant economic benefits to an individual user and terminal area routes should be designed to accommodate pilot selection of 3D climbs and descents. The use of fixed gradient VNAV routes for procedural separation and/or the requiring of 3D capability for entry into certain airspace does not appear to offer sufficient payoff in either airspace capacity or operational utility to warrant serious consideration and would impose a penalty on the user. A fixed-gradient VNAV design would provide a rigid structure which would reduce the controller's flexibility for impromptu changes and which would greatly complicate the controller's visualization of the traffic picture and his ability to monitor traffic.

High Altitude Route Structure

The Task Force recommended constant ± 4 nm route width for the 1977 system is achievable, with the current ground station accuracy. The expected capacity of the 1980s can also be served by constant ± 4 nm route widths.

The airborne equipment and FTE requirements stated in the Task Force Report are required, but VORTAC upgrading and installation cost is minimized. Slant range correction is required in the high altitude structure. The implementation of the RNAV Task Force pre-planned direct concept would require ATC automation features not yet developed, and would require a significant amount of flight checking of VORTACs to develop area coverage plots. Pre-planned direct routes will require radar monitoring until these requirements are fullfilled. The 1000 ft. vertical separation recommended by the Task Force is achievable with the altimetry improvements postulated, but only in level flight. VNAV separation is dependent upon along track error and 1000 foot separation is not generally achievable with the anticipated lateral accuracies.

Low Altitude Structure

Slant range correction should be required above 12,500 feet MSL. Aircraft operating above this altitude are required to have encoding altimeters, which is the major cost element in providing slant range correction. Slant range errors are negligible from a route design viewpoint below 8000 ft. AGL,

but a slight expansion of route width in the vicinity of the VORTAC is required from 8000 to 12500 ft. AGL. The low altitude structure should be a dual system with charted RNAV optimized and skeletal VOR structure accommodated. Pre-planned direct in the low altitude structure is dependent upon the same development features as in the high altitude structure.

Waypoint Designation

The RNAV concept involves both charted and pre-planned direct routes whose segments or, in some cases, the entire route are great circles. A variety of navigation sensors, ranging from VORTAC, to self-contained, to long range aids will be utilized. RNAV systems will range in complexity from single waypoint rho/theta analog computers to sophisticated digital multi-sensor dead reckoning systems. The charting of waypoints and the definition of impromptu waypoints, along with simplified waypoint descriptors, is necessary for the optimum use of these routes by pilots with the total range of airborne equipment. The optimum designation system is the combination of the radial/distance and pronounceable 5 alpha systems in use today.

Accuracy Tolerances

Route widths required to support airspace capacity requirements in the 1980's are not as stringent as envisioned by the Task Force, and the resulting route structure can be supported with upgrading the accuracy of only a few VOR stations. However, additional analysis of existing and planned data is necessary to completely resolve the issue of error budgeting and compliance with accuracy requirements.

RNAV Functional Requirements

The pre-RNAV Task Force operational elements necessary to provide area navigation capability were documented in RTCA publications [16,17]. The Task Force, however, detailed several additional functional requirements which are advantageous to the RNAV user and/or are required by the RNAV system operational concept, but which require additional airborne equipment capabilities. For controllers to effectively use RNAV, all operations based on like commands should result in like maneuvering of the aircraft over the ground. The basic functional capabilities issues - some as yet unresolved - are as follows:

- Waypoint Storage A minimum storage of more than one waypoint is required for high density terminal area operation. A method of readily and unambiguously checking waypoint storage data is required.
- 2) Turn anticipation requirements can probably be met procedurally, but some method of turn alert may be required.
- 3) RNAV systems should have the capability of parallel offsets in increments of one nm to a distance of 20 nm. Displaced needle CDI indications are not an acceptable method of parallel offset. Constant radius turns are not required, and paralleling through turns, when required, may be accomplished procedurally. The use of offsets inside a turn should be minimized.

- 4) When vertical guidance equipment is utilized, vertical maneuver anticipation of some type will be required. The anticipation requirement may be able to be accomplished procedurally in combination with an altitude alert signal.
- 5) The utilization of offsets during climb or descent, particularly when performed around a turn or in the vicinity of a crossing route turn point, should be based on a prior recognition of airspace limitations in order to insure that adequate vertical separation is maintained and that aircraft performance limits are not exceeded. When climbing or descending offsets around turns are utilized, turns may be handled procedurally in the plane of the parent route vertical path angle, or turn points may be computed on the bisector of the turn angle.

1.6 IMPLEMENTATION REQUIREMENTS

The successful implementation of Area Navigation requires a carefully coordinated, time phased, systems approach. It is particularly important that the design of RNAV structures for high altitude, low altitude, terminal and transition areas be closely coordinated. Optimum designs should be created to handle all high altitude and the majority of low altitude traffic. 2D RNAV routes which incorporate 3D altitude separation criteria should be designed for all high density and selected medium density terminals to maximize user Specific routes should be implemented from these designs (enroute and terminal) as public use routes in accordance with user requirements. Flight checking of routes should proceed in accordance with user requirements. but provisions should be made for the development of flight checked area coverage charts which will define areas and NAVAID coverage within which preplanned direct route may eventually be utilized without radar monitoring. A three phase implementation program, similar to that recommended by the Task Force, is required, but the timing of the phases should be based on user demand.

High Altitude Enroute

In Phase 1, VOR routes should be realigned as required as RNAV routes are implemented. Limited pre-planned direct flight will be accommodated, with radar monitoring, and pilot selection of 3D descents may be utilized where not in conflict with procedural descent requirements. Tactical use of parallel offsets by RNAV equipped aircraft should be initiated. In Phase 2, 2D RNAV utilizing the charted high altitude structure will be the system, and all VOR routes will be deleted. Procedures should be established to allow wider latitude in pilot selection of 3D descents. In Phase 3, the charted RNAV structure will be retained as the system, but completion of flight checking and implementation of automation improvements will permit widespread use of pre-planned direct flight without the requirement for radar monitoring. Use of VORTACs or existing waypoints will be required for pre-planned direct flight segments by station-referenced (non-geographic) systems and will normally be used by all aircraft unless a distinct user advantage is apparent.

Low Altitude Enroute

Implementation in Phase 1 should follow the same procedures outlined for the high altitude structure. In Phase 2, unnecessary VOR airways should be deleted, and the remaining VOR airways should be realigned to conform to RNAV routes to the extent practicable. Procedures should be developed to allow pilot selection of 3D descents. In Phase 3, 2D charted RNAV with a scaled down VOR airway structure will be the navigation system. Pre-planned direct may be utilized, with station referenced (non-geographic) systems using existing waypoints or VORTACs. Radar monitoring will be required where flight checking has not been completed.

Terminal Area

In Phase 1, 2D RNAV routes should be designed for all high and selected medium density terminals. 2D SIDs/STARs should be implemented in conjunction with corresponding enroute segments, and VOR/vector paths should be realigned to accommodate RNAV routes as practicable, or the RNAV routes may be realigned if necessary. 2D SIDs should be implemented to permit pilot selection of 3D user beneficial descents. 4D procedures should be established in M & S terminals. 2D/3D approaches should be established at all airports to the extent practicable, consistent with IFR requirements. In Phase 2, 2D RNAV routes should be designed for all low density and non-radar terminals. As RNAV routes are implemented, VOR/vector routes should be realigned as necessary to conform to the RNAV routes. 2D RNAV is the navigation system in high density terminals. Climb envelopes should be established to provide shorter departure routes for high performance aircraft. In Phase 3, RNAV will become the system in medium density terminals.

1.7 CONCLUSIONS

The results obtained from economic and operational impact analyses, and from various supporting system studies, indicate that the advantages of area navigation to both the users and the ATC system are sufficient to warrant implementation of the area navigation concept described in this report. This concept is based on the Task Force recommendations, but is modified as appropriate to insure that maximum benefits will accrue to both the system and the users. Although additional research and development work is still required in some areas, implementation of the area navigation concept can proceed in parallel with these efforts.

The overall objectives of this analysis of area navigation payoffs and development of operational concepts and implementation requirements were (1) to review and analyze the results of previous Task Force payoff studies and supporting system studies, (2) to update and complete payoff analyses which were initiated in previous study efforts, (3) to summarize RNAV payoffs and economic impact with respect to both the ATC system and the users of the National Airspace System, and (4) to develop an operational and implementation concept, based on the RNAV Task Force recommendations, which is most beneficial to the ATC system and a broad spectrum of users.

The FAA/Industry Task Force Report [1] described an area navigation system design concept which was based on certain assumptions concerning the application and capabilities of area navigation and the requirements of the Air Traffic Control System. The Task Force recognized that a significant amount of investigative work remained to be done in order to validate the recommended operational concept and the assumptions inherent in its design. The Task Force Report listed several problems associated with the implementation of area navigation, discussed regulatory, equipment, and system requirements, and described the research and development efforts which should be accomplished prior to implementation.

The Systems Research and Development Service (SRDS) of the FAA developed an Engineering and Development Plan [18] in response to the Action Plan detailed by the Task Force, and instituted a program of system studies and analyses in the areas of RNAV Payoff, RNAV Terminal Design, RNAV Enroute Design, RNAV avionics and other supporting studies. The area indicated as that of the highest priority by the Task Force was RNAV Payoff, and the other studies were designed to provide inputs to the payoff studies.

This report contains the results of a study effort to define the costs and to assess the economic impact which are expected to accrue to the ATC system and the various users of the National Airspace System as a result of the implementation of area navigation, and to develop an operational concept for area navigation which is based on the Task Force concept, but which utilizes the results of the subsequent analyses which were recommended by the Task Force.

This section is divided into two major parts: Section 2.1 - Supporting System Studies, and Section 2.2 - Expanded Payoff Analysis. The objectives and the overall approach used in previous and ongoing systems analyses, directed at identification and solution of potential RNAV implementation problems, are summarized in Section 2.1. The study approach utilized in the expansion of previous payoff analyses and the identification of payoff and the operational and economic impact of RNAV implementation on the user and the ATC system is described in Section 2.2.

The objectives, detailed methodologies, and results of the expanded payoff analyses are given in Section 3 (User Impact) and Section 4 (System Impact).

Section 5 presents first a summary and aggregation of RNAV payoffs to the user and the system (Sections 5.1 and 5.2), then a system design concept based on these payoffs as well as the Task Force concept and results of the supporting

systems analyses described in Section 2.1 (Section 5.3), and finally a detailing of the action required for the implementation of this system concept (Section 5.4). Conclusions concerning RNAV payoff, cost impact, system design concept, and implementation requirements are given in Section 6.0.

2.1 SUPPORTING SYSTEMS STUDIES

The objectives and overall study approach utilized in the several supporting system studies are summarized in the following paragraphs. Detailed methodologies and results are presented in separate reports which are referenced in the following discussion. A summary of these results is contained in the development of the RNAV system design concept in Section 5.3.

2.1.1 Waypoint Designation Standards

The RNAV concept involves both charted and preplanned direct routes and the utilization of both earth referenced (lat-long) and station referenced (rho-theta) airborne systems. The Task Force recognized the necessity for establishing a system of waypoint location, identification, and communication which would provide for optimum use of RNAV routes by either type of system while at the same time minimizing controller and pilot workload.

Reference 3 reports the results of a study which developed recommended way-point designation standards. The basic objectives of the study were to evaluate Task Force operational and procedural concepts and their impact on waypoint designation, to evaluate the advantages and disadvantages of grid and other alternative systems, and to develop a set of waypoint designation standards that will:

- 1. Present a communicable, easily remembered identifier of a unique geographical location for charting purposes for pilots and controllers.
- Provide an identifier compatible with computer input requirements for airborne navigation, flight planning, and ground based ATC systems.
- Apply to earth oriented as well as station oriented navigation systems.
- Provide an easily identified and transmitted navigation fix which can be utilized in the ATC system for controller/pilot voice communications.
- 5. Provide the most practical means for pilots and controllers to exchange information on impromptu waypoint locations for both station referenced and earth referenced RNAV equipped users.
- 6. Provide a set of waypoint location and identification standards for utilization during all three RNAV implementation phases.

The approach taken in this analysis was based on satisfying, to the extent possible, the requirements for all applications of the designators. There are several basic characteristics which a designator should satisfy:

- (1) Airspace Utilization
- (4) Charting Considerations
- (2) Communicability
- (5) Workload
- (3) Orientation Information
- (6) Equipment Implications

The precise specification of the desired waypoint designator characteristics produced a list of 32 designation criteria necessary to adequately satisfy the needs of the six basic characteristics. The waypoint designation criteria developed considered all phases of RNAV implementation.

The proposed waypoint designation techniques described and analyzed in this

study included:

GRID SYSTEMS

NON-GRID SYSTEMS

Coded Lat/Lon
Radix Lat/Lon
Delta Distance
Clock Grid
Cardinal Radial
Radial Distance
Computer Generated

Postal Code
Mnemonic 5 Alpha
Pronounceable 5 Alpha
VORTAC Referenced
Arrival/Departure
Standardized Location Designation
5 Numeric Designator

The preferred designation symbology derived from this analysis was a dual designator consisting of a combination of a grid and non-grid designation system. The recommended combination is the Radial Distance (polar grid) designator, and the Pronounceable 5 Alpha (non-grid) designator which are in use today for area navigation. The investigation failed to reveal any new technique which would be more advantageous. While certain other techniques are superior with respect to one or more of the criteria, the system chosen ranked highest in overall acceptability. In addition to the recommendation for waypoint designation, corresponding standards were developed in the following areas:

- (1) Facility Selection for RNAV Routes
- (2) Parallel Offset Considerations
- (3) Waypoint Charting
- (4) Waypoint Location and Route Definition

2.1.2 Avionics Standards

RNAV equipment functional and accuracy requirements to provide area navigation capability were documented in RTCA publications [16,17] and in Advisory Circular 90-45A [19]. The Task Force developed additional concepts which may be advantageous to the user and/or are required by the system operational concept, but which require additional functional capabilities. The Task Force also recommended system error budgets which are based on the RSS method of combining errors and which postulate an improvement in flight technical error over that currently attained. A need existed to validate, through flight test and simulation, the assumptions inherent in the Task Force Concept, and to develop a set of Avionics Standards and Minimum Operational Characteristics which are based on an operational concept more closely aligned to that recommended by the Task Force.

An avionics standards program was undertaken which included a series of flight tests and simulations designed to provide the data base necessary to resolve the issues raised by the Task Force concept. The basic method of approach for this study was to examine existing area navigation functional and accuracy requirements, to compare this set of information with the concepts described in the RNAV Task Force Report, and then to develop and categorize the minimum operational characteristics regarded as necessary for acceptable performance within the National Airspace System.

Flight tests were conducted using a variety of area navigation systems, from a single waypoint analog system to an ARINC-582 system, in aircraft ranging from an Aero Commander 500 to a DC-10. The flight tests were supplemented by simulations using GAT-II cockpit simulators. An interim report [7] which summarizes the form and substance desired for an updated avionics standards document has been prepared.

The standards developed were categorized into Basic Design Considerations, Standard Functional Requirements and Unresolved Equipment Capabilities. There were five unresolved issues solely related to 2D area navigation and two additional issues related to 3D area navigation.

Unresolved Equipment Capabilities

2D Only

Additional 3D Considerations

- (1) Waypoint Storage
- (1) Vertical Maneuver Anticipation
- (2) Turn Anticipation
- (2) Parallel Descent Maneuvers
- (3) Parallel Offset
- (4) Track Determination
- (5) Slant Range Error

Ongoing analysis of flight test and simulation data, coupled with consideration of equipment cost versus user and benefit, is designed to resolve these issues. The issues of system error budgets, error combination techniques and flight technical error are also undergoing additional analysis. The analysis of slant range error initiated in Reference 7 was expanded and is included in Section 4.1 of this report.

Preliminary conclusions concerning the unresolved issues have been made, based on analysis to date, wherever they have a bearing on the development of a recommended RNAV system design concept.

2.1.3 Terminal Area Design

The RNAV Task Force recommended a time-phased implementation of RNAV, and provided for two transition periods in which terminal area designs would be based on compromises between the post-1982 RNAV concept and the requirement for retaining VOR routes. The Task Force report contained a terminal area model providing for alternating arrival/departure octants, a final approach terminal maneuvering area, and specifying the location of arrival and departure waypoints.

A terminal area design and analysis program was initiated to validate the Task Force concepts.

The basic objectives of the program were twofold:

- (1) Evaluate the RNAV and VNAV (3D) Task Force terminal area design concept by applying it to various medium and high density terminal areas
- (2) Based on an analysis of the terminal area designs, recommend design techniques to be used in the pre-1977 and 1977-82 transition periods, and in the post-1982 period, for the development of both 2D and 3D RNAV terminal area designs

The major source of data and information on present terminal area designs was furnished by ATC teams from the National Aviation Facilities Experimental Center (NAFEC) who visited 14 terminal areas and collected all available information. This information was used to develop the current radar vector and VOR traffic flow patterns used in the terminal areas for both primary and satellite IFR airports.

The Task Force 2D RNAV terminal model was applied to seven terminal areas (13 airports). Preliminary time phased designs were created at all seven terminal areas. The first time period designs, pre-1977, were based heavily upon current terminal area procedures. RNAV routes in this time period often were essentially coincident with current radar vector and VOR routes. The second time period designs, 1977-82, were based upon accommodating both RNAV and VOR aircraft. The Task Force design was used to the maximum extent possible under the constraint of maintaining a satisfactory VOR traffic flow utilizing current navigation facilities at their present locations. The third time period design, post-1982, made use of the Task Force terminal area model to the maximum extent possible based upon the constraints of the characteristics of the terminal area.

The sequence of design development began with the design of the pre-1977 RNAV routes based on the present VOR route structure. Then the post-1982 RNAV routes were developed and the design effort for each terminal area concluded with the development of the mixed RNAV-VOR route structure for the 1977-1982 time period. The major perturbing factors in the application of the Task Force design to post-1982 designs were caused by multiple major airports in the terminal area and by complex runway layouts which necessitated modification of the terminal routes that were near the airport.

Terminal RNAV routes were developed for the following terminal areas for all three time periods: New York (LGA, JFK, EWR), Denver, Philadelphia, Chicago (ORD, MDW), New Orleans, San Francisco, and Miami. The 1972-1977 and post-1982 designs also included FLL, OAK, and SJC.

Two New York terminal transition period designs furnished the basis for a comprehensive real time simulation effort at NAFEC [9]. The simulation was designed to determine the controller's ability to operate in a mixed VOR RNAV environment during the transition period. Data on the impact of RNAV on the user and on controllers' workload were also recorded.

Nine airports in all seven designs (FLL, OAK, SJC, and MDW were excluded) were subjected to an analysis of route length and altitude restriction effects upon four types of turbojet aircraft. Economic comparisons were made between the post-1982 route structures and the current radar vector/VOR route structure.

The results of these analyses tended to point out apparent weaknesses in some elements of the Task Force terminal design model. In order to correct these weaknesses a modified Task Force terminal design procedure was developed. This procedure called for a greater use of the traffic density and direction of traffic flow information for the terminal area. In addition, the design of vertical departure route profiles for both 2D and 3D equipped aircraft which could accommodate varying aircraft climb performance was used. The arrival route vertical profiles were designed to maximize user benefits through pilot selected 3D descents. This design procedure was applied to the New York terminal area and the resulting design was also subjected to the route length and altitude restriction analysis program. The results of this analysis indicated that a considerable improvement in user benefits (time and fuel) could be expected if such a design were applied to the New York area operations as well as other areas. Selected high performance departure routes were developed for New York as well. These routes were developed in places where shorter route lengths to the boundary of the terminal area could be achieved if a minimum gradient can be attained by the aircraft using the route. Analysis of this design concept indicated that high performance departure envelopes provide a definite benefit to the user. Both time and fuel savings were achieved by aircraft using the envelopes rather than the corresponding RNAV route.

The terminal area designs were reviewed with Air Traffic Service personnel at the appropriate regions. New terminal area designs were then performed, based on both the region comments and considerations of the user related route length and altitude restriction economic analysis, and revised terminal area design guidelines were developed. The analysis also included determination of the incremental benefit realized through pilot selection of 3D descents. The new terminal area designs were subjected to the route length and altitude restriction analysis in an iterative procedure designed to optimize the designs for user economics, as reported in Section 3.1 of this report.

The new post-1982 terminal area design was utilized as the basis for a second real time simulation effort at NAFEC [11]. This simulation was designed to gather additional data on the controller's ability to operate in a mixed VOR/RNAV environment, to evaluate the impact of various levels of RNAV, VNAV, and radar-vectored traffic on system performance and the efficiency of operation of the system user, and to explore the effectiveness of vertically layered ("stacked") arrival routes. The simulation included the utilization of wind and navigation system error models.

2.1.4 Route Width Requirements

The RNAV Task Force report developed system accuracy requirements based on the assumption that route widths would have to be progressively reduced to ± 1.5 nm in the terminal area in order to accommodate traffic demand in the 1980s. Several analyses have subsequently been performed which addressed specifically the route widths required to meet the needs of forecasted traffic in an RNAV environment.

The requirement for route widths to satisfy traffic demand and the impact of route width on route design were addressed in three separate studies.

In one study [5], the objective was to quantify the current effective route width and the limitations which it imposes upon other aspects of the airway system, and to determine the requirements for reduced route widths. Route width was found to have an effect on the adequacy and interaction of route location, on the ability of the route structure to provide the required number of routes, and on route length. The impact on each of these areas was quantified through analysis and simulation and it was determined that traffic demand could be accommodated with constant ±4 nm route widths.

The impact of route width was also considered in the terminal area design process described in Section 2.1.3 and in the development of the high altitude route structure described in Reference 4, and the acceptability of ± 2 nm and ± 4 nm route widths respectively was verified.

2.2 EXPANDED PAYOFF ANALYSIS

The economic impact of area navigation on the ATC system and the users, as reported in References 11 and 12 and updated and expanded in Sections 3 and 4 of this report, was assessed as described in the following sections.

2.2.1 Terminal Area Analysis

The results of a route length and altitude restriction analysis of seven preliminary terminal area designs covering nine airports were presented in Reference 12. These results compared preliminary post-1982 designs with current VOR/radar vector routes, and quantified the savings in time and fuel available for several levels and projected mixes of traffic. These results, together with operational inputs from the appropriate FAA regions, were then used to develop new terminal area designs. The new terminal area designs [2] were subjected to the route length and altitude restriction analysis (described in Section 3.1) and the results of this analysis were extrapolated to a national scale, and further presented in terms of annual dollar savings in Section 5.1.3.

2.2.2 Enroute Impact Analysis

In the previous analysis [12] route length savings and conflict effects were computed for both high altitude charted RNAV, based on a 184 airport pair structure, and a high altitude direct structure, compared with the current VOR high altitude structure. Subsequently, an expanded high altitude structure was developed [4] and analyzed for route length and conflict effects. The results of this analyses are described in Section 5.1.2. The no-wind route length analysis of the high altitude structure was calibrated by a simulation and analysis of RNAV weather routes compared with existing VOR weather routes. This analysis is described in Section 3.2. Also, an analysis of low altitude enroute route length effects was performed as described in Section 3.3. The fuel and time savings due to all enroute route length effects are summarized in Section 5.1.2. Additionally, a discussion of the impact of more desirable altitude assignments in an RNAV environment is included in Section 5.1.2.

2.2.3 User Group Analysis

Previous payoff studies [12,13] were limited to the fuel and time effects of RNAV on air carrier jet aircraft operations. The current analysis expanded that effort in several areas. The terminal area and enroute impact analyses, described in Sections 2.2.1 and 2.2.2 above, included a broader spectrum of aircraft types, and estimates were made of dollar savings applied to business jet and other general aviation operations. The results of this analysis is given in Section 5.1.4, and the results of the expanded airline impact analysis is given in Section 5.1.3.

A series of interviews was conducted with various user groups to determine their respective views concerning RNAV implementation and to identify cost elements associated with that implementation as well as areas of potential benefit. Interviews were conducted with representatives of the National Business Aircraft Association, Aircraft Owners and Pilots Association, Helicopter Association of America, General Aviation Manufacturers Association, U.S. Army, U.S. Air Force, U.S. Navy, and U.S. Coast Guard. The information obtained was used to identify single event benefits, to calibrate the assumptions necessary in the analysis of enroute and terminal area fuel and time benefits, and to assess the overall impact of RNAV implementation on each of the user groups. An analysis of the impact of RNAV implementation on military aircraft is contained in Section 5.1.4.4.

The benefits accruing to each user, and the expected payback, is a function of both the cost and the capability of his airborne RNAV equipment, which in turn are related. Section 3.5 presents an analysis of capability versus cost for a postulated range of equipment complexity based on currently available hardware, and the relationship of cost to expected benefits is discussed in Section 5.1.

The Task Force recommended that RNAV capability be required in all medium and high density terminal areas as well as the high altitude structure. An assessment was made of the number and type of general aviation aircraft which would be required to purchase RNAV equipment as the "price of entry" to terminal areas as a function of various methods of defining those "terminal areas". The methods of defining the terminal areas ranged from the Task Force concept of a nominal 45 nm radius to a Terminal Control Area (TCA) concept where RNAV equipment would be required for only the primary airports and closely adjacent satellite airports. The analysis also included an assessment of VOR and DME equipment requirements to interface with the airborne RNAV equipment. The results of this analysis are given in Section 3.5.2. An analysis was also conducted of RNAV approach benefits attainable both through decreased minimums and mileage saved compared with existing approaches, and the results are discussed in Section 5.1.1.4.

2.2.4 ATC Impact Analysis

Previous efforts [12,13] investigated the ability of the controller to operate in a mixed VOR/RNAV environment, estimated controller productivity increases due to RNAV, studied the effects of RNAV on enroute and terminal area capacity, and made preliminary estimates of terminal area and high altitude enroute requirements for VORTAC coverage to support an RNAV structure compared with that to support a VOR structure expanded to accommodate an equal amount of traffic.

In the current effort, the enroute high altitude VORTAC requirements analysis was expanded to include upgrading and moving of VORTAC stations as well as the addition of new stations, and a parametric analysis was performed relating coverage requirements to accuracy, station upgrading cost, relocation cost, and new installation cost. Both VOR/DME and DME/DME coverage was analyzed. Results of this analysis are given in Section 4.2.

The preliminary analysis of terminal area VORTAC requirements [12] was expanded in Section 4.3 to include a more detailed assessment of the relationship between traffic density, fix requirements, and VORTAC requirements. The analysis of the effects of slant range error which was initiated in Reference 7 was expanded through flight tests and additional analysis, and the results applied to the assessment of VORTAC requirements in the terminal area. The potential effect of slant range error on low altitude enroute structure route placement was also determined. Results are given in Section 4.1. An assessment was made of RNAV route development requirements. The initial development of optimum high altitude, low altitude and terminal area structures, from which individual routes may be implemented in an evolutionary manner, was considered. An analysis also was made of the impact of RNAV route development on the related areas of charting, ATC automation, video maps, flight checking, VORTAC requirements and RNAV approaches. This analysis is described in Section 5.4.4.

The impact of charted and preplanned direct RNAV operations on both enroute and terminal area automation requirements was investigated as described in Section 4.4. Potential impact was considered with respect to storage and computational requirements for metering and sequencing, route definition, flow control, and automatic conflict prediction and resolution. Both hardware and software impact was assessed and an overview of potential problem areas, elements of probable impact, and cost impact was prepared.

Data from the terminal area real time simulations described in References 9 and 11 and from the conflict analysis of Reference 10 were utilized to analyze the impact of RNAV on the controller and on system capacity. Controller impact is discussed in Section 5.2.2 and airspace capacity is discussed in Section 5.2.4.

An analysis of the impact of VNAV on airspace capacity is given in Section 4.5. A comparison is made of the airspace required for crossing 2D routes, with procedural separation accomplished by altitude restrictions, with the airspace required for crossing 3D routes.

2.3 SYSTEM DESIGN CONCEPT AND IMPLEMENTATION REQUIREMENTS

The results of the payoff analyses, operational and economic impact analyses, and the several related systems studies were used in conjunction with the Task Force recommendations to formulate a system design concept and implementation scenario. The recommended concept is described in Sections 5.3 and 5.4, and includes an identification of the modifications made to the Task Force recommended concept.

This section addresses the impact of RNAV on several areas of user operation. Section 3.1 contains an analysis of fuel and time impact of RNAV operation in the terminal area, including both 2D RNAV and 3D descents. The sensitivity of these benefits to selection of descent procedure is also discussed. Section 3.2 presents an analysis of the effect of weather routes and restricted areas on high altitude route length savings and Section 3.3 estimates the RNAV route length savings available in the low altitude structure. The impact of 4D time control navigation on both capacity and delay is examined in Section 3.4. In Section 3.5 an analysis is made of RNAV capability versus cost of airborne equipment and finally, in Section 3.6, the RNAV savings available to six individual airlines is analyzed.

3.1 TERMINAL AREA ROUTE LENGTH AND ALTITUDE RESTRICTION ANALYSIS

In a previous analysis [12] preliminary terminal area RNAV designs for seven terminal areas (9 airports) were examined to determine the incremental fuel and transit time effects for several jet aircraft, compared with current VOR/vector procedures in the same terminal areas. The results of this analysis were reviewed by appropriate users and the FAA regional/facility offices affected. Following this review, the recommendations received were combined with the results of the user economic analysis in an iterative process to improve the designs. The design process, which is described in more detail in Section 5.3.1.2 and Reference 2, involved maximizing economic benefits on a traffic weighted basis for both 2D and 3D equipped aircraft, while providing an optimum RNAV design with respect to traffic flow, airspace capacity, and controller workload. Traffic demand data was taken from the North American Edition of the Official Airline Guide [20].

3.1.1 Analysis of Improved 2D RNAV Terminal Area Designs

An analysis of RNAV terminal area designs was performed for nine airports (JFK, LGA, EWR, MSY, MIA, PHL, ORD, DEN and SFO) in order to determine the impact of route length and altitude restrictions on direct operating cost. The incremental fuel and transit time effects were determined for several typical jet and turboprop aircraft (B-747, B-727, DC-8, DC-9, F-28, FH-227, DC-10, Lear 25C). Route length effects were determined for each terminal area in order to estimate the savings available to low altitude reciprocating engine traffic, whose fuel consumption and transit time is less dependent upon altitude than jet aircraft. The analysis compared Phase 3 (post-1982) terminal area designs with current VOR/radar vector routes. These designs are described in detail in Reference 2, and the methodology used in their evaluation is described in detail in Reference 12 and is summarized below.

Altitude restrictions imposed during climbs and descents have an immediate effect on the direct operating costs of turbine aircraft since they impact speed and fuel consumption rates. An automated model was developed for accurately evaluating the fuel and time penalties due to route length and altitude restriction effects in a given terminal area design, and for comparing competing designs in order to evaluate RNAV benefits. The analysis is based on traffic demand, and the results are traffic weighted averages of all routes in the terminal area. The aircraft climb and descent performance data used in the current analysis and the corresponding fuel and time penalty data are listed in Appendix A.

The final 2D RNAV results for each of the nine airports for which final designs were developed are listed in Table 3.1 by aircraft type and are expressed as the fuel (lb) and time (min) benefits per operation (average of arrival and departure). The benefits listed include those attained through use of departure envelopes for higher performance aircraft. The departure envelope concept provides a guidance "floor" representing a specified climb gradient above which each aircraft must operate. The guidance floor is specified by crossing altitude restrictions and may be flown by both 2D and 3D equipped aircraft. In some cases a fixed altitude "ceiling" must also be established in order to provide separation from overflying routes. The gradients defined by the floor and ceiling altitude restrictions may change as each restriction point is passed. This concept is potentially applicable to all terminal areas. However, among the nine airports analyzed, only three (EWR, JFK and LGA) designs resulted in configurations which could be improved through use of high performance departure envelopes. In the other six cases, minimum length departure routes were achieved which would accommodate all aircraft with generally unrestricted climb profiles which could not be improved by establishing a new route. The high performance departure envelopes can generally be utilized by all jet aircraft except heavy B-747s, B-707s and DC-8s.

In order to extend the RNAV terminal area benefits to terminal areas for which RNAV route designs were not developed, a regression analysis was performed. The objective of the analysis was to develop an equation in which the time and fuel benefits per terminal area operation could be computed for each of the eight aircraft types at all of the high and medium density terminal areas. Several terminal area parameters were selected as candidate estimation parameters. Several of these characteristics were related to traffic activity within the terminal area. These included:

- annual instrument operations at the primary airport
- total annual operations at the primary airport
- total annual operations at the satellite airports
- total air carrier operations at the primary airport
- total general aviation itinerant operations at the primary airport
- total itinerant operations at the primary airport

Several other characteristics were related to the existing VOR route structure. These included:

- number of VORTAC stations in the terminal area
- number of high altitude routes
- number of low altitude routes

An analysis of time and fuel benefits versus these individual characteristics of the seven designs (9 airports) yielded little correlation. However, it was determined that a good estimate for extending RNAV benefits to other terminal areas was a least squares fit of benefit versus the difference between air carrier operations and general aviation itinerant operations at the primary airport for five of the terminal areas. New York was excluded because of its singular nature insofar as terminal area complexity is concerned, and Chicago was excluded because of the extremely large number of operations at a multiple runway configured single airport. Although the remaining airports exhibit unique characteristics, summarized in Table 3.2, which make them representative

2D RNAV Benefit - Improvement of 1982 TMA RNAV Design Over 1972 VOR/Vector Design Table 3.1

Harmon H									20	Savings per operation	operation.		rage of a	irrival an	nd depart	(average of arrival and departure) (1) (2)	2)			
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E 30 23.1 23 22.9 24 64.9 25 7.6 7.6 24 60.4 25 9.0 11 14.5 20 6.6 7.2 W 30 25.1 27 3.6 27 7.6 5.6 2.6 2.6 2.6 2.6 2.6 2.6 2.6 2.6 3.0 11 3.1 47.4 3.9 27.0 3.0 11.4 3.7 3.2 4.6 3.0 3.0 3.7 3.0 3.2 3.0		direction	%use		Time (min)	Fuel (lb)	Time (min)	Fuel (Ib)	Time (min)	Fuel (Ib)	Time (min)	Fuel (Ib)	Time (min)	Fuel (Ib)	Time (min)	Fuel (Ib)	Time (min)	Fuel (Ib)	Time (min)	SAVED
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E 56 17.1 22 27.5 23 65.8 24 130.6 2.9 50.6 2.5 7.0 0 12.0 2.0 5.4 1.7 SE 50 13.8 .16 21.0 .17 35.1 .18 62.8 .18 38.7 .18 6.2 .15 10.3 .18 4.7 .18 6.2 .18 38.7 .18 6.2 .15 10.3 .18 4.7 .18 6.2 .15 10.3 .18 4.7 .18 6.2 .18 38.7 .18 6.2 .15 10.3 .16 4.7 .17 .17 .17 .25 .18 .38 .36 .45 .18 .36 .45 .17 .17 .36 .18 .38 .36 .45 .36 .36 .36 .36 .36 .36 .36 .36 .36 .36 .36 .36 .36 .36 .36 .36 .	N N N N N N N N N N N N N N N N N N N	≱ w	88	74.4	.39	42.4	.39	75.0	86.0	137.1	.35	95.2	.35	18.4	79.	27.2	.42	9.9	.40	2.34
SE 50 13.8 .16 21.0 .17 35.1 .18 62.8 .18 38.7 .18 6.2 .15 10.3 .18 4.7 .17 W 50 31.3 .33 .44.8 .35 60.4 .35 79.5 .36 .35 .78 .36 .37 .42 .15.9 .30 .78 .33 .44.8 .35 .45 .27.0 .46 .47.6 .45 .45 .47.6 .45 .47.6 .47.6 .45 .47.6 .47.6 .45 .47.6	MSY	ωZ	25 25	17.1	22.	27.5	.23	65.8	. 12. 17.	130.6	.71	50.6	.25	7.0	0 2.	12.0	.20	5.4	.68	3.22
E 70 43.3 .46 58.7 .46 90.3 .45 127.2 .46 89.6 .45 16.9 .30 27.8 .42 12.3 .44 W 30 67.2 .72 157.3 .71 266.4 .76 101.0 .72 22.0 .40 44.6 .69 19.3 .69 W 75 82.2 .72 93.6 .74 .78 343.4 .94 195.6 .89 30.4 .96 39.0 .40 19.3 .94 SE 25 45.8 .52 .95 .46 46.4 .50 102.3 .44 109.2 .44 28.0 1.68 34.6 .65 .93 .94 .95 .89 30.4 .96 .96 .97 .44 .96 .96 .96 .97 .44 .96 .96 .96 .96 .97 .44 .96 .96 .96 .96 .96 .96 <td>ORD</td> <td>S ≯</td> <td>88</td> <td>13.8</td> <td>33.5</td> <td>21.0</td> <td>.33</td> <td>35.1</td> <td>.35</td> <td>62.8</td> <td>.35</td> <td>38.7</td> <td>.36</td> <td>6.2</td> <td>.15</td> <td>10.3</td> <td>.33</td> <td>4.7</td> <td>.33</td> <td>0.66</td>	ORD	S ≯	88	13.8	33.5	21.0	.33	35.1	.35	62.8	.35	38.7	.36	6.2	.15	10.3	.33	4.7	.33	0.66
W 75 82.2 .92 95.2 .78 343.4 .94 195.6 .89 30.4 .96 39.0 .86 34.6 .87 .81 .87 .81 .89 .81 .89 .80 .89 .80 .89 .80 .89 .80 .89 .80 .89 .80 .89 .80 .89 .80 .80 .89 .80 .	MIA	w ≩	28	43.3	₹. Ľ.	58.7	4. K	90.3	.45 L.	127.2	.76	89.6	54.	16.9	.30	27.8	.42	12.3	4.%	1.24
SW 50 249.5 2.52 243.1 2.57 564.4 2.57 959.5 2.63 607.8 2.63 105.5 2.37 170.6 2.60 76.8 2.65 NE 50 276.3 2.86 385.2 2.81 1039.4 2.82 675.7 2.85 123.3 3.14 188.9 2.99 82.8 2.95 SW 50 309.0 3.17 455.3 3.24 732.8 1.169.0 3.35 764.3 3.37 132.6 2.08 230.1 3.23 109.9 3.37 NE 50 165.9 1.77 238.1 1.69 383.1 1.69 3.35 1.77 402.3 1.79 67.2 .67 111.0 1.45 54.5 1.56 SW 50 179.3 1.92 24.1 1.77 480.6 2.11 71.7 80 128.5 1.95 1.97 NE 50 72.6 1.01 135.3 1.05 <td>SFO</td> <td>S ≷</td> <td>75 25</td> <td>82.2 45.8</td> <td>25.</td> <td>95.2</td> <td>8.4</td> <td>169.4</td> <td>.50</td> <td>343.4</td> <td>2.4.</td> <td>195.6</td> <td>8. 4.</td> <td>30.4</td> <td>.96.</td> <td>39.0</td> <td>.85</td> <td>24.2</td> <td>.81</td> <td>4.10</td>	SFO	S ≷	75 25	82.2 45.8	25.	95.2	8.4	169.4	.50	343.4	2.4.	195.6	8. 4.	30.4	.96.	39.0	.85	24.2	.81	4.10
SW 50 309.0 3.17 455.3 3.24 732.8 3.22 1169.0 3.35 764.3 3.37 132.6 2.08 230.1 3.23 109.9 3.37 NE 50 165.9 1.79 238.1 1.59 383.1 1.68 586.5 1.71 402.3 1.79 67.2 .67 111.0 1.45 54.5 1.56 1.56 SW 50 179.3 1.92 264.3 2.03 489.2 1.97 898.8 2.13 480.6 2.11 71.7 .80 128.5 1.92 62.1 1.97 1.97 NE 50 92.6 1.01 135.3 1.05 254.2 1.05 461.8 1.14 230.5 1.07 33.2 .16 61.4 .88 29.1 .89	JFK	S S		249.5	2.52	243.1	12.83	564.4		959.5	2.63	607.8	2.63	105.5	2.37	170.6	2.60	76.8	2.65	9.91
SW 50 179.3 1.92 264.3 2.03 489.2 1.97 898.8 2.13 480.6 2.11 71.7 .80 128.5 1.92 62.1 1.97 898.8 2.13 480.6 2.11 71.7 .80 128.5 1.92 62.1 1.97 NE 50 72.6 1.01 135.3 1.05 254.2 1.05 461.8 1.14 230.5 1.07 33.2 .16 61.4 .88 29.1 .89	LGA	SW		309.0	3.17	455.3	.59	732.8		1169.0	3.35	764.3	3.37	132.6	2.08	230.1	3.23	109.9	3.37	8.37
	EWR	S Z		179.3	1.92	264.3	8.8	489.2	1.97	898.8	2.13	480.6	2.11	71.7	.80		1.92	62.1	1.97	2.99

(1) The positive values indicate a benefit to the airpsace user. (2) all aircraft types do not operate at all airports shown even though data is presented for that operation

TABLE 3.2

CHARACTERISTICS OF TERMINAL AREAS ANALYZED

NEW YORK	- Complex metroplex area which has three major airports and several major satellite airfields.
CHICAGO	 Metroplex area which is dominated by one airport with a complex runway structure.
SAN FRANCISCO	 One major airport with several major satellites lined up around San Francisco Bay. Noise and terrain problems as well.
MAMI	 No terrain problem but a very concentrated traffic area to the north. Unbalanced traffic distribution. One major satellite airfield at Fort Lauderdale.
PHILADELPHIA	- A major terminal area which lies between two metroplex areas. (New York and Washington)
DENVER	- Severe terrain problems to the west and a two directional (east-west) traffic flow.
NEW ORLEANS	 Medium density terminal area with a perpendicular runway structure and hemispherical traffic flow.

of most of the categories of large airports throughout the United States, the RNAV benefits at each was strongly correlated to the dominance of air carrier traffic over general aviation traffic. The amount of air carrier traffic is generally indicative of the amount of high altitude arrivals and departures, and the amount of general aviation traffic is generally indicative of the amount of low altitude traffic. Since fuel and time benefits are a result primarily of realignment of terminal area routes to better accommodate high altitude traffic, the correlation is not unexpected. The estimated benefits for each airport and each aircraft type are given in Table 3.3.

(Fuel and Time Savings per Operation) Terminal Area Benefits Table 3.3

	America d		۵	6 - 747			DC - 10	- 10	-		DC - 8	80			8 - 727		
	1974 Hinerant		uel-lb.	Į.	ime -Min.	3.4	Fuel-tb.	Time-Min.	-Min.	Fue	Fuel-ib.	Time	ime-Min.	Fuel-tb.		Time-Min.	in.
y indu	Carrier Minus General Aviation	IMA Analysis	Estimate	TMA	Estimote	TNA	Estimo	TMA te Analysis Estimate	stimate A	TMA Analysis 6	Estimate Analysis		Estimote	TMA Analysis Est	timate A	MA notypis Es	timate
EWR LGA JFK		683.9 877.8 989.5		3322		355.6 583.3 641.8 59.3		1.59 2.58 2.74 .27		371.7 558.0 585.3 47.8		2.45 2.69 2.09		346.7 346.7 32.9		* + 2 2	
MEN	57,000 45,000 61,000 156,000	213.5 116.7 79.7 169.0 283.1	13.4 124.4 136.4 207.5 259.8	8 2 2 2 8	ज्ञ स श्र	110.7 86.9 57.3 83.0	82.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.	\$ 5 5 5 K	28885	58.0 58.0 110.4	28.5 4.8 5.9 139.0	84. 35. 53. 17.	8 8 8 8		8.64 6.25	2 8 8 3 2	4488E

			ă	6-30			-	87 - 4			רביי ופר ופרו					
	Annual 1974 Itinerant		Fuel - ib.	17	Time - Min	Fu	Fuel - Lb.	-i.	Time - Min.	F.6	Fuel - Lb.	line - Min.		Fuel - Its.		e - Min.
Airport	Carrier Minus General Aviation	IMA Analysis	Estimate	7MA Analysis	Sstimate	TMA	TMA TMA Analysis Estimate Analysis		Estimate Analysis	TMA relysis	IMA Stirrate Analysis	A yais Estimore	A And	TMA TMA TMA TMA Analysis	IMA Analysis	Estimate
LGA JFK				2.48 2.69 2.69		95.0 170.6 179.8 15.8		2.34	4 8 7	5.8.5. 6.5.8.2.	1,43 2,46 2,80 2,80 ,25	5.90%	M X Z	2. 5. 4. 8. 9. 9. 9. 9. 9. 9. 9. 9. 9. 9. 9. 9. 9.	2.78	
MSY MSY MIA SFO	57,300 45,000 61,000 156,000	39.6 24.6 25.5 25.5 25.5 25.5 25.5 25.5 25.5 25	43.4 38.9 57.5 67.0	द घं थ घं छ	ল্পাসভাস	26.7 27.4 14.5 32.8	23.4 23.8 31.9 37.9	74.8. 22.08.88	8888	0.0 0 40 0.4 0.4 0.0	9.2 4.3 9.5 2.3 15.4 5.7 75	24886	- 28	16.4 15.1 78.6 14.3 9.2 15.4 16.4 22.2 29.8 27.1	# & E B #	* # 2 2 8

In order to apply the benefits for each aircraft type to each airport, it was necessary to develop an estimate of the number of operations expected by each aircraft type in each terminal area in the period of interest. Calendar year 1984 was chosen as the baseline period in which to assess the benefits of an all RNAV vs all VOR environment. This year was chosen since it falls within the post-1982 implementation phase described by the Task Force and represents a reasonable estimate of the earliest date at which an all RNAV environment might be expected to exist. Estimates of the distribution of aircraft operations by type of aircraft were developed from two sources. An equipment forecast for the top 25 air carrier airports, 1975-2000, was provided by the Office of Aviation Policy of the FAA, which was generated as a part of their study of the Upgraded Third Generation ATC System. This information, together with data from Terminal Area Forecast [21], was used to develop the estimates. An estimate was made for the primary airports in each of the terminal areas defined by the Task Force as high or medium density. Only CONUS terminals were considered, and Seattle and Ft. Lauderdale-Hollywood airports were added to the Task Force list due to traffic projections. The resulting list of 60 major airports, and the estimates of the number of operations in CY1984 for each aircraft type are given in Table 3.4. Since data were not available concerning distribution of business jet operations by type, they are included as a single category in Table 3.4. Estimated business jet operations were derived as a fixed percentage of forecast general aviation itinerant operation in CY1984. Information provided by the National Business Aircraft Association (NBAA) on the number of business aircraft flights was compared with total itinerant general aviation operations from Reference 21 to develop an estimate of this percentage (4.85%). The airports in Table 3.4 are grouped according to whether they are air carrier or general aviation operation dominant.

The regression equations were used to estimate the fuel and time benefits per operation at each airport for each type of aircraft, and the total annual benefits were determined by applying the per operation benefit data to the operations listed in Table 3.4. A summary of total annual fuel and time savings at the 60 major airports studied is given in Table 3.5. The benefits for each aircraft type (B-747, DC-10, DC-8, etc.) were assumed to be representative of the benefits available to all other aircraft within the same category:

4 engine Wide Body - B-747
3 engine Wide Body - DC-10
4 engine - DC-8
3 engine - B-727
2 engine - DC-9
business jet and turboprop-Lear Jet, F-28, F-227

The business jet and turboprop category represents a weighted average benefit of the three types studied, with each of the three representing a category of aircraft to which other types were assigned. The fuel and time savings of business jets at each airport were approximated by interpolating between the savings computed for the Lear Jet and the F-28 on the basis of average maximum engine thrust. Thrust data was obtained from Reference 22, and distribution of types of aircraft was based on NBAA data. The fuel and time savings of business turboprops were approximated in the same way as a percentage of F-227 savings.

Table 3.4 Estimated 1984 Terminal Area Operations by Type of Aircraft

				Number of	Operations	in 1984 (×1000)	
		AIRPORT	4 Engine Wide Body	3 Engine Wide Body	4 Engine Reg. Body	3 Engine Reg. Body	2 Engine Reg. Body	Business Jets and
		EWR LGA JFK ORD	13 71 50	45 42 71 118	16 43 25	99 135 85 235	108 113 64 192	Jets and turbprops 4 4 3 3
nt		MSY DEN PHL MIA SFO	3 6 8 22 42	24 46 29 76 76	3 11 16 29	64 107 84 129 155	78 126 132 72 80	2 2 5 3
Air Carrier Dominant	~	ATL BOS CLE DFW DTW IAH LAX PIT STL DCA BDL BUF CVG IND JAX MCI LOU MSP MCO PDX SMF TPA SEA	12 18 4 9 8 6 60 8 1 ——————————————————————————————————	100 40 28 56 42 27 116 20 31 28 ———————————————————————————————————	12 15 1 13 16 8 37 8 7 3 4 2 2 1	218 95 43 209 62 94 167 61 88 112 26 43 28 18 24 99 16 62 32 73 18 63 61	248 117 63 141 .72 57 83 164 127 93 47 31 67 75 27 61 56 54 31 46 28 23 20	6 3 4 6 5 3 3 4 5 5 4 6 5 5 7
General Aviation Dominant	~	MEM PHX SAT N SAT	12 18 4 9 8 6 6 6 8 1 	9 11 26 — — — — — — — — — — — — — — — — — —	2 17 3 38 15 6 ——————————————————————————————————	34 44 51 70 55 42 43 37 20 18 18 18 39 7 44 39 22 33 54 40 16 12 32 23 34 36 61	99 67 21 20 36 9 50 24 40 44 15 20 23 33 84 47 32 33 4 19 32 40 22 71 18 15 20	10 13 77 5 9 4 3 7 7 7 10 6 15 5 7 4 7 7 8 2 7 7 5 9 8 8 8 8 9 9 9 9 9 9 9 9 9 9 9 9 9 9

TABLE 3.5 2D RNAV FUEL AND TIME ANNUAL BENEFIT SUMMARY

AIRPORTS	4 Eng.	W.B.	3 Eng.	W.B.	4 Eng.	R.B.	3 Eng.	R.B.	2 Eng. R.B.		Business Jet and Turboprops	Jet and
	Fuel (1000lb)	Time (1000min)	Fuel (1000lb)	Time (1000min)	Fuel (1000lb)	(1000lb) (1000min) (1000lb) (1000min)	Fuel (1000lb)	Time (1000min)		uin)	Fuel (1000lb)	Time (1000min)
NEW YORK LGA EWR JFK	79,856	214	690'98	375	31,115	140	97,542	402	58,096	592	919	20
CHICAGO ORD	3,835	14	266'9	32	1,195	7	7,732	63	4,339	05	18	-
MSY, DEN, PHL, MIA, and SFO	17,584	51	28,608	133	609'9	33	35,612	265	24,086	213	124	2
23 AIR CARRIER DOMINANT AIRPORTS	37,172	1111	99,432	390	17,102	88	129,300	1,020	153,993	979	1,034	94
28 GENERAL AVIATION DOMINANT AIRPORTS	2,250	4	5,835	21	3,402	20	31,476	311	26,738	%	675	61
TOTAL	140,697	394	226,941	156	59,423	288	301,662	2,172	267,252	1,930	2,467	16

Projected annual savings in dollars, based on the fuel and time savings described in this section, are developed in Section 5.1.1, and additional benefits which are realized through the ability of an RNAV equipped aircraft to better adhere to the desired path are also discussed.

3.1.2 VNAV Descent Analysis

Vertical navigation (VNAV) provides the capability for optimizing the point where descent should be initiated. This can be very advantageous to the user over present procedures since the point at which descent is initiated would be later than would occur otherwise. Since a given type of descent requires the same distance regardless of when it is initiated, an early descent requires that the aircraft cruise at a lower altitude until the navigation fix where the descent altitude is required to be achieved is reached. Through the use of VNAV capability, this distance increment may be flown at the higher cruise altitude, which is always more advantageous in terms of fuel, and usually also in terms of time. This advantage applies both to the initial descent segment from cruise altitude to the initial transition point, and to descents from one level to another in terminal area operations. The VNAV advantage obtained for both of these descent cases has been evaluated and is presented in this section.

The initial descent segment (from cruise altitude) was evaluated by comparing fuel and time requirements of a VNAV-initiated high speed descent procedure to the requirements of candidate standard descent procedures. Present descent procedures range from the most costly, the initiation of descent when descent clearance is received from ATC, to the least costly, where standard airline "rules of thumb" regarding descent initiation point are used. In order to provide the most conservative estimate of VNAV advantage, "rules of thumb" for each of the five airline aircraft studied have been chosen very carefully in order to provide the best descent possible while insuring arrival at the fix at the desired altitude. The rules chosen do not reflect the wind estimation process, which requires either that wind effect estimates be included or that descents be routinely initiated somewhat earlier. The typical "rule of thumb" involves multiplying the altitude to be lost (in thousands) by some standard factor and adding a standard increment (e.g., "three times altitude difference plus ten"). These "standard factors" have been derived for each aircraft by computing a least-square linear fit to the aircraft descent performance. The resulting data shown in Table 3.6, includes, for each objective altitude, the altitude to be lost (AH in thousands of feet) and the distance required to lose it (nm). The slope and intercept parameters of the straight-line are shown below the data values. From these fit values the "rule of thumb" values were selected according to the following criteria:

Multipliers are even numbers

Added increments are multiples of five miles

Resulting descent distance must assure arrival at the desired altitude

Table 3.6 Derivation of Descent Rules of Thumb (ΔH in thousands of feet, ΔD in miles)

				AIRCE	RAFT					
CRUISE ALTITUDE		C-9 ,000		727 ,000	_	C-8 ,000	-	747 ,000		-10 ,000
Objective Altitude	ΔН	ΔD	ΔН	ΔD	ΔΗ	ΔD	ΔН	ΔD	ΔΗ	ΔD
0 5000 10000 15000 20000	31 26 21 16 11	139.5 127.0 114.4 99.6 89.2	35 30 25 20 15	109.0 96.5 82.7 61.8 52.1	39 34 29 24 19	141.5 129.0 112.8 94.1 81.7	39 34 29 24 19	122.5 111.5 95.6 73.5 61.0	39 34 29 24 19	128.0 115.5 98.9 77.8 65.1
Slope Intercept		.56 .2		.97	3 22	.09		.22		.27
Rule of Thumb	dif	ference thousands	dif in	altitude ference thousands = miles	dif in	ference thousands	dif in	ference thousands	dif in	ference thousands

In order to apply the rules of thumb and compare the results with the VNAV-initiated descent procedure for the initial descent segment from cruise altitude, the post-1982 RNAV terminal area designs described in earlier sections were used. For each terminal area, each arrival route was examined to determine the altitude of the first waypoint where a ceiling altitude restriction was in effect. That altitude was then used for the comparison. The results for each route were multiplied by the traffic demand on that route, summed and divided by total traffic to yield a traffic-weighted average VNAV benefit. Results for each of the two terminal route flow configurations were added in proportion to the percent time each configuration is in use in order to yield a single benefit value (fuel and time) for each aircraft at each terminal. These results are presented in Table 3.7.

The effect of using VNAV on terminal area descent segments (from one level to another in terminal area operations) were evaluated using the RNAV terminal area designs. For these descent segments, where usually only a few thousand feet of altitude are lost, it was assumed again that the VNAV-equipped aircraft would descend at the last timely moment, while the non-VNAV aircraft would initiate descent immediately after crossing each waypoint when the altitude bound at the next waypoint is lower than the present bound. This is analogous to the present common practice of beginning descent during terminal maneuvers immediately when cleared for descent. As before, the non-VNAV aircraft cruises at the lower altitude while the VNAV aircraft cruises higher, deriving a benefit. Each arrival route, from the first altitude-restricted waypoint (same as above) into the final approach fix was evaluated to determine VNAV benefit. Many of

Table 3.7 VNAV Descent Procedure Impact

(Fuel - Pounds, Time - Minutes)

			B-747	DC-10	DC-8	B-727	DC-9
MSY:	Fuel:	Init. Desc. Final Desc.	47.0 26.0	44.5	78.0 12.4	44.5 9.6	51.0 7.1
		TOTAL	73.0	54.8	90.4	54.1	58.1
	Time:	Init. Desc. Final Desc.	.133	.220	.337	.329	.502
		TOTAL	.175	.254	.372	.370	.546
DEN:	Fuel:	Init. Desc. Final Desc.	60.0 7.3	67.2 3.2	88.8	27.2 3.8	32.8 3.3
		TOTAL	67.3	70.4	92.4	31.0	36.1
	Time:	Init. Desc. Final Desc.	.104	.241	.295 .013	.176	.023
		TOTAL	.117	.252	.308	.205	.300
PHL:	Fuel:	Init. Desc. Final Desc.	18.9 41.1	33.9 17.7	65.3 21.0	41.1 14.4	49.8
		TOTAL	60.0	51.6	86.3	55.5	60.0
	Time:	Init. Desc. Final Desc.	.059	.161 .057	.286 .057	.0308 .030	.496
		TOTAL	.124	.218	.343	.368	.557
MIA:	Fuel:	Init. Desc. Final Desc.	26.0 37.7	42.0 15.7	68.0 17.3	32.0 13.8	38.0 10.1
		TOTAL	63.7	57.7	85.3	45.8	48.1
	Time:	Init. Desc. Final Desc.	.085	.203 .053	.304	.252	.394
		TOTAL	.149	.256	.358	.314	.459
SFO:	Fuel:	Init. Desc. Final Desc.	23.5 27.3	39.5 7.4	67.0 9.2	34.2 9.0	41.5 7.2
		TOTAL	50.8	46.9	76.2	43.2	48.7
	Time:	Init. Desc. Final Desc.	.079	.192 .024	.299 .026	.265 .040	.418 .046
		TOTAL	. 123	.216	.325	.305	.464
ORD:	Fuel:	Init. Desc. Final Desc.	26.5 45.1	42.5 16.6	68.5 19.2	32.0 14.7	37.5 10.7
en i		TOTAL	71.6	59.1	87.7	46.7	48.2
91	Time:	Init. Desc. Final Desc	.086	.205 .055	.305 .056	.249	.388
		TOTAL	.160	.260	.361	.312	.455
JFK:		Init. Desc. Init. Desc.	83.0 .163	73.0 .262	98.5 .332	43.5 .266	46.5 .401
LGA:		Init. Desc.	29.0 .099	45.5 .225	69.5 .313	27.0	33.5 .334
EWR:		Init. Desc.	40.5	54.0 .238	74.5 .308	25.5 .189	28.5

these routes have several descent segments. The benefit for each route was combined with other routes, as before, to yield a traffic weighted average for each terminal studied. Recent modifications to the New York terminal area designs prevented inclusion of final descent data for those three airports in Table 3.7. Their absence has no impact on the regression analysis for extrapolation purposes, described below, since New York and Chicago were not intended to be included. The results of the studies of VNAV benefits in each of the two cases described above are presented in Table 3.7, where fuel values are in pounds and time in minutes. Most of the fuel values are on the order of fifty to sixty pounds and most time values center around 0.3 minutes. As discussed before, these values are quite conservative since the rule of thumb initial descent procedures represent a very optimistic conventional descent; many airline operators do not adhere to such stringent and precise descent procedures.

An attempt was made to determine a regression equation which could be used to relate VNAV benefit to operation rates at each airport. The same methods as were used in Section 3.1.2 were applied, i.e., only five lower density airports were used, and the regression parameter used was 1974 itinerant operations minus GA itinerant operations. The resulting regression equations were almost totally insensitive to the operations count parameter. Therefore, it was not deemed to be advisable to apply the regression equations in the manner done in Section 3.1.2. Rather the traffic weighted combination for those five airports was formed, based upon the projected 1984 annual operations for each aircraft at each of the five airports. These values were then multiplied by the number of operations for each aircraft type projected in 1984 over the sixty major and medium density terminals. These results in terms of fuel and time, are presented in Table 3.8.

Table 3.8 Aggregate Annual VNAV Descent Benefit (Fuel-Pounds, Time-Minutes)

	B-7	47	DC-	10	DC-	-8	B-7	27	DC-	9
	Fuel	Time	Fue1	Time	Fuel	Time	Fue1	Time	Fue1	Time
Traffic Weighted Average	56.7	.132	55.8	.239	81.3	.340	44.7	.305	49.9	. 459
Sixty Airport Ops (1984)	406,	000	1,253	,000	419,	000	3,770	,000	3,634	,000
Total Aircraft Benefit	23.0m	54K	69.9m	300k	34.1	1142k	168.5	m 1150k	181.	3m 1668k
Aggregate Benefit: Fuel 47	76.8 Mi	llion	Pound	ls, Ti	me 331	4 Tho	usand	Minute	S	

3.1.3 Effects of Alternate Descent Procedures on Benefits

The 2D RNAV benefits and VNAV benefits computations for arrival aircraft were performed based upon the assumption that standard high speed descent procedures are used. Since airlines do not necessarily adhere to the high speed descent schedule, but conduct alternate types of descents, it is of interest to determine what, if any, effect the selection of alternate procedures would have on RNAV and VNAV benefits. The other standard descent procedure used in published handbook data is the long range (fuel economic) descent, which is considerably different and represents the probable opposite extreme which would ordinarily be used. Since performance data was available for this procedure (for four aircraft types only), it was selected as the other candidate for comparative purposes. To illustrate the different speed schedules used for the two procedures, the DC-8 procedures are as follows:

HIGH SPE	ED DESCENT	LONG RAN	GE DESCENT	
Descent Speed	Altitude	Descent Speed	Altitude	
Mach 0.83	to 26,600 ft	Mach 0.78	to 37,450 ft	*
340 KIAS	to 10,000 ft	250 KIAS	below 37,450	ft
250 KIAS	below 10,000 ft			

The differences in the procedures in terms of performance are that the high speed procedure requires less time but more fuel than the long range procedure. For example, for a DC-8 descent from FL390 to sea level, the long range descent saves 452 lbs of fuel, but extends descent time by 3.7 minutes.

The RNAV terminal area analysis presented in Section 3.1.1 for arrivals was repeated using the long range descent data for those aircraft for which such data was available (DC-9, B-727, DC-8 and DC-10). The purpose of this analysis was to determine if the RNAV savings over VOR are comparable for this descent procedure to the savings of RNAV over VOR for the high speed descent procedure. Since these two procedures represent the extreme in different descent procedures, the results would then be valid for other procedures. The results of this study are presented for each city and each aircraft along side of the original RNAV benefits computed using the high speed descent procedure in Table 3.9. In that table the data shown are the combined data for both flow directions. An examination of the resulting data shows that the differences are very minor and do not show any consistent trend. Therefore it may safely be assumed that the choice of either of these descent procedures, or others in between, would have no significant effect upon the benefits to be derived from RNAV terminal procedures over VOR procedures as presented in Section 3.1.1.

In Section 3.1.2 the benefits available to VNAV equipped users resulting from the use of VNAV guidance to initiate descent have been presented. In that analysis, the difference between the use of VNAV guidance as opposed to "rule of thumb" operational techniques has been computed. By allowing the initiation of descent at the last possible moment, VNAV prevents unnecessary cruise at a low altitude, which occurs when descent is initiated early as in a conventional descent procedure. If descent procedures other than the high speed descent, such as the long range descent, were to be substituted, VNAV would still provide the same advantage, i.e., it would allow more accurate control over the descent initiation point than is available through conventional means. Therefore, the VNAV benefit computed in Section 3.1.2 would be realized regardless of the particular descent procedure employed. Results of a real time terminal area simulation conducted at NAFEC [11] indicated that pilot selection of VNAV descents had no impact on the controllers handling of traffic or the ability of the aircraft to realize 2D benefits.

2D RNAV Arrival Benefit - Comparison of Benefits of RNAV Over VOR Achieved with High Speed and Long Range Descent Procedures Table 3.9

		DG	6-20			B-727				8-2g				ă	06-10	
	High	High Speed	Long	Long Range High Speed Long Range	High	peed	Long	Sange	High	High Speed Long Range	Long	Sange	High	High Speed	Long	Long Range
Airport	H. (4)	lime (min	Fue	Time Fuel Time	Fue E.E.	Time (min)	F. (4)	(min)	Fuel (Ib)	Time (min)	Fuel (Ib)	Time (min)	Fuel (Ib)	(min)	Fve (B)	Time (min)
FE	38.5	.417			53.8	.434	64.9	.420	91.3	1		.413		.443	1	1
DEN	-3.0	089	1.2	010.	2.8	2.8090	3.6	.012	.012 -17.6092	092	3.6	.002	-19.9	160	9.0	.002
MSY	32.9	.377	34.5	.425	50.7	.393	50.7	.346	74.5	.429	73.8	.436	94.8	.418	93.5	.438
ORD	51.8	.578	51.5	.572	76.3	.599	8	.602	124.8	.628	125.3	.620	137.9	.639	140.2	.636
MIA	6.98	.992	86.5	.942	124.8	1.015 147.3	147.3	876.	.978 202.9	896.	204.7	.949	208.1	.995	212.2	.958
SFO	123.9	1.420	125.0	1.438	178.7	1.403	6.161	1.403 191.9 1.400 263.3	263.3	1.388 264.0	264.0	1.390	301.7	1.378	378.0	1.385
JF.K	344.1	3,662	335.6	3.690	482.1	3.664	540.8	3.757	762.6	482.1 3.664 540.8 3.757 762.6 3.681 751.0 3.705	751.0	3.705	826.3	3.733	824.6	3.734
LGA	408.4		4.310 395.9	4.131	579.4	4,455	1.879	4.395	953.9	579.4 4.455 678.1 4.395 953.9 4.452 961.0 4.378 980.5	0.196	4.378	980.5	4.592	4.592 1006.6	4.479
EWR	134.3		131.2	1.346 131.2 1.249 193.6 1.421 226.4 1.361 323.8 1.443 323.7 1.379 326.2 1.491 330.0 1.413	193.6	1.421	226.4	1.361	323.8	1.443	323.7	1.379	326.2	1.491	330.0	1.413

3.2 HIGH ALTITUDE ROUTE LENGTH ANALYSIS

Studies aimed at quantifying RNAV enroute distance savings have been performed using a NAFEC designed charted RNAV structure compared with the existing VOR structure [4]. In these studies traffic was distributed over VOR routes between each airport pair according to historical data and was distributed over alternate RNAV routes between each airport pair according to criteria developed as part of the study. Since the traffic distribution over the VOR structure reflected actual, real world traffic situations, it included some percentage selection of VOR routes for purposes of minimizing wind mile distance. The distribution of traffic over RNAV routes, however, was based only on accommodating the traffic while minimizing ground mile distances, and also did not include the effect of restricted areas on the RNAV route structure. The RNAV benefits measured were based on the traffic weighted average difference in ground miles between the distributed VOR traffic and the distributed RNAV traffic between each airport pair in the structure. This method of analysis left unanswered two basic questions:

- (1) What is the traffic weighted average wind mile difference between the distributed VOR and RNAV traffic in the structures analyzed?
- (2) What would the traffic weighted average wind mile difference be between distributed VOR and RNAV traffic if the RNAV route design and traffic distribution had also been based on consideration of real world selection of weather routes and/or an RNAV structure which was constrained by restricted areas?

The current study was conducted to provide a calibration of the NAFEC study in order to determine what biases, if any, the consideration of weather route selection and restricted areas would have on the percent enroute distance savings expected from RNAV, as determined in the NAFEC study.

3.2.1 Alternate Weather Route Analysis

3.2.1.1 Study Approach

There are a variety of software capabilities which are currently being used to generate flight planning information, in a weather environment, whose results could be applicable to this study. Many of the major air carriers, as well as several independent service organizations have "flight planning" programs, one of the basic objectives of which is to compare the various alternate routes and establish that which is most favorable. Because of the existance of this flight planning capability, the option of developing the desired software was rejected pending verification that a suitable subcontractor could be found. Communications were initiated with several of the organizations with the appropriate capabilities for this study. These efforts resulted in isolating two organizations that were both willing and capable of providing the required support:

United Air Lines (UAL), and Lockheed (JETPLAN Service).

The capabilities of these organizations, designed to satisfy different requirements, also varied. The UAL program evaluates and selects a "best route" from a pre-defined set of routes; but in so doing, provides the results (wind miles) of each route comprising the given set. The JETPLAN program is more flexible in that a pseudo optimum route is developed for the prevailing conditions each time the program is run. This results in considerably more alternate routes being evaluated, nominally, but only the data relevant to the selected route is made available to the user. It was established that the JETPLAN service would provide a better means of analyzing a pre-planned direct environment, but that the UAL program was better suited to addressing a charted route environment, which was of primary interest in this study. The UAL program was therefore selected.

The definition of route structures to be simulated involved selection of appropriate airport pairs, and the specification of VOR and RNAV weather routes for each airport pair.

The selection of the airport pairs was constrained to some extent by UAL's desire that the results of this study be compatible with independent UAL evaluations of RNAV operations. A set of six airport pairs whose mix of distance, orientation and location was compatible with the study objectives was mutually agreed upon for the subsequent flight planning analysis. These airport pairs are listed below:

(1) Chicago-O'Hare (ORD) to/from Denver (DEN)

(2) Chicago-O'Hare (ORD) to/from Los Angeles (LAX)

(3) New York-Newark (EWR) to/from Chicago-O'Hare (ORD)

4) Washington, D.C.-Dulles (IAD) to/from Los Angeles (LAX)

(5) Miami (MIA) to/from Cleveland (CLE)

(6) New York-Kennedy (JFK) to/from Seattle (SEA)

Having selected this set of airport pairs to be examined in this study the next task was to define the corresponding route structures. The VOR alternate routes currently stored and flown by UAL seemed to satisfy the VOR route structure requirements. Since UAL developed and utilized a sophisticated flight planning capability over a period of years, it is not unreasonable to expect that their choice of VOR routes retained for evaluation by their flight planning program would approximate an optimum set of VOR routes. However, since a complete RNAV alternate route structure does not currently exist, the best method to define a reasonable and representative RNAV structure for use in this study was not obvious. The definition and justification of a single representative RNAV structure would have been difficult, if not impossible to accomplish. As a result several RNAV route structures were included in the analysis.

In all, two VOR (differing only in the terminal area) and four RNAV route structures were created for each airport pair. An additional VOR and two additional RNAV structures were created as combinations of the basic sets subsequent to the processing of the results. While the redundancy of structures was considered necessary to minimize the possibility of unrepresentative results, the various structures were also developed based upon different constraints and objectives in an effort to extract the maximum information from the study. The characteristics of each structure and the manner in which they were developed are presented in the next section.

3.2.1.2 Data Base Development

The data base associated with the alternate weather route analysis included that information required to describe each route of each postulated route structure and the data necessary to characterize the weather of those days selected to be used for flight planning simulations. To place the weather requirements in the proper perspective it is helpful to summarize the UAL flight planning program. This program can best be described as a two phase process. The first phase is used to select the desired route from the set of candidate routes of a given structure stored by UAL for the designated airport pair. The second phase defines the optimum vertical flight profile for the route selected in Phase 1, considering type of aircraft, gross weight at takeoff, criteria for optimization such as minimum fuel or minimum flight time and other factors. While this final flight profile was of interest, it was the results of the first phase which were directly utilized in this study. The output format of both Phase 1 and Phase 2 are illustrated in Appendix B of this report.

Weather Sample

The effect of winds are simulated for the Phase 1 process by defining a wind vector for each route segment. The characteristics of these vectors are determined by processing weather data obtained from the National Weather Service at 12 hour intervals, updated when appropriate, based upon communications from UAL weather reporting flights. This information is processed, producing a weather data base 500, 400, 300, 250 and 200 millibars (approximating 18, 24, 30, 35 and 39K foot altitudes, respectively) on a grid of roughly 2-1/2° latitude by 5° longitude. For the Phase 1 (route selection) process this information is converted in an average wind by weighting the winds at 300, 250 and 200 millibars 22, 50 and 28 percent, respectively. Linear interpolation between the grid points (2-1/2° lat-5° long) is utilized to obtain the desired wind information at the beginning and end of each route segment. The average of these values is then applied to the entire segment. This process is repeated for winds during each cruise segment of the route. The climb portion of each flight is simulated as identifying the wind vector at this fix which is closest to a point 175 nm from the departure airport, but is not greater than 175 nm from the airport. The magnitude of this vector is halved and it is applied during the first 30 minutes of flight. The aircraft heading during climb is approximated by setting it equal to the heading between the airport and the first fix. The descent leg of each flight is simulated in a similar manner except that the distance is changed from 175 nm to 140 nm and the duration of the descent is set at 15 minutes. A more sophisticated and accurate technique is used in the Phase 2 program which generates the actual flight plan used by UAL in their operations.

Route Structure Design

The route structure data base requirements are summarized below:

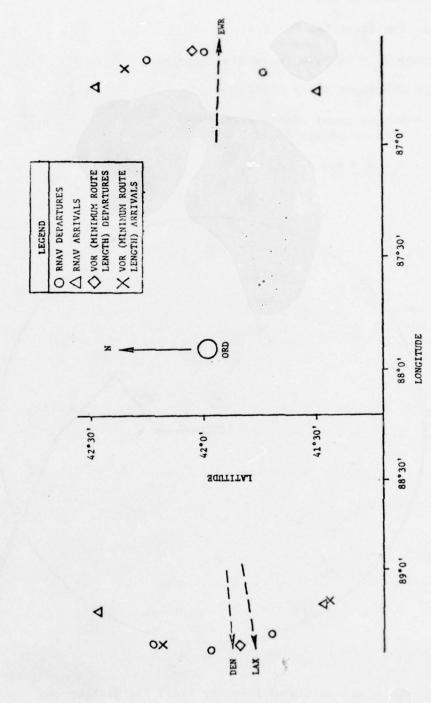
- Definition of the terminal waypoints (both RNAV and VOR).
- (2) Normalization of the VOR terminal points and terminal segment lengths.
- (3) RNAV route structure development.
- (4) Route structure implementation at UAL.
- (5) Validation of the data base.

The purpose of the first two tasks was to assure that the ground rules and assumptions underlying the RNAV and VOR structures were compatible. The RNAV route structure development involved both the satisfying of specific objectives associated with each of the structures, as well as specifying the routes.

To insure compatibility between studies addressing RNAV benefits which occur in the terminal areas and those of this study dealing with enroute benefits, efforts were made to utilize a common interface. In this case, this interface was the terminal area waypoints. Information was obtained identifying terminal waypoint locations which reflected the prevailing wagon wheel terminal area design philosophy for each of the nine airports examined in this study. An example of the resulting arrival and departure locations is illustrated in Figure 3.1. The location of the VOR arrival and departure fixes associated with the minimum ground mile VOR route is also shown. The location of these points was approximated by the intersection of the terminal area circle (at a radius of either 45 or 60 nm, depending upon the specific terminal area) and the estimated flight path between the airport and UAL's first VOR enroute fix outside of the terminal area circle.

Since the UAL flight planning program considers the entire flight, including both terminal area and enroute components, it was necessary to adjust the terminal area route structure to null out any bias which could influence the route selection process which, for purposes of the analysis, were considered to be an exclusive enroute effect. This was accomplished by simulating the airport location at the center of the terminal area and providing a great circle route segment from each terminal area fix to the airport. These simulated conditions resulted in all routes having common terminal area route lengths, either 45 or 60 miles depending on the terminal area.

One exception was made to this approach in order to reflect the impact of the wide variation in VOR terminal route lengths that are prevalent at some airports. This was accomplished by incorporating a terminal area route distance into the UAL route structure data base which, when added to the great circle distance between the terminal area waypoint and UAL's first enroute fix, would equal the actual distance identified by UAL between the airport of the first fix of each VOR route. This procedure is illustrated in Figure 3.2. Thus two VOR route structures different only in terminal area route lengths, were defined for each airport pair: The VOR A structure with terminal area routes lengths set equal to those of corresponding RNAV routes (45 or 60 nm) and the VOR B structure which approximated the actual distance between the airport and UAL's first fix, as currently flown. To account for the VOR terminal route distance variations without jeopardizing the validity of the VOR-RNAV enroute comparisons required the use of both structures. The VOR B structure was used to select a VOR route on any given day based on minimum wind miles, while the VOR A structure was used to define the RNAV compatible wind miles for that route corresponding to the VOR B selected route. In most cases the same route produced the minimum wind miles in both the VOR A and VOR B structures. However, the results of this process was to produce average VOR wind miles slightly larger than the values which would have been derived using only the VOR £ minimum wind mile routes. A VOR C route structure name was established to identify the results of the



Alternate Weather Route Study, Chicago Terminal Area Arrival and Departure Points Figure 3.1

Point A is the intersection of UAL VOR route and 45 or 60 mile circle.

Point B is the first UAL VOR fix.

 d_1 through d_5 = UAL VOR route distances

 d_{6} d_{7} = SCI input route distances

d₇ = actual great circle distance between points A and B

 $d_6 = d_1 + d_2 + d_3 + d_4 + d_5 - d_7$

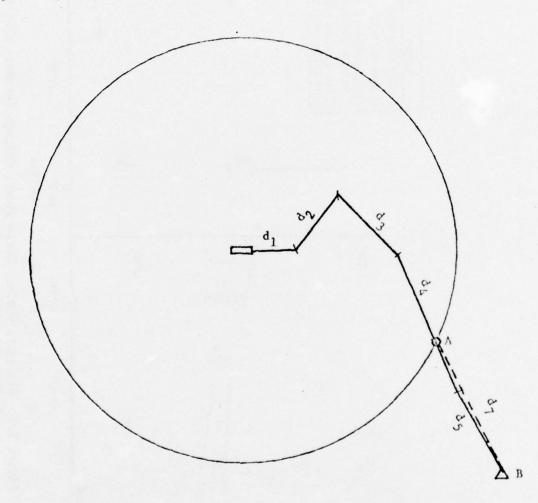


Figure 3.2 Determination of First or Final Fix Distance to Airfield VOR B Structure

aforementioned process of using both VOR A and VOR B data to establish a realistic, yet RNAV route compatible, set of VOR wind mile characteristics.

RNAV Route Structures

Four distinct RNAV route structures resulting primarily from different design philosophies were developed for this study.

The first structure is what will be referred to as the UAL/RNAV route structure. This route structure was developed by United Air lines, but subject to the constraints imposed by the terminal waypoints previously defined. The structure was intended to reflect a practical, achievable, yet near optimum RNAV structure. An "optimum" RNAV alternate routes structure would generally be defined as one which included the "optimum" route on any given day. As such, a virtually infinite number of routes would be required. As will later be shown, for given airport pairs, a greater number of routes will reduce the average route length (as measured in wind miles). A point of diminishing returns exists, however, as the incremental benefit of additional routes is outweighted by the data management costs. The UAL/RNAV route structure reflects the UAL's subjective solution to this trade-off problem with regard to both the number of routes and their general design.

Inherent within their designs are two primary assumptions. First, they adhere to the principle that, in the long run, the great circle arc (connecting the appropriate terminal waypoints) is the average candidate "preferred" route. The UAL/RNAV structure for each airport pair therefore includes a route of this form. The second assumption pertains to the geometric shape of the other (alternate) routes. Specifically, the preferred shape was assumed to be a smooth curve (with its maximum displacement from the great circle arc occurring at the mid-point) connecting the terminal waypoints. With the various alternate routes being distinguished by their deviation from the great circle arc. The UAL structures are described best as a "family of concentric football-shaped routes". The maximum displacement (from the great circle arc) of the outermost routes, the number of routes and consequently their separation were determined from specific characteristics of the airport pairs. This determination was based upon UAL's considerable practical experience. The UAL/RNAV route structure for the airport pair LAX-ORD is presented in Figure 3.3.

The UAL structure was developed from its conception to provide a set of alternate routes and did so without regard to the preferred and/or alternate routes of any neighboring airport pairs. This design philosophy differs from that utilized in FAA/RNAV route design studies. Specifically the FAA approach emphasizes the efficient design of "preferred" RNAV routes. The best example is the RNAV route structure created by NAFEC. This structure illustrated in Figure 3.4, includes some 429 airport pairs (2-way) and is expected to be adequate to accommode 92% of high altitude traffic with minor extensions to include airports in close proximity to routes in the structure. It was designed, however, based upon preferred (great circle arc) routes only and with extensive consideration given to overall traffic manageability. Of particular interest in this study, was whether or not a structure designed according to these guidelines could provide an adequate set of alternate weather routes as a part of the basic route structure. To address this question, the SCI/NAFEC and FAA/NAFEC structures were constructed by utilizing the routes

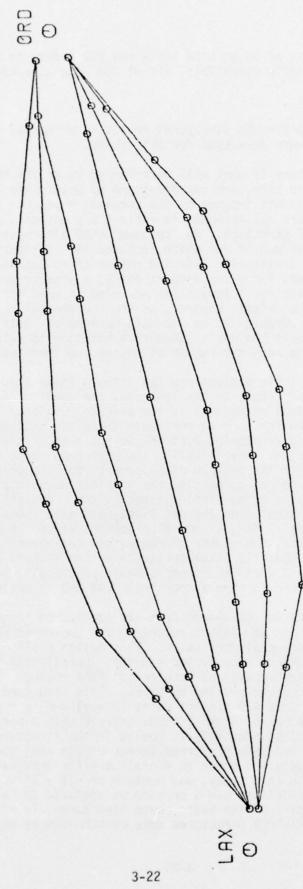


Figure 3.3 Alternate Weather Route Study UAL RNAV Route Structure LAX to ORD

Figure 3.4 "429" Airport Pair NAFEC RNAV Route Structure

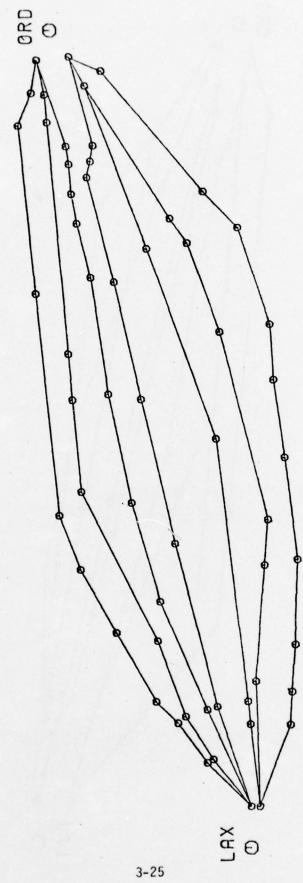
and/or route segments of the NAFEC RNAV structure. Additional segments were added to the NAFEC structure only when "deemed necessary and not detrimental" by the design agencies (SCI and the FAA/SRDS, respectively). The redundant effort on the part of SCI and the FAA was considered appropriate due to the high level of subjectivity inherent in the development of any route design. In addition to the subjective differences, there are other characteristics which distinguish the resulting structures. In general, the SCI/NAFEC structure was intended to reproduce the UAL/RNAV structure in so far as possible. The numbers of routes were identical and their shapes were as similar as was possible within the constraints imposed by the NAFEC structure. The FAA/NAFEC routes were not directly influenced by the UAL designs and/or design philosophy, and therefore, constitute an independent solution. The FAA structures also encompass a greater number of routes. The LAX-ORD routes for each of these structures are presented in Figures 3.5 and 3.6, respectively.

Each of the previous structures was developed based upon specific design philosophies. To preclude the possibility that these philosophies might in some respect unduly handicap the RNAV potential, and thereby jeopardize the credibility of the results, a fourth RNAV route structure, referred to as the FAA "Direct", was developed. Its primary objective was to provide routes of various configurations so as to fill any potential voids in the other route structures. Its value was, therefore, complementary in nature and it was not necessarily intended to be a complete structure within itself. The FAA "Direct" routes for LAX-ORD are shown in Figure 3.7.

In addition to the four primary RNAV route structures, two additional RNAV structures were created by means of combining routes of the original structures.

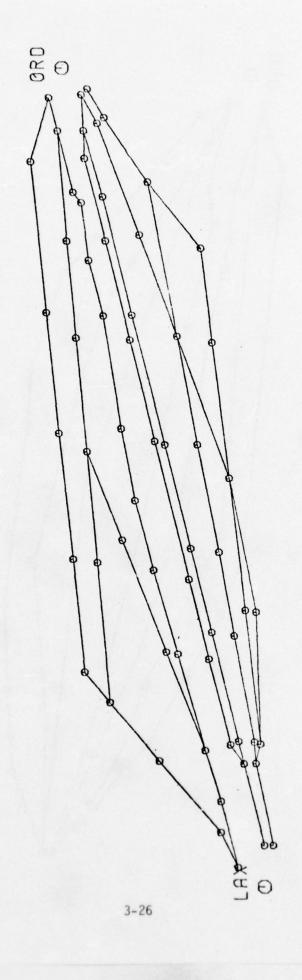
The first of these is referred to as the SCI/FAA/NAFEC structure. The initial wind mile results indicated that while the overall performance of the SCI and FAA/NAFEC structures were very similar, among the various airport pairs, instances were found when one of the structures performed noticeably better than the other. For each airport pair, this structure is identical to either the SCI or FAA/NAFEC structure, whichever of the two provided the larger benefits (i.e., short wind mile route lengths) over the entire sample period. The SCI/FAA/NAFEC structure is not comparable to the union of the two individual structures because the selection of which structure was better, was not done on a day-by-day basis.

The second additional structure is referred to as the "pre-planned" structure. In constrast to the combining procedure used for the SCI/FAA/NAFEC structure, the "pre-planned" structure is the union of all routes of all structures (VOR A included). The wind mile results for a given airport pair and sample day were obtained by utilizing whichever route of any set that produced the minimum wind miles. The "pre-planned" structure was intended to approximate the environment which its name implies to the maximum extent that was possible within this study. It is important to note that the "pre-planned" results, from this view point, would be conservative because the selection process was limited to a finite number of routes, none of which was designed to provide an optimum route for a given weather pattern.



Alternate Weather Route Study SCI/NAFEC RNAV Route Structure LAX to ORD Figure 3.5

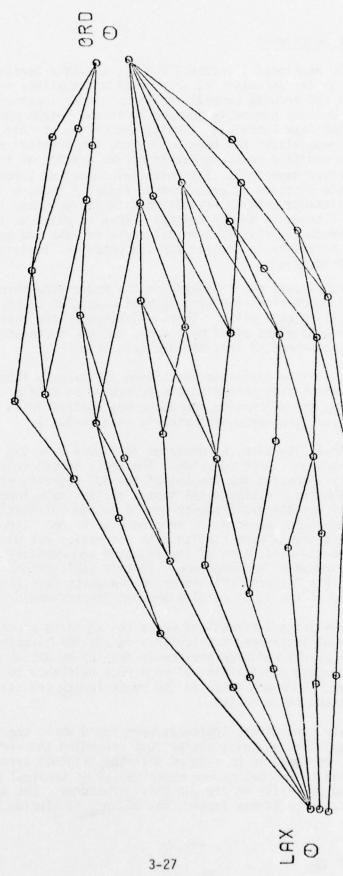
100, 200 NMI



NMI

Alternate Weather Route Study FAA/NAFEC RNAV Route Structure LAX to ORD

Figure 3.6



Alternate Weather Route Study FAA "Direct" RNAV Route Structure LAX to ORD Figure 3.7

100 200 NWI

Terminal Waypoint Adjustment

As previously mentioned (Section 3.2.2.1), the RNAV terminal waypoints were located, in so far as possible, to provide a compatible interface between the terminal area and enroute benefit analyses. These locations were predicted on terminal area designs generally conforming to the "wagon wheel concept" as defined by the RNAV Task Force [1]. During the course of this study after UAL's flight planning computations has been completed, the terminal area design ground rules were modified so as to permit closer spacing of the terminal arrival and departure waypoints. The potential impact of terminal waypoint location on enroute distance is presented in Figure 3.8 which illustrates the maximum enroute distance penalty associated with the waypoint location conforming to the wagon wheel terminal design concept. Thus to maintain the desired compatibility between the revised terminal route designs and the enroute structure it was necessary to account for the potential improvement in the terminal waypoint locations.

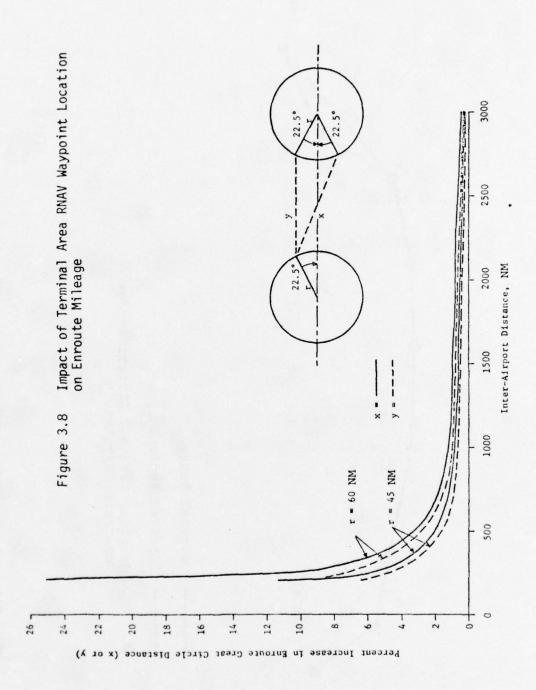
This was accomplished not by modifying the route structures themselves, as the wind mile computations had already been completed by UAL, but by altering the resulting wind miles. These adjustments were designed to estimate what the wind miles would have been, had the route structure alterations been incorporated into the UAL data base.

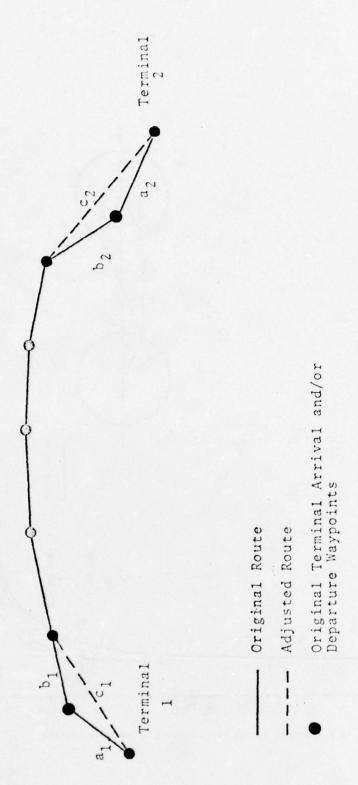
This process will be explained based upon the example illustrated in Figure 3.9. The first step is applicable to either an RNAV or VOR route. To insure consistency, the alteration procedure was applied in an identical manner for all route structures evaluated in this analysis.

For the general situation, as depicted in Figure 3.9, the lengths of the 'a', 'b' and 'c' segments were computed. The route length reductions were obtained by simply comparing the lengths of the 'c' segments with the sum of the 'a' and 'b' lengths. Although the impact of the route length reductions (summarized later) was discernible with regard to the VOR/RNAV comparison, their absolute magnitudes were small, from zero to 15 nm. Since the general directions of the segments were similar, the assumption was made that the difference between the wind miles of the adjusted and unadjusted routes should equal the ground mile difference. In this application, a ten percent error produced only a 2nm wind mile error in the worst case situation. Thus, the validity of the study results are not jeopardized by this procedure.

There were two primary situations where the standard procedure alone was altered. In several instances, particularly in the VOR structures, the original terminal points had been previously deleted by UAL so that certain flight planning program constraints, of no direct relevance to this study, could be satisfied. Care was taken in the route length reduction process not to unduly shorten these routes.

In other route structures, instances were found where the waypoint just after the departure waypoint and/or just preceding the arrival waypoint were specifically designed to be aligned with the original terminal waypoints and would have been ill-suited to any other choice of terminal points. This situation occurred primarily in the UAL RNAV structures. UAL automatically subdivided long segments (those larger than 250 nm) by placing waypoints





Torminal Waypoint/Route Length Adjustment Example Figure 3.9

every 250 nm. For example, the automated process would place a waypoint 250 nm along a 251 nm segment. This resulted in distinct waypoints in very close proximity to one another, specifically, in this example, 1 nm. In instances of this type, the terminal point and the adjacent point were both eliminated in the adjustment process. However, the general characteristics of the routes were never altered.

3.2.1.3 Alternate Weather Route Study Results

A total of 39 daily weather samples (based upon their diverse characteristics) were selected by UAL from between March 23 and May 14, 1975 for use in this study. A total of six route structures (UAL/RNAV, SCI/NAFEC, FAA/DIRECT, FAA/NAFEC, VOR A and VOR B) were defined for each of the 12 unidirectional airport pairs selected for evaluation in this study. Thus for each weather sample UAL produced a total of 72 flight plans for the selected routes of each structure plus identifying the ground miles for every route in the set.

Before presenting summaries of ground mile and wind mile results from the UAL data, it is appropriate to again define briefly the VOR and RNAV route structures which were utilized to aid the reader in the interpretation of those results. Table 3.10 presents a summary of these route structure definitions.

In the analysis of the results of this study, two factors are of primary concern: (1) the total RNAV benefit, expressed in terms of percentage wind mile savings on the selected individual RNAV weather routes compared with corresponding selected individual VOR routes; and (2) the relationship of the RNAV weather route benefit measured in this study of twelve airport pairs, on an individual route basis, to the ground mile RNAV benefit measured in the previous NAFEC study which compared traffic distributed over several VOR and several RNAV routes between the same airport pairs in a 429 airport pair structure. The first factor is indicative of the benefit available through RNAV when capacity and traffic flow considerations do not preclude the selection of either the shortest RNAV route or the shortest VOR route. The second factor provides a calibration of the no wind analysis which was based on real world constraints of required traffic distribution, and will allow an interpretation of those benefits which includes consideration of requested weather routes in the distribution of that traffic. In other words, the first factor is indicative of best case VOR route assignment (and worst case RNAV benefit), and the second factor is representative of distributed route assignment and corresponding traffic weighted average RNAV benefit.

The basic data output from the UAL simulation which was used in the analysis of these two factors are summarized in Tables 3.11, 3.12 and 3.13. Table 3.11 presents the terminal center to terminal center ground miles of the best ground mile routes for each route structure and each airport pair. The figures in parentheses are the percent of average ground miles over great circle distances. The "pre-planned" column is the best of the other 6 structures over the period, and should be compared with the VOR C column, as indicated in Table 3.10. The ground mile distance in Table 3.11 may differ from corresponding route lengths in the NAFEC analysis [4] since they are measured prior to terminal waypoint adjustment (see Section 3.2.1.2),

TABLE 3.10

WEATHER ROUTE ANALYSIS ROUTE STRUCTURE DEFINITION

Route Structure

Definition

עסמיב פרו מרימו ב	
Vor A - used to define terminal distance for RNAV comparison	UAL VOR weather routes with terminal lengths set equal to those of corresponding RNAV routes
VOR B - used to select minimum wind mile route	UAL VOR weather routes with terminal route lengths set such that the distance from the airport to UAL's first enroute fix approximates actual distance currently flown
VOR C - used for enroute wind mile comparison with RNAV	VOR A route corresponding to selected VOR B route (in most, but not all cases, the shortest wind mile VOR A route was the same as the shortest wind mile VOR B route)
UAL RNAV	UAL designed RNAV weather routes - great circle plus family of "con-centric footballs"
SCI/NAFEC	SCI modification of NAFEC 429 airport pair structure routes to approximate UAL "footballs"
FAA/NAFEC	FAA modification of NAFEC 429 airport pair structure routes to encompass VOR weather routes
FAA/Direct - not used in SCI/FAA/NAFEC since another RNAV route was always equal or better	FAA designed structure encompassing wider excursions from the great circle than other structures, and providing for "S" shaped routes
SCI/FAA/NAFEC	Composite of SCI/NAFEC and FAA/NAFEC on an airport pair basis, utilizing best (shortest wind miles) of structures over entire sample period
Pre-planned	Composite (union) of all routes of all RNAV plus VOR A structures

and due to slight differences in definition of the terminal center points.

Table 3.12 presents the average enroute wind miles of the shortest ground mile routes ("preferred" routes) and the average enroute wind miles of the daily selected shortest wind mile routes ("best" routes) over the 39 day sample period.

Table 3.13 presents the average shortest ground mile and wind mile route RNAV benefit over the 39 day sample period, the RNAV benefit of the shortest RNAV vs. shortest VOR route in the NAFEC analysis and the average RNAV benefit when comparing the traffic weighted average of traffic distributed over RNAV routes and over VOR routes between the same airport pairs in the NAFEC analysis.

Columns 1, 2 and 3 of Table 3.13 present ground mile benefit, wind mile increment, and total benefit of the shortest ground mile routes. Column 4 presents the ground mile benefit of the shortest ground mile routes, and is the same as Column 1 except for computer rounding differences. Column 5 is the increment between shortest ground mile benefit and total ground mile plus wind mile benefit on the selected route which is given in Column 6.

The CLE-MIA, MIA-CLE and SEA-JFK airport pairs were excluded from Table 3.13, since they were not a part of the NAFEC 429 pair route structure analysis. The weather route analysis benefits in Table 3.13 are based on a comparison of the best routes from all the RNAV structures over the sample period ("pre-planned"). Tables of individual airport pair benefits from which Table 3.13 was compiled are included in Appendix C, together with day by day plots of wind miles on each of the simulated routes. Although the "pre-planned" route structure is a composite of all the route structures used in the analysis, and was originally intended to be representative of a pre-planned direct structure, the number of routes actually selected was not as large as anticipated.

Table 3.14 lists the number of routes available and selected in each of the structures, including the "pre-planned". Although a detailed analysis of the interaction of the selected "pre-planned" routes between each airport pair was not performed, it can be seen that the number of selected routes is small enough to indicate that they could form the basis of a charted structure (rather than pre-planned) which could be expected to yield similar benefits. The relatively small number of routes selected is attributed to the fact that all selected routes were those closer to the "center" great circle route, and the routes on the periphery of each structure were not selected. A more accurate approximation of a pre-planned direct structure would have been a more fine-grained definition of additional routes within the periphery of the structures defined by those routes actually selected in the weather route analysis.

As stated earlier in this section, two primary factors are of concern in the analysis and interpretation of the results of this study. The benefits available through RNAV when capacity and traffic flow considerations do not preclude selection of either the shortest wind mile VOR or the shortest wind mile RNAV route are represented by column 6 of Table 3.13, which yields an average route mile benefit of 0.92% based on the NAFEC analysis traffic sample.

Route Structure Summary Data
Shortest Ground Mile Route/Ground Mile Output
Route Structure

TABLE 3.11

AP PAIR	UAL. RNAV	SCI NAFEC	FAA NAFEC	SCI/FAA NAFEC	VOR A	PRE-PLANNED	VOR C
ORD DEN	783.9 (.08)	783.8 (.04)	783.8 (.06)	783.8 (.06)	784.0 (.09)	783.8 (.09)	784.0 (.09)
DEN ORD	784.8 (.19)	786.8 (.45)	788.3 (.63)	788.3 (.63)	790.0 (.85)	784.8 (.19)	790.0 (.85)
ORD LAX	1510.0 (.06)	1509.6 (.03)	1514.4 (.35)	1509.6 (.03)	1517.0 (.52)	1509.6 (.03)	1517.0
LAX ORD	1510.6 (.10)	1516.3 (.48)	1509.6 (.03)	1509.6 (.03)	1524.0 (.99)	1509.6 (.03)	1524.0
EWR ORD	638.1	640.3 (.74)	6 4 0.7 (.81)	640.3 (.74)	637.0 (.22)	637.0 (.22)	637.0
ORD EWR	636.1 (.08)	636.8	636.1 (.08)	636.1	636.1 (.08)	635.0 (09)	636.1
IAD LAX	2005.9	2004.2	2019.1 (.90)	2019.1 (.90)	2006.9	2001.5	2006.9
LAX IAD	2004.0 (.16)	2004.9	2011.5	2004.9	2016.0	2004.1 (.16)	2016.0
MIA CLE	939.1 (.08)	944.1 (.60)	944.8 (.68)	944.1 (.60)	949.0 (1.13)	939.1 (.08)	949.0 (1.13)
CLE MIA	940.5	944.5 (.65)	945.0 (.71)	944.5 (.65)	950.0 (1.24)	940.5 (.22)	950.0 (1.24)
JFK SEA	2098.0 (.10)	2099.2	2099.6 (.18)	2099.2	2118.2 (1.10)	2098.0 (.10)	2118.9 (1.10)
SEA JFK	2096.4	2099.4	2097.8	2097.8	2126.8 (1.47)	2096.4	2126.8 (1.47)
AVERAGE	1329.0	1330.8	1332.6 (.40)	1331.4	1338.0	1328.3	1338.0

TABLE 3.12

Route Structure Summary Data
WIND MILE RESULTS: SHORTEST GROUND MILE ROUTE AND SELECTED WEATHER ROUTES
Data Based on Runs 1 through 39

AP PAIR	UAL RNAV	SCI NAFEC	FAA NAFEC	SCI/FAA NAFEC	VOR A	PRE-PLANNED	VOR C
ORD DEN	871.8 871.3	870.4 870.4	871.3 871.1	870.4 870.4	874.4 870.9	870. 4 867.6	874.4 872.5
DEN ORD	721.6 721.6	723.5 723.5	724.0 721.8	724.0 721.8	730.1 728.1	721.6 720.6	730.1
OPD LAX	1694.6 1672.2	1691.4 1673.4	1703.4 1676.4	1691.4 1673.4	1717.9 1683.6	1691.4 1669.5	1717.0
LAX ORD	1372.8 1366.6	1382.8 1368.2	1372.5 1365.7	1372.5 1365.7	1389.2 1372.8	1372.5	1389.2 1374.5
EWR ORD	723.6 720.0	727.2 724.4	726.8 725.0	727.2 724.4	728.5 725.7	728.5 719.5	728.5 727.9
ORD EWR	577. 4 577. 4	577.2 577.2	577.7 576.1	577.7 576.1	577.7 577.7	576.9 575.7	577.7 577.7
IAD LAX	2336.5 2287.9	2333.8 2290.7	2300.1 2288.9	2300.1 2288.9	2323.6 2303.9	2312.5 2275.4	2323.6
LAX IAD	1777.4 1773.5	1773.5 1771.1	1783.2 1778.3	1773.5 1771.1	1784.3 1781.4	1777.4 1768.3	1784.3 1781.4
MIA CLE	982.1 982.1	982.1 982.0	985.5 985.5	982.1 982.0	986.4 986.4	982.1 980.4	986.4 986.4
CLE MIA	923.8 923.8	921.3 919.7	921.9 921.7	921.3 919.7	943.0 943.0	923.8 919.0	943.0 943.0
JFK SEA	2266.6 2251.0	2273.3 2253.8	2264.6 2254.6	2273.3 2253.8	2297.5 2269.3	2266.6 2247.6	2247.5
SEA JFK	1985.0 1976.9	1985.4 1979.8	1985.9 1977.9	1985.9 1977.9	2009.5 1995.2	1985.0 1974.5	2009.5
AVERAGE	1352.8 1343.7	1353.5 1344.5	1351.4 1345.2	1349.9	1363.5 1353.2	1350.7 1340.2	1363.5

Comparison of RNAV Benefits from Weather Route Analysis and NAFEC Analysis Table 3.13

	average & RNAV benefit over VOR	benefit over	VOR	average & RNAV benefit over VOR	benefit over	VOR	ground mile benefit % FNAV benefit over	ground mile benefit % FNAV benefit over VOR
Airport	(1)ground mile benefit	(2)wind mile increment	(3) total benefit	(4) ground mile benefit	(5)wind mile increment	(6) _{total} benefit	(7) shortest routes*	(7) shortest (8) traffic weighted routes* average of distributed routes
ORD- DEN	.02	.43	.45	.02	. 54	95.	0	.50
DEN-	17.	.45	1.16	27.	.42	1.14	99.	**99.
ORD- LAX	.45	1.11	1.54	. 44	07.	1.14	.50	76.
LAX- ORD	1.04	71.	1.21	1.05	32	.73	1.43	1.50
ORD- EWR	.19	06	.13	61.	.15	.34	0	.37
EWR-	0	0	0	0	1.15	1.15	3.20	3.20**
IAD-	.23	.25	.48	.23	1.31	1.55	3.13	3.13**
LAX	.67	28	.39	.67	.07	27.	1.51	1.51**
JFK	16.	.43	1.34	.92	.04	96.	.63	.63**

Table 3.14
NUMBER OF ROUTES AVAILABLE/NUMBER OF ROUTES USED
39 DAY WEATHER SAMPLES

AIRPORT			R	ROUTE STRUCUTURE	in in	
	UAL/RNAV	SCI/NAFEC	FAA/NAFEC	FAA/DIRECT	VOR A	PRE-PLANNED**
ORD-DEN	5.0/2	5.0/2	10.0/3	20.0/4	5.0/3	45.0/8
DEN-ORD	5.0/1	5.0/1	10.0/2	20.0/2	5.0/2	45.0/6
ORD-LAX	7.0/3	7.0/3	10.0/3	20.0/4	11.0/4	55.0/10
LAX-0RD	7.0/3	7.0/3	10.0/3	20.0/2	9.0/2	53.0/7
EWR-ORD	4.0/2	4.0/4	9.0/3	11.0/2	3.0/2	31.0/5
ORD-EWR	3.0/1	3.0/1	10.0/3	11.0/2	3.0/1	30.0/6
IAD-LAX	9/0.6	9/0.6	10.0/5	20.0/9	12.0/5	11/0.09
LAX-IAD	9.0/5	9.0/3	10.0/5	20.0/5	9.0/4	57.0/11
MIA-CLE	3.0/1	3.0/2	10.0/2	16.0/3	3.0/1	35.0/5
CLE-MIA	3.0/1	3.0/2	10.0/2	16.0/3	3.0/1	35.0/5
JFK-SEA	9.0/5	8.0/5	10.0/5	20.0/9	14.0/3	62.0/10
SEA-JFK	9.0/4	9.0/5	10.0/5	20.0/9	9.0/5	6/0.79
AVERAGE	6.1/2.8	6.1/3.1	9.9/3.4	17.8/ 4.5	7.2/ 2.8	47.1/7.7
TOTAL	73.0/34	73.0/37	119.0/41	214.0/54	86.0/33	565.0/93
RATIO OF	.47	15.	.34	.25	.39	.16
NO. ROUTES USED TO NO. AVAIL.						

** The pre-planned structure is the aggregation of all routes from each structure.

In comparing the ground mile benefit of the shortest ground mile route in the weather analysis (Column 1 of Table 3.13) with the ground mile benefit of the shortest routes on which traffic was placed in the NAFEC study (Column 7), it can be seen that in six cases the benefits are comparable (ORD-DEN, DEN-ORD, ORD-LAX, LAX-ORD, ORD-EWR and JFK-SEA). The benefits are also comparable when comparing selected weather route ground mile benefits (Column 4) with Column 7. In the other three cases the benefits differ enough to warrant further investigation (EWR-ORD, IAD-LAX, and LAX-IAD). each of these cases traffic was placed on only one VOR and one RNAV route in the NAFEC analysis, and it is important to determine if the individual VOR routes were the shortest ground mile routes, which differed markedly from the shortest RNAV routes due to traffic flow peculiarities of the VOR structure, or if they were selected for wind mile optimization. The EWR-ORD VOR traffic utilized the preferred route published in Part 4 of the Airman's Information Manual for all 20 flights in the NAFEC analysis, and the route assignment can be considered to be for traffic flow purposes rather than requested weather routes. In the LAX-IAD case, all 7 flights were placed on a northerly routing (J-146, J-64, J-30) and in the IAD-LAX case, all 7 flights were placed on an extreme southerly routing (J-42, J-46, J-78). It can therefore be assumed that these routes assignments were in accordance with requested weather routings in the traffic sample.

The large discrepancy between the ground mile benefits of RNAV over VOR in the EWR-ORD case between Columns 1 and 7 of Table 3.13 is due to the large difference between the preferred VOR route for traffic flow and the shortest VOR route. In the case of the NAFEC RNAV structure, bi-directional traffic flow can be accommodated between these airport pairs with high traffic density, without the necessity of unduly increasing route length in one direction, as is the case in the VOR structure. The RNAV structure can therefore provide more flexibility, in cases like EWR-ORD, in accommodating high density bidirectional traffic and in allowing the selection of routes which optimize wind miles rather than constraining traffic to long ground mile routes which may in turn be also less desirable from a wind viewpoint. The wind mile increment over ground miles of the benefit on the shortest wind route between EWR and ORD (Column 5 of Table 3.13) would therefore tend to be additive to the ground mile benefit which allows selection of that route in the first place (Column 7).

On the other hand, the discrepancy between Column 1 and Column 7 for IAD-LAX and LAX-IAD is due to selection of minimum ground mile routes in one case (Column 1) for both VOR and RNAV, and selection of minimum ground miles for RNAV and minimum wind miles (longer ground miles) for VOR in the other case (Column 7). The RNAV benefit given in Column 7 and 8 is therefore overstated for these airport pairs. In these cases the actual RNAV benefit is more accurately represented by the total benefit over the shortest wind mile route (Column 6). The NAFEC 429 airport pair RNAV structure which was used in the no wind, traffic distributed RNAV vs. VOR route mile analysis may be characterized as consisting of two groups of airport pairs together with their connecting RNAV routes:

- (1) Airport pairs between which VOR traffic is currently routed without concern for wind effects. This traffic included that on minimum ground mile routes, on ATC published preferred routes which in many cases are much longer than the minimum ground mile route, and traffic placed on other than minimum ground mile routes on an ad hoc basis for the purpose of traffic control.
- (2) Airport pairs between which VOR traffic is currently routed primarily in response to requested routings established to minimize wind miles.

In both cases traffic is currently distributed over one or more VOR routes between each airport pair, and over one or more RNAV routes between each airport pair for benefit comparison.

Since the NAFEC analysis was reasonably realistic in terms of traffic distribution, it is required only to separate the 429 airport pairs into the two groupings described above in order to calibrate the second group with the results of the weather route analysis. In order to perform this separation, it was arbitrarily assumed that any airport pair separated by a great circle enroute distance of more than 500 miles in the NAFEC structure had traffic distributed over VOR routes between these airport pairs on the basis of minimum wind miles. An exception was made if a published ATC preferred route existed for that airport pair and the great circle distance was less than 800 miles (13 airport pairs). An exception was also made if traffic was distributed over more than one VOR route and the great circle distance was less than 800 miles (3 airport pairs). In these cases, traffic was allowed to remain on the routes used in the NAFEC analysis. A total of 203 airport pairs out of the 429 met these criteria. It was assumed that, for routes of less than 500 miles, the characteristics of the VOR route structure, coupled with the distribution of traffic over more than one route per airport pair, were dominant over wind effects with respect to the realization of benefits on RNAV routes. This assumption was validated by an examination of the ground mile benefits on these shorter routes in the NAFEC analysis which showed that the average ground mile benefits on the shorter routes were three times larger than the wind mile increments experienced in the weather route simulation. The total flight mile benefit for each of these routes in the NAFEC analysis was then adjusted to represent an average wind mile benefit as determined below.

The 39 day average RNAV percent route structure combination benefit for each two way airport pair relative to the VOR C structure with the exception of the CLE-MIA airport pair whose results were significantly affected by computational anomalies, appear to be a reasonably well behaved linear function of the great circle distances.

Lines were fit through these data points utilizing the least squares technique, the results of which are summarized in Table 3.15.

TABLE 3.15

PARAMETERS FOR THE RNAV PERCENT BENEFIT EQUATION

 $P = a \cdot G + b$

where G = Great Circle Distance (terminal center to terminal center)

P = RNAV Route Length Benefit In Percent of VOR Distance

ROUTE STRUCTURE	RELATIVE to VOR C	
	a	Ь
Pre-Planned	0.000296	0.569
UAL/RNAV	0.000273	0.363
SCI/NAFEC	0.000361	0.123
FAA/NAFEC	0.000271	0.251
SCI/FAA/NAFEC	0.000316	0.264

Using the benefit equation from Table 3.15 for the "pre-planned" (representative of optimum charted) structure, the flight mile benefit for each of the 203 airport pairs as follows:

total VOR flight miles between 429 airport pair total RNAV flight miles between 429 airport pai	
flight mile benefit from NAFEC analysis adjusted flight mile benefit from 203 airport p	62,794 pairs -15,042
adjusted flight mile benefit over 429 airport p	

RNAV flight mile (traffic weighted) benefit relative to VOR = 1.79%

The weather route calibration of the NAFEC no-wind analysis is conservative (in favor of VOR) for two reasons:

- (1) It does not consider real world impromptu deviations from optimum weather routings, larger than corresponding RNAV routing deviations, which are currently required for traffic control, or due to traffic density.
- (2) It does not include the additional wind mile benefits available through the ability to select an RNAV route which is more favorably oriented than a current VOR preferred route.

3.2.2 Restricted Area Analysis

All restricted and warning areas which would potentially affect the high altitude enroute RNAV structure were identified in the Federal Register, Part 73, "Special Use Airspace". These areas were then plotted on a projection of the NAFEC 429 airport pair RNAV route structure. All route segments impinging upon, or in close proximity to, the plotted restricted areas were then identified and were plotted on a high altitude enroute airways chart for verification. A section of the high altitude airways chart showing RNAV route segments crossing restricted areas is illustrated in Figure 3.10.

The airport pairs associated with each route segment which violated a restricted (or warning) area were identified from a computer listing provided by NAFEC [4]. A listing of restricted and warning areas which affect the NAFEC structure and the corresponding affected route segments is given in Table 3.16. The high altitude route structure airport pairs which utilize the affected route segments are given in Table 3.17.

Table 3.16 NAFEC Design Route Segments Affected by Restricted Areas

Restrict- ed/warn- ing Area	Ro	ute Se	ment:	s Affe	ected	
R2306	345G	379W	379E	345B		
R2308	345G	379W	379E			
R2501	007E	010E				
R2502	003W	3800				
R2505	380C					
R2507	345G	F414	379W	345B		
R2508	400A	256W	446A	155G	177A	
	380C	400B				
R2510	379W					
R2515	177A	177P				
R2521	3458	F414				
R2524	177A	177P	380C			
R3005	420A					
R4806	155G	177A	177P			
R4807	155G	449A				
R4808	155G	C340				
R4809	449A					
R5103	174C					
R5107	174C	034A	217W	217E		
R5111	217W	217E				
R5503	2250	3890	318A	3700	3820	453D
R5504	2250	3890				
R6402	014H	014P	005A			
R6404	365A					
R5405	014H	005A				
R6406	258A					
R6407	014P					
R6714	2980					
R6903	258B	304A				
W106	117A					
W107	117A	006A	006P			
W122	117A	0068	051B			
W132	126A					
W151	367A					
W157	126A					
W158		051B	126A	126B		
W177	126A					

TABLE 3.17 ROUTE SEGMENTS IMPINGING ON RESTRICTED/ WARNING AREAS AND CORRESPONDING AIRPORT PAIRS

Segment			Airport	Pairs			
003W	BURLAS L	AYRUR	LAXMSP	ONTLAS			-
005A	DTWSFO C				SEOMDW	SECORD	
0001	SFOPHL S		1116-21-0	31.001.4	3. 0.10	51 00110	
006A	BOLMIA F		IL JEK N	ITARDI N	ATAFWR N	ATAJEK	
OUGH	MIALGA	LLC MIC 1	LLUI K	II NODE I	ILINE MILL	II NOT IN	
0068	BOLMIA E	WRELL	FLIFUR	FILL JEK	JEKELL	JEKMIA	
0000	MIABOL M				OINILL	ULIVITA	
0060	BOLMIA E						
006P	EWRFLL J				1 GAORE		
007E	LAXCLE L				LIMIONI		
010E	LAXBAL L				LAYJER	LAYJEK	2
0.05	LAXLGA L						
014H	EWRSFO J						
01411	DAKJEK C					1.043.3	
	SFOJEK S						
014P	JFKSF03						
034A	DALLAX (MSVLAY	PHYDAI	
0518	EWRMIA F						
03.0	PBIJEK F					THAT THE	
117A	BOSFLL B						
126A	BALMIA D				MIADCA	MIATAD	
126B	BALMIA D			114730016			
155G	ABOSFO D			SEDARD	SECONAL	SEOMIA	
1740	IAHLAX						
177A	LAXSLC S		Erio Erio	C. C. C. C.			
177P	LAXSLC	JE O ET IN					
217E	ABOELP						
217W	ELPABO						
2250	ATLDTW D	ITAWTO					
256W	BURRNO L		LAXRNO	LAXSEA			
258A	BOSSED S		Limite	Printer.			
2588	BOSSEO S						
298C	DENSEA S						
304A	BALSEA I		DTWMSP	TADSEA	MSPDCA	MCDDTW	
3044	SEABAL S		DIMIG	Incorn	nor ben	Marota	
318A	GSOORD (
345B	ELPLAX S		THELAY				
345G	LAXDAL I	AYELD	LAYIAH	LAYMIA	LAYMSP	LAYPHY	
3430	LAXSAT I	AXTUS	ONTPHY	SANPHY	Ser test sof	SELECT TIM	
365A	MSPSFO S		G.T.I.IIA	JAMES IN			
367A	MSYTPA 1						
3700	ORDRDU F						
379E	SANPHX	NDO OND					
379W	PHXSAN						
3800	PHXSFO S	SEOPHY					
3820	PITSDE S						
3890	DTWTPA						
400A	IAHSFO S						
400B	IAHSFO						
420A	CMHMIA I		MIACMH	MIANTH			
446A	PSPSF0						
449A	ABOMCC I						
453A	BOSDAL I						
C340	RNOLAS						
F414	SANJEK	SANORD					

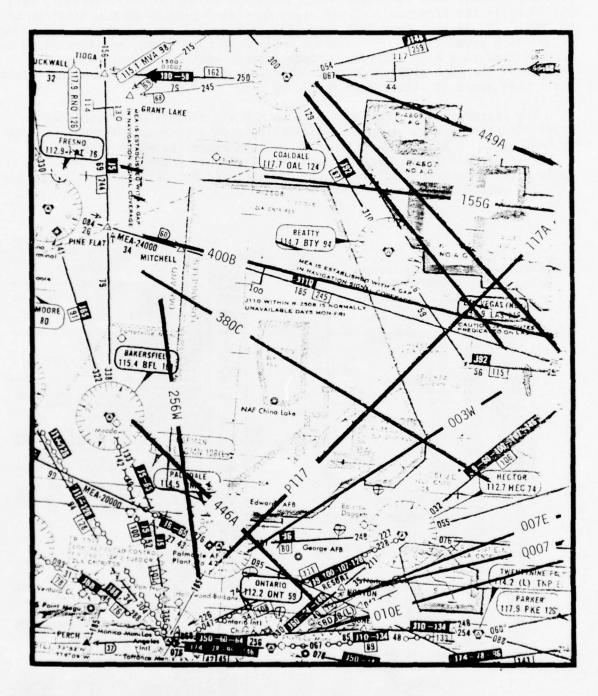


Figure 3.10 NAFEC High Altitude Route Structure-Route Segments Crossing Restricted Areas

The modification of the NAFEC route length analysis described in Section 3.2.1.3 recognized the RNAV savings in route length which would be available through the approximation of desired weather paths by RNAV routes rather than VOR routes. These RNAV benefits represent the average differences in path length between RNAV and VOR in traversing a desired curved path (in that case a "weather route") which is represented by a series of great circle segments which closely approximate the desired path in the RNAV case, and a series of VOR segments which less accurately approximate the desired path in the VOR case. This RNAV benefit was found to be a linear function of a great circle distance.

In order to calibrate the NAFEC route length analysis for restricted area effects, it was assumed that "curved paths" necessary to avoid restricted areas would be analogous to the "curved paths" representing weather routes, and that the same route length dependent benefits would apply. As in the weather route calibration, airport pairs separated by more than 500 miles were assigned RNAV benefits in accordance with Table 3.15. All airport pairs separated by less than 500 miles whose connecting route penetrated a restricted area were assigned zero RNAV benefit. The benefit over the 429 airport pair structures was then computed as follows:

Total VOR flight miles between
429 airport pairs from
NAFEC analysis -----2,674,014

Flight mile benefit from
NAFEC analysis (RNAV vs VOR)-- 62,794
RNAV benefit over VOR = 2.35%

Flight mile benefit from weather route calibration of NAFEC analysis (section 3.2.1) ------47,752 RNAV benefit over VOR = 1.79% for following assumptions:

- VOR routes do not violate restricted areas
- RNAV routes not constrained by restricted areas
- Airport pairs separated by distances of 500 mi or less, and by 800 mi or less if VOR route is a preferred route, assigned benefit from NAFEC analysis
- Airport pairs separated by more than 800 miles, or more than 500 miles if VOR route is not a preferred route, assigned benefit based on weather route analysis

Flight mile benefit from weather route and restricted area calibration of NAFEC analysis (Section 3.2.2) ---43,092 RNAV benefit over VOR = 1.61% for following assumptions:

- VOR routes do not violate restricted areas

- RNAV routes do not violate restricted areas

 Airport pairs separated by distance of 500 mile or less:

 Assigned zero benefit if RNAV route violates a restricted area

 Assigned NAFEC derived benefit if RNAV route does not violate a restricted area

- Airport pairs separated by distance of 500 to 800 miles:

 Assigned benefit derived from weather route analysis if RNAV route violates a restricted area or VOR route is not a preferred route

 Assigned benefit derived from NAFEC analysis if RNAV route does not violate a restricted area and VOR route is a preferred route

- Airport pairs separated by distance of greater than 800 miles assigned benefit derived from weather route analysis

In summary, a conservative estimate of traffic weighted average flight mile in a high altitude enroute RNAV structure, compared with the current VOR jet route structure, is 1.61%. This estimate includes the effects of wind on selected weather routes, the effects of real world traffic distribution, and the assumption that the RNAV structure will be constrained by existing restricted areas, and is based on an RNAV structure capable of accommodating 92% of the estimated 1977 traffic on an average day, with minor extensions to include airports in close proximity to the simulated structure. A listing of the airport pairs utilized in the NAFEC analysis, calibrated for both weather route and restricted area effects is given in Table 3.18. In cases where route mile benefits are shown but flight mile benefits are zero, no traffic was applied to that airport pair in the NAFEC analysis. It should be emphasized that the NAFEC structure used in this analysis has not been optimized, and that some increase in benefits could be expected in a properly optimized structure.

TABLE 3.18
ROUTE MILE AND FLIGHT MILE SAVINGS: RNAV OVER VOR

FLIGHT MILES VOR -RMAY

VOR ROUTE MILES

AIRPORT PAIR FROM TO

FLIGHT	MILES VOR -RNAV	22 20 20 20 20 20 20 20 20 20 20 20 20 2
ROUTE	MILES VOR -RNAV	28 20 20 20 20 20 20 20 20 20 20 20 20 20
YOR	ROUTE	140 434 485 485 485 485 485 485 485 485 485 48
T PAIR	10	PER
AIRPORT	FROM	ABB

T PAIR TO	BB
FROM	33333333333333333333333333333333333333
P. IGHT MILES VOR -RNAV	180 - 40 - 40 - 40 - 40 - 40 - 40 - 40 -
ROUTE MILES VOR -RNAV	23
VOR ROUTE MILES	436 534 534 102 102 103 103 103 103 103 103 103 103 103 103
TO TO	PTT
ATRPOR FROM	8AL 8AL 8BAL 8BAL 8BAL 8BAL 8BAL 8BAL 8B

51	MILES VOR -RNAV	252
ROUTE	VOR -RNAV	-01-68-88-0-8-0-8-8-0-8-8-8-8-8-8-8-8-8-8-8
CY.	ROUTE	168 8334 1202 1202 1202 1202 1202 1202 1202 120
T PAIR	10	BBBBBBBBBBBBBBBBBBBBBBBBBBBBBBBBBBBBBB
AIRPOR	000	\$

PHX
SSSATURE

SECTION OF THE SECTIO

TABLE 3.18 (Cont'd.)

ROUTE MILES VOR -RNAV

YOR ROUTE MILES

AIRPORT PAIR FROM TO

FLIGHT MILES VOR -KNAV	23 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3
ROUTE MILES VOR -RNAY	74 - 72 - 6 - 6 - 6 - 6 - 6 - 6 - 6 - 6 - 6 -
VOR ROUTE MILES	531 2158 675 675 871 874 874 874 874 874 873 873 873 874 873 873 873 874 873 874 873 874 873 874 873 874 873 874 873 874 874 874 874 874 874 874 874 874 874
TO TO	PROSECTION OF STATE O
FROM	EN SON SON SON SON SON SON SON SON SON SO

DOTA THE REPORT OF THE REPORT

9	MILES VOR -RNAV	19 254 67 61 61 61 61 61 61 61 61 62 63 64 64 64 64 64 64 64 64 64 64
ROUTE	70R	222 222 223 223 223 223 223 223 223 223
OR	ROUTE	634 2034 2035 2037 2037 391 104 406 838 838 838 838 838 838 838 834 837 837 837 838 838 838 838 838 838 838
IT PAIR	10	ATT LASS SECOND
AIRPORT	FROM	LAXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXXX

FLICHT HILES VOR -RHAV	138 1557 1157 1117 1117 1117 1117 1117 111
ROUTE MILES VOR -RNAV	80446 w800544 v w 4 m 4 0 m 5 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
VOR ROUTE MILES	513 2023 1913 1913 1937 1937 1936 1936 1936 1937 1937 1938 1939 1939 1939 1939 1939 1939 1939
TO TO	I I I I I I I I I I I I I I I I I I I
FROM	AAR SET TE T

FROM		######################################
FLIGHT	VON -RNAV	4-7-4-4-4-4-4-4-4-4-4-4-4-4-4-4-4-4-4-4
ROUTE VOR -RKAV		
NO.	MILES	126 183 4633 6833 6833 6833 6833 6833 683 137 201 87 101 1132 1148 1151 1151 1151 1151 1151 1151 1151
PAIR	10	6 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2
ALRPORT	FROM	MANUAL MA

FL IGHT MILES VOR -RHAV	255 1 2 2 3 3 3 3 4 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
ROUTE MILES VOR -RUAY	~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~
VOR ROUTE MILES	883 878 878 878 878 878 878 878 878 878
T PAIR TO	LCAX NO
A1RPOR FROM	SAN

post.	VOR -RNAV	10 110 110 122 229 229 235 102 102 102 102 102 102 103 103 103 103 103 103 103 103 103 103
5	VOR -RNAV	
VOR	MILES	264 264 264 199 199 199 199 199 199 199 19
PAIR	T0	OSSEA SEE SEE SEE SEE SEE SEE SEE SEE SEE
AIRPORT	FROM	000 00 00 00 00 00 00 00 00 00 00 00 00

FLIGHT MILES VOR -RHAV		240 327 327 327 327 327 327 327 327 327 327
ROUTE	VOR RHAV	24 4 2 4 2 4 2 4 2 4 2 4 2 4 2 4 2 4 2
VOR ROUTE MILES		136 804 804 1175 1160 1150 1150 1150 1150 1150 1160 116
PAIR	10	ST CEEF ST CEE
AIRPORT	FROM	88888888888888888888888888888888888888

TABLE 3.18 (Cont'd.)

VOR ROUTE
- ROUTE MILES
- MILES VOR
- RNAV

FLIGHT MILES VOR -RNAV		255 2177 2177 255 255 255 257 277 277 277 277 277 2
ROUTE	VOR -RITAV	40 60 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6
108	MILES	185 267 267 267 267 267 269 269 269 269 269 269 269 269 269 272 269 273 273 273 273 273 273 273 273 273 273
RPORT PAIR	T0	MAGA MAGA MAGA MAGA MAGA MAGA MAGA MAGA
	MOX	TITLE TO THE TENT OF THE TENT

ROUTE FLIGHT MILES MILES VOR -RNAV -RNAV		386 356 356 356 356 356 356 357 375 375 375 375 375 375 375 375 375
		~ 88 m 8 4 ~ 4 6 m 8 L 4 m 0 m 0 m 8 m 8 m 8 m 8 m 8 m 8 m 8 m 8
VOR	MILES	2010 2129 1878 1888 1888 1888 1889 1880 1880 1880 188
PAIR	ro	SE S
AIRPORT	FROM	\$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$

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3.3 ENROUTE LOW ALTITUDE RNAV BENEFITS

The analysis described in Section 3.1 and 3.2 focused on quantifying those benefits resulting from the use of RNAV in the high altitude (at or above FL 180) enroute environment. This companion analysis, examining the low altitude regime, was undertaken to provide an assessment of the potential RNAV benefits that could be obtained by general aviation owners and operators, commuter carriers, air taxi operators and other user groups who normally operate below 18,000 feet.

The high altitude studies made use of a comprehensive 429 airport pair RNAV route structure, developed by NAFEC, to directly identify RNAV route lengths for subsequent comparison with corresponding VOR routes. However, a comparable low altitude RNAV structure had not been designed prior to initiation of this task, and the design of such a structure on a national scale was beyond the scope of this effort. An approach was developed whereby a low altitude route structure was developed for a limited geographical area, the route length benefits determined, and the results extrapolated to a national result based on common airport pair characteristics. This approach is summarized in the flow diagram of Figure 3.11 and described in detail in the following sections.

3.3.1 Sample Area Selection for Low Altitude RNAV Route Design

The selection of a specific geographic region, was established as a ground rule to ensure that the routes of the resulting low altitude RNAV structure would have realistic interactions not only with the surrounding VOR routes, but with other RNAV routes as well.

The method ultimately adopted for this analysis was to select a geographical area based on its diverse characteristics which, if possible, should reflect all conditions found on a national scale, but not necessarily in the same proportion. In this case, the individual airport pair payoffs were to be related to certain characteristics prevalent in the airport pair by means of a regression analysis.

The resulting regression equation was then used in combination with the appropriate characteristics developed for each airport pair having a low altitude IFR exchange [23] to predict individual airport pair payoffs. Those results were then aggregated to produce a national low altitude payoff.

Data from the IFR Peak Day Tape, supplemented by airport location information obtained from FAA airport directory tapes was used to establish the distribution of low altitude operations among the states. To facilitate the analysis, nine states and the District of Columbia were merged into three "Larger States", where (A) relatively light traffic existed over several small adjacent states, (B) traffic was concentrated in a "metropolitan" area encompassing parts of two states, and (C) several adjacent states encompassed a relatively small geographical area. This resulted in a total of 42 states including the following merged states:

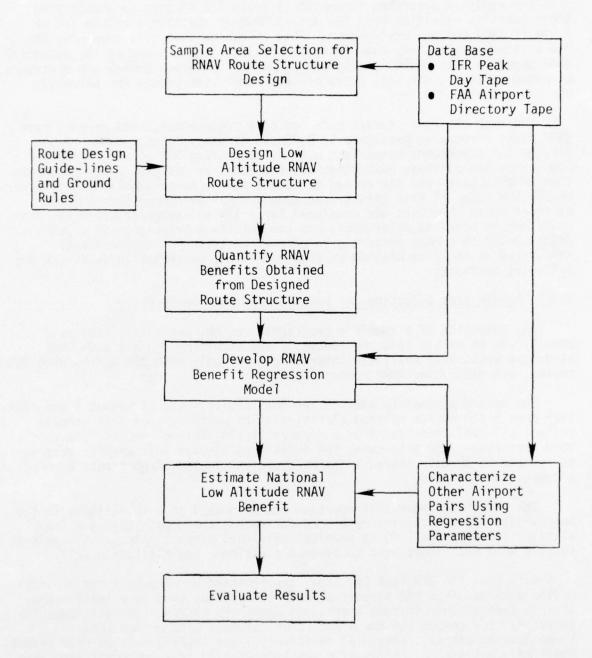


Figure 3.11 Approach to Determine RNAV Low Altitude Route Length Benefits

"MERGED STATE"

CONSISTS OF

(A)	Connecticut, New Hampshire, Vermont, Rhode Island, and Massachusetts
(B)	New York and New Jersey
(C)	Delaware, District of Columbia

and Maryland

The IFR tape processing produced a total of 20950 low altitude flights, encompassing 8076 airport pairs and 1985 airports. The airport directory provided the locations (latitude, longitude and state) for 1336 of these airports. The discrepancy between the number of airports contained on each data source can be attributed primarily to the fact that the IFR tape includes Alaska and Hawaii flights, as well as departures to foreign airports. The airport directory does not contain foreign airports, and Alaska and Hawaii airports were not processed since only those states within the contiguous United States were included in this analysis.

The basic state and state pair results are summarized in Table 3.19. It can be seen that the State of California has by far the greatest intra-state traffic exchange with Texas ranked number two.

Table 3.19 Low Altitude Intra-State Traffic (Top Twenty)

	States	Peak Day Exchange
1.	California	1272
2.	Texas	751
3.	New York/New Jersey	668
4.	Florida	502
5.	Ohio	439
6.	Conn/VT/NH/RI/Mass	415
7.	Michigan	401
8.		369
9.	Pennsylvania	355
10.	N. Carolina	224
11.	Georgia	210
12.		192
13.	Tennesse	181
14.	Kansas	181
15.	Wisconsin	172
16.	Colorado	170
17.	Delaware/DC/MD	159
18.		149
19.	Louisiana	140
20.		131

Since the average hop length of general aviation flights is only 160 miles and low altitude traffic is dominated by general aviation, it was felt that a large state with a large amount of intra-state traffic would be the most appropriate for analysis of a low altitude RNAV route structure, and thus California was chosen. California contains two high density terminal areas (SFO, LAX), several of medium density, and an abundance of low density airports. In general, the large number of airport pairs (275) and flights (1272) should insure the maximum possible diversity among route characteristics.

3.3.2 California Low Altitude RNAV Route Design

The primary objective of the route design effort was to establish the low altitude enroute length benefit data for a specific area, which would subsequently serve as a baseline for extrapolation. The terminal area effects were, therefore, not addressed. For the small terminals, the terminal area was defined simply as a radius of 5 nm which approximated the control zone, airport traffic area, or airport advisory area. Routes passing over these terminal areas were assumed to be unaffected by the terminal. In the two high density terminal areas, encompassing Los Angeles and San Francisco regions, respectively, the RNAV routes were designed so that they began and/or ended at current VOR compulsory reporting points. Although the RNAV arrival and departure waypoints may be expected to be in slightly different locations in an RNAV environment, the impact of these new locations on the aggregate results of this analysis would be negligible.

The RNAV structure was permitted to overlay the existing VOR airway segments if appropriate and necessary to produce an efficient RNAV structure. Thus, the resulting RNAV route structure is the composite of a portion of the VOR structure plus unique RNAV route segments. As a result the RNAV routes were not adversely impacted by the remaining VOR structure. It was generally found, as expected, that whenever the VOR routes were inordinately long, and therefore, inferior to the RNAV routes, they tended to deviate sufficiently from the great circle arc so as not to interfere with the resulting optimal RNAV design.

Intersecting segments, both within the RNAV structure and between RNAV and VOR routes, were designed in a manner so as to maximize the intersection angles. The 15° nominal minimum intersection angle, used as a guideline for high altitude RNAV design produced by NAFEC, was also applied to this study. However, as in the high altitude designs, a few exceptions are necessary. The primary cause of small intersection angles in the California low altitude design stems from the RNAV/VOR merging or demerging at VOR stations. In many instances, the number of VOR routes intersecting at a VOR station was such that no space existed for the design of an additional route with a 15° separation. When there was no reasonable alternative, the 15° requirement was relaxed. The primary intent of the intersection angle requirement was to facilitate the separation of routes. Larger angles result in less common airspace. Routes which are intended to be procedurally separated (i.e., by their route width) would be separated over a longer distance when larger

intersection angles are used. For this reason, however, the 15° requirement was not found to be as appropriate in the low altitude airspace as at the higher altitudes.

The number of intersections in combination with the small average route segment lengths and the lack of predominant directions of flow virtually prohibit the procedural separation of segments. In several areas of California, the current low altitude VOR structure is already so complex that procedural separation is rarely evident. In general, segments which are at all close to one another intersect. While certain RNAV segments were designed intentionally to achieve separation (based on a \pm 4 nm route width), the primary and most restrictive constraint was the overall structure complexity.

The final ground rule applicable in the route design effort was that airport pairs with a great circle distance of less than 50 nm were not considered. The rationale for this decision was based on two factors. First, short flights where either of the airports were within a major terminal area would produce only a minimal amount of enroute travel. If both terminals are in high density areas, as is frequently the case along the east coast, travel outside the terminal area could be completely eliminated. Although subjective, this was the rationale for the choice of 50 nm as the specific exclusion criteria. The second reason for excluding these airport pairs was that the peculiarities of the terminals themselves can cause a considerable variation in the conventional (VOR) route lengths. If both of the terminals have VORs a "direct" route is available. The actual flight length, however, may be considerably longer due to the VOR/terminal orientation. These route length variations occur on longer flights as well, but their relative impact on the total route length is reduced. It was known that these route length variations could not be adequately accounted for in the regression analysis, and their inclusion would, therefore, not be advantageous. Of the original 275 California airport pairs listed on the peak day tape [23], 197 had distances of 50 nm or more.

VORTAC coverage and altitude restrictions were also considered in producing the RNAV design. However, it was considered important not to influence the California design or its routes length results by coverage or terrain phenomena peculiar to California. As a rule, the VOR routes tended to fly directly over mountainous areas, being constrained only by minimum altitude requirements. In these areas, the RNAV routes were similarly designed, assuming that comparable minimum altitude requirements would be imposed.

Of greater concern were the cases where the VOR routes circumvented specific geographic areas. In these instances, determination of the cause of the deviation was necessary. If the VOR's were not suitably located, this was assumed to be the sole cause of the route bending. Since complete coverage data was not available, an estimate was made as to whether or not adequate RNAV (VORTAC) coverage existed. If coverage was reasonably certain, an RNAV route was designed. If coverage was unlikely, a conservative approach was taken and the RNAV route was redesigned to utilize known coverage. If coverage was not known, the airport pair was eliminated from the analysis. Airport pairs were also eliminated when the cause of the VOR deviation could not be ascertained.

If coverage and terrain factors indicated that a VOR route shorter than the charted VOR route could be established, the RNAV route was considered to have an unfair advantage. On the other hand, setting the RNAV benefit equal to zero would unfairly reduce the overall RNAV payoff averaged over all routes. This occurred in a number of instances, due to the fact that a large number of airport pairs considered included a great many of very low traffic exchange. In all, 17 airport pairs were discarded on these grounds, leaving the 180 which were subsequently used in this analysis.

The route design procedure consisted of sequentially analyzing each airport pair and adding RNAV-peculiar segments as appropriate. The aggregate structure become more complete as the process evolved. As each airport pair was analyzed, the first consideration was whether or not the current composite structure (i.e., the existing VOR structure plus any newly added RNAV segments) provided a route which did not significantly deviate from the direct path. If this was the case, no additional RNAV segments were deemed necessary. However, when an acceptable route did not exist, RNAV segments were added where feasible, either so as to build the entire route or to cut corners on an existing route.

In certain instances, a previously designed RNAV segment would be found to be detrimental to the design of another airport pair. In these instances, the airport pair exchange and the relative value of each segment was evaluated in order to arrive at a reasonable compromise which would potentially provide the greatest RNAV payoff for the most traffic.

The sequential nature of this procedure was such that the results improved with subsequent iterations. Routes which did not appear to justify additional RNAV segments were able to take advantage of additional segments which were added for other airport pairs. The entire process was repeated several times until additional changes did not appear to produce a more efficient design. In all, 108 of the 180 California airport pairs retained as candidates for RNAV routes were able to benefit from the design of RNAV segments.

The resultant RNAV structure is depicted in Figure 3.12. The dotted lines represent exclusive RNAV route segments. The heavy solid lines are existing VOR routes, which are also proposed as RNAV route segments to supplement the unique RNAV segments, in order to provide a complete RNAV structure serving each of the 180 airport pairs.

The route length computations were based on determining the shortest RNAV and VOR routes which were made available in each of the structures. All routes within the terminal areas were assumed to be great circle arcs from the point of intersection with the terminal area boundary (or reporting point) to the airfield. Table 3.20 presents the final California route length results for each airport pair. The average RNAV benefit relative to VOR is 2.92% on a route mile basis and 2.61% on a flight mile (traffic weighted) basis. The total VOR enroute route length, used only as the denominator in the RNAV percent benefit computation, was obtained by assuming individual enroute route lengths equal to the airport-to-airport lengths minus 10 nm (5 miles at each end).

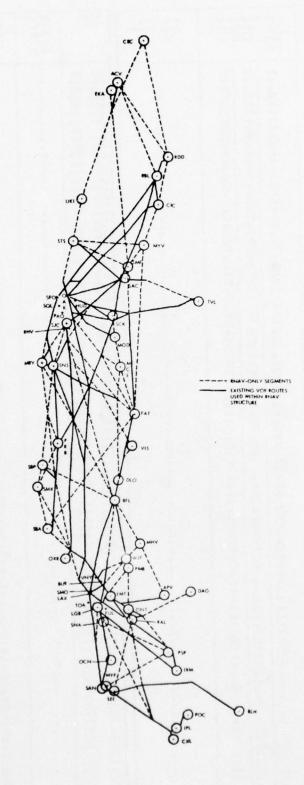


Figure 3.12 California Low Altitude RNAV Route Structure

Table 3.20

California Low Altitude RNAV Route Structure Results

PORT	NO. OF FLTS. PEAK DAY (REF.1)			RNAV ROUTE AIRPORT PAIR	PEAK	ROUTE	AIRPORT TO AIRPORT ROUTE LENGTH (NM) GREAT RNAV VOR		RNAV ROUTE LENGTH BENEFIT		
		-	RNAV VOR	-	(NM)		DAY (REF.1)	CIRCLE	KNAV	VOR	(NM)
V OAK	1	214	216	227	11	LG8 SBA LG8 TRM	1 5	91 99	94	98	4 0
V RBL	2 2	98 86	88	105	17	MOD RDD	1	183	191	191	0
V SAC	i	190	192	205	13	MOD SFO	3	67	68	68	0
V SFO	7	217	218	229	11	MOD SAF	1	70 237	73 241	73 247	6
SMF 2 575	2	179	180	193	13	MRY OAK	6	70	71	77	6
V UKI	2	118	118	126	8	MRY PRB	1	80	81	88	7
MFL.	2	105	105	113	8	MRY SAC	2	117	118	124	6
SNA	1	63 81	70	70 83	0 2	MRY SEO	26	66	66	68	2
FAT	54	87	87	87	0	MRY SMX	2	121	123	125	2
LAX	21	94	95	95	0	MRY VNY	1	217	104	235 110	17
LG8 MCE	3	106	106	108	2	MYP POC MYP SNA	1	83 63	67	67	0
MRY	i	152	153	161	8	MYV RDD	1	91	91	94	3
OAK	2	205	208	214	6	MYV STS	2 2	144	145	146	6
RDD	1	341	68 342	349	0 7	OAK PRB	2	167	170	170	0
SAN	2	186	188	188	0	OAK SAC	4	58	58	58	0
58P	1	78 181	78 182	87 188	9	OAK SMF	5	65	70	69	0
SCK SFO	2	207	209	215	6	OAK STS	3	55	56	63	7
VIS	1	55	58	58	0	OAK TOA	1	301	305	305	0
SAN	2	90 135	93	93 148	0	OAK TVL	3	126 277	130	130 280	0
IPL	2	161	179	183	4	ONT PSP	4	56	59	62	3
MCE	1	213	214	216	2 2	ONT SEO	2	112 315	113 317	115 326	2 9
MHV	2	53 102	55	57 111	0	ONT SNS	1	250	254	260	6
OAK	2	262	282	282	0	ONT WJF	3	51	52	56	4
PAO	1	268	271	271	0	PAO SNS	1	242 53	248 55	250 56	2
PSP SAN	18	105	96 106	96 108	0 2	PRB SFO	4	143	146	146	0
S8A	1	74	74	76	2	PRB SJC	2	119	120	122	2
SBP	1	129	134	137	3	PSP SAN	2	73 97	102	102	12
R SFO	1	283 257	287 260	287 260	0	PSP SNA	2	68	68	73	5
E SMX	3	111	113	122	9	RAL SJC	1	299	300 70	70	10
TVL CIC	1	292 161	292 163	192	0 29	RAL TRM		168	175	175	0
C SFO	2	264	264	279	15	RBL SMF	1	92	92	92	0
C OAK	1	125	126	126	0	RDD SAC	10	125 173	126	126	0
C SAC		78 133	137	137	2 0	RDD SMF	3	113	114	174	0
C SMF	i	67	68	69	1	RHV SAC	1	72	73	73	> 0
L SFO	1	448	457	467	10	RHV STS SAC SEO	2	84 68	68	68	0
G RAL	2	100	102	75	11	SAC SJC	4	71	72	72	0
A UKI	i	109	112	123	11	SAC SNS	3	111	111	117	6
T SAN	2	91	94	94	0	SAC SQL		69	63	71 63	0
T LAX	2 2	181	182	182	1	SAC TVL	2	73	76	87	11
T MOD	4	78	79	89	10	SAN SME	1 !	417	419	422	3
T MRY	2	103	105	108	3	SAN SMO	8	65	67	67	0
T SAC	4	134	134	134	0	SAN TOA	1	86	86	86 110	0
SBA	5	141	142	156	14	SAN VNY	3	109	110	76	0
SCK SEO	17	136	137	151	14	SBA SFO	1	227	233	233	0
T SJC	11	111	114	123	9	SBA SMO	2	72	75	79	4
SMF	1 1	145	93	96	3	SBA TOA	1	83 68	87 69	71	2
MYP	1	76	87	87	0	SBP SFO	1	165	168	178	10
SAN	1 1	79	86	86	0	SBP SNS	14	97 56	98 63	63	0
SBA	!!!	98 244	101 253	114	13	SCK SMF	2	50	51	51	
SNS VIS	1 :	162	164	166	2	SCK SQL	2	53	55	55	0
D MRY	2	65	65	71	6	SCK TVL	2	83 258	263	84 270	
D RBL	1 1	150 257	152 259	152 265	0	SFO SMF	49	74	78	78	
D WJF	2	156	167	172	5	SEO SMX	1	187	189	201	12
ONT	1	125	138	144	6	SFO SNS	17	67 57	70 58	70 65	
MAZ	16	125	87 134	141	0 7	SEO TOA	1	301	306	306	
X MHV	1	68	69	71	2	SFO TYL	5 2	136	98	107	0
X MRY		231	234	71	13	SFO VIS	1	162	167	175	8
X OCN X PSP	14	94	96	96	0	SEO VNY	1	278	281	281	0
x SAN	113	94	94	94	0	SJC SMF	9 2	81 262	83 266	83 266	
X SBA	7 7	76 292	78 295	82 297	2	SJC SMX	3	164	165	174	4
X SFO	1	267	268	270	2	SJC TVL	3	129	133	133	
X SMF	1	324	324	327	3	SMF SMX-	1	319 234	235	322 240	
X SMX X TRM	15	116	117	123	6	SMF STS	4	58	59	67	
B MRY	1	245	250	263	13	SMX SNS SQL STS	1 1	120	121	12E 72	
8 OAK	2	306 82	310	312 83	2 0	SQL STS		65 332	337	344	7
	1 1	87	83	0.0			1 1	120	124	124	

3.3.3 Regression Model Development

The RNAV low altitude route length benefit potential within the State of California, identified in Section 3.3.2, was used as a basis for extrapolation to a national scale. This was accomplished by applying a regression analysis to the California data base in order to develop an analytical expression for use in estimating low altitude RNAV route length benefits as a function of specific airport pair characteristics.

The capability to extrapolate the regression analysis results to include the thousands of airport pairs for which route designs were not made is the key element which governed the methods used in the regression analysis itself. If the sole task of the regression analysis was to characterize the California benefits, an accurate and feasible approach would include quantitative consideration of the terrain features, VORTAC locations, their orientations relative to the terminals and perhaps even the interaction between the high and low altitude route structures. The computation of these characteristics for each of the 8,000 national airport pairs, however, was beyond the scope of this endeavor. Therefore, to facilitate automation and computational efficiency, the potential regression variables were confined to the data obtained from the initial IFR Tape and Airport Directory tape processing.

Thus, the first step in this regression analysis was to extract from the data base as many potential independent variables as possible which may have a bearing on the RNAV benefits. The following characteristics (independent variables) were computed for each of the 180 airport pairs in the California design:

- the number of departures from each airport; (2 characteristics)
- the number of airports and their total departures within 25 nm of each of the arrival and departure airports; (4 characteristics)
- the number of airports and their departures in the region connecting the two 25 nm radius terminal area circles (swath width of 50 nm); (2 characteristics)
- the number of airport pairs and their total flights whose great circle arcs intersect that of the airport pair being considered; (2 characteristics)
- the airport pair exchange; and
- the great circle distance.

The first item above is a measure of the activity at the specific airport. The second item relates to the general traffic density within each terminal area. The 25 nm value was used, as this generally coincides with the location of low altitude arrival and departure waypoints at the larger terminals. This is the average radius for low altitude arrival waypoints based upon recent RNAV terminal design studies [2]. Logically, smaller terminals do not exert influence over this great a distance. However, by including the number of departures of these terminals, the regression model is provided the necessary information to distinguish the larger terminals from the smaller (assuming that the distinction improves the model). The same comments also apply to item 3, the airports overflown. The characteristics in item 4 are indicative

of the general design complexity of the area. The number of flights is potentially indicative of the suitability of the existing VOR route.

The characteristics listed above, in combination with the route length results of the design effort, were the fundamental data base used in this regression analysis. It is apparent that there are a great many potential regression models which can be derived from these data. In order to converge upon a satisfactory model in an expeditious and cost-effective manner, the regression attempts were designed to achieve the following:

- Determine the most suitable dependent variable and the independent variable which contributes most significantly to the regression fit.
- (2) Determine the improvement in the regression fit of nonlinear variations of the significant variables.
- (3) Compare the resulting models and select that which is the most appropriate.

A detailed description of the regression attempts and the selection of the regression model is given in Appendix D.

3.3.4 Extrapolation to a National Scale

The expression relating the RNAV absolute mileage benefit to (1) the great circle distance, (2) number of airports within each terminal area, and (3) number of airports overflown (using a swath width of ±25 nm about the great circle ground track) was used to estimate the RNAV benefit for each airport pair whose great circle distance was greater than 50 nm and which had at least one low altitude exchange listed on the Peak Day Tape. This procedure required the computation of each of the four aforementioned regression variables for each airport pair analyzed, insertion into the regression equation and the computation and aggregation of the resulting benefits.

The only instances when the regression equation was altered were when negative RNAV benefits resulted. This may occur, according to the regression model, on short flights in very high density areas. Within the ground rules of the route design effort, however, negative RNAV benefits cannot occur (the option to duplicate VOR routes always exists) and a zero RNAV benefit was assumed under these circumstances.

The absolute RNAV benefit results provided the necessary data to estimate the per route, per flight and total route and flight mile savings. In order to express these results in terms of RNAV percent savings relative to VOR, it was necessary to estimate the specific VOR route lengths, which were not directly available from the extrapolation. Since the percent savings are derived by dividing a small mileage saving by a large route length, the sensitivity of the results to this estimation process is

negligible, and a sophisticated procedure was not required. Instead, the relationship of RNAV to great circle route length in the California structure was assumed to be representative of that relationship in the entire U.S. structure. The VOR route lengths could then be estimated by adding the difference in RNAV and great circle distance plus the difference in VOR and RNAV distance to the great circle distance. Within California, the average RNAV route was 1.89% longer than the great circle distance. Thus, for each airport pair over which the extrapolation was made, the RNAV route length of that pair was assumed to be the great circle distance plus 1.89%. The VOR route length was therefore derived by adding to the RNAV route length the estimated RNAV benefit over VOR. For the percent computations, 10 nm (5 nm in each terminal area) was subtracted from each route to account for the terminal areas.

Tables 3.21 through 3.23 present the national, inter-state benefit, and intra-state summaries, respectively. The regression equation coefficients are described in Appendix D. Of primary importance is the 2.3% per route RNAV benefit. The estimates were derived from consideration of individual airport pairs, with negative RNAV benefits replaced with zero. For this reason, the average benefits shown in the tables cannot be derived by analysis of the average parameter values.

Of particular interest is the variance between the inter-state and intra-state results. The estimated percent benefit of intra-state RNAV traffic is roughly twice that of inter-state traffic. This phenomenon can be attributed to several factors, not all of which necessarily pertain to inherent RNAY capabilities. It can be seen that the average terminal area density for inter-state traffic is somewhat higher than for intra-state. The number of airports overflown is approximately four times that of the intra-state results; more importantly, twice as many airports are overflown per flight length. One would expect that the longer flights would tend to interconnect areas of higher than average polulation density. A reduced RNAV benefit is therefore not so much the result of RNAV inefficiency, but more probably the result of a more comprehensive VOR structure which would be expected in such areas. The complexity of the VOR structure and its consequent interference with the RNAV routes may also be a factor. Further, the inter-state results tend to be consistent with the intra-state results for these regions of the country.

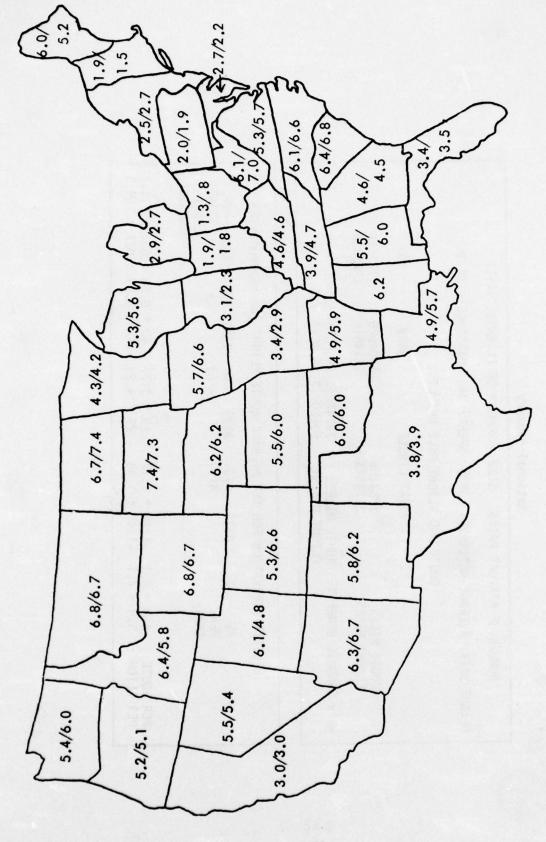
Figure 3.13 presents the average RNAV per route/per flight benefits % for each state based upon the intra-state results. The California route mile results are the same as those obtained in the route design effort. This is a necessary consequence of the regression analysis, although a slight discrepancy arises from the fact that the results presented here include the 17 airport pairs which were not considered in the design. The flight mile percent benefit, however, exhibits estimation error (3.0% compared to 2.6%), attributed to the inability of the regression model to provide a perfect fit.

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IMPLEMENTATION OF AREA NAVIGATION IN THE NATIONAL AIRSPACE SYST-ETC(U)
DEC 76 W H CLARK, E H BOLZ, H L SOLOMON DOT-FA72WA-3098
FAA-RD-76-196 AD-A039 225 UNCLASSIFIED 2 of 5 AD 39225

The results of this study indicate that higher RNAV benefits are provided in low density areas and vice-versa. The average flight length, however, also had a marked impact on the percent results. Delaware/Maryland/DC, for example, wherein the flights are necessarily short, actually received a very small absolute benefit (2 nm per route). The average great circle distance of only 79 nm caused the percent benefit to be unexpectedly high (relative to the subject state).

Table 3.24 summarizes the intra-state benefit derivation for the average per route savings (see Appendix D for regression parameters). The extent to which each characteristic is affected is shown. The states are listed in order of their absolute benefit. The most pronounced trend is found in the benefit degradation due to the smaller terminal area. This indicates that RNAV can produce a substantial benefit when only one of the terminals is in a high density area but this benefit is reduced considerably when both are high density. The apparent inability of RNAV to provide an enroute benefit for airport pairs where both terminals are of high density can be attributed to the fact that a good portion of these routes are contained within the terminal areas and generally good VOR routes already exist between these areas.

Based on this extrapolation, it was estimated that the average per flight mile benefit of RNAV over VOR in the low altitude structure is 2.36% (Table 3.21).



RNAV Route Mile/Flight Mile Percent Benefit Relative to VOR Routes Low Altitude Intra-State Routes Figure 3.13

Table 3.21

National Summary

				96 36	
14111	: 154 NM		VOR	1129666 2268436	
NUMBER OF AIRPORT PAIRS: 5122 NUMBER OF FLIGHTS: 14111	FLIGHT DATA: AVERAGE ALTITUDE: 8539 SPEED: 244 DISTANCE: 154 NM		RNAV	1104926 2214816	19 PCT) 36 PCT)
UMBER OF	ED: 244	RESULTS			24741 (2.19 PCT) 53620 (2.36 PCT)
5122 N	8539 SPE	ROUTE AND FLIGHT MILE RESULTS	GREAT CIRCLE	1084486 2173845	
T PAIRS:	TITUDE:	TE AND FL	GRE	210	ROUTE MILES: FLIGHT MILES
OF AIRPOR	VERAGE AL	ROL		LES	SENEFIT:
NUMBER	F DATA: A			ROUTE MILES FLIGHT MILES	RNAV ENROUTE BENEFIT:
	FLIGH				RNAV

A)	AVG. GREAT CIRCLE	* 211.7 * 154.1
pendix	B(4)	+.031
See Ap	AVG.	8.56
IGHT (B(3)	43 *
UTE/FL	AVG. A(L)	3.77
RAGE RO	AVG. B(2) AVG. B(3) AVG. B(4) AVG. A(S) A(L) θ GREAT CIRCLE DISTANCE	* 62
BENEFIT DERIVATION FOR THE AVERAGE ROUTE/FLIGHT (See Appendix A)		* 1.17
ON FOR	C B(1)	-1.45
RIVATI		= 4.1
FIT DE	AVG. BEN. (NM)	3.8
BENE		PER ROUTE: 4.8 = 4.1 -1.45 * 1.1729 * 3.7743 * 8.56 +.031 * 211.7 PER FLIGHT: 3.8 = 4.1 -1.45 * 1.3829 * 4.2343 * 6.24 +.031 * 154.1
		PER

Table 3.22

Inter-State Summary

NUMBER OF AIR	NUMBER OF AIRPORTS PAIRS: 3336		NUMBER OF FLIGHTS: 7855	7855	
FLIGHT DATA: AVERAGE ALTITUDE: 9172 SPEED: 247 DISTANCE: 191 NM	ALTITUDE: 9172	SPEED: 247	DISTANCE:	191 NM	
	ROUTE AND FLIGHT MILE RESULTS	T MILE RES	ULTS		
	GREAT CIRCLE	RNAV	VOR		
ROUTE MILES FLIGHT MILES	864084 1497315	880374 1525542	897123 1554310	- 77	
RNAV ENROUTE BENEFIT:	ROUTE MILES: 16749 (1.87 PCT) FLIGHT MILES: 28767 (1.85 PCT)	5749 (1.87 3767 (1.85	PCT)		

NET 11	DERIV	ATION	FOR	BENEFIL DEKIVATION FOR THE AVERAGE ROUTE/FLIGHT (See Appendix A)	WGE RO	UTE/FL	IGHT (See Apl	pendix	A)
	AVG. BEN.		B(1)	C B(1) AVG. B(2) AVG. B(3) AVG. B(4) A (S) A (L) O	B(2)	AVG.	B(3)	AVG.	B(4)	AVG. GREAT CIRCLE
UTE: IGHT:	5.0 =	4.1.	-1.45	* 1.33	* 62	4.02	43	11.38	+.031	PER ROUTE: 5.0 = 4.1 -1.45 * 1.3329 * 4.0243 * 11.38 +.031 * 259.0 PER FLIGHT: 3.7 = 4.1 -1.45 * 1.7029 * 4.7043 * 9.02 +.031 * 190.6

Table 3.23

Intra-State Summary

•	NA NA				
6256	100		~	40	
HTS:	INCE:		VOR	232550	
FLIG	JIST#	SULTS			PCT.)
R OF	40	E RE	RNAV	224558 689287	3.44
IOMBE	D: 2	MIL M	~	224	992 (353 (
186	SPEI	LIGH	ш		: 24
3:	7745	ROUTE AND FLIGHT MILE RESULTS	GREAT CIRCLE	220402 676530	ROUTE MILES: 7992 (3.44 PCT) FLIGHT MILES: 24853 (3.48 PCT)
PAI	UDE:	UTE /	EAT (220	TE M
NUMBER OF AIRPORT PAIRS: 1786 NUMBER OF FLIGHTS: 6256	FLIGHT DATA: AVERAGE ALTITUDE: 7745 SPEED: 240 DISTANCE: 108 NM	8	8		
F AIR	AGE A			LES: ILES:	H.
ER O	AVER			ROUTE MILES: FLIGHT MILES	RNAV ENROUTE BENEFIT:
S S	ATA:			ROUT FLIG	OUTE
	HT D				ENR
	FLIG				RNAV

BE	NEFIT	DERI	BENEFIT DERIVATION FOR THE AVERAGE ROUTE/FLIGHT	FOR 1	HE AVE	MGE RO	UTE/FL	IGHT.		
	AVG. BEN. (NM)	U	AVG. C 3(1) BEN. (NM)		AVG. B(2) AVG. B(3) AVG. B(4) A(S) A(L) θ	AVG.	B(3)	AVG.	B(4)	AVG. GREAT CIRCLE DISTANCE
PER ROUTE: 4.5 = 4.1 -1.45 * .8729 * 3.3043 * 3.28 +.031 * 123.4	4.5	4.1	-1.45	* .87	29	3.30	43	3.28	+.031	* 123.
PER FLIGHT:	4.0 =	4.1	-1.45	* .98	29	* 3.63	43	2.76	+.031	* 108.

Table 3.24

Average Intra-State RNAV Induced Per Route Benefit for Each State (Peak Day Traffic)

	10:31	To the			ROUNV Benet	fit Compo	nents (App	endix D)		The Survey	11002200
State	No. of Air- port Pairs	No. of Plts.	Aver- age Great Circle Dist.	Const- ant	Smlr. Ter- minal Area	Lrgr. Ter- minal Area	Over- flown Air- port	Route Length (3% of Great Circle	Corr. to Acc. for Zero Benefit Routes	RNAV Abso- lute Benefit (nmi)	& Saving Relative for VOR
NEV	6	15	158	4.06	.00	.00	21	4.96	.00	8.80	5.51
101	7	11	143	4.06	.00	.00	12	4.49	.00	8.43	5.85
ID	7	25	117	4.06	.00	04	24	3.69	.00	7.46	6.37
MYO	9	25	111	4.06	.00	10	.00	3.47	.00	7.44	6.75
MOA	13	56	112	4.06	.00	.00	13	3.50	.00	7.43	6.69
HOM	25	67	109	4.06	.00	.00	14	3.44	.00	7.36	6.76
ARI	7	39	118	4.06	21	20	.00	3.70	.00	7.34	6.26
UTA	3	9	114	4.06	.00	19	57	3.58	.00	6.88	6.09
TEX	170	593	179	4.06	80	89	-1.12	5.62	.00	6.87	3.83
SDA	13	43	93	4.06	.00	.00	16	2.92	.00	6.82	7.44
KED	18	68	108	4.06	24	27	29	3.39	.00	6.65	6.24
RIS	15	29	102	4.06	19	30	51	3.21	.00	6.26	6.23
LA	26	125	127	4.06	84	38	72	3.99	.00	6.10	4.86
W	14	45	102	4.06	10	-,39	70	3.21	.00	6.08	6.06
ALA	15	56	111	4.06	58	53	40	3.50	.00	6.04	5.52
EAN	38	114	111	4.06	38	69	47	3.50	.00	6.01	5.49
TEN	25	169	153	4.06	-1.11	65	-1.23	4.80	.00	5.87	3.87
COL	17	140	111	4.06	60	87	25	3.50	.00	5.83	5.34
IA	41	130	102	4.06	28	31	95	3.21	.00	5.73	5.73
MIN	38	108	134	4.06	38	-1.46	74	4.20	.00	5.67	4.30
HC	39	176	94	4.06	30	31	83	2.96	.00	5.57	6.08
088	34	88	106	4.06	68	70	59	3.34	.00	5.42	5.23
ec	7	26	96	4.06	42	29	73	2.71	.00	5.33	6.41
ARK	15	44	111	4.06	97	~.57	66	3.47	.00	5.33	4.94
GA	48	186	115	4.06	70	72	-1.04	3.61	.00	5.20	4.64
PLA	131	435	156	4.06	-1.29	77	-1.73	4.89	.01	5.18	3.37
100	13	26	88	4.06	78	35	59	2.77	.00	5.10	6.01
VA	21	87	98	4.06	90	58	65	3.08	.00	5.01	5.27
OKT	19	74	96	4.06	77	44	63	2.71	.00	4.94	5.95
104	31	81	92	4.06	61	95	61	2.88	.02	4.79	5.44
WIS	45	140	93	4.06	78	66	84	2.94	.04	4.77	5.30
KY	26	102	96	4.06	-1.51	59	72	3.01	.00	4.24	4.82
NO	45	167	123	4.06	-1.91	-1.15	77	3.85	.00	4.08	3.43
CAL	197	794	137	4.06	-1.57	-1.33	-1.42	4.32	.01	4.06	3.03
MIC	99	277	116	4.06	-1.75	93	-1.89	3.65	.05	3.19	2.80
ILL	72	285	106	4.06	89	-1.78	-1.60	3.33	.06	3.18	3.14
NJ	146	484	129	4.06	-1.44	-1.51	-2.66	4.06	.63	3.13	2.51
PA	86	245	122	4.06	-2.23	-1.35	-2.68	3.84	.66	2.30	1.96
DEL	16	54	79	4.06	-2.45	93	-1.26	2.49	.05	1.97	2.70
IND	36	96	97	4.06	-2.71	88	-1.96	3.06	.13	1.70	1.86
COM	74	249	90	4.06	-2.73	-1.10	-2.26	2.84	.80	1.60	1.92
CHI	79	271	103	4.06	-3.06	-1.10	-2.38	3.23	.51	1.26	1.31

3.4 IMPACT OF 4D RNAV CAPABILITY ON ARRIVAL CAPACITY

The purpose of this analysis is to evaluate the impact of 4D RNAV capability on arrival runway capacity and, therefore, arrival delays through a comparison of 4D RNAV time-control capabilites with presently planned metering and spacing automation improvements. Metering and spacing techniques are intended to improve the degree of precision of control of interarrival spacing from final approach fix to runway threshold. If these spacings may be controlled more precisely, smaller values for the buffer distance required in addition to the nominal in-trail spacings to insure that the nominal spacings are not violated may be used. While these buffer values are not explicitly a part of control procedure, they exist in reality as the margin for error provided routinely in final approach control. As a result, the average interarrival spacing during periods of arrival saturation may be reduced if the precision to which threshold arrival time is controlled is improved.

Results of analytical studies and recent real time cockpit simulation and fast time simulation studies have shown that 4D RNAV techniques may be employed to significantly reduce arrival control error in comparison with the control error expected with advanced metering and spacing technology. The control error values found to be representative of open-loop control systems (M&S) are on the order of 10-15 seconds, while 4D RNAV has been shown to yield accuracies of approximately five seconds (one sigma). The discussion which follows derives the effect of such control error improvement on runway arrival capacity.

3.4.1 Runway Capacity Analysis

The techniques used for this analysis are based upon the relationships derived in Reference 24. That document describes the methods for quantifying arrival runway capacity covering a variety of conditions and assumptions. In order to keep the present analysis brief and oriented towards the primary problem of concern, the subject airport (San Francisco) was chosen in order to eliminate many of the complicating factors which impact arrival capacity. San Francisco serves a high volume of traffic, and under most circumstances all arrivals are served by one dedicated landing runway (28L), while departures are served by a dedicated runway of their own(1). The primary data sources used for the analysis concern the nature of the arrival traffic. Gross traffic data [25] were used for determining the overall traffic demand, while detailed data concerning the types of air carrier aircraft which serve SFO [26] and the scheduled arrival times[20] were used for determining runway capacity and to provide a baseline demand pattern for comparison with the capacity values determined to represent the cases of M & S, of improved control with 4D RNAV, and the baseline case of no arrival error.

In the present ATC environment, arrival runway capacity is limited primarily by the interarrival spacings required for IFR operations. The required spacings are three miles, or, for wake turbulence avoidance*, five miles for a light aircraft following a heavy transport, or four miles for

*Note that these separations have been further modified since this analysis was performed.

a heavy aircraft following a heavy aircraft [15]. The separation which results in terms of time is dependent upon the speeds of the the two aircraft of concern, and is significantly different depending on which aircraft is the faster. When the leading aircraft is faster, the minimum separation must be provided at the outer marker (five or six miles out), since the outer marker is the latest point at which a common path to the threshold can begin, and so the spacing at the threshold is longer. However, in the case where the trailing aircraft is faster, the minimum separation must be provided at the threshold. The minimum separation time, m, is [24]:

$$m (V_2, V_1) = \begin{cases} \delta_{12}/V_2 & \text{where } V_2 \ge V_1 \\ \delta_{12}/V_2 + \gamma(\frac{1}{V_2} - \frac{1}{V_1}) & \text{where } V_1 > V_2 \end{cases}$$

where:

m = Minimum required time separation at the threshold

V, = Speed of leading aircraft

 V_2 = Speed of trailing aircraft

 δ_{12} = Minimum separation (depending upon the aircraft types)

y = Distance from outer marker to threshold

In order to apply the equation the characteristics of the arrival traffic at SFO must be determined. These characteristics include the mix of aircraft types and the final approach speed of each type (1.3 V_{SO} was used for this analysis). Using the data from Reference 26 plus aircraft performance (stall speed) data for each aircraft type, the compilation of air carrier traffic shown in Table 3.25 was developed. Since no reliable data concerning GA activity other than total operation count was readily available, the distribution shown in Table 3.26 was used. The average traffic demand in FY71 [25] was 1032 itinerant operations, of which 815 (79%) were air carrier and 217 (21%) were general aviation. Applying these ratios to the data in Tables 3.25 and 3.26 yielded the overall distribution shown in Table 3.27. The percent distribution value for each speed class may be considered to be the probability that any given arrival aircraft is of the class indicated. These probabilities may be applied to the values for minimum separation time in such a manner that the average expected time separation for all arrivals may be computed. From the average separation, the average operations per hour (landing capacity) may be computed. The following relationships are derived in Reference 24:

$$s = pMp^T = \sum_{ij} p_i M_{ij} p_j$$

where

s = Average expected time separation

 $p = Vector of probabilities, where <math>p = [p, p_2, \dots, p_n]$

M = Matrix of minimum separation times

 $M_{ij} = m (V_i, V_j)$

superscript T indicates matrix transpose

Runway capacity, c, is then the reciprocal of s.

Table 3.25 Air Carrier Arrival Traffic at SFO

Aircraft Type	Annual Arrivals	1.5V so (kt)	Landing Category*	Speed Class
707-100B	10032	153	D	5
707-300B	3846	144	D	4
707-300C	4633	144	D	4
720B	7570	151	D	4 5 4 4 4 7 5 6 7
727-100	12512	141	C	4
727-200	8481	148	C C C	4
737-200	15328	150		4
747	6032	156	D	7
DC-8-10/20/30	5103	157	D	5
DC-8-50	5710	162	D	6
DC-8-61	5398	166	D	7
DC-8-62	1494	166	D	7
DC-8-63	1696	166	D	7
DC-9-10	937	150	C	4
DC-9-15F	3	156	C	4 5 4 7
DC-9-30	3423	145	C	4
DC-10-10	2120	156	D	7
L-100-20	253	115	В -	7 3
L-1011	224	148	D	7
CV-580	359	127	В	3
CV-880	1505	162	D	6
F-27	4108	105	В	1
TOTAL	100767	en de son-	osqui et zo	E E H

Speed Classes:	Speed Range	Average Spee
	1< 110 kt	105
	2< 120 kt	115
	3≤ 140 kt	128
	4< 150 kt	145
	4 <u><</u> 150 kt 5 <u><</u> 160 kt	155
	6 > 160 kt	165
	7 All Heavies (> 300,000 lb)	160

^{*} Category A. Speed 50-90 knots or weight 30,000 lbs. or less. Category B. Speed 91-120 knots or weight 30,001-60,000 lbs. Category C. Speed 121-140 knots or weight 60,001-150,000 lbs. Category D. Speed 141-165 knots or weight over 150,000 lbs.

Table 3.26 General Aviation Traffic at SFO (Assumed)

Aircraft Type	Per Cent Distribution	Speed Class
Twin Piston	20%	1
Slow Turboprop	30%	2
Fast Turboprop	15%	3
Slow Turbojet	15%	3
Fast Turbojet	20%	4

Table 3.27 SFO Traffic Distribution

Speed Class	Per Cent Distribution	Average Approach Speed
1	7.43%	105 kt
2	6.50%	115 kt
3	6.59%	128 kt
4	42.74%	145 kt
5	17.80%	155 kt
6	5.65%	165 kt
7	13.29%	160 kt

Capacity was computed for the data given in Table 3.27. The resulting values are shown below:

$$p = \begin{bmatrix} .0743 & .0650 & .0659 & .4274 & .1780 & .0565 & .1329 \end{bmatrix}$$

$$M = \begin{bmatrix} 102.9 & 120.8 & 139.8 & 159.6 & 169.2 & 242.1 & 177.7 \\ 93.9 & 93.9 & 113.0 & 132.8 & 142.4 & 209.3 & 150.8 \\ 84.4 & 84.4 & 84.4 & 102.4 & 113.8 & 174.4 & 122.2 \\ 74.5 & 74.5 & 74.5 & 74.5 & 84.1 & 138.1 & 92.5 \\ 69.7 & 69.7 & 69.7 & 69.7 & 69.7 & 120.5 & 78.1 \\ 67.5 & 67.5 & 67.5 & 67.5 & 67.5 & 90.0 & 71.6 \\ 65.5 & 65.5 & 65.5 & 65.5 & 65.5 & 109.1 & 65.5 \end{bmatrix}$$

s = 92.44 sec/operation, c = 38.94 operations/hour

This result, 39 IFR operations per hour, is the arrival capacity of San Francisco based upon the assumption of zero time control error. There are, however, a few reasons why this estimate is believed to understate actual arrival capacity. First of all, during periods of high demand, controllers will often exercise a rudimentary form of speed class sequencing, which tends

to improve landing rate. Even more importantly, however, is the fact that controllers will often request slower aircraft to conduct their approach at a higher speed than desired. This not only increases overall average speed but also has the effect of speed class sequencing on arrival capacity. Although understated, the computed capacity value serves as a useful baseline for the following analysis of the impact of arrival time control error on landing capacity.

The effect of arrival time control error may be represented as an added buffer zone over and above the nominal in-trail separation requirement. The nominal size of this buffer zone is dependent upon three factors, presuming that the arrival control error tends to behave in a gaussian manner. These are the one-sigma values for control error, the desired degree of confidence in not exceeding the specified buffer zone, and the velocities of the aircraft involved. The equations for the buffer time are as follows [24].

$$b(V_2, V_1) = \begin{cases} \sigma_o & q(p_v) \\ \text{Larger of 0 or} \left[\sigma_o & q(p_v) - \delta_{12} \left(\frac{1}{V_2} - \frac{1}{V_1}\right)\right] & \text{when } V_1 > V_2 \end{cases}$$

where $b(V_2, V_1) = Buffer separation time$

 $\sigma_{o}q(p_{v}) = one-sigma control error value for the desired level of confidence <math>P_{v}$.

In that reference, the value used for P $_{\rm V}$ under present separation criteria is 5% (i.e. based on a 95% confidence level). The resulting value of q(p $_{\rm V}$) is 1.645, from standard cumulative distribution function tables for normal distributions. The average contribution of these buffer times may be found in a manner identical to that used previously for finding average separation time.

$$s_b = pBp^T = \sum_{ij} p_i B_{ij} p_j$$

where

s_b = Average expected buffer separation

B = Matrix of Buffer separation times

$$B = b (V_i, V_i)$$

The runway capacity, c_t , is the reciprical of s_t , where:

$$s_t = s + s_b$$

$$c_t = \frac{1}{s + s_b}$$

In order to compute new values for total capacity, c_{t} , values for control error must be determined. The expected control error has been the subject of analytical studies, real time cockpit simulation studies, and fast time simulation studies. In Reference 27 the three cases of manual spacing, computer aided spacing and fully automated closed-loop spacing are analyzed. The expected interarrival errors determined were 20 seconds for manual control, 11 seconds for computer aided control, and 5 seconds for automated time control (one-sigma values). Actual cockpit simulator experiments using subject pilots [6] have been conducted recently which demonstrated that open-loop time control errors of 15.7 seconds (corresponding somewhat to the computer aided M & S case), and 4D RNAV time control errors of 5.4 seconds (one-sigma values) can be achieved. The open-loop technique used in those experiment was based upon the automated Metering and Spacing principle. However, a larger time control error (15.7 sec.) than expected (11 sec.) resulted due to the fact that M & S capabilities were not exercised to the greatest possible extent. Rather than to issue corrective vectors throughout maneuvers to the final approach course, the aircraft was allowed to proceed without a final correction vector (openloop) after departing the last fix prior to final approach intercept. Therefore, the resulting time control errors are larger than those expected of a fully-implemented M & S system. In confirmation of the expected performance of 4D RNAV time control, a recent fast-time simulation study [28] of advanced 4D RNAV time control systems has shown that delivery error to the outer marker would be expected to be on the order of five seconds. Therefore, for purposes of this analysis, the error times determined in Reference 27 (20,11 and 5 seconds) will be used. The matrices of buffer separation times, B, for these three cases are as follows:

Case 1: $\sigma_0 = 20.0 \text{ seconds} \quad \underline{\text{Manual Control}}$

$$\sigma_{Q}q(p_{V}) = 32.90$$

 $s_b = 27.56$, $s_t = 92.44 + 27.56 = 120.00$ seconds/operation

 $c_{+} = 30.0 \text{ operations/hour}$

Case 2: $\sigma_o = 11.0 \text{ seconds} \quad \underline{\text{Metering and Spacing}}$ $\sigma_o q(p_v) = 18.10$

 s_b = 13.86, s_t = 92.44 + 13.86 = 106.30 seconds/operation c_t = 33.87 operations/hour

Case 3: $\sigma_{o} = 5.0 \text{ seconds } \frac{4D \text{ RNAV M & S}}{4D \text{ RNAV M & S}}$ $\sigma_{o} q(p_{v}) = 8.23$

 s_b = 5.60, s_t = 92.44 + 5.60 = 98.04 seconds/operation c_t = 36.72 operations/hour

The values which have resulted for average buffer separation, $s_b,$ are relatively independent of the factors discussed earlier which tended make the earlier error-free runway capacity number conservative. The only variable of significance which would affect the s_b computation is the desired value for buffer violation probability, $p_{\nu},$ for which 5% was used here.

The basic metering and spacing capability will improve capacity by 12.9%, from 30.00 to 33.87 operations per hour. The degree of improvement which results from improving arrival time control accuracy through the use of 4D RNAV is expressed by the increase in arrival capacity, from 33.87 to 36.72 operations per hour, an improvement of 8.4%. This is a very significant increase in runway capacity, particularly in view of the fact that no runway improvements or significant ATC automation improvements are involved in this comparison (both presume metering and spacing automation), but that the entire improvement is due to the addition of time-control navigation capability (4D) to the basic airborne 2D or 3D RNAV computer system.

The capacity improvement is put in perspective in Figure 3.14, which shows the three computed capacity values overlaid on a plot of the estimated total hourly arrival demand for San Francisco. The hourly demand data for

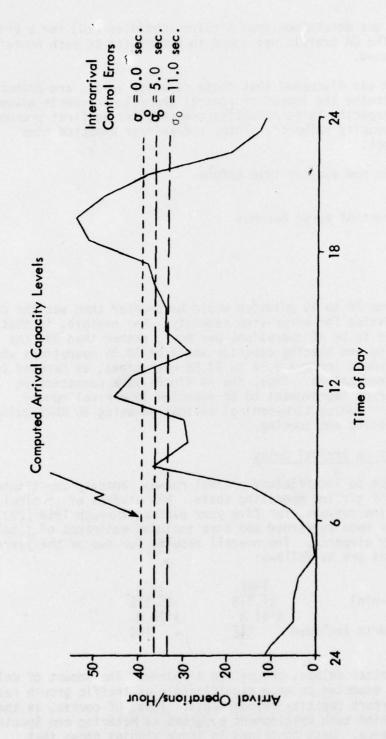


Figure 3.14 Arrival Operations at San Francisco

air carrier aircraft was determined from airline schedules [20] for a Friday in December, 1974. The GA traffic was added in proportion to each hours' data to get total demand.

Earlier the fact was discussed that these capacity values are probably understated. To determine the impact of control error improvements given that the error-free capacity is in actuality greater, let us first presume a greater error free capacity value c'. Then, the average expected time separation, s', becomes:

$$s' = \frac{1}{c'}$$
, which is now smaller than before.

Total capacity with control error becomes

$$c'_t = \frac{1}{s' + s_b}$$

So the improvement from 2D to 4D guidance would be greater than was the case with the understated value for error-free capacity. For example, if that capacity were presumed to be 45 operations per hour, rather than 39, the computer-aided metering and spacing capacity would be 38.36 operations while the 4D RNAV capacity would improve 9.6% to 42.06 operations, as opposed to the 8.4% gain determined before. Thus, the 8% figure is a conservative estimate of the degree of improvement to be expected in arrival runway capacity as a result of adding time-control navigation using 4D RNAV guidance to computer aided metering and spacing.

3.4.2 Capacity Impact on Arrival Delay

Arrival delays due to insufficient arrival runway capacity constitute an important segment of airline operating costs. FAA studies of terminal area delays, based on airline surveys, for five year periods through 1969 [29] and through 1974 [30] have been performed and have included estimates of total airline delay at major airports. The overall results for two of the years within the time periods are as follows:

Year	1968	1973
Total Delay (K-min)	21,518	25,922
Delay Cost	\$141 M	\$176 M
Number of Airports Include	ed 117	103

These include all terminal delays, ground and airborne. The amount of delay and related costs are expected to grow significantly as traffic growth rates continue to exceed airport capacity improvements. This, of course, is the primary motivation behind such development programs as Metering and Spacing, and Wake Vortex Avoidance. Data contained in these studies shows that approximately one-half the delay time is due to ATC causes (as reported by the airlines) and half is due to airport congestion (Table A-3 in Reference 30).

Nearly all of the ATC terminal delays are to arrival aircraft and so one-half of total delay appears as arrival holding and vectoring delay. In order to account for the occasional severe weather delays and so to consider only those normal, recurring delays due to demand temporarily exceeding normal IFR capacity, it has been assumed for purposes of this analysis that one-third of total terminal area delay is normal arrival holding delay which results from the natural capacity limitations of the runway configuration and arrival traffic type mix.

Nominal per-minute delay costs as used by the participating airlines are also reported in [30]. The costs stated are averages across the airline fleets. The values stated are (1974):

ATC delays \$9.46/minute Airport delays \$6.54/minute \$7.91/minute

The ATC delay costs are, of course, higher than airport delay costs since they occur while airborne (for the most part), while the airport delay cost are primarily ground delay-related. The \$9.46 airline delay cost figure will be used in the following since it is consistent with data used elsewhere in this report and since it simplifies the analysis.

As explained in the previous section, the derived capacity figures understate actual arrival capacity at San Francisco. Therefore, it is desirable to determine approximately what the actual arrival capacity is under present circumstances. Once known, the improvements in capacity which would result from the implementation of M & S and 4D RNAV capabilities may be computed. After consideration of the several contributors to arrival delay, which include the inability to serve randomly arriving aircraft immediately due to the necessity of assembling them into an orderly final approach queue, and the quite sizeable impact of periods of temporary excess demand where aircraft must be held until approach slots are available, it was determined that temporarily excessive demand is far and away the largest cause of delays at airports where long delays are ordinarily experienced regardless of weather conditions. A straightforward model which uses hourly arrival rate data, such as shown in Figure 3.14, has been developed which computes total 24 hour delay based upon a presumed arrival capacity value. The model presumes that no aircraft are delayed when capacity exceeds demand. This assumption simplifies the model considerably, produces conservative results, and does not have a large impact on model accuracy when applied to the busier airports. When capacity is exceeded, that excess number of aircraft are delayed into the next hour where they are landed, and others are delayed if capacity is still exceeded. This process continues until demand decreases to the point where all waiting aircraft have been landed. For each problem studied, delay values are computed over a range of presumed capacity values so that a relationship of delay to capacity may be developed.

The model used to compute total daily delay operates by examining the number of arrivals scheduled for each hour, adding any surplus arrivals left over from the previous hour, computing delay for that hour and computing the number of aircraft left (if any) at the end of the hour. Daily delay is then the sum of the individual hourly delays. The relationships are as follows:

 $A = presumed hourly capacity \\ D_i = arrival demand for ith hour \\ S_{i-1} = surplus from previous hour \\ \delta_i = total delay during ith hour (in hours) \\ if D_i < A: \\ SL_i = surplus arrivals landed in ith hour \\ SL_i = Min \left[A - D_i \text{ or } S_{i-1}\right] \\ S_i = S_{i-1} - SL_i \\ \delta_i = \frac{SL_i^2}{2(A - D_i)} + S_i \\ \end{cases}$

if
$$D_i \ge A$$
:

$$S_i = S_{i-1} + D_i - A$$

$$\delta_i = S_{i-1} + \frac{D_i - A}{2}$$

 δ = total daily delay = $\Sigma \delta_i$ (converted to minutes)

The first step towards ascertaining SFO IFR arrival capacity from the delay data in References 29 and 30 has been to run the model for demand values derived from the annual demand data presented in those studies. The delay-capacity results for those two cases are shown in Figure 3.15. The various total daily demand values have been modeled by modifying the arrival pattern shown in Figure 3.14 in direct proportion to the demand value. The data used were determined as follows, where air carrier operations are assumed to represent all IFR operations:

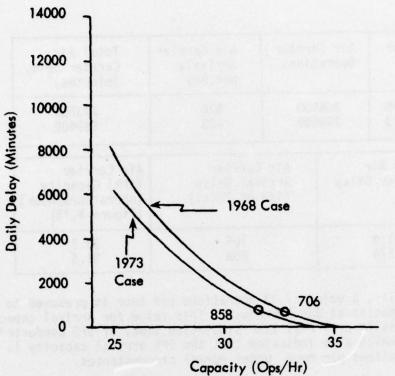


Figure 3.15 Delay Data Cases for SFO [29,30]

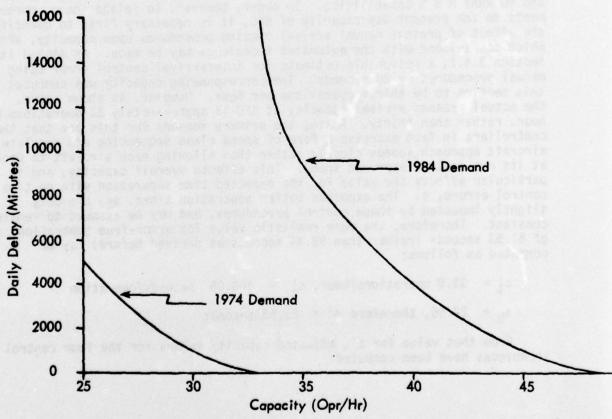


Figure 3.16 SFO Delay Projection for 1974 and 1984 Air Carrier Demand 3-7.7

Referen	ce	Year	Air Ca Operat		Air Carrier Arrivals per Day	Total Air Carrier Delay (minutes)	
29 30		1968 1973		306900 420 294500 403		773500 939400	
Year	Daily Air Carrier Delay		Arri	Carrier val Delay of Total)	Air Carrier (IFR) Capacity (operations/hour) (Figure 3.15)		
1968 1972		2119 2574			706 858	32.9 31.5	

Based upon these results, a value of 33 operations per hour is presumed to be the normal arrival capacity at San Francisco. This value for arrival capacity is verified by the results of a real time simulation study of SFO conducted at NAFEC in 1971[31] which also indicated that the IFR arrival capacity is approximately 33 operations per hour under normal circumstances.

Now that a reasonable estimate for San Francisco arrival capacity is available, it is possible to relate capacity increases to Metering and Spacing and 4D RNAV M & S capabilities. In order, however, to relate these improvements to the present day capacity of SFO, it is necessary first to quantify the effect of present manual arrival spacing procedures upon capacity, after which comparisons with the automated techniques may be made. As stated in Section 3.4.1, a reasonable estimate for interarrival control error using manual procedures in 20 seconds. The corresponding capacity was computed in this section to be thirty operations per hour. However, as shown above, the actual present arrival capacity at SFO is approximately 33 operations per hour, rather than thirty. Again, the primary reasons for this are that the controllers in fact exercise a form of speed class sequencing and regulate aircraft approach speeds closely rather than allowing each aircraft to proceed at its own desired approach speed. This affects overall capacity, and in particular affects the value for the expected time separation with no time control errors, s. The expected buffer separation times, sb, are only slightly impacted by these control procedures, and may be assumed to remain constant. Therefore, the more realistic value for error-free separation, s', of 81.53 seconds (rather than 92.44 seconds as derived before) may be computed as follows:

- $c_t^{\prime} = 33.0 \text{ operations/hour, } s_t^{\prime} = 109.09 \text{ seconds/operation}$
- $s_b = 27.56$, therefore s' = 81.53 seconds

From that value for s', adjusted capacity values for the four control techniques have been computed:

Case	Time Error	Computed Capacity	Adjusted Capacity
Manua1	20.0	30.0	33.0
M&S	11.0	33.9	37.7
4D RNAV	5.0	36.7	41.3
Idea1	0.0	38.9	44.1

What this table essentially says is that the theoretical arrival capacity at SFO given its runway conditions, traffic type mix and demand pattern is 44.1 operations per hour, while the imprecision of manual sequencing techniques reduces the effective capacity to 33 operations per hour, 25% less. This, of course, presumes that runway occupancy would not in fact constrain actual capacity below 44.1 operations, were that many able to land.

In order to establish the effect of these potential improvements on delays with present and projected traff.ic levels, demand data for 1974 and 1984 have been taken from the forecasts in Reference 32. These data have been used to provide daily air carrier arrival rates which were used as before as inputs to the delay-capacity model. The results of the model runs for each case are presented in Figure 3.16. The resulting delay values are as follows:

Case	Capacity	Daily Demand (No.of Arrivals)				Daily AC Cost (Dollars)		Annual AC Cost (Dollars)	
		1974	1984	1974	1984	1974	1984	1974	1984
Manual	33.0	397	575	50	15910	\$0.5K	\$150.5K	\$0.2M	\$54.9M
M&S	37.7	397	575	0	6124	0	57.9K	0	21.1M
4D RNAV	41.3	397	575	0	3038	0	28.7K	0	10.5M
Ideal	44.1	397	575	0	1459	0	13.8K	0	5.0M

The airline (IFR) delay costs are computed from the \$9.46 per minute rate discussed earlier.

The results of this analysis are obviously quite significant. In 1984, 4D RNAV used in combination with Metering and Spacing capabilities can save \$44,400,000 annually in terms of airline delay costs, while M & S alone could save \$33,800,000. It is best at this time to review the assumptions used in arriving at these numbers in order to view them from the proper perspective.

- a) Runway occupancy is not a limiting factor -- it is assumed that runway occupancy constraints would not be a limit for operations at the 4D rate (41.3/hour). This is reasonable since current limitations are due to the wake vortex and control error problems, not runway and taxiway problems.
- b) No significant arrival runway improvements -- no significant runway improvements (other than taxiway improvements) are currently planned at SFO [46].

- c) Peak flattening was not considered -- Peak flattening is inevitable as operation rates increase. Airlines would not tolerate the delay (27 min per arrival) shown in the 1984 figures without taking action to adjust schedules. Therefore the delay would be lower than that. However, other penalties would be suffered (lost revenue from contracted schedules and inconvenient arrival times) in the process, and so it is not unrealistic to considered non-peak-flattened delay as representing operating penalty for comparative purposes.
- d) Traffic growth projections -- Latest available projection data is being used [32].
- e) Aircraft buffer separation times are based on a 5% buffer violation probability -- If a larger value were presumed, M & S and 4D RNAV would make lesser improvements to overall delays, although unimproved delay values would remain the same.

In order to test the effect of the last assumption, the arrival rates and delay values have been recomputed presuming a 10% buffer violation probability. The results are as follows:

10%	Buffer	Violation	Probability
IUR	Duilei	VIOIACION	11 ODGD111 Cy

Case	Capacity	Daily D			Delay utes)	Daily A	C Cost		AC Cost lars)
		1974	1984	1974	1 1984	1974	1 1984	1974	1 1984
Manua 1	33.0	397	575	50	15910	\$0.5K	\$150.5K	\$0.2M	\$54.9M
M&S	36.4	397	575	0	7548	0	71.4K	0	26.1M
4D RNAV	38.8	397	575	0	5047	0	47.7K	0	17.4M
Ideal	40.7	397	575	0	3472	0	32.8K	0	12.0M

The total annual 4D RNAV with M & S savings for air carriers is now \$37.5 million rather than \$44.4 million, a decrease of 16% even though the allowable buffer violation probability has been doubled. To be conservative, 10% buffer violation probability figures will be used for the remainder of the analysis.

3.4.3 Capacity Impact at the Major U.S. Terminals

In order to be able to extrapolate the M & S and 4D RNAV M & S benefits over the major U.S. terminals, a simple procedure for estimating the delay/capacity relationships at other terminals has been developed. The only additional assumption required is that the average buffer separation time values, sb, at the other terminals are similar to those at San Francisco. The sb, values depend primarily on traffic type mix, which does not remain contant from terminal to terminal. However, the sb values are not expected to change significantly.

The airports specifically studied here include Denver, Miami, Chicago, New York JFK, New York LGA, Newark and Philadelphia. In order to establish a

demand pattern for each airport, Reference 20 was used to obtain the air carrier arrival traffic sample (as illustrated in Figure 3.14 for SFO). The same procedure as before, using References 29 and 30, was used to estimate existing capacity at each of the seven airports. Then the projected capacity using Metering and Spacing and using 4D RNAV M & S was computed from the sb values derived for San Francisco (based upon the 10% buffer violation probability level). Where more than one dedicated arrival runway exists (ORD & JFK, also MIA was modified to account for independent parallel runways with mixed operations) one-half of the sb values were used in order to represent the split arrival situation. The results of these evaluations are presented as Table 3.28. In order to estimate, conservatively, the overall U.S. impact, estimates have been derived for the impact at the twenty-five highest delay airports in the U.S. [30]. The method used here for estimation was to determine the least-square exponential relationship of delay per operation as a function of 1984 daily air carrier arrival operations for each case: M & S versus manual control, and 4D RNAV with M & S versus manual control. The resulting curve fit relationships (Table 3.28) were applied to the twenty-five top delay airports, with the results given in Table 3.29. The values shown in Table 3.29 for the eight airports studied are curve fit values, which differ somewhat for each individual airport from the values in Table 3.28. However, by virtue of the curve fit process, the totals of either set of values for those eight airports are almost identical. The exponential relationship of total delay versus operations count was selected not only since it provided a reasonable result, but since the exponential relationship has been shown to properly fit substantial delay data compiled in other studies [65].

The 4D RNAV benefit column in Table 3.29 was computed as the difference of the other two values for which the curve fit technique was used. The total resulting 4D RNAV arrival delay benefit for the 25 high delay airports is \$85,500,000 annually, based upon projected 1984 air carrier traffic levels.

Table 3.28 1984 Eight Airport Delay Projection (minutes)

					Minutes	Savina	-minute	1
Airport	Case		Estimated	Daily	Delay per			peration
All post		Arrivals	Capacity	Delay	Operation	M&5	4D	M&S +40
ORD	Manual	849	55.0	8040	9.47			
	MAS		59.7	813	0.96	8.51		
	4D		62.9	149	0.18		0.78	9.29
h (teles)	Ideal		65.3	10	0.01		1	
JFK	Manual	488	42.0	5702	11.68			
	M&S		44.7	2564	5,25	6.43		
	4D		46.4	1443	2.96		2.30	8.73
	Ideal		47.8	865	1.77			
LGA	Manual	441	28.5	5946	13.48			
	M&S		31.0	2092	4.74	8.74		
	4D		32.8	447	1.01		3.73	12.47
	Ideal		34.1	112	0.25			
EWR	Manual	436	28.0	8220	18,85		10 19	WW/ 95
199-51	M&S	13.19	30.4	4908	11.26	7.60	1 100	
	4D		32.1	2931	6.72	1	4.53	12.13
	Ideal		33.4	1683	3.86			
PHL	Manual	362	24.0	0033	8.38			
	M&S		25.8	876	2.42	5.96	TARE	
	4D	101	26.9	453	1.25		1.17	7.13
	Ideal		27.8	232	0.64		10	07,000
MIA	Manual	432	33.0	5271	12.20			
	M&S		35.5	2805	6.49	5.71		
	4D		37.2	2107	4.88		1.61	7.32
	Ideal		38.5	1688	3.91			
DEN	Manual	421	29.0	7475	17.76			
	M&S		31.6	3802	9.03	8.72		
	40		33,4	2608	6.19		2.84	11.56
	Ideal		34.8	1960	4.66			
SFO	Manual	575	33.0	15910	27.67			
	MAS		36.4	7548	13.13	14.54		
	4D		38.8	5047	8.78		4.35	18.89
	Ideal		40.7	3472	6.04			

Regression Equation: Savings/Operation = Ae^{Bx} , x = Arrivals M & S: A = 1.16873, B = 0.0040863 A = 1.31224, B = 0.0044385

Table 3.29 1984 Annual Airline Arrival Delay Savings Projection

e i Limb	1984 Daily		s per Ope Minutes)	eration	Airline Annual Savings (\$ Millions)			
Airport	Airline			M&S			M&S	
	Arrivals	M&S	4D	+ 4D	M&S	4D	+ 4D	
ORD	849	13.27	4.04	17.31	38.9	11.8	50.7	
ATL	808	12.18	3.59	15.77	34.0	10.0	44.0	
JFK	488	8.58	2.87	11.45	14.5	4.8	19.3	
LGA	441	7.08	2.21	9.29	10.8	3.4	14.2	
SFO	575	12.25	4.59	16.84	24.3	9.1	33.4	
LAX	634	15.59	6.29	21.88	34.1	13.8	47.9	
DEN	421	6.53	1.97	8.50	9.5	2.9	12.4	
PHL	362	5.13	1.41	6.54	6.4	1.8	8.2	
EWR	436	6.94	2.15	9.09	10.4	3.2	13.7	
MIA	432	6.83	2.10	8.93	10.2	3.1	13.3	
DAL/FTW	585	12.76	4.85	17.61	25.8	9.8	35.6	
DCA	319	4.30	1.11	5.41	4.7	1.2	6.0	
PIT	347	4.83	1.29	6.12	5.8	1.6	7.3	
BOS	422	6.56	1.98	8.54	9.6	2.9	12.4	
CLE	192	2.56	0.52	3.08	1.7	0.3	2.0	
DTW	274	3.58	0.85	4.43	3.4	0.8	4.2	
MSY	237	3.08	0.68	3.76	2.5	0.6	3.1	
LAS	230	2.99	0.65	3.64	2.4	0.5	2.9	
HNL	301	4.00	0.99	4.94	4.2	1.0	5.2	
STL	356	5.01	1.36	6.37	6.1	1.7	7.8	
FLL	110	1.83	0.31	2.14	0.7	0.1	0.8	
TPA	158	2.23	0.42	2.65	1.2	0.2	1.4	
MSP	205	2.70	0.56	3.26	1.9	0.4	2.3	
SEA	182	2.46	0.48	2.94	1.6	0.3	1.9	
BAL	151	2.17	0.39	2.56	1.1	0.2	1.3	
TOTAL					265.8	85.5	351.3	
25 Airport	4D RNAV I	mpact = \$8	5,500,00	0				

3.5 RNAV EQUIPMENT CAPABILITIES AND COSTS

This section investigates the capabilities offered by the spectrum of RNAV equipment presently available and the costs of each of these levels of capability, and relates system equipment costs to the several types of airspace users. Many existing RNAV systems do not meet proposed Minimum Operational Characteristics (MOC) requirements, and so the cost of providing such systems which do are estimated. Even though the total span of capabilities available is extremely broad, full 2D RNAV benefits are available to aircraft which equip with the most basic systems which meet MOC requirements.

3.5.1 RNAV Equipment Costs

This section considers the various levels of RNAV capability which are available and the incremental costs necessary to obtain each level. Both those minimum capabilities which will probably be required (Section 5.3), and additional capabilities which would serve to satisfy other user needs and desires are identified and cost estimates made. However, it should be recognized that each capability is not necessarily obtainable (or quantifiable) separately from all other capabilities since they typically come in groups or packages. All cost data were taken from published manufacturers' price lists.

For purposes of this analysis it is useful to separate capabilities into three type categories: functional capabilities, interface capabilities and data storage and management capabilities. The functional capabilities include the types of computations performed, modes of operation and output data generated. Interface capabilities include types of sensors usable with a system, compatible sensor manufacturers and models, and types of control systems and displays which can be driven. Data storage and management capabilities include multiple waypoint storage, automatic data entry, and mass data storage and retrieval. In detail, these capability levels are as follows:

Functional Capabilities

- Basic 2D area navigation capability
- Track angle computation
- 3D (VNAV) guidance capability
- 4D (time control) guidance capability
- Earth-oriented versus station-oriented computation
- Parallel offset capability
- Along track offset capability (VNAV mode)
- Direct-to-waypoint capability
- Approach mode sensitivity selection
- Slant range error compensation
- Computed data outputs (time to waypoint, ground speed, winds, command descent rate, etc.)
- Map display capability

Interface Capabilities

VOR and DME interface

Altimeter input (slant range error compensation)

Altimeter input (3D RNAV)

Air data input

Multisensor capability (DME/DME, VLF, OMEGA, LORAN)

ARINC system interfaces

• Non-ARINC interfaces, manufacturer and model limitations

Display interface capabilities (CDI, HSI, RMI)

• Control system interface capabilities (flight director, autopilot)

Data Storage and Management Capabilities

Single waypoint system

Multiple waypoint storage

Multiple waypoint storage with track storage and frequency selection

Automatic data entry (cards, magnetized cards)

• Data conversion capability (rho/theta and lat/lon compatibility)

Mass route data storage and retrieval

Each of the capabilities described above can impact RNAV system acquisition cost, although, as stated before, it is not always possible to separately list the incremental cost of each individual capability, and so costs must be estimated for functional groups of capabilities. Acquisition costs can also include costs of equipment other than the RNAV computer system. A prime example is the requirement for DME equipment, which not all affected general aviation aircraft would otherwise have. Also, VNAV capability, for example, requires that an altimeter compatible with the VNAV computer be available. In addition to equipment costs, there are other costs required for system installation and routine maintenance.

Tables 3.30 and 3.31 present matrices of RNAV system functional capabilities. Table 3.30 shows those capabilities which are normally available on existing basic systems, and the cost associated with these systems. Nearly all RNAV systems have certain basic capabilities, such as Distance to Waypoint and Cross Track Deviation computation, Enroute/Approach mode selection, normal VOR operating mode, normal localizer operation, direct-to waypoint navigation capability, and the ability to interface with RMI's, flight directors and autopilots (in some cases additional cost options are required for such interface capability). Different manufacturers provide different levels of compatibility with other manufacturer's sensors, with greater flexibility usually costing more. Lowest cost systems only interface with specific sensors manufactured by the same company. Functional capability increments available for basic RNAV systems include multiple waypoint capability and basic VNAV guidance. These systems typically are either analog or digital hybrid in mechanization.

In order to meet proposed MOC requirements, the basic system will be required to provide additional capabilities as shown below:

Basic RNAV System

Existing Capabilities

Additional MOC Capabilities

Station-oriented 2D RMAV, DTW, CTD computation, Approach/Enroute Modes, Direct-To Capability Multiple Waypoint Storage, Parallel Offsets

In Table 3.30 the estimated costs of the Basic MOC System are given (with and without VNAV). These costs were estimated based upon the assumption of integral multiple way-point and parallel offset capabilities, not add-on packages.

More complex RNAV systems, which are summarized in Table 3.31, are of two basic types: complex station oriented systems and earth-oriented systems. Typically the former are of digital hybrid or all digital mechanical while the latter are all digital (except for required sensor signal processing for digitization). The station-oriented systems are intended for business and private aircraft applications and air taxi and commuter airlines, while the earth-oriented systems are intended mainly for airline applications. Note however that this is the present case and that there is no outstanding reason why a lower-cost earth-oriented system could not be developed, except that a Flight Data Storage Unit (FDSU) is almost a requirement for efficient system operation due to the large amount of information required for earth-oriented system operation. The FDSU would substantially increase system cost.

The capabilities which the complex station-oriented systems have over and above the basic systems include wind and ground speed computation, slant range correction, parallel offsets, dead reckoning capability and multiple waypoint, track bearing and station frequency storage. VNAV, with along track offset capability, is optional in some cases, standard in others. Some systems also have an Automatic Data Entry Unit available as an option. Earth-oriented systems possess the additional capabilities of operating in terms of latitude and longitude, performing track computation, operating with multiple sensor inputs, and operating with an FDSU (usually the FDSU is an option, although operation without it is difficult). More sophisticated systems also provide sophisticated control/display and data base management, including automatic station selection capability, at higher prices.

3.5.2 RNAV Equipment Costs for User Groups

The basic system requirements for the several types of aircraft operators are dictated by their types of operations and operational environment. Low altitude aircraft do not require slant range correction, and so may obtain RNAV capability through installation of basic MOC systems, whereas high altitude aircraft (including pressurized piston aircraft repeatedly involved

Matrix of Basic RNAV Equipment Capabilities and Associated Costs **Table 3.30**

RNAV System Type	Functional Capabilities	Interface Capabilities	Data Management	RNAV	Other Costs
BASIC SYSTEMS *					
Basic 2D RNAV	Station-oriented 2D RNAV, DTW, CTD computation, Approach/Enroute Modes, Direct-To Capability	VOR/DME,CDI, RMI, Flight Director, Auto- Pilot	Single Maypoint, \$2000 Manual OBS	\$2000	DME \$2500 (where not equipped)
plus Multiple May- points	ı	ı	Multiple Way- point	add \$500/ MP or \$3000 for multi(10- 16 MP) storage box	
plus 30 RNAV	Basic VNAV (VTD)	Altimeter, VDI, Flight Director, Autopilot	Manual Gra- dient & Alti- tude Select	add \$1500	add \$1500 Altimeter \$2000 to \$5000
BASIC NOC SYSTEM:					
2D RNAV	Station-oriented 2D RNAV DTW,CTD, comput- ation, Approach/ Enroute Modes, Direct- To Capabllity, Paral- Tel Offset Capability	VOR/DME, CDI, RMI, Flight Director, Auto- Pilot	Multiple Way- point, Manual OBS	\$3000	DME \$2500 (where not equipped)
plus 30 RNAV	Basic WAN (VTD)	Altimeter, VDI, Flight Director, Autopilot	Manual Gradient & Altitude Select		add \$1500 Altimeter \$2000 to \$5000

*does not meet MOC

Table 3.31 Matrix of Complex RNAV Equipment Capabilities and Associated Costs

RNAV System Type	Functional Capabilities	Interface Capabilities	Data Nanagement	RNAV	Other Cost
COMPLEX STATION- ORIENTED SYSTEMS:					
2D RNAV System	Station-oriented RWAY, DTW, CTD,Wind,TIS,etc., Approach/Enroute Nodes, Direct-To-Capabilities, Parallel Offsets, Slant Range Correction, Dead Reckoning Capability	VOR/DME, Altitude TAS, Compass, CDI, RMI, Flight Direc- tor Autopilot	10 waypoints, Track Bearings, Station Frequen- cies	\$18000 to	\$18000 to DWE \$5000 \$18000 (where not equipped)
plus 3D RNAV	VMAV along track offset	Altimeter, VDI, Flight Director	Manual Gradient add \$2000 Alltimeter & Alltitude Select, Some Systems have integral VNAV	add \$2000	Altimeter \$2000 to \$5000
EARTH-ORIENTED SYSTEMS:	Company of the control of the contro				
Basic Systems	Earth-oriented 2D RNAV, Track Computation, DTM CTD, TAS, Wind, TTS, etc., VNAV, Along Track Offsets, Parallel Offsets, Slant Range Correction. Approach/Enroute Nodes, Direct-To-Capability, Dead Reckoning Capability	VOR/DWE Altitude Air Data Systems Multisensor Op- tions CDI, RMI, Flight Director, Autopilot	Multiple May- point Storage	\$20000 to \$25000	
plus ADEU or FDSU	(above)	(above)	ADEU/FDSU	add \$10000 to \$150000	-
Sophisticated Data base System (with FDSU)	(above) plus Automatic Station Selction	(above)	Sophisticated Data Base and Control/Display Unit	\$80000 to \$100000	
plus CRT Map Display	-	CRT Map Output	:	add \$35000	

in operations in metropolitan terminal areas) would probably obtain multiple waypoint capability to ease cockpit workload. Airline aircraft, repeatedly flying the same route structure, would in most cases obtain dual systems capable of storing entire route structures, such as the earth-oriented systems. Table 3.32 lists the typical minimum RNAV equipment costs for several aircraft classes, based upon the costs in Tables 3.30 and 3.31. DME and VOR costs are listed in order to obtain total cost for RNAV equipage for aircraft not presently possessing those capabilities.

Table 3.32 Minimum RNAV Capability Equipment Costs

Aircraft Category	Min	imum Equipme	nt Cost	Required Capability	
e produce de la compania del compania de la compania de la compania del compania de la compania del la compania de la compania del la compania de la compania de la compania del la	RNA	/ DME	VOR	person was shartdis	
Single Engine, < 4 places	\$ 30	000 \$2500	\$2000	Basic MOC RNAV	
Single Engine, > 4 places	30	000 2500	2500	the positions when the	
Multi-Engine, < 12,500 lb	50	000* 5000	2500	Multiple Waypoint	
Multi-Engine, > 12,500 lb	50	000* 5000			
Turboprop, < 12,500 1b	100	000 5000		Slant Range Correction	
Turboprop, > 12,500 lb	100	5000			
Turbojet	100	000 5000			
Turbojet (Air Carrier)	600	000		FDSU, Dual Installatio	

^{*} Pressurized, high altitude multi-engine aircraft would require slant range correction.

This section discusses the fleet equipage costs for the three basic user groups: air carrier aircraft, business aircraft and non-business GA aircraft. It is assumed that all air carrier jet aircraft will be equipped with dual RNAV installations, whereas business jet and turboprop aircraft are assumed to be equipped with single installations, although many would be expected to be dual equipped. Other GA aircraft are treated separately in a parametric manner based upon several candidate levels of RNAV requirement.

The projected 1984 civil aircraft fleet is listed below, based upon estimates taken from Reference 33.

Air Carrier	Bus	iness GA		Non-B	Susiness GA	
(A11) 3,100	(Turbine) 6550	(Recip-Mult 21550	i)(Recip-Single) 38500	(Turbine)	(Recip-Mult	i)(Recip-Single) 135900

3.5.2.1 Air Carrier Costs

It is indicated above that there are projected to be 3100 air carrier aircraft in 1984. Table 3.32 lists a cost of \$60,000 as being the minimum cost required to outfit each air carrier aircraft with dual RNAV installations. Based upon these figures, the one-time cost required to outfit the entire 1984 fleet would be on the order of \$186 million.

3.5.2.2 Business and Other General Aviation Costs

To the General Aviation segment, and particularly the business aircraft operators, it is of significant interest to know the impact which a requirement for RNAV for operations in certain airspace would have on user costs for equipment. Likewise, that cost is of interest to the FAA when making such an airspace-restriction decision. Two basic classifications of such airspace which would be affected by RNAV are the high altitude environment and certain terminal areas.

High Altitude

According to RNAV Task Force guidelines and the implementation plan proposed in this report, RNAV would eventually be required in the high altitude environment and in certain terminal areas. The minimum cost of equipping the existing high altitude GA fleet is listed in Table 3.33. The aircraft counts in Table 3.33 represent estimates of fleet size as of May 1975 and were derived in the manner discussed in the following sections.

Table 3.33 High Altitude GA Fleet Equipage Cost (\$ Millions) for Existing

A/C Type	A/C Affected	RNAV Egpd.	Remainder	No DME	No VOR	Cost
Multi > 12,500 lb	646	80	566	193		\$3.80M
Turbo < 12,500 1b	1336	396	940	39		9.60
Turbo > 12,500 1b	264	61	203	5		2.06
Turbojet	1200	377	823	12		8.29
TOTAL:	3446	914	2532	249		\$23.75M

These numbers in Table 3.33 are projected into 1984 and broken down into business operators and non-business operators through use of the projected fleet data stated earlier, as shown in Table 3.34.

Table 3.34 High Altitude GA Fleet Equipage Cost (\$Millions) for 1984 Fleet

	Busines	ss A/C	Non-Business A/			
A/C Type	Number	Cost	Number	Cost		
Turbine Multi > 12,500	6550 467	\$46.7M 2.7	1250 179	\$8.9M 1.1		
Tota1	7017	\$49.4M	1429	\$10.0M		

(Note: Piston Multi > 12,500 lb numbers were not increased from 1974 to 1984 levels since they are no longer being manufacturered)

Terminal Area

In order to determine the cost impact of requiring RNAV capability for entry into terminal area airspace (as transponders and altitude reporting capability are now required at some terminals), it is necessary to make some assumptions regarding the airspace to be so set aside. Therefore, this cost study has been performed in a parametric manner in order to present user cost data in a form which shows the effects of including progressively more airspace as requiring RNAV capability. Given that the airspace serving a particular airport is set aside requiring RNAV capability for entrance, the user impact can only be determined from knowledge of the numbers of each type of aircraft which operated at that particular airport. Since this information was not available, a substitute source of like information was sought. The Aircraft Registration Master File, maintained by FAA through the use of annual aircraft registration questionnaires, contains information concerning aircraft type, base airport, and avionics complement for each registered aircraft in the United States. The base airport data was used for this analysis as representative of the airport at which the aircraft operates. Naturally, this would understate the number of aircraft of each type which would operate at any given airport, since common destination airports are not included. However, this does represent a certain subset of total operations; appropriate multipliers may be estimated to attempt to determine the total effect for any given case.

The registration file of May, 1975 was used to provide the most recent data available. This file contains just short of 200,000 aircraft of which 140,000 have been estimated to be currently certificated and active (References 22,34,35,36 contain historical active aircraft data from which the estimates were made). A search of the file has yielded 110,000 aircraft as showing active status. The remaining thirty thousand are shown as inactive on that tape since the 1975 registration questionnaires for them had not yet been processed. The numbers of aircraft in each category at each airport were adjusted upward to account for this discrepancy. The resulting tables contained the numbers of each type of aircraft which was fitted with a given avionics complement, at each group of airports selected.

Eight categories of airports were selected for the parametric analysis: Primary airports at high density hubs (as defined by the Task Force) and airports within fifteen, thirty and forty-five miles of the center of each high density hub, and medium density hub airports and airports within fifteen, thirty and forty-five miles of them. The analysis of costs required to RNAVequip each aircraft based within a certain group of airports was performed by counting the total number of aircraft of each type (e.g., Single Engine, four place or greater) and subtracting the number which are presently RNAVequipped. This remainder was multiplied by the appropriate equipment cost (Table 3.32). Then the numbers of aircraft which are presently DME equipped or VOR equipped were multiplied by their appropriate costs, with all results summed to yield total cost for RNAV equipage at the particular set of airports of interest. The results are presented in Tables 3.35 (high density hubs) and 3.36 (medium density hubs). These results are given for single engine aircraft, by seating capacity, and for light twins. Other (high altitude) aircraft have been treated separately in Table 3.33. Note that

the cost figures listed in each of Tables 3.35 and 3.36 are not additive; that is to say that each category (e.g., within 30 nm) contains all aircraft within the previous category (within 15 nm). The numbers in these tables understate somewhat the actual costs since they do not include aircraft based outside of such terminal areas which conduct operations within such areas in the normal course of business. Also, some aircraft listed in the Aircraft Registration Master File do not indicate a base airport (1730), some are not based at an established airport (10023), and some reported a non-standard or unidentifiable airport (3840), out of the 140,000 active aircraft. The degree to which the equipage costs are understated is greater for the case of hub airports only, and becomes lesser as more airports are included.

Based upon the fleet forecasts stated earlier the data in Tables 3.35 and 3.36 may be projected to 1984 levels, and expressed in terms of business and non-business usage as shown in Table 3.37.

Since the cost of requiring RNAV capability in certain airspace could in some cases be a burdensome requirement, it is possible to approach the design of RNAV terminal airspace from a Terminal Control Area point of view. That approach is to require that only a small radius of airspace around each major terminal be restricted to aircraft with certain capabilities at ground level; hence, only operators at the major hub airport (and in a few cases other very close airports) would be required to be so equipped. The radius of airspace so affected becomes larger at higher altitude levels, but is terminated at a fixed altitude (usually 7000 or 8000 feet) to allow VFR flyovers. An example of a TCA is illustrated in Figure 3.17, which is taken from the Airman's Information Manual. In that figure the altitude bounds of each segment of the TCA are given. The only airport affected by the ground level requirement (inner circle) is the major hub airport (Boston Logan). Operations may be conducted at other nearby airports without affecting the TCA airspace, and so operators at those airports need not have the avionics capabilities required at the major hub airport(s).

The cost of equipping business and non-business general aviation aircraft, based on 1984 fleet size, is summarized in Table 3.37. The cost of equipping these aircraft for operation only in the high altitude structure, plus at high and medium density hub airports (through implementation of a TCA concept at these airports) is given in Table 3.38. In that table, it is presumed that all operators (IFR and VFR) would be required to have RNAV capability. The results shown are quite significant, in that the minimum requirement for achieving the major portion of all RNAV benefits may be realized by requiring RNAV in the high altitude airspace and at the twenty major terminal hub airports and the forty medium density hub airports, and that the one-time cost involved for business operators would be \$66.4 million, and for non-business operators would be \$37.3 million, only fractions of the total annual benefits to all users. And, most of those GA operators who would be required to equip would also be in a position to derive significant RNAV benefits as a result.

Table 3.35 RNAV Equipage Costs at High Density Terminal Hubs for Existing Fleet

A/C Class	A/C Affected	W1th RNAV	Remainder	No DME	No VOR	Cost
	At High Der	sity Hut	Airports:			
Single Engine < 4 Places Single Engine > 4 Places Multi Engine < 12,500 Lbs.	964 2266 766	21 139 99	943 2127 667	874 1705 140	79 11 0	\$ 4.11M 8.53 4.04
TOTAL	3996	259	3737	2719	90	\$16.58M
	Within 15	m of Hig	h Density	ubs:		
Single Engine < 4 Places Single Engine > 4 Places Multi Engine < 12,500 Lbs.	2581 5522 1729	86 371 240	2495 5151 1489	2200 4245 334	344 46 11	\$11.18M 21.00 9.14
TOTAL	9832	697	9135	6779	401	\$41.32M
	Within 30	m of Hig	h Density	lubs:		
Single Engine < 4 Places Single Engine > 4 Places Multi Engine < 12,500 Lbs.	5004 10288 3102	158 688 458	4846 9600 2644	4219 7973 554	715 82 20	\$21.69M 39.39 16.05
TOTAL	18394	1304	17090	12746	817	\$77.03M
	Within 45	m of Hi	h Density	Hubs:		
Single Engine < 4 Places Single Engine > 4 Places Multi Engine < 12,500 Lbs.	8070 15152 4207	281 1011 606	7789 14141 3601	6601 11909 781	1380 136 26	\$34.84M 58.33 21.99
TOTAL	27429	1898	25531	19291	1542	\$115.16M

Table 3.36 RNAV Equipage Costs at Medium Density Terminal Hubs for Existing Fleet

A/C Class	A/C Affected	Wi th RNAV	Remainder	No DME	No VOR	Cost
	At Medium D	ensity	Hub Airports	:		
Single Engine < 4 Places Single Engine > 4 Places Multi Engine < 12,500 Lbs.	399 1329 864	10 91 175	389 1238 689	370 985 94	18 11 7	\$1.74M 4.96 3.93.
TOTAL	2592	276	2316	1449	36	\$10.63M
	Within 15	m of Me	dium Density	Hubs:		
Single Engine < 4 Places Single Engine > 4 Places Multi Engine < 12,500 Lbs.	2371 4779 1803	76 318 315	2295 4461 1488	1940 3708 269	402 62 11	\$10.24M 18.32 8.81
TOTAL	8953	709	8244	5917	475	\$37.37M
	Within 30	m of Me	dium Density	Hubs:		
Single Engine < 4 Places Single Engine > 4 Places Multi Engine < 12,500 Lbs.	3494 6284 2064	116 409 352	3378 5875 1712	2681 4980 333	781 88 11	\$15.02M 24.38 10.25
TOTAL	11842	877	10965	7994	880	\$49.65M
refración de la sectión de	Within 45	m of Me	dium Density	Hubs:		
Single Engine < 4 Places Single Engine > 4 Places Multi Engine < 12,500 Lbs.	5585 9391 2742	165 626 462	5420 8765 2280	4154 7532 462	1383 137 14	\$23.99M 36.63 13.75
TOTAL	17718	1253	16465	12148	1534	\$74.37M

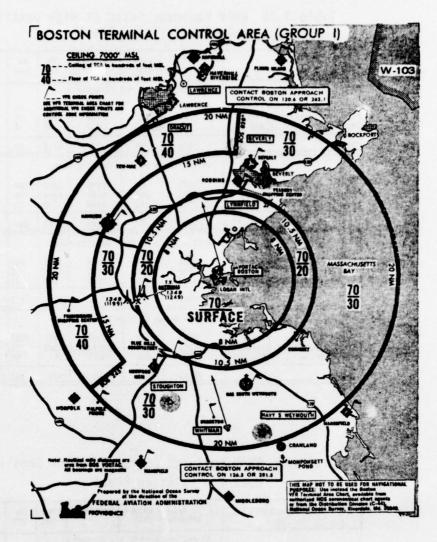


Figure 3.17 Boston Terminal Control Area

Table 3.37 Summary of 1984 Terminar Area RNAV Equipage Costs

	High De	nsity Hubs	Medium	Density Hubs
No. of the	Business	Non-Business	Business	Non-Business
Hub Airports	\$ 9.58M	\$17.23M	\$ 7.41M	\$10.03M
Within 15 mm	22.91	43.29	21.24	38.77
Within 30 nm	60.49	81.52 123.11	26.82 38.63	52.54 79.82

Table 3.38 RNAY GA Requirement 1984 Cost Summary - TCA Concept, Terminal and Enroute

Requirement	Business Operators	Non-Business Operators
High Density Hub Airports Only Medium Density Hub Airports Only	\$ 9.58M 7.41	\$17.200 10.03
TOTAL TERMINAL	\$16.99M	\$27.201
High Altitude GA Operators	49.4	10.0
TOTAL GA	385.39M	\$37.26A

3.6 INDIVIDUAL AIRLINE BENEFITS

A set of computer programs has been developed for purposes of evaluating the benefits which RNAV usage can provide for individual airlines. In order to provide a complete analysis of potential RNAV benefits, the entire route structure of a subject airline is analyzed in order to determine benefits in the enroute and terminal phases of flight, and to isolate 3D (VNAV) and 4D (time control) benefits in arrival operations.

3.6.1 Data Sources

The basic source of data for an airline RNAV benefits analysis is the airline schedule. The basic items of interest are, for each scheduled flight, the origin and destination city, the frequency of service (daily, etc.), and the type of aircraft used. All of this information is available in the Official Airline Guide (OAG) listings [20]. These data were utilized to create files of identical form for each of six (6) airlines. All flight segments either originating or terminating at non-CONUS airports were deleted from the files.

A data base concerning performance for several popular aircraft types (B-747, DC-10-10, DC-8-63, B-727-100, DC-9-30) was generated from manufacturer's aircraft performance data handbooks. The basic items of interest were true airspeed and fuel consumption rate under typical cruise conditions. However, selected cruise altitude is strongly influenced by the range between city-pairs, since it is not economical to climb to maximum cruising altitude on the shorter range flights. Tables of optimum cruise altitude as a function of intercity distance were generated during an earlier study [12], and so were used here. For each airline city-pair in the airline schedule file, the intercity distance was determined in one of two ways: If the city-pair had been evaluated for RNAV benefits (if included in the NAFEC route structure), the enroute distance was taken from the RNAV benefits file (explained later), to which ninety (90) miles were added, representing terminal distance, to estimate intercity distance; If the city-pair were not included in the NAFEC design, intercity distance was computed by looking up the latitude/ longitude of each airport in an airport coordinates file created for this purpose, and using great circle equations to find distance. Only those citypairs represented in the NAFEC structure were evaluated directly for RNAV benefits by this program. In these cases the intercity distance estimate was used in a lookup table for the particular aircraft type to find optimum cruise altitude, and corresponding fuel and time consumption parameters. These parameters were multiplied by the RNAV route length benefit for that route to get fuel and time saved. These values were then multiplied by flight frequency to get annual RNAV benefits for that city-pair for that flight. Enroute distance taken from the RNAV benefits file was also multiplied by flight frequency to get annual route miles for that city-pair for that flight. In those cases where the city-pair was not in the NAFEC structure, enroute distance was estimated by subtracting ninety (90) miles from the computed intercity distance. However, no cruise altitude was

estimated since no estimated RNAV benefit was available. Instead, annual RNAV time and fuel benefits and annual RNAV route miles were aggregated over the various routes; also, annual route miles for those segments not in the NAFEC structure were aggregated separately. In this manner, total annual flight mileage on NAFEC structure routes is known for each aircraft type, and therefore the RNAV benefits achieved on the NAFEC structure could be extrapolated to an entire airline's route structure, presuming that the final operational RNAV route structure expected to be implemented would serve all of the airline's routes.

In order to accommodate the several jet aircraft types in operation other than those five types analyzed, conversion factors were derived for each type which compensated for cruise speed and fuel consumption differences among the various types. For each aircraft type, the most similar aircraft for which data existed was selected as the baseline (e.g. the B-737 used the DC-9-30 as its baseline). Data for deriving conversion factors for each model were obtained from the operational fuel consumption/block speed reports collected by the CAB [37].

3.6.2 Enroute 2D RNAV Benefits

The basic enroute RNAV benefits file was constructed to show net enroute distance benefit for each city-pair represented in the NAFEC structure. The file contained origin and destination airport codes, RNAV benefit and VOR enroute route length. The net RNAV benefit was taken from the results derived at NAFEC [4], as modified by the weather routes study (Section 3.2.1) and the restricted areas study (Section 3.2.2).

The enroute RNAV benefits for each city-pair for each airline studied are shown in Tables G.1 through G.6 in Appendix G. A sample of the results for Eastern Airlines are shown in Table 3.39. In the first entry of that table, the origin airport (ATL-Atlanta) code, followed by the destination code (BAL-Baltimore), is followed by the EAL flight number (126). Subsequently, the enroute distance (388.0 nm) and RNAV benefit (14.00 nm), taken from the RNAV benefits file, are listed. The aircraft type is the stretched DC-9 (D9S). Flight frequency is daily (7 days/week). The typical cruise altitude derived directly from the intercity distance is 34480 feet. Naturally, this is an unrealistic altitude since flight levels at four thousand (4000) foot increments are assigned. However, the odd value derived represents a reasonable average flight level given the various flight levels which would be assigned over a period of time. The last two items listed are the RNAV benefits in terms of time saved (minutes) and fuel saved (gallons).

3.6.3 Terminal 2D RNAV Benefits

The primary data sources used in the analysis of the terminal area 2D RNAV benefits are the results of the analyses in Section 3.1.1, where peroperation RNAV benefits for eight (8) aircraft types at nine (9) large airports are derived. Also, methods of extropolating these benefits to fortynine (49) other airports were developed. Individual airline terminal 2D

TABLE 3.39 EASTERN AIRLINES ENROUTE RNAY BENEFITS ANALYSIS

ORG DST	FLT	RANGE	RNAV	A/C	F CRUISE	TIME	FUEL	ORG	DST	FLT	RANGE	RNAV	A/C	F CRUISE	TIME	FUEL.
A/P A/P	NO	EMI	BENE		ALT	BENE	BENE		A/P	NO	NMI	BENE		AI.T	BENE	BENE
ATL SAL	124				7 34480. 7 34440.	1.89	23.45		040	141				7 40400.	-5.15	-3.07
ATL SAL	134	338.0	14.00	D95	7 34480. 1 34480.	1.89	23.45	ATL	184	269	339.0-	-33.00	151	7 35000.	-5.09	-89.25
ATL BAL	135	333.9	14.00	095	7 34440.	1.89	23.45	ATL	P31	329	304-3-	v	725	7 35000-	-5.15	-97.51
ATL SAL	424				1 35000. 6 35000.	1.85	32.05		PHL	120				7 35000.	1.70	29.76
ATL HVA	244	98.0	1.00	045	7 27640.	.13	1.95	ATL	PHL	155	536.0	13.40	295	35000.	1.76	21.59
ATL BYA	520	98.0			7 25640.	.13	3.06		PHL	140				7 41300.	1.76	59.03
ATL SVA	575	98.0			7 27640.	.13	1.95		PHL	320				7 35000.	1.72	32.50
ATL 205	115	770.0	6.00	727	7 35000.	.78	13.74	ATL	PIT	336	376.0	5.00	727	7 35000.	.65	11.45
ATL 805	123	770.0			7 41000. 6 35000.	.74	26.78		SOF	252				7 35000-	40	-5.33
ATL SUF	334	770.0			7 35000. 7 35060.	.78	9.16		SOF	250				7 31667.		-5.33
ATL BUF	632	535.0	4.00	725	7 35000.	.53	10.00	ATL	SOF	462	190.0	-3.00	095	7 31667.	40	-5.32
ATL CLT	322	535.0			7 41000. 7 27090.	1.01	21.43		SUF	712				7 31667.	3.59	-5.32
AIL CLT	324	116.0	2.00	095	7 29040.	1.05	15-01	ATL	STL	270	366.6	29.60	152	35000.	3.63	72.51
ATL CLT	344	116.0			7 29040.	1.05	23.41		STL	272	360.0	29.00	095	1 34200-	3.90	49.82
ATL CLT	476	115.0			7 27090. 7 27080.	1.02	23.41		STL	452				34200.	3.70	66.40
ATL CCA	130	355.0	3.00	725	7 15000.	.40	7.50	ATL	TPA	289	277.0	11.66	725	7 350co.	1.45	27.50
ATL CCA	345	356.0			7 35000. 7 35000.	.40	7.50		TPA	365				7 35000.	1.45	27.53
ATL BEA	924	346.0	3.00	725	7 35000. 7 35000.	.40	7.50		TPA	485				35000.	1.45	27.50
ATL GF.	451	555.0	1.00	DC9	7 35000.	.13	1.57	ATL	TPA	727	277.0	11.00	725	7 35000.	1.45	27.50
ATL OF	573	555.0			7 35000. 7 35000.	.13	2.29		ATL	131				1 34470-	1.21	15.03
ATL DF.	577	555.0	1.00	727	7 35000.	.13	2.29	HAL	ATL	633	387.0	9.00	095	34470.	1.21	15.08
ATL END	106	555.0			7 35000. 7 35000.	.52	9.16		ATL	424				34840.	2.22	39.07
ATL EN	112	585.0			6 35000.	.53	10.00	RAL	825	434	261.0	17.00	152	34840.	c.24	42.66
ATL ERK	ile	555.0	4.00	LIO	7 41000.	.49	17.85	401	DCA	593	107.0	23.00	295	31587. 31587.	3.06	40.69
ATL 550	368	178.0			7 31347. 7 31027.	0.00	0.00		MIA	181				12600-	4.05	49.84
ATL GSU	500	178.0	0.00	045	7 31347.	0.00	0.00	BNA	ATL	299	100.0	4.00	152	25800.	.51	15.15
ATL SSO	607		-6.00	727	7 31027.	76	-15.16	RNA	ATL	959		4.00	725	25800.	.52	12.19
ATL JAX	231				7 29320. 7 29320.	77	-16.56		ATL	671	100.0			27800.	.52	7.75
ATL JAX	533	140.0	-0.00	095	7 34249.	79	-10.94	AVA	090	258	258.0	6.00	727	7 34720-	./8	13.83
ATL JAX	578				7 30240.	77	-10.94		ATL	129	740.0			7 33180.	.79	12.50
ATL JEK	108	691.0			7 35000.	.52	9.16		ATL	149	748.0			7 35000-	.65	11.45
ATL LAX	83	1542.0	17.00	727	7.35060.	2.22	34.92	ROS	ATL	533	745.0	5.00	Dos .	7 35000.	. 58	8.33
ATL LAK					6 35000.	2.22	38.92		ATL	537	748.0			35000.	.66	12.50
ATL LAK	89	1542.0	17.00	727	7 35000.	2.22	9.16	435	DCA	199	244.0	0.00	727	7 34140.	0.00	0.00
ATL LOA	100	567.0	4.00	095	7 35000.	.54	6.64	905	DCA	393	244.0	9.00	727	7 34160.	0.00	0.00
ATL LOA	102	567.0			7 35000. 7 35000.	.52	6.64		DCA DCA	507	244.0			7 34160.	0.00	0.00
ATL LGA	544	547.0	4.00	727	7 35000. 7 35000.	.52	9.16	435	JFK	491	76.0	17.00	725	. 08865	2.20	53.83
ATL HEM	572	217.0	12.00	DC9	7 32397.	1.57	19.72	905	MIA		1030.0	0.00	725	7 35000.	2.17	0.00
ATL WEN	930	217.0	12.00	727	7 32397.	1.60	28.70		AIM		1030.0			7 35000.	0.00	0.03
ATL HEM	392	217.0	12.60	725	7 33000.	1.57	31.34	405	M[4	419	1030.0	G . 00	727	7 35000-	0.00	0.00
ATL MIA	219				7 35000. 7 35000.	5.35	93.87		ATL	345				7 31933.	.52	9.16
ATL PIA	255	470.0	41.00	CYS	7 35000.	5.54	68.05		ATL	951	551.0	4.00	725	7 35000-	.53	10.03
ATL MIA	367	470.0	41.00	727	7 35000. 7 35000.	5.35	68.05 93.87	HUF	PHL	125	191.0	34.00	151	7 31720-		92.77
ATL WST	147				7 35000. 7 33480.	.54	69.05		UFK 040	354				7 34500. 7 34680.	3.37	16-63
ATL MST	144	265.0	4.00	121	7 35000.	.52	9.10	CL1	020	214 218	423.0	10.00	DYS	7 34880.	1.35	16.63
ATL MST	-24				7 35000. 7 35000.	.53	9.16	CL1	040	225	428.3	10.00	095	7 34880.	1.35	16.63
ATL MST	793				7 33490.	26	-3.14		PHL	540				7 33720.	1.61	20.37
ATL URD	242	424.0	-2.00	725	7 35000.	26	-5.00	CLI	DH.	596	312.0	12.00	121	7 35000.	1.57	27-47
47L 040	244				5 350GO. 2 350UO.	26	-5.00	CL	PIL	313				7 35000.	1.34	17.19
ATL 920	245	424.0	-2.0u	DC9	7 14840.	26	-3.14		PIT	334				7 33000.		17.19
415 049	249	424.0	-2.00	064	7 34840.	46	-3.14									

TABLE 3.39 EASTERN AIRLINES ENROUTE RNAV BENEFITS ANALYSIS Continued

ORG DST	FLT	RANGE NMI	RNAV		F CRUISE	TIME	BENE		DST PA/I				A/C	F CRUISE		FULL
											NMI	BENE	nos	ALT	BENE	BINE
CLT PIT	476				7 33000. 7 34000.	1.34	25.58		IP					7 27000-	52	-7.97
OCA 305	372				7 34840.	2.22	39.07		TP		55.55			7 27000-	52	-7.92
OCA HOS	398				7 34840.	2.22	34.07	G\$	LG	365	304.0	11.00	095	7 33640.	1.48	10.70
DCA BOS	566				7 33210. 7 33210.	2.28	29.12		LG					7 33640 -	1.48	18.70
OCA BOS	878				7 33210.	2.26	29.12		0 020					7 33640.	0.00	0.00
DCA CLT	375	206.0	12.00	095	7 32093.	1.00	21.01		0 080					7 34940-	0.00	0-00
DCA CLT	385				7 32093.	1.60	21.01		E					7 25720 -	38	-8-19
DCA CLT	391				7 32520.	1.55	24.95) JF					7 25860.	2.09	36.63
	199	465.0			7 35000.	.91	16.03		ATL					7 29427.	.26	5.51
	869				7 35000.	.95	11.62		ATL		148.0			7 29427.	-26	5.51
DCA LGA I					5 26040.	19	-6.10 -6.10		TATE		148.0			7 29427.	.26	5.51
DCA LGA 1					7 20040.	39	-6.10		ATL		148.0			2 29427.	.25	5-05
DCA LGA 1					7 26040.	39	-6.10		ATL		148.0			7 30320-	.26	3.66
DCA LGA 1					7 26040.	39	-6.10		C DC		148.0			7 29427.	-26	11.62
OC4 LG4 1					7 250-0.	39	-5.10		DC		462.0			7 35000-	.95	11.62
DCA LGA 1					7 26040.	39	-6.10		JF		634.0	5.00	095	7 35000-	.68	8.10
DCA LGA 1					7 25040.	39	-5-10		HIA		203.0			7 32360.	0.00	0-00-
DCA LGA I					7 26040.	39	-6.10		ATL					7 35000-	2.43	29.87
OCA LGA 1					7 26040.	39	-5.10		ATL		572.0	18.00	095	7 35000-	2.43	29.87
OCA LGA 1					7 26040.	39	-6.10		ATL					6 35000.	2.38	45-00
DCA LGA I					6 26040.	39	-6.10		C FLL		856.0			7 26280. 7 41000.	6.30	31-25
	175	718.0			7 35000.	.68	8.30		FLI		858.0			7 35000.	.92	17.50
PCA MIA	177	718.0	5.00	095	7 35000.	.68	8.30	JF	(FLI	. 755	848.0	7.00	D95	7 35060-	.95	11.62
DCA HIA	195	718.0			7 35000.	.65	11.45		140		104.0			7 26120.	.38	8.29
DCA MIA	197	718.0			7 35000.	.68	8.30		C MI		874.0			7 41000.	0.00	0.00
DCA SOF	370	320.0	7.00	095	7 33800.	.94	11.87	JF	(MIM	23	874.0	0.00	727	7 35000-	0.00	0-00
OCA SOF	509	350.0			7 35000.	.92	17.50		HI		874.0			7 41000-	0.00	0-00
	114	563.0			7 35000.	.54	6.64		CHIA		874.0			7 41000-	0.00	0-00
DEH ATL	120	553.0			7 35000.	.52	9.16		451		952.0			7 35000-	1.04	18.32
DF# ATL	286	563.0			7 35000.	.53	6.27		(MS1		952.0			7 35000.	1.04	18.32
	596	563.0			7 35000.	.54	9.16		TPA		749.0			7 35000. 7 35000.	.79	13.74
	804	101.0			25460.	50	-11.13		TPA		789.0			7 35000.	.81	9.96
	341	85.0	-1.00	727	7 24600.	13	-2.87	LG	ATL	. 67	550.0	4.00	727	8 35000-	.52	9.16
DIS PIT	349				7 26800.	~.13	-1.99		ATL		560.0			1 35000.	.52	9.16
DIM TPA	739 335	-0.0			7 18200.	0.00	0.00		ATL		560.0			7 35000.	.52	9.16
	453	-0.0			18200.	0.00	0.00		ATL		560.0			7 15000-	.53	10-00
	639	-0.0			7 18200-	0.00	0.00		A ATL		560.0			7 35000.	.54	6.64
	135	569.0			7 35000.	.53	6.27		A CL		387.0			7 35000.	80.5	36.63
	806	569.0			7 41000.	.49	17.85		CLI					6 35000-	2.11	40.00
	353	377.0			7 35000.	.78	13.74		CL					1 35000.	2.11	40-00
	359	377.0			7 34370.	.81	10.07		CL					7 35000.	2.09	26.63
	407	269.0			35000.	.42	17.50	LG	N UCA	1401	105.0	4.00	095	5 28200.	.52	7-67
	745	855.0			7 35000.	.91	16.03			1411	105.0			6 28200.	.52	7.67
	759	858.0	200		7 35000.	.95	16.03			1421	105.0			7 28200-	.52	7.67
	559	60.0			22200.	0.00	0.00	LG	DC	1441	105.0	4.00	095	7 28200.	.52	7.67
AIM SES	3	A56.0	7.00	725	7 35000-	.92	17.50			1451	105.0			- 00265	.52	7.67
EAR HIA	5	856.0			35000.	.91	11.62			1461	105.0			7 28200.	.52	7.67
AIM FINS	,	856.0			7 35000.	.92	17.50	LG	A DC	1491	105.0	4.00	095	7 28200.	.52	7.67
and the same of th	403	856.0	7.00	151	7 35000.	.91	16.03			1001	105.0			7 28200.	.52	7.67
ENP TPA	165	811.0			7 41000.	.87	31.25			1511	105.0			7 28200.	.52	7.67
FLL BOS		1029.0			7 35000.	1.17	20.61	LG	OCA	1531	105.0	4.00	093	.00585 6	.52	7.67
FLL ROS	880	0.9501	9.00	151	7 35000.	1.17	20.61			1541	105.0	4.00	095	7 28200-	.52	7.67
FLL EWR	405	874.0			7 35000.	.91	17.50		I HI		627.0			7 35000. 5 35000.	1.55	17.50
FLL ENH	742	874.0			7 35000.	.45	11.62	LG	MIN A	11	865.0	7.00	725	1 35000.	.92	17.50
FLL ENN	758	874.0	7.00	727	7 35000.	.91	16.03	LG	MIA	11	865.0			1 35000-	.91	16.03
FLL JFK	412	864.0			7 41000.	.65	22.32	LG	1	21	865.0			7 41000-	.41	16.03
FLL JFK	750	854.0			7 35000.	.68	4.30	16	HI	50	865.0	7.00	LIO	7 41000.	.87	31.25
FLL UNU	458	951.0	8.00	727	7 35000.	1.04	18.32	LG		415	865.0	7.00	LIO	7 41000.	.87	₹5.1€
FLL 090	790	957.0			7 35000.	1.04	18.32		99		A03.0			7 35000-	.95	17.50
FLL PHL	795	853.0			7 35000.	1.04	16.03		ATI		0.105	-1.00	045	7 31960.	13	-1.76
	875				7 35000.	.91	16.03		. ATL		501.0	-1.00	064	7 31940.	13	-1.05

TABLE 3.39 EASTERN AIRLINES ENROUTE RNAV BENEFITS ANALYSIS Continued

ORG DST	FLT	NMI	RNAV	A/C	F CRUISE ALT	TIME	BENE	ORG A/P		FLT	RANGE NM1	BENE		F CRUISE ALT	TIME	TULL.
WEN ATL	981			726	7 32253.			050		791						BENT:
ALA MEN	933				7 32253.	13	-2.65	080		795	960.0			7 35000. 7 35000.	1.04	19.32
MIA ATL	252				7 35000.	3.11	39.17	050		207	-0.0			7 18200.	0.00	0.00
MIA ATL	564				7 35000. 6 35000.	3.00	57.50	040		75	953.0			7 18200-	1.04	18.32
MIA ATL	564	453.0	23.00	725	1 35000.	3.04	57.50	020	MIA	79	953.0	6.00	725	7 35000.	1.06	20.00
WIA ATL	615				7 35000. 7 35000.	3.00	57.50	020		433	953.0			5 35000 . 2 35000 .	1.06	20.00
HIA ATL	678				7 35000.	3.00	52.66	020		993	953.0			7 35000.	1.04	19.32
414 94L	172	762.0			7 35000.	.79	15.00	040		203				7 35000.	6.49	79.67
MIA BOL	132	977.0			7 35000. 7 35000.	1.19	22.50	020		205	804.0			7 35000. 7 35000.	.78	13.74
MIA BOS		1081.0			7 35000.	1.19	22.50	OPD	TPA	259	904.0	6.00	151	7 35000.	.78	13.74
414 C44	302	814.0			7 35000. 7 35000.	.91	16.03	090		105	391.0			7 35000. 7 35000.	0.00	0.05
MIA DCA	159	747.0	6.00		5 35000.	.78	13.74	164		142	391.0			7 35000-	0.00	0.00
HIA DCA	158	747.0			2 35000. 7 35000.	.81	9.96	169		242	341.0			7 35000.	0.00	0.00
ADG AIM	190	747.0			7 35000.	.78	9.96	169		632	391.0			7 35000. 7 35000.	1.06	20.00
MIA OCA	192	747.0			7 35000.	.79	15.00	169	JFK	146	877.0	H.00	110	7 -1000.	.99	35.71
MIA DEM	976	320.0			7 35000.	1.43	25.19	169		294	877.0 823.0			7 35000-	1-04	11.62
WIA OT#	422	924.0	8.00	727	7 35000.	1.04	18.32	PHL	ATL	121	500.0	9.00	727	7 35000.	1.17	20.61
MIG AIM	952	924.0			7 41000. 7 35000.	.99	17.50	PHL		123	500.0			7 35000. 7 35000.	1.17	20.61
MIA ENA	5	889.0	7.00	095	7 35000.	.95	11.62	PHL		127	500.0	9.00	727	7 35000-	1.17	20.61
PES AIR	402	889.0			7 35000. 7 35000.	.92	17.50	PHL		561	500.0			7 41000-	1.11	47.17
MIA JEK	14	881.0			7 35000.	1.04	18.32	PHL		556				7 32627-	5.55	103.53
MIA JEK	14	881.0			7 35000.	1.04	14.32	PHL		646				7 32733.	6.85	127.47
MIA JFK	5.	981.0			7 35000.	1.09	35.71	PHL		373				7 33390.	2.82	35.55 35.55
414 JFK	-00	881.0	8.00	L10	7 41000.	. 49	35.71	PHL	CLT	579	329.0	21.00	095	7 33390-	2.82	35.55
414 LGA	16	0.568			7 41000. 6 35000.	.97	17.50	PHL		603	329.0			7 33490.	2.82	35.55
HIA LGA	20	892.0	7.00	727	1 35000.	.91	16.03	PHL	MIA	37	916.0	7.00	727	7 35900-	-91	16.03
MIA LGA	24	0.568			7 35000. 6 35000.	.91	16.03	PHL		-11	H10.0			7 35000. 7 35000.	.91	15.03
WIA LGA	414	892.0	7.00	L10	7 41000.	.87	31.25	PHL	MSY	575	-0.0	0.00	727	7 12600.	0.00	0.00
MIA LOA	524	9.568			7 41000.	.87	31.25	TIG		327				7 34480.	2.56	31.33 47.50
MIT AZA	907	509.0			7 35000.	.26	5.00	PIT	ATL	999	398.0	19.00	725	7 35000-	2.51	47.50
GEO AIN	72	959.0			7 35000. 7 35000.	1.04	18.32	TIG		311	235.0			7 33800.	.39	5.17
WIA DEC	430	959.0			7 35000.	1.04	18.32	PIT	CLI	341	235.0			7 33800.	.39	7.05
MIA DAL	34	820.0			7 35000.	-91	16.03	PIT		349	235.0			7 32867.	2.22	34.92
ALV DAF	36	820.0			7 35000. 7 35000.	.91	16.03	PIT	MIA	303	809.0	6.00	725	7 35000-	.79	15.00
WIA PHL	955	820.0	7.00	L10	7 41000.	.87	31.25	PIT		483	809.0			35309.	-78	9.95
MIA DIT	300	811.0			7 35000. 7 35000.	.79	9.96	200		219	223.0			7 33320-	18.	4.76
MIA STL	222	861.0	7.00	095	7 35000.	.95	11.62	ROU		361	223.0			6 33320.	.26	5.19
HIA TOA	614	861.0			7 35000. 7 27000.	52	-7.92	ROU		361	553.0			1 33320.	.26	4.76
ALT TOA	572	90.0	-4.00	095	7 27000.	52	-7.92	RDU		549	223.0			7 33320.	.26	5.19
WE ATL	787				7 35000. 7 35000.	3.38	57.24	400		585 585	223.0			5 33320.	.26	5.19
HSY ATL	103	289.0	0.00	727	7 35000.	0.00	0.00	RDU	040	204	489.0	0.00	095	7 35000-	0.00	0.00
MSY ATL	555	0.665			7 35000. 7 33480.	0.00	0.00	HIC		208 562	484.0			7 35000-	0.00	0.00
MSY ATL	534	268.0	0.03	725	7 35000.	0.00	0.00	RIC	LGA	898	145.0	0.00	U95	7 30240.	0.00	0.00
MSY ATL	739	288.0			7 33480.	0.00	0.00	SOF		261				7 32013.	1.60	21.05
USY UFK	521	951.0			7 35000. 7 35000.	.91	16.63	SDF	ATL	269	203.0	12.00	DC9	7 32013-	1.56	19.87
MSY MIA		505.0			7 35000.	.91	16.03	SOF						7 32013.		21.05
OND ATL					7 35000. 7 34950.	2.30	28.24	SOF	DCA	254	321.0	12.00	095	7 33910.	1.61	20.34
JTA CFC	234	435.0	17.00	095	7 34950.	2.30	28-24	SDF	DCA	509				7 35000-	1.58	30.00
OPO ATL	245				7 35000. 7 34950.	2.22	26.65	STL		99				7 37608.	.13	4.93
DID ATL	249	435.0	17.00	095	7 34950.	2.30	28.24	STL	ATL	271	334.0	1.00	UYS	7 33940.	.13	1.69
020 ATL	225				7 40840.	2.10	76.26	STL		273				7 33940.	.13	2-50
0-0 544					7 34560.	.13	2.31	STL	ATL	699	334.0	1.00	295	7 33940.	-13	1.69
AVE CLC					7 33140. 7 35000.	2.70	1.62	STL		200				7 35000-	•54 •.70	59.54
090 CLT	217	443.0	20.00	095	7 35000.	2.70	33.19	TOA	ATL	476	301.0	35.00	725	7 35000-	4.62	87.51
120 050	155					2.70	33.19	TPA		546				7 35000-	4.57	80.13
020 014						1.08	13.28	TPA	ATL	572	301.0	35.00	725	7 35000-	4.62	87.51
								TPA	ATL	624	301.0	35.00	725	7 35060.	4.62	£7.51

TABLE 3.39 EASTERN AIRLINES ENROUTE RNAV BENEFITS ANALYSIS Continued

OF	G	DST	FLT	PANGE	RNAV	A/C	F	CRUISE	TIME	FUEL	OKG	DST	FIT	RANGE	DVAV	AIC	F CRUISE	TIME	
A/	P	A/F	NO	NM1	BENE			ALT	BENE	BENE		A/P	NO	IMA	BENE	,,,	ALT	BLNE	BENE
10		ATL	476	301.0	35.00	725	7	35000.	4.62	87.51									
TP		ATL	488	301.0	35.00	721	7	35000.	4.57	80.13									
TP		ATL	546	301.0	15.00	725	7	35000.	4.62	67.51									
15		ATL	572	301.0	15.00	125	7	35000.	4.62	87.51									
TP		ATL	624	301.0	35.00	725	7	35000.	4.62	87.51									
10		CLE	316	728.0	5.00	095	7	35000.	.68	8.30									
19	4	DT	340	757.0	7.00	DYS	7	35000.	.95	11.62									
12	A 1	w10	344	797.0	7.00	095	7	35000.	.95	11.62									
19		DT.	464	777.0	7.00	095	7	35000.	.45	11.62									
TP		EMA	168	916.0	6.00	LIG	7	41000.	.74	26.78									
19		FLL	189	103.0	18.00	725	7	26040.	2.29	54.43									
10		FLL	553	103.0	13.00	095	7	24040.	2.36	34.67									
12		FLL	339	103.0	18.00	095	7	28040.	2.36	34.67									
		FLL	485	103.0	18.00	725	7	26040.	2.29	54.43									
		FLL	529					28040.	2.31	32.73									
		JFK	160	928.0				35000.	1.04	18.32									
		JFK	162	0.659				35000.	1.04	18.32									
		JFK	426	928.0				35000.	1.04	18.32									
		HIA	127					20040.	2.27	49.84									
		MIA	437					20040.	2.36	34.67									
		HIA	645					26040.	2.29	54.43									
		HIA	645					26040.	2.29	54.43									
		080	230	793.0				35000.	.78	13.74									
		050	262	793.0				35000.	.79	9.40									
IP			466	793.0				35000.	.78	13.74									

RNAV benefits were derived from the airline schedule files described earlier and the benefits results for the fifty-eight (58) airports. Potential benefits at other airports served by an airline were not considered. Benefit determination at any given airport proceeded as follows: For each flight, the origin airport was examined to determine if it is one of the nine (9) primary airports. If so, the benefit data for the aircraft type most nearly like the actual aircraft were taken from Table 3.3 and multiplied by the fuel and time conversion factors for that aircraft type (as discussed earlier). Note that only origin airports need be examined; total benefits are then twice the per operation benefits, accounting for the arrival and departure at that airport.

If the origin airport is not among the nine (9) primaries, it was examined to determine if it is one of the other forty-nine (49). If not, the next flight is selected and the procedure repeated. If so, then the proper extrapolation equation was selected for the aircraft type and the results were multiplied by the aircraft conversion factors, as before, to get fuel and time benefits for that flight.

3D RNAV Benefits

The benefits due to 3D RNAV (VNAV) in descent operations are discussed in Section 3.1.2. In that section it was shown that VNAV capability may be used to conduct descents more advantageously in terms of fuel and time usage, regardless of where the descent is conducted. Therefore, if VNAV descent procedures are adhered to by an airline throughout its route structure, this benefit is available for each descent conducted. Therefore, airline VNAV benefits computation was a rather straightforward process. The annual descent operations conducted by each type of aircraft were counted, and the totals were multiplied by the benefit values shown in Tables 3.8, as modified by the appropriate individual aircraft type conversion factors, as done before, to get aggregate VNAV benefits for each aircraft type in an airline's inventory.

3.6.4 4D RNAV Benefits

The benefits due to 4D RNAV in terminal arrival operations are discussed in Section 3.4.3. In that section, Table 3.29 was created which shows the potential per operation benefits available to operators at each of the twenty-five (25) highest delay airports given that 4D operations are in extensive use. These benefits are in terms of delay reduction per operation, and are independent of aircraft type. Only time, not fuel, savings were calculated; fuel savings may be derived from time savings for a given aircraft type from low altitude cruise performance data. The cruise condition selected as being appropriate in representing the condition where such a delay reduction would be experienced was a holding pattern at 15,000 ft. Fuel consumption (gpm) rates were determined for the five (5) primary aircraft types from the performance manuals at that condition, and so fuel benefits could also be calculated.

In order to assess 4D RNAV benefits for a given airline, each operation was examined to determine if the destination airport was among the twenty-five (25) listed in Table 3.29. If not, no contribution to total 4D benefit was made. If so, the time savings per operation was multiplied by the fuel consumption rate, as modified by the appropriate aircraft type conversion factor, as done before, to determine the contribution to aggregate 4D RNAV benefits for that aircraft type.

3.6.5 Data Aggregation and Extrapolation

It was desired to be able to express RNAV benefits individually for each airline and for each aircraft type within each airline on a per aircraft basis. The per aircraft mode of expression is extremely important since it shows the degree of payoff to be expected for each individual RNAV installation. In order to make that computation, fleet composition data was obtained from each airline. It is further significant to show RNAV benefits individually for each aircraft type since the particular types of operations and routes flown by individual types of aircraft for individual airlines have a very strong influence upon the magnitude of the resulting benefits due to enroute and terminal 2D/3D RNAV and 4D RNAV.

In order to express benefits by airline and by individual aircraft type, the analysis programs were designed to accumulate total time and fuel benefits separately for each individual aircraft type and each airline. Furthermore, in the case of the 2D RNAV benefits evaluation, total route miles flown were aggregated, as were total route miles flown on remaining routes for which benefits data was not directly available. As discussed earlier, the percent of operations conducted on the NAFEC structure by each aircraft could then be computed. This was very useful for extrapolating total benefits presuming that a comprehensive RNAV structure were implemented. Since basic RNAV benefits are sensitive to the particular routes flown, it follows that the operations flown by a particular aircraft type on certain routes within the NAFEC structure are more representative of total RNAV benefits available to that aircraft type than the entire NAFEC structure would be as a whole. Therefore, extrapolation on an individual aircraft type basis is preferable.

In the case of the VNAV benefits analysis, no extrapolation is necessary since the VNAV benefit for each arrival has been included. The 4D RNAV benefits analysis concerned only operations at the twenty-five (25) airports with highest total delay. However, it is not appropriate to extrapolate these results to the remainder of operations for two reasons. First, the benefit to be realized diminishes rapidly as operations rate and total delay decreases at the smaller airports making extrapolation tenuous. Second, and more important, the 4D benefit can only be realized at airports employing a sophisticated form of Metering and Spacing automation. The number of airports with such services available will most certainly be limited to the higher delay airports. Likewise, 2D terminal area benefits were not extrapolated beyond the fifty-eight (58) airports.

Table 3.40 lists the total annual flight miles, based on the I February 1976 OAG, by aircraft type for each airline studied as well as the percent of those flight miles which would occur on the NAFEC 429 airport pair high altitude structure. The flight hour cost range for each aircraft, also listed in Table 3.40 was computed as follows: The high end of the range was based on 1973 Direct Operating Cost (DOC) minus fuel [37], inflated to 1975 dollars at 5.5% per year, plus fuel at \$.30 per gallon. The low end of the range was based on 1973 DOC minus fuel, depreciation, and rentals. This figure was then arbitrarily reduced by 20% (to allow for further variation of flight time sensitive DOC among individual airlines) prior to inflating to 1975 dollars. Fuel at \$.18 per gallon (representing lowest current prices) was then added.

The projected annual total and per aircraft benefits for EAL are given in Table 3.41 for 2D and 3D and in 3.42 for 4D. The projected benefits for the other airlines studied are given in Appendix G.

Table 3.40
Airline Flight Hour Cost Range
<u>UAL</u>

AIRCRAFT	TOTAL ANNUAL ENROUTE FLIGHT MILES	% OF ANNUAL ENROUTE FLIGHT MILES ON NAFEC STRUCTURE		OUR COST ESS FUEL LOW
727	86,061,307	45%	\$ 612	\$ 356
727-222	20,487,791	61%	647	353
737-222	18,606,697	58%	565	346
DC-10	33,633,192	84%	1297	602
DC 8	41,249,912	58%	768	396
DC 8-61/62	30,032,832	81%	877	374
747	6,604,780	100%	1433	660

TWA

AIRCRAFT	TOTAL ANNUAL ENROUTE FLIGHT MILES	% OF ANNUAL. ENROUTE FLIGHT MILES ON NAFEC STRUCTURE	FLIGHT H RANGE (L HIGH	OUR COST ESS FUEL LOW
L1011	21,421,754	94%	\$1373	\$542
727	27,869,383	65%	636	373
727-200	33,363,355	56%	642	310
DC-9	8,708,373	51%	644	363
707	63,241,652	72%	744	392
747	1,533,168	100%	1907	563

NAL

AIRCRAFT	TOTAL ANNUAL ENROUTE FLIGHT MILES	% OF ANNUAL ENROUTE FLIGHT MILES ON NAFEC STRUCTURE	FLIGHT HO RANGE (LI HIGH	
727	7,651,302	64%	\$ 574	\$ 334
727-200	18,527,601	39%	610	363
DC-10	20,929,614	63%	1038	516

Table 3.40 (Cont'd.)

EAL

AIRCRAFT	TOTAL ANNUAL ENROUTE FLIGHT MILES	% OF ANNUAL ENROUTE FLIGHT MILES ON NAFEC STRUCTURE	FLIGHT H RANGE (L HIGH	OUR COST ESS FUEL) LOW
L-1011	13,911,792	58%	\$1614	\$ 499
727	64,878,577	49%	668	414
727-200	32,388,897	49%	597	339
DC-9-14	7,409,371	44%	466	250
DC-9-31	51,856,801	44%	500	301

DAL

AIRCRAFT	TOTAL ANNUAL ENROUTE FLIGHT MILES	% OF ANNUAL ENROUTE FLIGHT MILES ON NAFEC STRUCTURE	FLIGHT HOUR RANGE (LESS HIGH	
DC 8-51	7,686,915	67%	\$ 629	\$404
747 L1011	21,455,556 16,902,233	27% 81%	1907 1677	563 523
DC 8-61 727-95/232/295	10,030,994 70,020,915	73% 57%	812 635	436 326
DC-9	34,291,234	43%	438	230

AAL

AIRCRAFT	TOTAL ANNUAL ENROUTE FLIGHT MILES	% OF ANNUAL ENROUTE FLIGHT MILES ON NAFEC STRUCTURE		HOUR COST LESS FUEL) LOW
707-123/323 727 DC-10	62,486,119 86,379,812 24,227,667	70% 62% 94%	\$ 905 624 1225	\$ 450- 338 590
747	1,482,208	100%	2710	1395

Table 3.41

RNAV Benefits over VOR for 2/1/76 Schedule and Route Structure

		AMMILAI	N.		-		d	DED ATRCDAFT		
Aircraft		1		\$ Ra	Range	₽ of	Fuel	Time	S R	Range
Type	Benerit	(gal)	(mtn)	LOW	High	Aircraft	(ga1)	(min)	101	High
	20 Enroute	Ж169	18K	273K	847K		23.1K	607	9,202	23,248
917	20 TK	1498K	41K	610K	1552K	30	49.9K	1382	20,475	52,146
	30 THA	227K	9 ek	91K	230K		7.6K	198	3,015	1,606
	Total	2416K	65K	974K	2569K		80.6K	2187	32,692	83,000
	2D Enroute	2.20M	122K	1.24H	2.02M		29.4K	1631	16,544	26,975
121	20 THA	3.10M	162K	1.68M	2.73M	75	41.3K	2161	22,344	36,449
	30 TWA	м68.	41K	.44M	.72H		11.9K	244	968'5	9,627
	Total	6.19м	325K	3,36м	5.47M		82.6K	4336	44,784	73,051
	20 Enroute	1. 39M	701	647M	1 1144		35.5K	1800	16,566	28,570
725	20 TWA	2.36м	120K	1. 103M	•		60.5K	3066	28,213	48,657
	30 THA	M29.	28K	. 279M		39	17.2K	729	7,215	12,414
	Total	4.42H	218K	2.029M	3.496м		113.2K	5655	51,994	89,641
	20 Enroute	147K	12K	77K	138K		18.3K	1471	9.427	16.922
600	20 TMA	322K	25K	162K	291K	~	40.2K	3169	20,440	36,672
	30 THA	179K	11K	78K	139K		22.4K	1429	986'6	17,819
	Total	648K	48K	317K	568K		80.9K	6909	39,853	71,413
	20 Enroute	1.97M	157K	1.15M	1 90M		27.1K	2157	15,692	26,094
260	2D THA	3.37M	229K	1.76M	2.92M		46.2K	3136	24,048	39,993
	3D TWA	1.44K	89K	.71M	1.17M	73	19.7K	1213	9,631	16,018
	Total	6.78м	475K	4 63 W	700		92.9K	6506	49 371	82,105

Table 3.42

EA

4D RNAV PROJECTED BENEFITS

	Range	High	47,497	39,504	157,18	59,442	38.077
	S Ra	Low	18,067	24,260	29,935	32,952	22,901
PER AIRCRAFT	Time	(min.)	1333	2533	3487	2500	3301
PER A	Fuel	(gal)	38.8K	37.7K	56.8K	55.8K	35.2K
	• of	Aircraft	30	75	39	8	.73
	Range	High	1425K	2963K	2018K	476K	2780K
	\$ Ra	LOW	542K	1820K	1167K	264K	1672K
UAL	Time	(min.)	40K	190K	136К	44K	241K
ANNUAL	Fuel	(gal)	1163K	2825K	2217K	441K	2571K
	Aircraft	Type	017	727	22.2	600	\$60

4.1 SLANT RANGE EFFECTS

The problem of the slant range error effect on DME measurement accuracy has long been recognized. Provisions for additional protected airspace requirements near VORTACs have been made in the airspace handbook (Handbook 7110.18[38], based upon earlier analyses. The RNAV Task Force has recognized the slant range problem as both an airspace problem and as an equipment requirements and cost problem. It is therefore the intent of this section to investigate the slant range error problem in order to determine the actual impact of slant range error on aircraft track deviations and cockpit procedures in the lower altitudes (below 18,000 ft). It is furthermore the purpose of this section to investigate the airspace allocation criteria and equipment requirements criteria stated in Handbook 7110.18 and the Task Force Report [1], and to develop new criteria based upon the results of these new analyses and flight tests. Finally, the low altitude, high altitude and terminal area environments are evaluated as to the impact of slant range error and these newly developed criteria on the flexibility of the route designer in the placement of routes, and the operational flexibility of the controller in utilizing RNAV maneuvers.

The major task of this analysis is to examine the influence of slant range error on deviations from desired track and on cockpit procedures. The determination of airborne equipment requirements and/or effects on airspace design procedures are affected by slant range induced track deviations and pilot reactions. One potential solution to the slant range error problem is the requirement for slant range correction for all users or for admission to certain airspace. This is a step to be taken only after careful deliberation since the impact on low-cost RNAV systems is significant, due to the fact that increased capabilities of both the altimetry and RNAV computational systems are required. Alternatively, if the slant range effects on airspace requirements or cockpit operational problems are severe, slant range correction or restrictive route design procedures which could impact route efficiency may be required. It was therefore the intent of this analysis to concentrate the study of the slant range error problem where it is least severe, where encoding altimeters are not necessarily required (below 12,500 ft.), and where the overwhelming majority of lower capability general aviation users operate: the low altitude and terminal area airspace. Using the data and techniques so developed, operations in the intermediate (12,5000 to 18,000 ft.) airspace and high altitude airspace have also been evaluated.

4.1.1 Methodology

The slant range error effect has been the subject of analysis for quite some time. The magnitude of the guidance error is easily determined under any specified conditions. However, the resulting effect upon actual deviations from intended track is very difficult to assess, although it is widely accepted to be significant for aircraft at high altitudes. Likewise, the degree to which the guidance fluctuations disturb the flight crew is not readily apparent. Very little flight test work has been performed in this area. As a direct

result of this, a specific test program dedicated to the slant range error effect in the terminal area environment was conceived and carried out, and the results were used to calibrate a simulation algorithm so that flight regimes other than the ones tested could be analyzed.

The first step in the procedure was to plan and conduct a dedicated slant range error flight test program. The flight test plan is presented in Appendix E. Briefly, a twenty mile long track was specified with the VORTAC at the middle. Four combinations of altitude and nominal tangent point distance were selected, with two pilots flying each combination twice plus one additional flight, for a total of seventeen flights. The data recording and recovery techniques used were identical to earlier test programs at Denver and Miami [8], except that DME range, TO/FROM flag and CDI valid flag data were also recorded. Data reduction and computer processing procedures were also similar to those used for the earlier test programs, except that an additional error analysis program was developed to process the DME data and isolate the slant range error components, and compute statistical values. Diagrams showing actual aircraft track and RNAV indicated track were created for each flight, and are included in Appendix E.

The resulting flight test data was analyzed in an attempt to determine the effect of slant range error as it disturbs achieved track as compared to hypothetical identical flights with no error. This effort was complicated considerably by the presence of other errors in the navigation system and VORTAC which typify such RNAV operations. Pilot and observer comments and the data were reviewed to assess the operational significance of the slant range effect, particularly in comparison to other operational problems including navigation errors and signal dropouts.

In order to further refine the analysis of the flight test data and to be able to determine the effect on other types of aircraft operating at different speeds, a dynamic computer simulation program was developed. This program was similar to, and used the same aircraft/pilot control system as, the simulation reported in Reference 39. The specific control system sensitivities and parameters were adjusted in order to match closely the slant range error induced effect apparent in the flight test data. The flight test aircraft model was then simulated over a wide range of altitudes and tangent point distances. More importantly, a very slow aircraft representative of the class of aircraft most seriously affected by the slant range error was simulated over a broad range of conditions. A faster, less maneuverable aircraft was also modeled in order to confirm that the effect is less significant in that case. The results of these simulation studies have been expressed in two ways which are critical to the understanding of the slant range error effect on airspace requirements. First, the additional airspace required to assure that 95% of the aircraft remain within boundaries is discussed. Alternatively, plots are shown which depict the decrease in level of confidence of remaining within a fixed route width as a function of nominal tangent point distance.

4.1.2 Terminal and Low Altitude Analysis

It is obvious from an examination of the aircraft track plots, a typical example of which appears as Figure 4.1, that the slant range effect is well hidden amongst all of the other errors and disturbances affecting each flight. While this made analysis rather difficult, it is perhaps an early indicator of the relative magnitude of the slant range effect under 12,500 ft. The most obvious errors are an angular bias in the RNAV/VORTAC system, and a scallop effect. The scalloping was reported by the flight crew, particularly on the west side of the station when flying west. The slant range effect experienced by an aircraft is dependent wholly upon the actual track it is making good, not the intended or nominal track. Also, since the nominal track is deviated from, the slant range error effect on track deviation must be measured with respect to the track which probably would have been achieved had there been no slant range error, rather than the nominal track. In the case of simulation there is no problem since no other errors exist to disturb the track. Flight test data is quite another story, however. In order to attempt to discern the probable track without slant range error, the trend of the aircraft position at intervals along the inbound (to the VORTAC) portion of the track was determined by a least square curve fit to a straight line. Two measurements were then made with respect to this straight line: the actual tangent point distance of the achieved track, and the maximum deviation of the aircraft from the line immediately following VORTAC passage. The first measure, along with altitude, is the basic parameter affecting slant range error. The second was an attempt to quantify the track deviation due to slant range error. If no other errors were present, these two measures should be related in some consistent manner. However, certain flaws are obvious. First, the curve fit is not necessarily an accurate representation of the actual situation and, second, the deviation value measured includes the effects of other navigation errors and winds. The results of these tests are summarized in Table 4.1. The table lists, for each flight, the subject pilot, direction of flight, altitude, tangent distance to the nominal track, the angle corresponding to that distance, the tangent distance to the curve-fit track and the corresponding angle, the mean and standard deviation of system cross track error and the maximum deviation ("track bulge") attributed to slant range error. Note that in two cases (flights 6 and 8) the curve fit could not be reliably applied. These maximum deviation results, versus tangent point distance, are plotted in Figure 4.2. The data is noisy and this is highlighted by the fact that the 8000 ft points show in general a greater deviation than the 12000 ft points, which is contrary to expectations. The simulation, as discussed below, was subsequently used to remove much of this noise from the flight test data, revealing a much more reasonable result.

4.1.2.1 Simulated Aircraft Control System

The basic control system equations are documented in Reference 39. The heart of the system is a third order autopilot/aircraft model which uses cross track deviation, heading and command heading inputs. The parameters of significance include the cross track deviation gain, the maximum aircraft

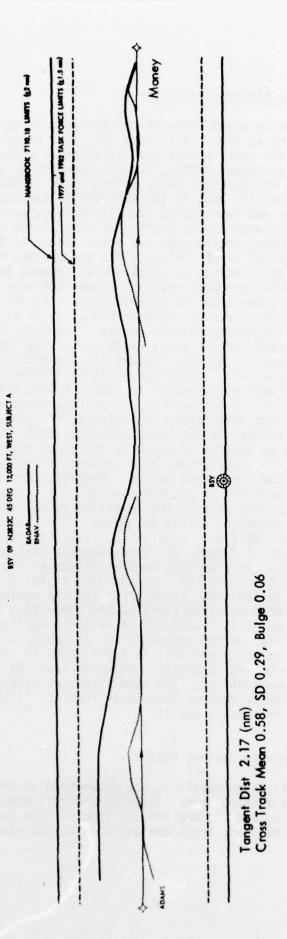


Figure 4.1 Example Aircraft Track and RNAV Guidance Plot

4-4

Table 4.1 Slant Range Effect Flight Test Results

											_		_				
Track Bulge	9.6	0.29	0.49	0.25	91.0	0.25	0.07	0.26	0.25	90.0	0.10	1	0.00	0.21	0.11	1	9.0
Cross Track	0.72 0.39			0.85 0.40													
Tangent Angle (Actual)	38.0°	35.0	60.70	76.30	4.90	47.20	89.10	æ.*.	43.30	42.30	42.20	1	76.10	57.90	57.90	:	84.20
Tangent Dist. (Actual)	1.69	1.88	0.74	0.32	1.32	1.22	9.0	6.13	2.10	2.17	2.18	!	0.49	1.24	1.24	1	0.20
Tangent Angle (Nominal)	45°	450	450	450	009	009	009	000	45°	45°	450	45°	450	009	009	009	009
Tangent Dist. (Nominal)	1.32	1.32	1.32	1.32	0.76	0.76	0.76	0.76	1.98	1.98	1.98	1.98	1.98	1.1	±	7.7	1.14
Altitude	*	*	*	*	*	*	*	*	12K	12K	12K	12K	12K	12K	12K	12K	12K
Flight Subject Direction	3	3	3	w	*	*	w	w	3	3	3	w	w	3	3	w	w.
Subject	4		4	*	<	8	4	•	۷	4	•	4		4		<	•
ght	-	10	2	=	3	12	+	13	7	6	91	8	21	2	7	9	15

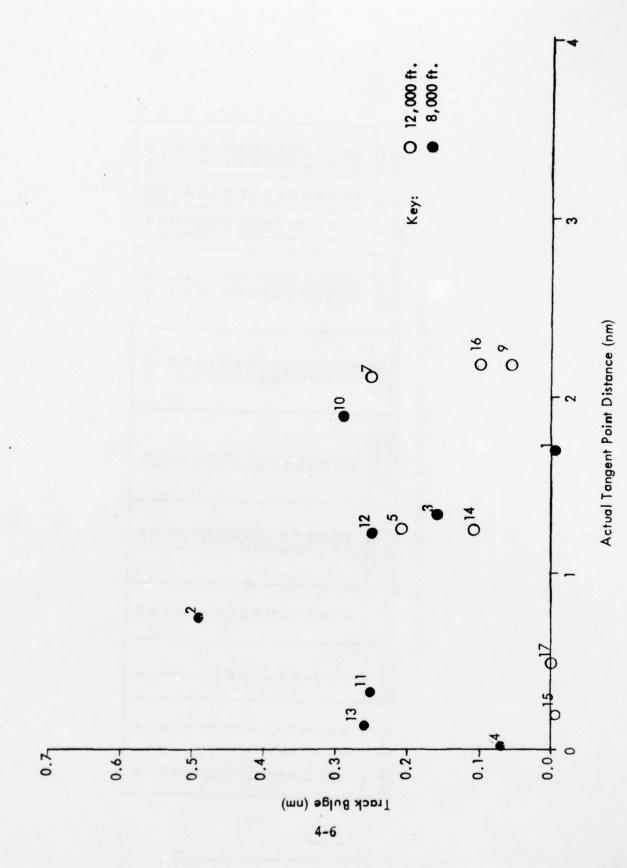


Figure 4.2 Slant Range Flight Test Data

bank angle, and the track acquisition intercept angle. The bank angle limit selected was twenty-five degrees. The cross track deviation gain was selected differently for different aircraft speeds, as discussed below. While the cases simulated are track-following cases rather than track-acquisition cases, the track intercept angle was useful for limiting the heading change which a pilot would take to correct course. After selecting these parameters a stability parameter was adjusted to get the fastest non-oscillatory aircraft response available. To more accurately model the performance of a human pilot, a cross track deviation dead-band was also simulated.

Several aircraft response characteristics were tested in order to ascertain the behavior of the control system. The characteristics which most accurately recreated the behavior of the slant range effect flight tested was the set based upon pilot interviews and an analysis of the track plots. The parameters used were a speed of 160 kt (the speed of the flight test aircraft), a cross track sensitivity of ten degrees heading change per dot of needle deflection, no heading change limit and a dead-band of 0.35 dot (nm). The dead-band value was estimated by examining the aircraft track plots, whereupon it was found that a deflection of at least 0.25 nm and sometimes as much as 0.5 nm was required before a heading change resulted. Likewise, the average heading change resulting from one dot needle deflection was found to be about ten degrees. The track deviation or bulge characteristics of this modeled aircraft system are shown in Figure 4.3 for several values of tangent point distance (note that the VORTAC stations are shown). The resulting bulges, with and without dead-band, are illustrated in Figure 4.4 as a function of tangent point distance.

4.1.2.2 Simulation Calibration

Using the parameters stated above, the track deviations or bulges produced were of the same magnitude as those experienced in the flight tests. It was, however, desired to more accurately correlate the two. A major weakness in the analysis of the flight test data was in determining the effective tangent point distance for each flight, since the value selected additionally impacts the calculation of maximum track deviation. Therefore, means for relating the simulation results to the flight test results were sought. After examining several of the statistical measures available, it was found that the standard deviation of the along track component of slant range error was useful since it implicitly contains tangent point distance information. It is additionally useful since the mean value is always very nearly zero, and therefore one number can characterize the entire situation. The simulations were purposely run over identical track lengths, etc., as the flight tests so that all statistics collected would be comparable. The curves relating the standard deviation of the along track component of slant range error to tangent point distance, as derived from simulation results, are shown in Figure 4.5. Note also that these curves are very insensitive to control system parameter changes of a moderate degree. The standard deviation values change less than 6% when the cross track deviation gain is doubled to twenty degrees per dot deflection at 12,000 ft at a tangent point distance of one mile, for example.

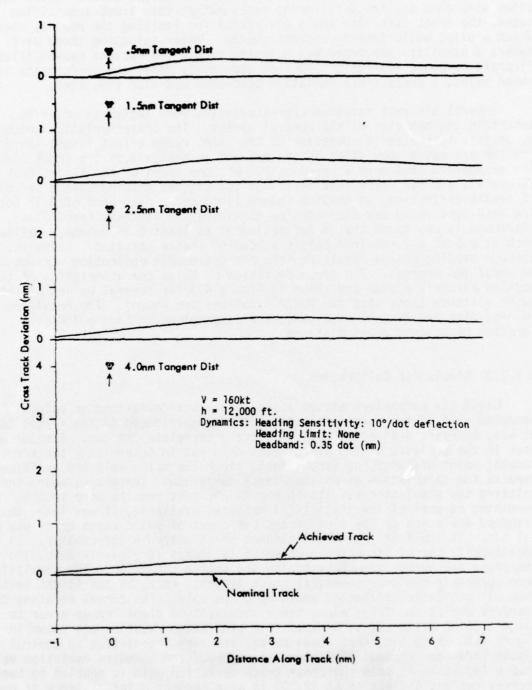


Figure 4.3 Track Deviation Due to Slant Range Error

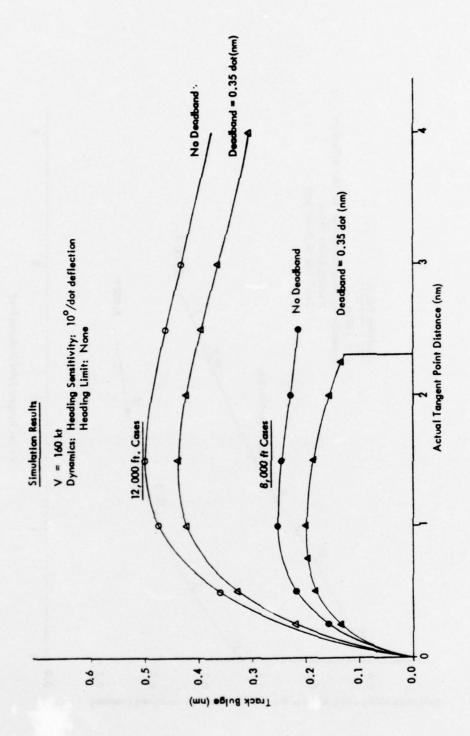
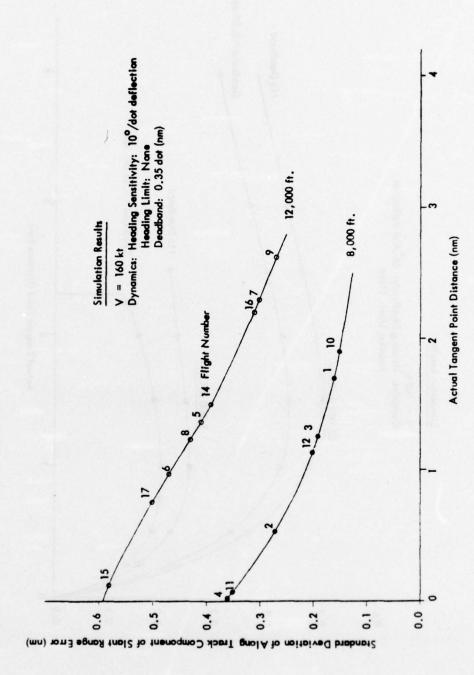


Figure 4.4 Maximum Track Deviation versus Tangent Point Distance



Standard Deviation of Along Track SRE versus Tangent Point Distance Figure 4.5

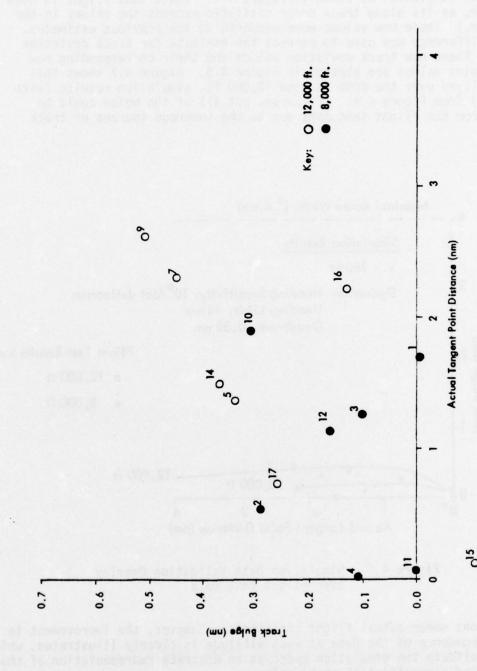


Figure 4.6 Slant Range Data Using Corrected Tangent Point Distance

In order to better evaluate the flight test results, a new tangent point distance was selected for each based on their slant range along track error statistic, as shown in Figure 4.5. (Note that flight 13 does not appear, as its along track error statistic exceeds the values in the simulation.) These new values were compared to the previous estimates, and the difference was used to correct the estimate for track deviation (bulge). These new track deviation values and their corresponding new tangent point values are plotted in Figure 4.6. Figure 4.7 shows this data overlayed over the 8000 ft. and 12,000 ft. simulation results (with dead-band) from Figure 4.4. Of course, not all of the noise could be removed from the flight test data due to the numerous sources of track

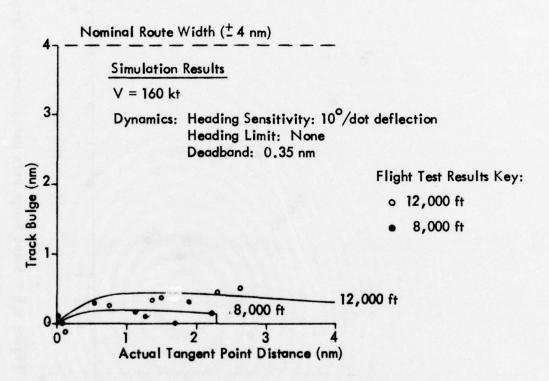


Figure 4.7 Simulation Data Validation Overlay over Flight Test Data

perturbations under actual flight conditions. However, the improvement in the correspondence of the data at each altitude is clearly illustrated, which tends to validate the simulation model as an accurate representation of the flight-tested aircraft/pilot combination.

4.1.2.3 Simulation of the Worst Case Aircraft

In order to ascertain the most serious track deviations to be expected in the low altitude airspace, a slow aircraft with correspondingly severe maneuvers was modeled. It is important to outline the characteristics of this type of aircraft very carefully, and to substantiate the choice of it as being the worst case. As a general rule, larger heading changes are taken by a pilot of slower aircraft in course correction maneuvers than would be the case with faster aircraft. The major reason is that, because of its slow speed, larger angles are required to get back on track in a reasonable period of time. If a fast aircraft were to make such large corrections for normal course deviations, it would overshoot its path and could end up farther off track than before. Even the course corrections used by the pilots of the flight test program (approximated by 10°/dot deflection) were stated by those pilots to be greater than, and to have been applied more rigorously than, the course maneuvers they would apply in the course of normal work as air charter pilots. Thus the 10°/dot figure is probably conservative for the class of light twin used in the test (the 160 kt capability is slower than most twins). The FAA Instrument Flying Handbook [40] recommends, for low speed aircraft, that a heading change of 20° be used for course correction purposes. While no threshold value or specific needle deflection is stated, it is implied by the text and figures that this is for correction of significant deviations from track. In order to interpret this in a conservative manner, the control system gain was set to a response of 20°/dot deflection; however, a heading change limit of 20° maximum was also invoked in order to correspond to the recommendations of the handbook. The course deviation dead-band remained at 0.35 dot (nm), as before, to complete the model of the worst case aircraft.

The slow, responsive aircraft has long been thought to be the one most seriously affected by the slant range problem simply because its maneuverability and speed allow it to respond to the slant range perturbation as the VORTAC is passed, while higher speed aircraft would tend to miss the short-lived effect altogether. This has turned out to be the case, as illustrated by the curves at 8000, 10,000 and 12,000 feet in Figure 4.8. Note that the maximum deviation of 0.72 nm at 12,000 feet is 64% greater than the 0.44 nm deviation of the 160 kt aircraft. For purposes of illustration, the track profiles at 10,000 ft. are shown in Figure 4.9 for several tangent point distances. A faster (200 kt), less maneuverable aircraft was also studied to confirm that the slant range effect on track deviation is less significant in that case than for the flight test case.

The role which this "worst case" type of aircraft plays in IFR operations is of importance here. Data taken from the IFR Peak Day Survey for 1971 [41] is summarized in Table 4.2. The table contains data for only those aircraft which we are presently concerned with: low altitude aircraft which would probably not be required to have an encoding altimeter. As may be seen from this table, 72% of all aircraft concerned file flight plans below 8000 ft., and so do not really have a slant range problem of any significance. The "worst case" type aircraft filing for altitudes above 10,000 ft constitute only 0.4% of those flight plans. This residual number is on the order of

three sigma expectations (99.75%), while route widths are designed around two sigma expectations (95.44%). The maximum deviation from track at 10,000 feet for the "worst case" aircraft is 0.52 nm, and at 12,000 feet is 0.72 nm. These may be considered to be the largest likely deviations due to slant range error in operations below 12,500 feet, since the maximum for the 160 knot aircraft is smaller, 0.44 nm.

TABLE 4.2 IFR PEAK DAY FLIGHT PLAN DATA

Altitude Range	Single Engine 1 to 3 Places (worst case)	Single Engine 4 or more Places	Multi Engine Under 12,500 lb.
Number of Aircraft: 0 to 7,999 ft. 8,000 to 9,999 ft. 10,000 to 12,999 ft. (Total Aircraft: 8749	180 35 35	2056 317 101	4092 1226 707
Per Cent of Total:	Toolage Sec.	auto or sensor or	
0 to 7,999 ft. 8,000 to 9,999 ft. 10,000 to 12,999 ft.	2.05% 0.40% 0.40%	23.50% 3.62% 1.15%	46.77% 14.01%

4.1.2.4 Impact on Protected Airspace Requirements

The basic standard governing RNAV route widths is Handbook 7110.18 [38]. It includes criteria for expanding route widths due to the slant range effect for low and high altitude enroute and terminal area operations. As stated before, this discussion is limited to operations below 12,500 feet. The additional airspace requirements stated are as follows:

Phase	Nominal Route Width	Altitude	Tangent Point Distance	Expanded Route Width	Along Track Distance
Enroute Terminal	+4.00 +2.00	<17,000 10,000-17,000 <10,000	2.0-3.0 0.9-5.2 0.9-2.0	4.90 4.00 2.85	±10. ± 5. + 4.

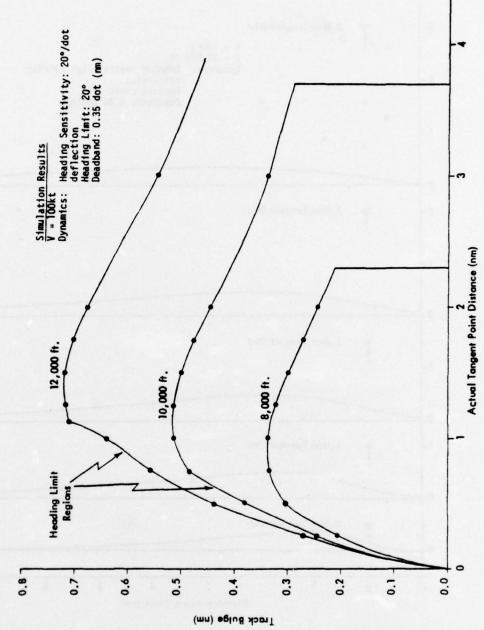


Figure 4.8 Maximum Track Deviation versus Tangent Point Distance

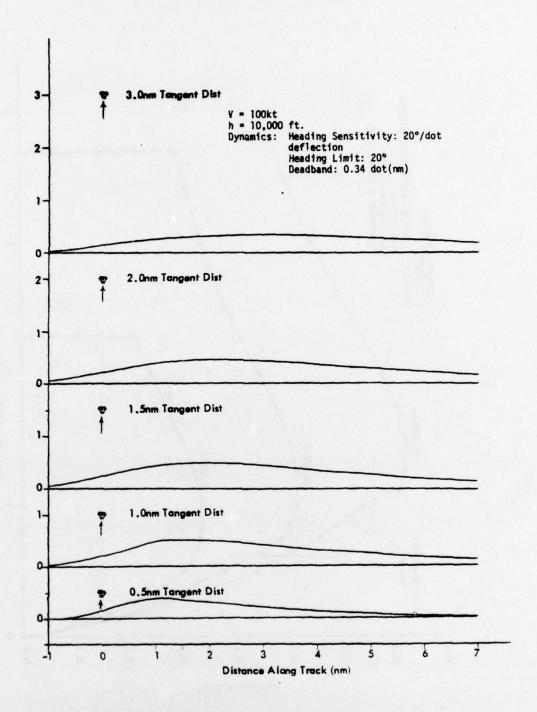


Figure 4.9 Track Deviation Due to Slant Range Error

The expanded route widths are on the VORTAC side of the route centerline only, and extend the indicated along track distance both prior to and subsequent to the station. These values are based upon a mathematical analysis of the slant range guidance error. The flight test/simulation data introduced here is in conflict with these earlier findings, since this data concerns actual aircraft performance under slant range error conditions, not simply guidance error values.

Three major points are raised by this new data. First, the additional protected airspace requirements below 12,500 feet need not be as large as indicated in Reference 38, but may be limited to 0.5 nm (to 10,000 ft) and 0.75 nm (to 12,000 ft). The value indicated in the Handbook for terminal operations at or below 10,000 feet is, for example, 0.85 nm, and would certainly be greater at 12,500 feet. Furthermore, the slant range effect becomes inconsequential below 8000 ft.; therefore SID or STAR route segments constrained at such altitudes when near the reference facility need no additional protected airspace. The second major point is illustrated in Figure 4.10. Note that, except in the case of extremely small tangent point distances where the effect is small, the overall deflection from desired track is less than the actual tangent distance from the actual aircraft track (before passing the VORTAC) to the station, as indicated by the 45° reference line. This simply means that, wherever the aircraft track would have been if there were no slant range error, the aircraft would not be pulled to the other side of the VORTAC by the slant range effect. This means that the presence of a station anywhere between the route centerline and the airspace boundary would not cause any aircraft which would otherwise stay within the protected airspace boundary to exceed that airspace. Therefore, additional slant range airspace need only be provided when the station location is outside of the protected airspace region. This is not reflected in the handbook data, which specifies airspace expansion for VORTAC locations within the normal route width in all three cases listed. The third point of importance is illustrated in Figures 4.3 and 4.9. In all cases the aircraft deviates from track subsequent to VORTAC passage. Therefore, additional protected airspace need only be provided on one side of the station for one-way routes, SIDs and STARs. These figures also illustrate the fact that such protection should be provided for at least five, and probably eight miles along track from the station.

There is an additional way of assessing the protected airspace requirements problem as it is affected by slant range error. It is instructive to look at the impact of slant range error on the probability of remaining within the nominal route width (without expansion). Usually, only a small proportion of the aircraft would be caused to exceed the airspace boundary. For example, if the parameters of the situation were such that a 0.4 nm deviation from track would be expected, only those aircraft which would normally be within 0.4 nm of the airspace boundary would be caused to violate the airspace boundary. Therefore the probability of exceeding the protected airspace (nominally 1-.9544 =.0456) would increase only by the probability of an aircraft being within 0.4 nm of the airspace boundary on side near the VORTAC. This level of probability depends upon the nominal route width, assuming that in each case the route width represents the

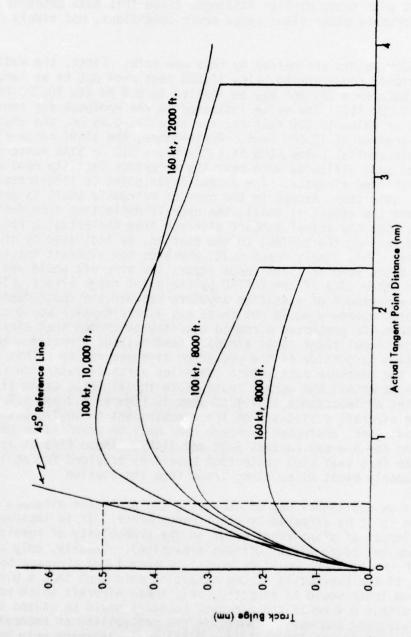
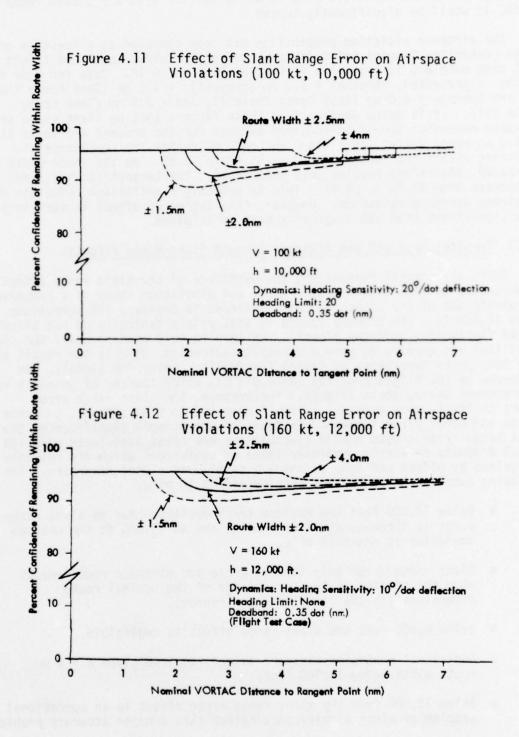


Figure 4.10 Maximum Track Deviation versus Tangent Point Distance



two sigma probability level. Using a \pm 4 mile route width, the probability of being within the outer 0.4 mile is quite small. With a \pm 2 mile route width, it would be significantly larger.

The airspace violation probability has been computed as a function of route centerline tangent point distance using the "worst case" and flight test case data and is presented in Figures 4.11 and 4.12. Data for four route widths is presented: Terminal \pm 2.0 nm (present), \pm 1.5 nm (Task Force Phase II) and Enroute \pm 4.0 nm (Task Force Phase II) and \pm 2.5 nm (Task Force Phase III). It is quite obvious from these figures that no slant range error airspace expansion should be required enroute for the present and Phase II (\pm 4.0 nm route width) since the error only decreases the confidence of remaining within the route width from 95.4% to 93.6%. As the route width is decreased, the effect becomes more pronounced. The largest (at \pm 1.5 nm) is a decrease from 95.4% to 88.4%. This is probably a sufficient reduction for requiring airspace expansion. However, this degree of effect is certainly less significant than had originally been anticipated.

4.1.3 Terminal Area and Low Altitude Enroute Slant Range Effects

There are several reasons why the magnitude of the slant range effect demonstrated in this flight test program and simulation study is a conservative representation of the effect to be experienced in ordinary IFR operations below 12,500 ft. The primary reason is that pilots typically do not blindly follow fluctuating guidance signals, but tend to hold heading until the signal stabilizes and appears to give a reliable indication. This is the result of many long years experience at flying fluctuating, noisy VOR signals. As evidenced by the flight profiles (Appendix E), other sources of guidance noise were present during these flights. Furthermore, the slant range error effect on track, when present, is small compared to wind effects, guidance biases and other error sources. In terms of operational significance, the slant range error caused needle fluctuations are often associated with VOR signal dropouts or erratic behavior (zone of confusion) which are commonly recognized by pilots and ignored until a stabilized signal reappears. The following conclusions bear these considerations in mind.

- Below 12,500 feet the maximum track deviation due to slant range error is three-quarters of a mile, and at 10,000 ft the maximum deviation is one-half mile.
- Slant range error only impacts protected airspace requirements when the station is located outside of the nominal route dimensions (in the low altitude airspace).
- Below 8,000 feet the slant range effect is negligible.
- Additional protected airspace is not required given a \pm 4 nm route width below 12,500 feet.
- Below 12,500 feet the slant range error effect is an operational problem of minor significance rather than a major accuracy problem.

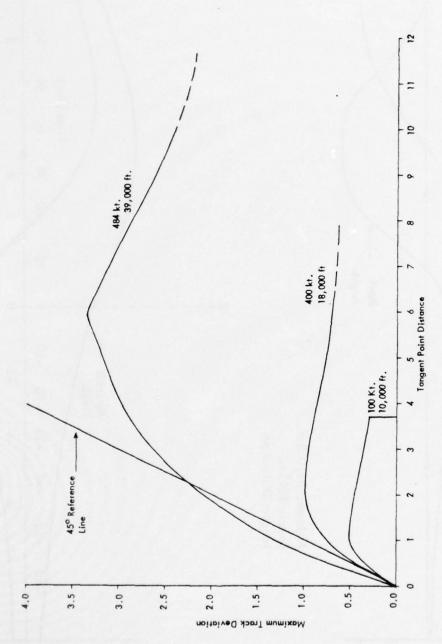
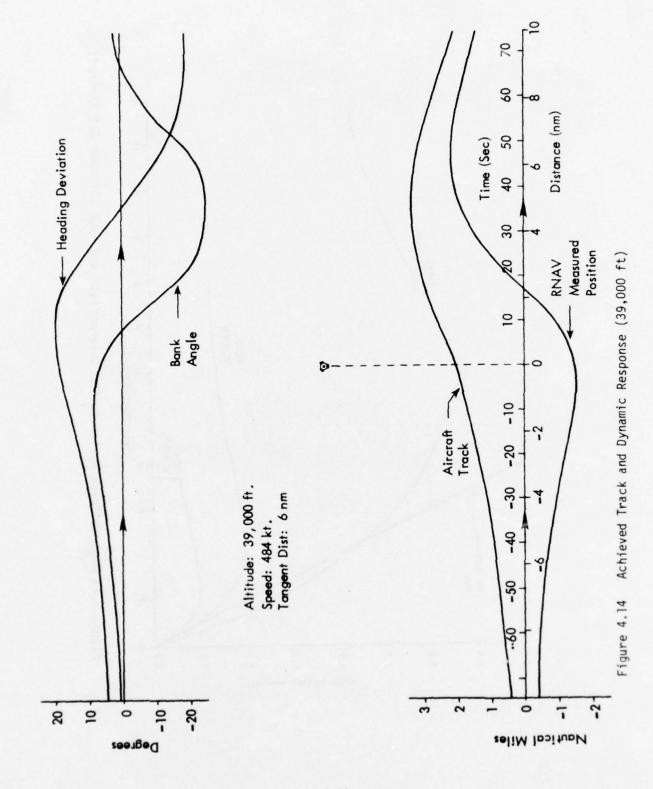
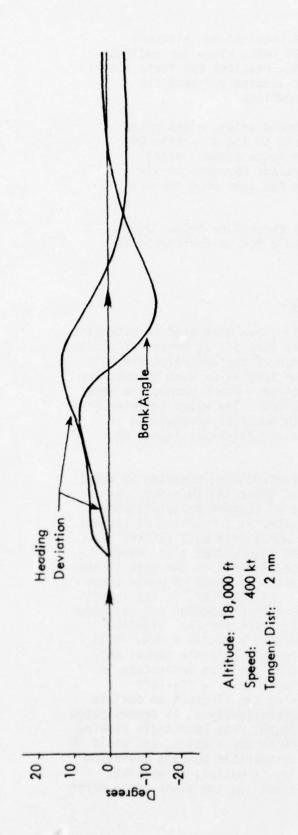
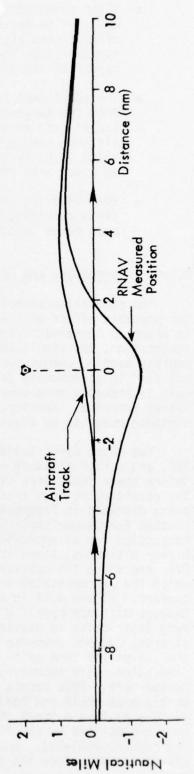


Figure 4.13 Maximum Track Deviation for Intermediate and High Altitude Jet Aircraft







- Slant range correction should not be required for aircraft confined to the airspace below 12,500 feet, since the addition of protected airspace near the VORTAC, required for route widths narrower than four miles, should not adversely impact the layout of the enroute and terminal routings.
- The additional protected airspace should extend eight miles from the tangent point along the route in the direction of travel, and should be added when the route tangent point distance ranges from the nominal airspace boundary to six miles (e.g. from two to six miles in the case of + two mile terminal area route width).
- Even with the absence of slant range correction below 12,500 feet, resulting track deviations should not be noticeable in most cases to controllers.

4.1.4 Intermediate and High Altitude Analysis

The simulation techniques discussed earlier have been used to determine the probable effect of slant range error on jet aircraft at higher altitudes. In order to conservatively model the performance of the autopilot/aircraft combination, the simulation gain was set to the same value used before, ten degrees heading change per mile guidance deviation. Other parameters were adjusted to accommodate the higher speeds involved. The value selected for gain is probably more sensitive than that which would be encountered in actual practice. However, this allows the maximum potential impact on airspace usage to be assessed.

Two cases were studied, a jet aircraft at 18,000 ft. cruising at 400kt TAS, and a high altitude cruise (39,000 ft.) at 484kt TAS (M = .84). As before these cases were simulated over a range of tangent point distances. The resulting maximum track deviations are plotted as a function of tangent point distance in Figure 4.13. Note that the worst case data (100kt) is plotted for comparison. It is obvious from the figure that slant range correction (or an approximation to it) would be required in the case of the higher altitudes, since track deviations in excess of three nm would occur. Over and above the question of airspace boundaries, however, is the impact which the uncompensated error would have on cockpit procedures and passenger comfort. Figure 4.14 is a plot of the aircraft track for the six-mile tangent distance case. While the aircraft track is a smooth curve, it is very instructive to examine the behavior of the RNAV guidance signal and aircraft dynamic response more closely. The RNAV position indication is plotted on the same axis. The slant range effect causes a fly-left indication while approaching the station, causing the aircraft to deviate to the left. This occurs quite smoothly and unnoticeably as is demonstrated by the bank angle and heading plots in that figure. The bank angle remains at four or five degrees throughout the approach to the station, and would remain unnoticed by the flight crew until a considerable heading deviation had been accumulated. However, once the station is passed, a "whiplash" effect occurs: since the aircraft is left of track and the slant range error component is diminishing rapidly, a large (two mile) fly-right indication is produced quite quickly. In addition, the heading deviation is now in the wrong direction and combines with the RNAV indication to cause a roll to the right to the bank limit (25°) in order to produce the prescribed change in heading. Obviously, this is not suitable as a routine enroute procedure, and so indicates a requirement for slant range error compensation.

In addition to the disturbing dynamics associated with nearby passes to the VORTAC station, there are additional reasons for requiring slant range correction in the high altitude environment. Primarily, at high altitudes the slant range error effect is not a localized effect, but would impinge upon airspace requirements for almost all high altitude routes. For example, at 39,000 ft. an RNAV guidance error of one mile (at the tangent point) exists at a distance of 21 miles from the station, one-half mile at 41 miles, and one-quarter mile at 83 miles away. Since nearly all RNAV routes pass within fifty miles or so of the reference stations, and since the error levels change so slowly at those distances that the aircraft/control system would track them, additional protected airspace would be required.

The case for slant range correction for aircraft flying at intermediate altitudes (12,500 to 18,000 feet) is not nearly so clear-cut. Figure 4.15 shows that the maximum course deviation at 18,000 feet to be one mile for a 400 knot aircraft. (A slower aircraft would deviate much further, however). One mile of additional protected airspace in the region of the VORTAC between 12,500 and 18,000 feet would probably be sufficient to resolve the slant range error problem from an airspace point of view. However, the aircraft dynamics associated with the guidance error (see Figure 4.15), while not nearly so severe as in the high altitude case, still supports the case for requiring slant range correction. This is particularly true in light of the fact that encoding altimeters, which are the most costly part of the slant range correction process, will be required in that airspace. If necessary in order to minimize costs, approximate correction techniques could be allowed. Approximations could include an assumed mean station elevation, or the use of curve-fit approximations to the root-difference-square correction computation. Existing systems could thus be upgraded with the simple addition of a DME signal pre-processor or similar device separate from the existing RNAV equipment. These approximate techniques would be suitable due to the limited range of altitudes to be accepted (0 to 18,000 ft).

4.1.5 Intermediate and High Altitude Slant Range Effects

- If left uncorrected, slant range error would produce unacceptable airspace expansions affecting nearly all high altitude enroute segments.
- Above 12,500 feet, the slant range error can produce signal fluctuations and dynamic reactions which could be disturbing or misleading to the flight crew.
- Slant range correction is required above 18,000 both to conserve airspace and for purposes of passenger comfort and guidance signal integrity.

- Some form of slant range correction should be required above 12,500 feet.
- Approximate techniques for slant range correction should be considered for suitability for operations between 12,500 and 18,000 feet.

4.1.6 Effect of Slant Range Error on Route Placement and RNAV Maneuvers

The results of the analysis in this section indicates that if slant range correction is required above 12,500 ft. MSL, the only area where slant range error will have any impact is between 8,000 ft. and 12,000 ft. in the terminal area. The seven terminal area designs described in Reference 2 were examined to determine if the requirement for additional route width in the vicinity of VORTACs, or described in Section 4.1.3, would have any impact on route placement, and it was determined that there was no impact on any of the designs. The potential impact of slant range error on the utilization of RNAV maneuvers in the terminal area is also negligible. Of the two route deviation maneuvers used, "direct to a waypoint" and "parallel offsets", only the latter has any route width connotation. As pointed out in Section 4.1.2, route widths are established on the basis of assuming that aircraft will remain within the specified route width 95.4% of the time. Slant range effects between 8,000 and 12,500 feet will cause only a slight reduction in that percentage and will be indistinguishable to the controller from other navigation error induced excursions.

4.2 ENROUTE HIGH ALTITUDE VORTAC REQUIREMENTS

This phase of the ATC system impact analysis was conducted to identify the modifications to the existing VORTAC system, and the associated implementation costs, which would be required to support both charted and preplanned direct structures in the high altitude enroute environment. These modifications were determined specifically for route width-error budget combinations recommended by the RNAV Task Force [1], for operations at altitudes between 18,000 and 45,000 feet. The modifications considered included addition of new stations and upgrading of existing stations to provide high altitude coverage. This study assessed only the overall scope of the requirements for high altitude VORTAC coverage. Determination of the specific, individual station requirements must be based on an implementation route structure yet to be developed. This study also did not consider the removal of high altitude stations not needed for high altitude coverage because of their possible use in providing full low altitude coverage, a subject which was not addressed in this study.

In a previous study [12], preliminary estimates of potential modifications to the existing VORTAC system necessary to support charted and preplanned direct RNAV were made based on the assumption that certain postulated improvements to the VORTAC system would be accomplished [1]. Several alternatives were considered, including addition of new stations, bending of routes around coverage gaps, selectively increasing minimum enroute altitudes, and the utilization of variable route width requirements. In the current study, in addition to further consideration of minimum enroute altitudes, five other techniques were identified and evaluated as candidates to enhance the current VORTAC system to the point where the required navigation coverage could be achieved:

- Converting low altitude VOR/DMEs to high altitude operations.
 Converting low altitude VOR/DMEs to high altitude operations.
- and upgrading bearing error performance through the use of DVOR.
- (3) Upgrading the bearing error performance of existing high altitude VOR/DMEs through conversions to DVOR.
- (4) Establishing new VOR/DME stations.
- (5) Establishing new VORTACs with improved bearing error performance attained through the use of DVOR.

The VORTAC system requirements, subsequently presented in this section, were determined by estimating the most cost-effective combination of these five system enhancing techniques for each postulated set of RNAV system conditions.

4.2.1 Methodology

The study was structured to identify the impact on the VORTAC system necessary to provide coverage for both charted and preplanned direct (total CONUS) route structure coverage in the high altitude enroute environment for certain combinations of route width and navigation error budgets. The charted RNAV route structure utilized in the analysis was the 429 airport pair structure developed by NAFEC [4]. The route widths selected

were a constant ± 4 nm and ± 2.5 nm, which correspond to the Task Force recommended route widths for the 1977 and 1982 implementation periods respectively. The corresponding error budgets recommended by the Task Force are given in Table 4.3.

The 1977 case was analyzed with the primary goal of providing coverage at an altitude of 18,000 feet for the 429 airport pair RNAV route structure designed by NAFEC. However, this case was also evaluated for full CONUS coverage and at altitudes of 24,000 and 30,000 feet in addition to the baseline altitude. The 1982 case was evaluated to full CONUS coverage at 18,000 feet. Implementation cost sensitivities as a function of route width and flight technical error for each of four selected error budget sets were derived from a parametric analysis. Information is also presented to permit the reader to derive VORTAC system requirements for virtually any other error budget-route width combination and to modify the total implementation cost should it become necessary to adjust one or more of the unit costs used in this study. VORTAC frequency protection was examined primarily for the 1977 case. Because of the long term possibility of closer channel spacing and the potential elimination of high altitude NAVAIDs which may not be required in a total high altitude RNAV environment, a sophisticated frequency protection analysis for the "1982" period was not conducted.

The possibility of using ρ - ρ in lieu of ρ - θ navigation was also explored with requirements defined for both the NAFEC 1977 RNAV route structure and full 1982 CONUS coverage at 18,000, 24,000 and 30,000 feet, respectively.

Table 4.3 Task Force Recommended Route Widths and Error Budgets

Implementation Period	1977	1982
Recommended Route Width	+4 nm	+2.5 nm
Recommended Error Budget		The state of
Flight Technical Error	1.0 nm	1.0 nm
Ground VOR	1.5°	1.0°
Airborne VOR	1.5°	1.0°
Ground DME	0.1 nm	0.1 nm
Airborne DME	0.5 nm	0.25 nm
Airborne Computer	0.25 nm	0.25 nm

The methodology adopted to estimate the VORTAC requirements for any given set of route width and navigation accuracy constraints consists of the following steps:

- Convert the route width and navigation accuracy requirements, including station bearing error, into a maximum VORTAC coverage distance (radius);
- (2) Determine limits on coverage patterns of each high altitude VORTAC based upon terrain, signal strength and frequency protection considerations;
- (3) Impose the results of Step 2 as an additional constraint on the coverage patterns obtained in Step 1;

- (4) Aggregate the station coverage patterns to determine which portions of designated RNAV route structures or areas of the CONUS airspace do not have adequate coverage; and
- (5) Examine the alternative methods to fill the coverage gaps and identify characteristics and costs associated with those that produce the most cost-effective solutions.

This process was repeated as necessary to address the various system requirements. The specific methodology applied in each of these areas will be described in detail in the following subsections.

4.2.1.1 Individual Station Coverage

Fundamental to the VORTAC requirements study was the determination of the effective coverage area of each ground VORTAC station. The means by which this coverage can be determined is a direct result of the requirements imposed upon the VOR and DME signals. These are as follows:

- (1) The signal errors must be sufficiently low to satisfy the specified navigation accuracy requirements.
- (2) The signals must be free from interference, specifically interference from other stations whose signals would produce erroneous information.
- (3) The signals must be receivable within the station's service volume.

Navigation errors result from a variety of sources in addition to the ground station. The current error budgeting practice (outlined in AC-90-45A) [19] assumes independent normal error distribution, which allows the use of a root-sum-square (RSS) technique to estimate overall cross-track error based on the error components. Large VORTAC station errors logically produce large navigation errors. The DME signal error, as well as its cross track impact, is generally accepted not to be a function of range. Such is not the case with the VOR signals, however. While the signal itself may degrade only slightly as the range increases, its impact on aircraft position error increases linearly with range. Based upon the RSS equations, if values for all of the component errors are assumed, the usable range of a station can be determined by the following equation:

$$R = \frac{\sqrt{W^2 - C^2 - \rho_G^2 - \rho_A^2 - F^2}}{\sqrt{\theta_A^2 + \theta_G^2}}$$
 (4.1)

where

R = range (nm)
W = desired route width (nm)
C = airborne computer error (nm)
PG = ground DME error (nm)
E = airborne DME error (nm)
F = flight technical error (nm)
B = airborne VOR receiver bearing error (radians)
B = VOR transmitter bearing error (radians)

The route width, W, is assumed to be synonymous to the 2 σ value for cross track error. The equation provides appropriate results when 2 σ values for the component errors are utilized.

The second requirement imposed upon VOR/DME signals stems from frequency protection considerations. The current practice is to insure frequency protection for high altitude VORTACs out to a distance of 130 nm. This implies that an aircraft anywhere within 130 nm of its selected station would not be able to receive signals from any other station of the same frequency, nor would the signals to be interfered with by stations on adjacent frequencies (the manner in which this frequency protection is insured is discussed in Section 4.2.1.5). Only extreme combinations of large route widths and small navigation errors permit the use of VORTAC stations at distances beyond the current regulatory limit. While a reduction in the frequency protection radius would have no impact on the current stations, an expansion of the service area, as would be necessary to provide a greater coverage radius, may induce frequency protection violations which would not otherwise exist. For these reasons, the current frequency protection radius of 130 nm was imposed as an upper bound for all station coverage addressed in this study.

The previous two requirements serve to insure that the VORTAC signals are sufficiently accurate and free from interference. Whether or not the signals are receivable is the final area of concern. Terrain effects and signal strength must be considered. The data necessary to establish the coverage patterns for each of the high altitude VORTACs was available, at least in part, from two independent data sources; random coverage data (RACO) [42] and Electromagnetic Compatability Analysis Center data (ECAC) [43].

The ECAC data was generated by an analytical procedure using topographical information measured every 30 seconds of latitude and longitude (1/2 nautical mile). This grid-oriented information was used to establish the horizon angle for each one degree radial surrounding each station. The coverage distance along each radial was then computed based upon the horizon angle and the altitude at which coverage is desired, of which a total of eight were considered. The RACO data were derived primarily from flight inspections predicated on flying the 5 micro-volt signal strength contour around the stations at an altitude of 18,000 feet.

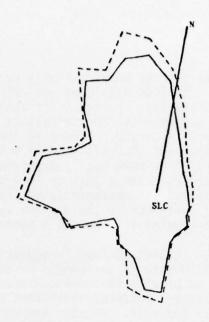
While both of these data sources provide highly reliable information, each has its disadvantages. The ECAC data does not reflect signal strength, accuracy or scalloping peculiarities. Further, its ability to properly predict coverage based upon geographical considerations is constrained by the grid size and the topographical accuracy. Errors in the RACO data can be caused by the inability of the test aircraft to follow the 5 micro-volt contour. This is particularly true when highly irregular contours are encountered and is most severe when the signal is lost due to obstructions. The point of "regained signal" is not necessarily on the 5 micro-volt contour and the measured radial distances to the station could be somewhat erroneous until the aircraft regains its desired 5 micro-volt course.

Since both of these data reflect contour irregularities caused by terrain obstructions, but only the RACO data is influenced by signal strength, the RACO data would be expected to provide the more conservative (and accurate) results. Two examples of coverage patterns consistent with this expectation are presented in Figures 4.16 and 4.17. Counter examples, however, are also available. These plots are based upon a data point every 10 degrees. This granularity was utilized for the VORTAC study for reasons of computer efficiency, even though the ECAC data is available in one degree increments. The VORTACs illustrated in Figures 4.16 and 4.17 have areas where the RACO coverage radii were apparently foreshortened by signal strength rather than terrain (south and west of ABQ., for example). In other directions, the ECAC and RACO data are reasonably consistent. In theory, the RACO contour should be contained within the ECAC contour. Inaccuracies in either of the data bases, however, can result in longer RACO radii, examples of which can be found in both of the previously mentioned figures.

There were 194 high altitude VORTAC stations for which both RACO and ECAC data were available. The data for these stations were compared to ascertain both their degree of consistency and to fully establish that the RACO data is the more conservative. The results are summarized in Table 4.4.

For reasons of frequency protection, coverage distances in both data sets were truncated at 130 nm. The larger differences between the ECAC and RACO data anticipated to occur at distances greater than 130 nm from the stations were not considered and, therefore, did not influence the comparison results. While most of the results indicated that the RACO data provided, as expected, smaller more conservative coverage patterns, it can be seen that the ECAC data did in fact produce a smaller coverage area for virtually one quarter of the stations considered. The resulting average discrepancy of the overall radii was only 3.5 nm, a reasonably small figure.

This study focused on the coverage patterns at FL180, the lower limit of the high altitude airspace. Both sources of data, RACO and ECAC, provided data directly usable at this altitude. For certain portions of the study, the coverage patterns at higher elevations were obtained via linear extrapolation. This was a slightly conservative approach in that the tendency of the radio waves to bend toward the earth as they ascend in altitude (thereby providing greater coverage) was not taken into account.



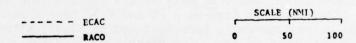


Figure 4.16 RACO and ECAC Coverage Patterns for the VORTAC (SLC), Salt Lake City

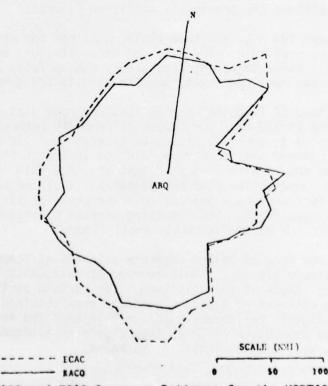


Figure 4.17 RACO and ECAC Coverage Patterns for the VCRTAC (ABQ), Albuquerque

Table 4.4 ECAC/RACO COMPARISON SUMMARY

(based on data from 194 high altitude VORTACs)

	RACO	ECAC
Average Coverage radius (nm)	121.2	121.4
Average Coverage area (sq.mn)	40,059	47,225
Average discrepancy between	0.5	
individual radii Percent of RACO radii which	= 3.5 nm	
were shorter than ECAC	= 87.7	
Number of Stations with smaller average RACO radius Number of stations with	= 146 (75.3%)	
smaller RACO coverage area	= 148 (76.3%)	

	Comparison of the with equivalent	coverage	Stations
		constant 130	irregu
	Constant 130	107	18
RACO	Irregular	6	63

At the time that the VORTAC study was initiated, with the consideration of the current high altitude VORTAC Stations, the ECAC data was only partially available. As a result of the aforementioned comparions and data availability factor, this study used the more conservative RACO data with the ECAC being used, subject to its availability, in a supplementary manner when RACO data was not available.

4.2.1.2 Coverage Gaps

Having established the procedures whereby the coverage pattern of a given station can be determined, the next task involved the identification of the coverage patterns produced by an aggregated set of individual VORTACs. From this information the areas where no coverage occurred, i.e., coverage gaps, could be defined. Two distinguishable sets of VORTACs were used in this study. Specifically, high altitude VORTACs used for navigation between 18,000 and 45,000 feet and low altitude VORTACs providing navigation signals for aircraft below 18,000 feet. Selected Canadian VORTACs which can be used to provide coverage within the CONUS are also included.

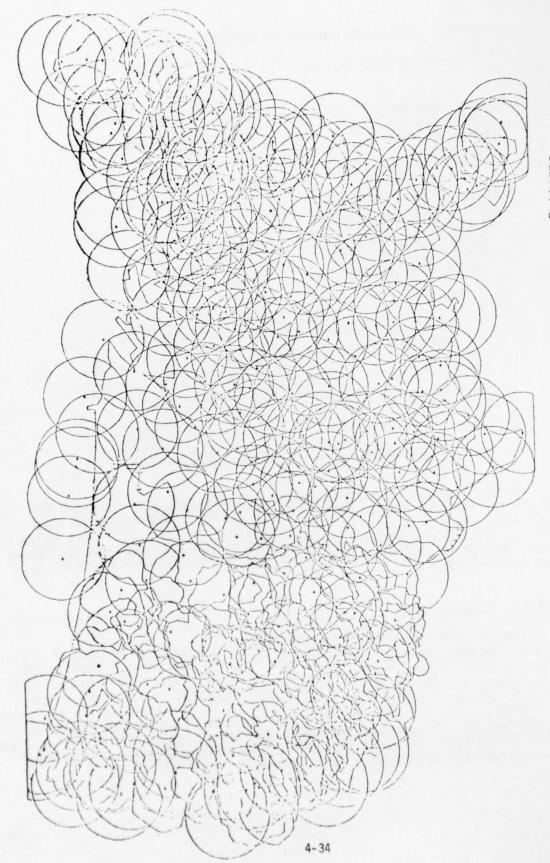
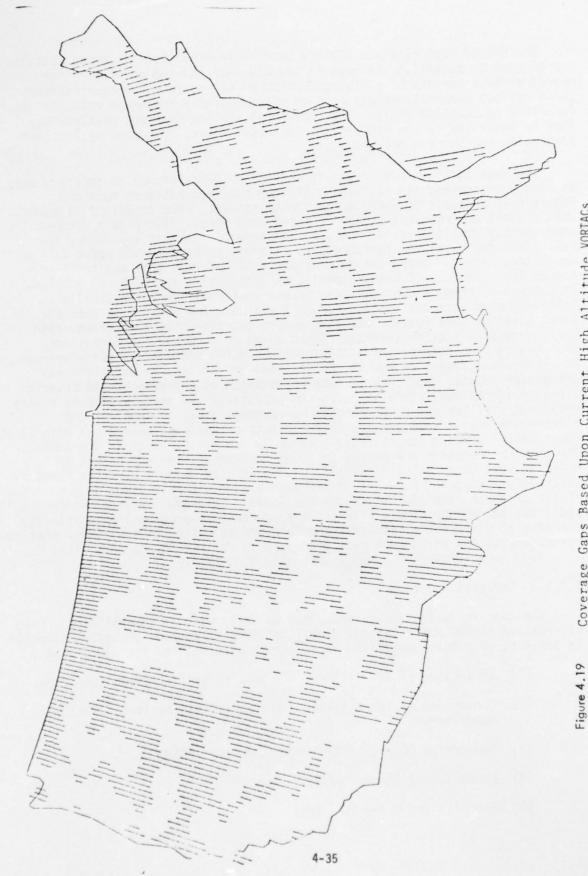


Figure 4.18 High Altitude NAVAID Coverage Patterns 130 nm Radius



Coverage Gaps Based Upon Current High Altitude VORTACs With 50 nm Maximum Coverage Radius (lines represent coverage gaps)

The set of high altitude VORTACs with their associated coverage patterns can be used to establish the gaps in the CONUS or charted coverage which result under a variety of postulated error budget-route width combinations. These various combinations may each be expressed as a "coverage radius", as described by Equation (4.1). The individual station coverage patterns may then be described by the appropriate coverage radius, truncated by frequency protection and terrain effects.

The most straightforward way to determine the existence of coverage gaps is simply to overlay the individual station coverage patterns on a map. Figure 4.18 presents an example for 130 nm radius. This technique is useful when a small geographical region is being considered. It rapidly becomes unreasonably complicated, however, as the area of interest approaches the full CONUS due mainly to the irregular coverage patterns which occur in the mountainous areas. Since the determination of coverage gaps was to be performed for a variety of cases (radii), an alternative approach was adopted. In a previous study a computer program was developed for automatically selecting waypoints and VORTACs for RNAV routes. The program accepts as input a set of RNAV routes, the station and terrain-oriented coverage data and the maximum station coverage radius based on accuracy considerations. The program then selects the NAVAID stations which provide the best coverage of the routes (fewest changeover points) and identifies the waypoints and changeover points. Portions of each route which do not have VORTAC coverage are also identified. This last program capability was used to produce a very efficient tool for the determination of coverage gaps. A total of 168 north-south routes were drawn with end points on the U.S. border and with 20 minutes longitudinal spacing (approximately 13 nm in the north, 18 nm in the south). The program processes the routes in 10 nm increments, the end result of which is that every point associated with a 10 x 15 nm grid over the CONUS is discretely analyzed. Figure 4.18 presents a sample result based on a coverage radius of 50 nm (the minimum radius examined). Coverage plots for different radii can be made by simply altering the appropriate program input. Finally, the gaps at different altitudes are determined by using the individual station coverage patterns which have been processed to reflect the desired altitude.

4.2.1.3 Corrective Options

The gaps in navigation coverage can be dealt with in several ways. Five alternative NAVAID modifications were identified to provide increased coverage in order to eliminate navigation gaps. These alternatives are as follows:

- (1) Conversion of low altitude VOR/DMEs to high altitude operation.
- (2) Conversion of low altitude VOR/DMEs to high altitude DVOR/DME operation.
- (3) Conversion of high altitude VOR/DMEs to DVOR/DMEs.
- (4) Establishing a new VOR/DME.
- (5) Establishing a new DVOR/DME.

Specific cases (1977 ρ - θ and ρ - ρ) were evaluated at three altitudes, 18,000, 24,000 and 30,000 feet in order to establish the VORTAC requirements and cost sensitivities to minimum altitude restrictions.

The primary emphasis, however, was focused on achieving the desired coverage by converting, upgrading and/or adding facilities to the baseline VORTAC system. An iterative process was used to estimate the most cost-effective set of modifications for a given set of route width-error budget conditions (reflected in the values of initial VOR and improved DVOR coverage radii).

4.2.1.4 Unit Costs

The estimated implementation cost to achieve the desired coverage performance was derived in this analysis by aggregating the projected unit costs associated with each individual VORTAC change or addition. These costs [44] reflect the expected average implementation cost of the designated conversion, addition or upgrading within the CONUS. This approach was selected in preference to the development of site peculiar costs for each installation, an effort which was considered to be beyond the scope of this study. The costs associated with reducing individual station bearing error, $\theta_{\rm G}$, as recommended by the Task Force for the 1977 period from 1.9° to 1.5° or the 1982 period from 1.5° to 1.0° are believed to be achievable through the use of DVOR if the initial value of 1.5° can be attained by the use of VOR equipment. The unit costs used in this study are presented in Table 4.5, for both the ρ - θ and ρ - ρ elements of this analysis, and discussed in Sections 4.2.2 and 4.2.3, respectively.

4.2.1.5 Frequency Protection

As was previously addressed in Section 4.2.1.1, frequency protection is one of the primary problems associated with the addition of new VOR and/or DME stations. The problem was of such magnitude that the FAA at one time proposed an eventual reduction in the separation for VOR frequencies from 100 to 50 kHz, thereby providing the potential to double the number of available frequencies. The potential for DME communication capabilities were likewise doubled by providing an additional interrogation rate (Y) for each existing channel. Second, frequency protection includes protection not only from those of adjacent frequencies. Thus, while use of the new frequencies will ease the burden of finding a vacant frequency for a given station, both adjacent frequencies will be of the original set and adjacent channel protection, although improved, will remain a problem.

Co-channel frequency protection for a station is based upon two factors; the desired radius of the frequency-protected airspace and its maximum altitude. Protection is assured when there are no stations of the same frequency within the radio horizon distance of any point within the desired frequency-protected space. The region wherein no co-channel stations are allowed is generally defined to be a circle with radius equal to the sum of the frequency protected service area radius and the radio horizon distance at the maximum protected altitude. For high altitude VORTACs, this results in a radius of 390 NMI (130 + 260).

Table 4.5 VORTAC Implementation Unit Costs, 1975 Dollars

New Facilit	ties
Dual VOR-DME Dual DVOR-DME Dual DME Dual VORTAC	\$279,900 \$372,000 \$154,000 \$274,300*
Modification of Exist	ting Facilities
Convert low altitude VOR-DME to high altitude operations	\$7,500
Add DME to VOR (dual)	\$86,400
Convert VOR to DVOR (dual)	\$160,500

*Since the cost of a new VORTAC installation is nearly the same as a VOR-DME (due to use of existing TACAN equipment) an analysis of VORTAC vs VOR-DME requirements was not attempted.

Adjacent channel frequency protection is based upon totally different considerations. Specifically, the signal from adjacent channel stations cannot exceed that of the desired station by more than 46 dB. For stations of similar power, this relative signal strength is a function of the distances from the aircraft to each of the stations. Figure 4.20 illustrates this relationship. The interference zone is the region wherein the signal strength of the undesired station is expected to exceed the desired station by the previously stated amount. The spacing (S) provides the necessary radius to insure that the interference zone does not intersect the service area. For high altitude VORTACs, the interference zone extends from the undesired station toward the desired station approximately 15 nm (with a margin for error) necessitating a 145 nm spacing. If both stations are close together (specifically radius B, 14 nm, then the signal from the undesired station can never exceed that of the desired. This is the only exception (barring special cases) to the 145 nm adjacent channel spacing requirements for a high altitude VORTAC.

It should be noted that frequency protection in general is not a reciprocal relationship. Station A can be protected from B but not vice versa. This stems from the fact that A is protected from B via consideration of A's service volume but with no consideration of the service volume of B. If the service volume of B is larger than that of A, then B may not be protected. The frequency protected service volume of high altitude stations, however, have both the largest radius and highest altitude of any of the station types (i.e., high, low, terminal, localizer). Thus, the addition of a frequency protected high altitude station will not jeopardize the

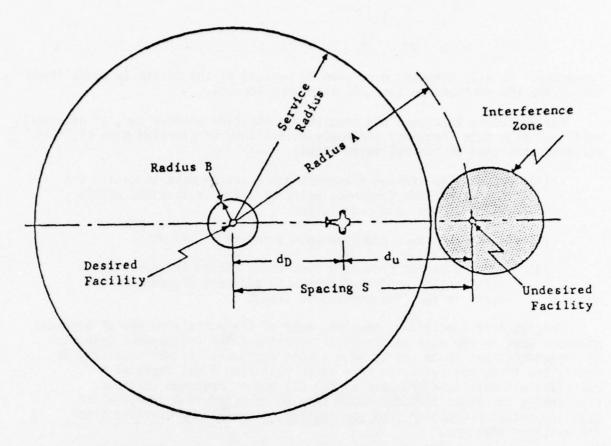


Figure 4.20 Geometry of Adjacent-Channel Interference

protection of any other station unless the other station has a service area which has been extended beyond normal high altitude limits.

This study addressed frequency protection in detail for only one case of interest, the 1977 Task Force recommended route width and navigation accuracy conditions (Table 4.3). Analysis of the frequency protection problems for other situations was beyond the scope of this study. The results are presented in Section 4.2.2.1. The methodology adopted for this effort was constrained to some extent by the available data. The first task, in evaluating frequency protection, consisted of determining the current station frequencies and locations. This necessitated the merging of the frequency data obtained from the current SAFI* listing and station location data which was derived from the FAA NAVAID tape acquired during RNAV studies. This merging effort produced two problems. First, prior processing of the NAVAID tape did not consider localizers. The effect of localizers on the adjacent channel frequency protection of VOR stations could, therefore, not be considered. This problem was minimized by avoiding frequency selection within the localizer region where possible. The second problem was that locations were not found for all stations on the SAFI listing. The vast majority of the missing stations were military, particularly Navy TACANs. Most of these stations are expected to be in coastal regions where the addition of new stations was not found to be

^{*}Semi-Automatic Flight Inspection

necessary. In all, however, more than 90 percent of the initially established set of the VOR stations on the SAFI list were located.

The procedure to assess the frequency protection problem and, if necessary, select an alternate frequency for newly established or upgraded high altitude stations consisted of the following steps:

- (1) Process the station frequency and location data to obtain the list of VOR/DME frequency pairs utilized by stations within 390 nm of the selected station.
- (2) Repeat the above premised upon a radius of 145 nm.
- (3) Select a vacant frequency from those listed in Step 1 which satisfies the condition that both adjacent frequencies are vacant in the list derived in Step 2.

For the 1977 conditions, however, many of the postulated new or upgraded stations were in the same geographical region and the independent selection of frequencies was found not to be a viable approach. It was necessary to modify the frequency selection process so that Step 3 was repeated to determine all available frequencies for all new or upgraded stations. Frequencies for these stations could then be selected in a manner which provided mutual protection from one another, as well as protection from the existing stations.

With regard to high altitude VORTACs, major frequency protection problems would be expected only in the event that a great many new stations are required. However, it is qestionable whether those route width and error budget combinations which would require the addition of many stations will ultimately be implemented. The requirement to add many new VORTACs in an RNAV environment can result only if the usable range of the existing stations are considerably reduced. Furthermore, should it ever become apparent that the addition of many stations will be necessary, long range planning and the shuffling of existing frequencies can in all likelihood provide satisfactory results. Thus, for the aforementioned reasons, specific consideration of frequency protection for other than the 1977 considerations did not appear to be necessary and was therefore not included in this analysis.

4.2.1.6 p-p Alternatives

For reasons which will be explained, DME/DME navigation, while having certain drawbacks, inherently facilitates very accurate navigation and therefore warrants consideration whenever reduced route width requirements are being addressed. DME/DME navigation, hereafter referred to as $\rho\text{-}\rho$, implies that the aircraft location information is resolved by processing "distance measuring" signals from two non-collocated DME stations. A single DME signal allows an aircraft to ascertain that it is somewhere on a circle surrounding the station with radius equal to the DME reading. Unless the aircraft location is collinear with the stations, the circles defined by

two DME signals will intersect at two points. If the stations are appropriately selected, the two intersection points will be sufficiently separated so that an adequate dead-reckoning system can readily identify the proper intersection.

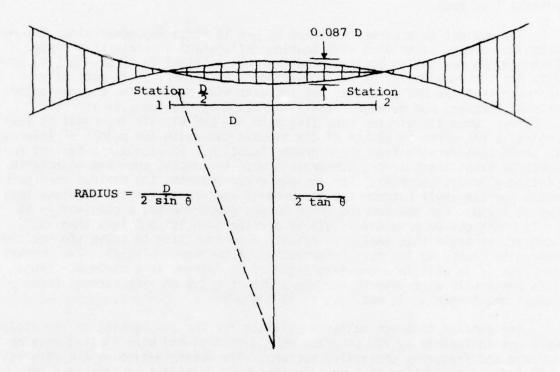
The primary advantage of $\rho\text{-}\rho$ is that the range dependent VOR signals are not required. Its disadvantages are that two ground stations are necessary for navigation rather than only one as with the $\rho\text{-}\theta$ system and specialized $\rho\text{-}\rho$ airborne navigation equipment would be required, including a method of position ambiguity solution. These factors tend to compensate for one another, but it is the intent of this study to compare, under a variety of conditions, both $\rho\text{-}\rho$ and $\rho\text{-}\theta$ systems in an RNAV environment in order to assess the relative potential of each.

The navigation accuracy afforded by ρ - θ is range dependent since the geographical errors associated with bearing measurement increase with range. Although the navigation accuracy requirements outlined in AC 90-45A [19] provide for the possibility of range dependent DME errors, it is generally accepted that the DME errors do not increase with range. The $\rho-\rho$ navigation errors, however, can be significantly affected by station pair-aircraft geometry. When the circles resulting from two DME signals intersect at small angles, a DME error in either of the signals can cause the points of intersection to shift considerably from their proper location. To maintain a desired route width or cross track error, constraints must be imposed upon the acceptable station-aircraft geometry. For various error budgets, the minimum permissible value for the angle between the measurement vectors to the two stations $(\Delta \theta)$ may be found. For the current error budget requirements, a constant +4 nm route width can be provided if $\Delta\theta$ is greater than 10° and less than 170°. However, an angle this small can result in an inability to solve the position ambiguity caused by the dual intersection of the range circles. The current practice is to utilize a crossing angle of 30 degrees as a minimum. Since this angle will also support a route width of + 2.5 nm with current error budget requirements, it was used in this study.

The station coverage patterns utilized for the $\rho\text{-}\rho$ portion of the study were not influenced by VOR accuracy considerations and were limited only by terrain and frequency protection factors. The determination of the coverage gaps for $\rho\text{-}\rho$ navigation is a considerably more difficult process than for $\rho\text{-}\theta$ navigation. The existence of dual DME coverage from stations satisfying the geometrical requirements must be insured for each geographical location (grid point) of concern. Figure 4.21 illustrates the region associated with any pair of stations where coverage is not provided as a result of the $\Delta\theta$ constraint.

The CONUS and charted RNAV coverage gaps for $\rho-\rho$ navigation were determined through the use of a previously developed computer program for $\rho-\rho$ NAVAID selection which was modified for this application. The resulting program was applied to obtain the $\rho-\rho$ results in a manner similar to that of the $\rho-\theta$ VORTAC selection program when used to identify $\rho-\theta$ coverage gaps. This process is described in Section 4.2.1.2.

COVERAGE IS UNACCEPTABLE IF THE ANGULAR SEPARATION OF THE STATIONS, RELATIVE TO THE AIRCRAFT, IS WITHIN θ DEGREES OF 0 OR 180. PLOT BELOW IS FOR θ = 10°.



- STATION SEPARATION = D CURVES GENERATED BY TWO CIRCLES OF RADIUS $\frac{2}{2 \sin \theta}$
- CIRCLE CENTERS AT A DISTANCE OF $\frac{D}{2 \tan \theta}$ FROM MIDPOINT BETWEEN STATIONS

FIGURE 4.21 P-P COVERAGE LIMITATIONS

Since station coverage distances in excess of 130 nm were not considered, ρ - ρ coverage gaps could be eliminated only by establishing additional high altitude DME stations. For reasons of cost, however, a selection priority system pertaining to the alternatives available for establishing the new stations become apparent. The most cost effective solution stems from the fact that there is no physical or operational difference between low and high altitude DME stations. The "upgrading" of a low altitude VORTAC, or even only the DME portion thereof, consists only of insuring frequency protection and performing a flight inspection. The unit cost of this activity has been estimated by the FAA [44] to be \$7500, primarily a flight inspection expense. Two additional corrective options exist, both of which require the use of new, rather than converted, DME equipment. These involve the addition of a DME to an existing VOR station and the installation of DME stations (without VOR) at new sites. The costs of these corrective options are included in Table 4.5.

The only difference between the methodology adopted for the elimination of the $\rho\text{-}\rho$ coverage gaps as compared to $\rho\text{-}\theta$ was that it was necessary to rely more extensively upon the computer program due to the terrain and $\Delta\,\theta\text{-}$ induced coverage irregularities. The first step was to establish the current high altitude $\rho\text{-}\rho$ gaps. In view of the previously cited costs, the second step was that of identifying all low altitude VORTACs whose upgrading would be expected to either fill or reduce the gaps. These were then added to the high altitude station set and the coverage program rerun. The additional DMEs required for each of the situations addressed were determined by an iterative process of manual selection followed by computer verification. This process was continued until all the gaps in the stipulated coverage areas were filled.

4.2.2 ρ-θ NAVAID System Requirements

The RNAV Task Force recommended error budgets and route widths to be attained in the 1977 and 1982 implementation periods. This section identifies the modifications to the existing $\rho - \theta$ NAVAID system that would be required in order to meet these objectives. The associated implementation costs are presented based on the unit costs summarized in Section 4.2.1.4. A means is also provided for the reader to derive implementation costs based on a new set of VORTAC characteristics and costs through use of the cost sensitivity data contained in Section 4.2.2.2.

4.2.2.1 1977 Charted RNAV/CONUS Coverage Requirements

The 1977 error budget and route width requirements, as recommended by the Task Force, are summarized in Table 4.3. The ground VOR error, $\theta_{G}=1.5^{\circ}$, was to have represented an improvement from the assumed current 2 σ value of θ_{G} , namely 1.9°. This improvement was to be achieved by converting existing VORs (either high or low altitude stations) to high altitude DVORs. Those VORTAC modifications made up two of the five corrective options assumed to be available for use by the VORTAC system planner, as defined in Section 4.2.1.3. An examination of the current set of high altitude VORTACs revealed that the

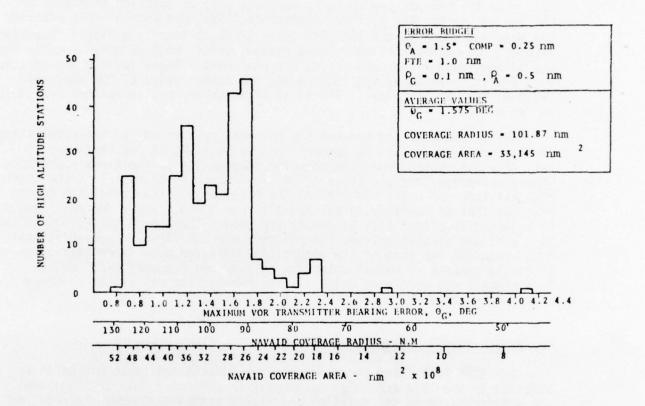


Figure 4.22 Distribution of Current High Altitude Stations As A Function of Maximum Observed Station Bearing Error θ_G (Maximum Route Width - 4 nm)

Table 4.6
SUMMARY OF NAVAID REQUIREMENTS TO SATISFY
1977 TASK FORCE CONDITIONS

REQUIREMENT	S FOR C	OVERA	GE OF N	AFEC I	ROUTE S	TRUCTUR	
FLIGHT LEVEL	18	0	24	e I	300		
NEW STATIONS	NS1 NS2	} 2	NS2	} 1	NS2	}1	
LOW ALT. CONVERTED TO HIGH ALT. OPS	CKW DYR LBF RHI RLG	5	DYR LBF RHI	3	DYR LBF RHI	}	
IMPLEMENTATION COST ('75 dollars)	597,300		302,	400	302,400		

FLIGHT LEVEL	180		24	0	300		
NEW STATIONS	NS1 NS2 NS3 NS4	6	NS2	3	NS2	2	
	NSS NS6		NS5	-)	
LOW ALTITUDE STATIONS CONVERTED	BJI BTM		ВЈІ	1	ВЈІ	1	
TO HIGH ALTITUDE OPERATIONS	CKW CTB		CKW CTB		CKW CTB DNW		
	DYR ISO ISN		DYR ISO ISN		DYR 150 15N		
	LBF	16	LBF	13	LBF	12	
	MOT MQT MSO		MOT		MOT	1	
	PHP RHI		PHP		PHP		
	SDO SDO		SDO	1		1	
VORS UPGRADED TO	MLP }	1	MLP	} 1	MLP	} 1	
IMPLEMENTATION COST ('75 doilars)	1,959,9	200	1,097	,700	810,	300	

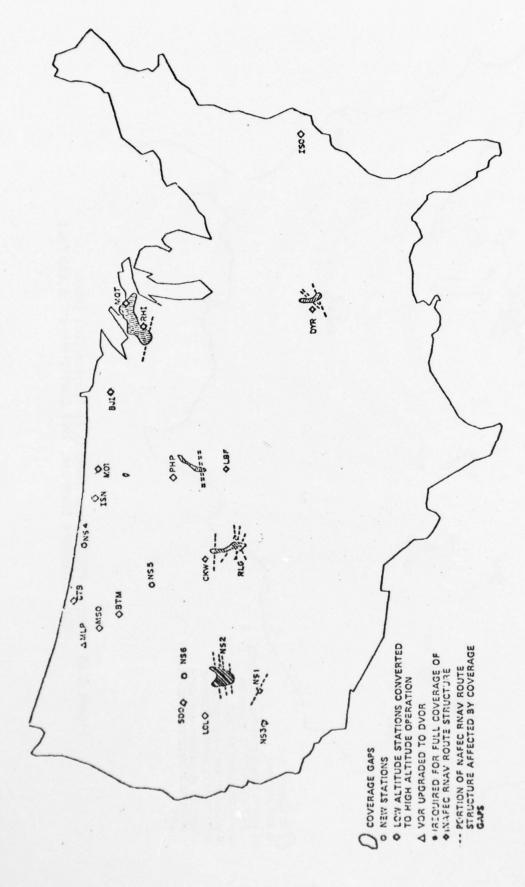
average $\underline{\text{maximum}}^*$ (not 2σ) bearing error was 1.575 degrees, only 0.075 degrees greater than the improved value stipulated by the RNAV Task Force for the 1977 case. This information was extracted from Column 8 of the SAFI VORTAC Systems Characteristic report [45], and was used in the 1977 NAFEC route structure and CONUS coverage analysis. An analysis of CONUS coverage, using 2σ bearing error, which is presented in Section 4.2.2.2, produced similar results.

The SAFI VORTAC System Characteristic data base was used to identify the maximum bearing error for each high altitude VOR station examined in this study. Distribution of this parameter and the associated individual station coverage radius and area is illustrated in Figure 4.22. These values of θG were used in combination with line-of-sight limiting terrain factors to identify individual station coverage patterns (Section 4.2.1.1) and to subsequently identify the resulting navigation coverage gaps (Section 4.2.1.2) with respect to either the NAFEC RNAV route structure or full CONUS coverage. Figures 4.23 and 4.24 identify the gaps prevalent in the current VORTAC system at 18,000 feet. Candidate low altitude VOR/DMEs were selected from the set of low altitude VORTACs based on their proximity to the gaps shown in Figures 4.23 and 4.24. The maximum bearing error value was determined for each VORTAC from the SAFI VORTAC Systems Characteristic report and used in combination with the appropriate terrain features to produce the coverage contours shown in Figure 4.25. These contours were then used to identify which of the candidate low altitude stations should be converted to high altitude operations, and an estimate was made of the most cost-effective combination of the five corrective options which would provide the desired coverage.

Table 4.6 summarizes the resulting requirements to provide either RNAV NAFEC route structure (Figure 4.26) or full CONUS coverage at altitudes of 18,000, 24,000 or 30,000 feet, with the symbols of Figures 4.23, 4.24, 4.27 4.28, 4.29 and 4.30 identifying the locations of the recommended VORTAC system modifications. Because of the relatively low values of θ_{G} , only one station, MLP, was upgraded to a DVOR. In that case its current θ_{GMAX} of 1.9° was assumed to be reduced to a value of 1.5°.

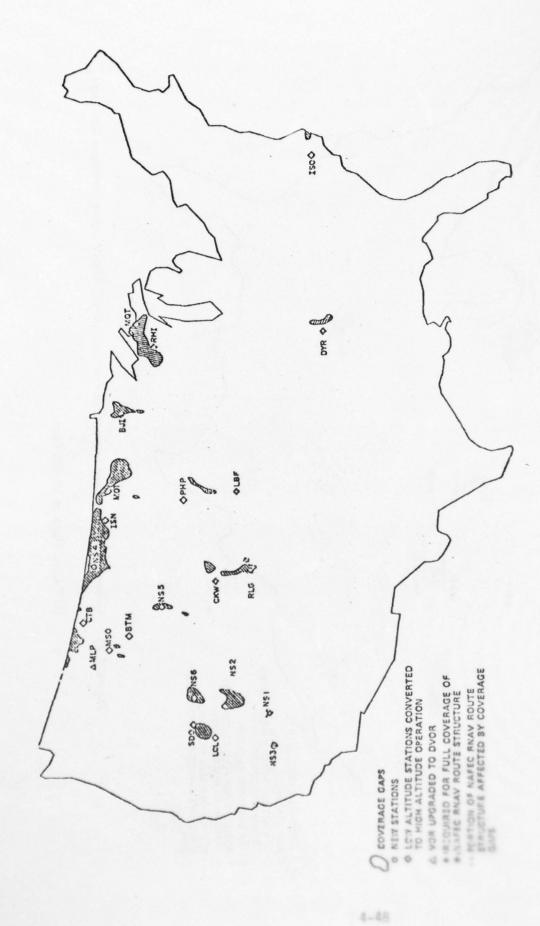
Implementation costs are \$597,300 for a charted RNAV structure and \$1,959,900 for full CONUS coverage with +4 nm route widths and the Task Force 1977 recommended error budget. Although the study did not specifically consider the removal of high altitude VORTACs whose coverage is entirely redundant, it can be seen from Figure 4.18 that considerable redundancy appears to exist. Determination of which stations could be removed will be dependent upon a detailed analysis of coverage requirements for the route structure designed for implementation. The NAFEC structure upon which this analysis was based was designed as an input to these payoff studies and, while representative of the techniques to be employed in the development of an implementation structure, is not an optimum structure which has been coordinated with both low altitude and terminal area designs. Removal of only 12 high altitude VORTACs would provide an annual maintenance cost savings equal to the one-time upgrading cost required for charted RNAV coverage (See Section 4.3).

^{*}based upon a total of 720 observations taken along each 1/2° radial



Ground VOR/DME Requirements to Provide NAFEC RNAV Route Structure Coverage, Figure 4.23

1977 Conditions at 18,000 Feet

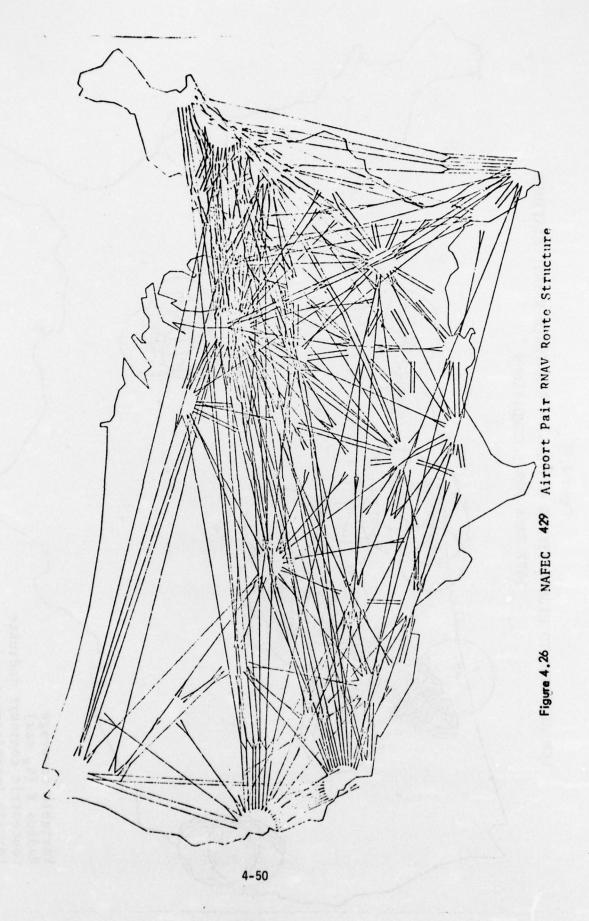


Ground VOR/DME Requirements To Provide Full CONUS Route Structure Coverage, 1977 Conditions at 18,000 Feet Figure 4.24

SYSTEMS CONTROL INC PALO ALTO CALIF
IMPLEMENTATION OF AREA NAVIGATION IN THE NATIONAL AIRSPACE SYST--ETC(U)
DEC 76 W H CLARK, E H BOLZ, H L SOLOMON DOT-FA72WA-3098
FAA-RD-76-196 AD-A039 225 UNCLASSIFIED 3 OF 5. AD 39225 T. C. all.

LOW ALTITUDE TO HIGH ALTITUDE CONVERSION CANDIDATE NAVAID STATIONS 1977 TASK FORCE CONDITIONS Variable coverage Radius f $(\theta_e \text{ max})$ Concentric contours indicate torrain interference and reflect coverages at 18,000 24,000, and 30,000 feet, respectively.

Figure 4.25



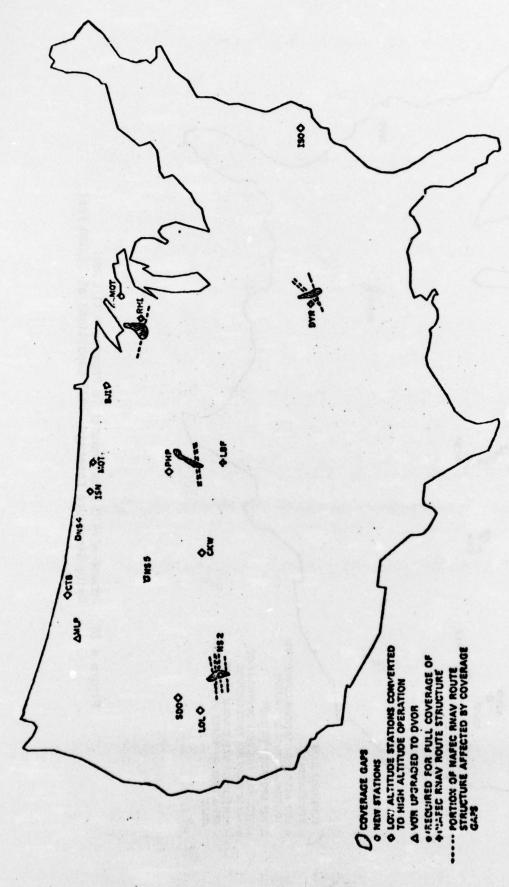
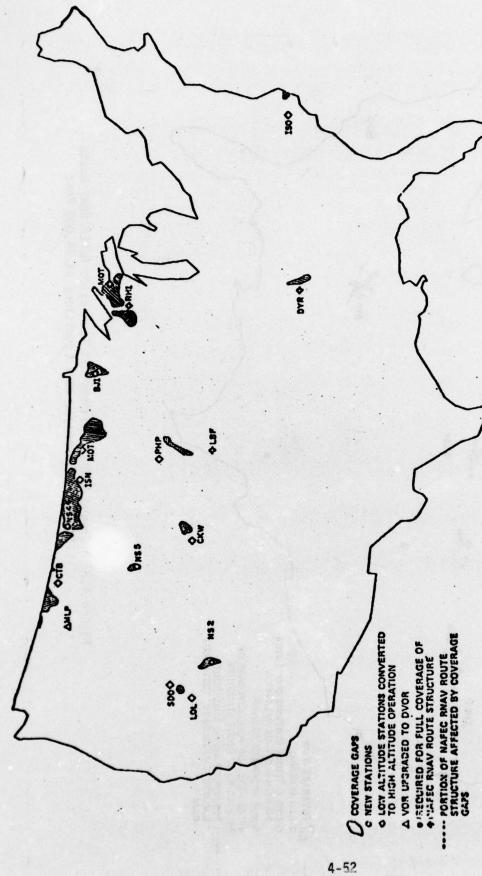
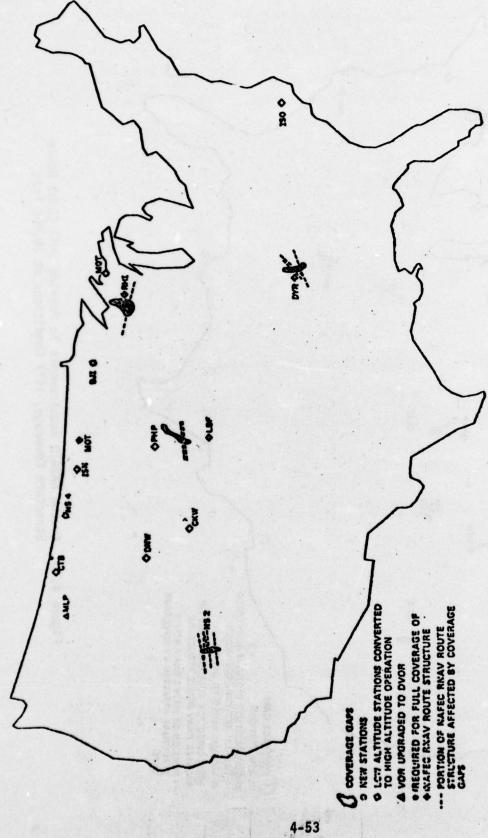


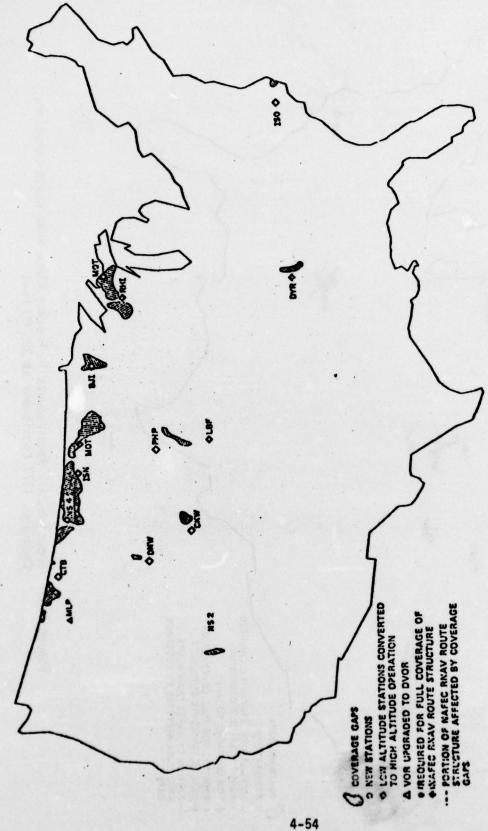
Figure 4.27 Ground NAVAID Requirements To Provide NAFEC RNAV Route Structure Coverage, 1977 Conditions at 24,000 Feet



Ground NAVAID Requirements To Provide Full CONUS Route Structure Coverage, 1977 Conditions at 24,000 Feet Figure 4.28



Ground NAVAID Requirements To Provide NAFEC RNAV Route Structure Coverage, 1977 Conditions at 30,000 Feet Figure 4.29



Ground NAVAID Requirements To Provide Full CONUS Route Structure Coverage, 1977 Conditions at 30,000 Feet Figure 4.30

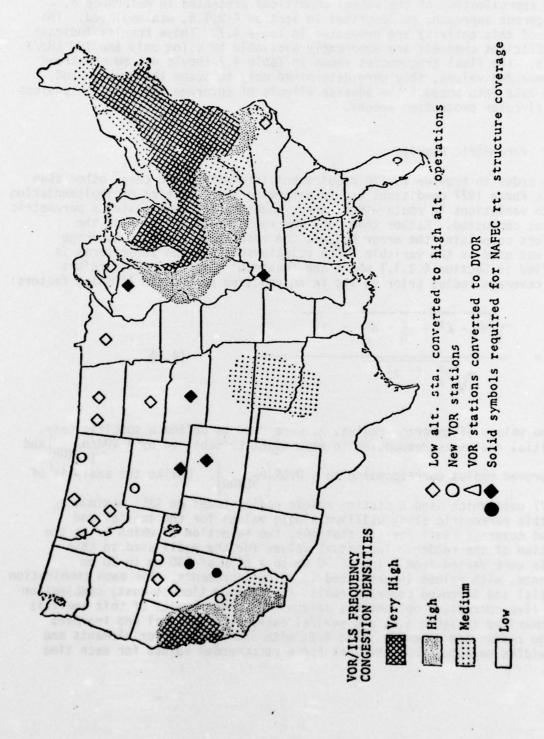


Figure 4.31 1977 VORTAC Requirement Impact on Frequency Congestion

The locations of the modified stations would appear to minimize the problems that might be associated with frequency protection. When these locations are superimposed on a frequency congestion map (Figure 4.31), approximating the regions of frequency congestion, all relevant stations appeared to be located in the low density regions. Since this map was only an approximation of the actual conditions presented in Reference 4, a more rigorous approach, as described in Section 4.2.1.5, was employed. The results of this activity are presented in Table 4.7. These results indicate that sufficient channels are apparently available by using only the 100 kHz/X channels. The final frequencies shown in Table 4.7 should not be construed as recommended values, they were determined only to scope the problem and did not take into account the adverse effects of coverage keyholes (extensions out of circular protection areas).

4.2.2.2 Parametric Results

In order to provide VORTAC requirement information for cases other than the Task Force 1977 conditions and to determine sensitivities of implementation costs to variations in route width, error budgets, and unit costs, a parametric study was conducted. Rather than vary the route width and each of the parameters comprising the error budget, the resulting "maximum" coverage radius was used as the variable. The relationship between these terms is identified in Section 4.2.1.1 where the "maximum" coverage radius refers to the coverage radius prior to its being adjusted by terrain limiting factors:

$$R = \frac{\sqrt{W^2 - C^2 - \rho_G^2 - \rho_A^2 - F^2}}{\sqrt{\frac{\theta_A^2 + \theta_G^2}{A}}}$$
(4.1)

Two values of coverage radius, R, were used to define a specific case, the initial radius corresponding to what might be achieved by a VOR θ_{GVOR} and the improved radius corresponding to a DVOR θ_{GDVOR} . Unlike the analysis of

the 1977 case which used a station unique radius based on SAFI maximum θ_A data, this parametric study utilized single values for the initial and improved coverage radii for all stations, the selection of which is at the discretion of the reader. The actual values for the radii used in this analysis were varied from a low of 50 nm to a high of 130 nm in 20 nm increments, with values interpolated in 10 nm increments. For each combination of initial and improved coverage radii, the optimum (lowest cost) combination of the five corrective options was determined. The results of this analysis are summarized in Table 4.8, and several examples of initial and improved coverage radius are given in Table 4.9, with individual error elements and route widths selected from the Task Force recommended values for each time period.

The sensitivity of the implementation costs identified in Table 4.8 with respect to initial and improved radius is illustrated in the plot of Figure 4.32. The range of coverage radii examined produced a three order of magnitude variation of the implementation costs required to achieve full coverage at 18,000 feet.

Additional implementation cost sensitivities with respect to the parameters of route width and flight technical error, for discrete sets of error budgets, were derived by computing the corresponding initial and improved coverage radius, locating the appropriate intersection on Figure 4.32 and reading the corresponding implementation cost. These sensitivity results are presented in Appendix F.

In the event it is desired to modify one or more of the unit costs presented in Section 4.2.1.4, it is necessary to have knowledge of the number of each type of corrective option that would be required, for specified combinations of initial and improved radii, in order to recompute total implementation costs. Figures presenting this information, based on the data presented in Table 4.8, for each of the five corrective options employed in this study are also included in Appendix F.

The cost to provide CONUS coverage for 1977 conditions may be computed from the parametric relationship using the initial and improved radii given in Example C of Table 4.9. In this example it is assumed that the ground DME and airborne equipment error budget has been achieved, and that the improved radius is due to an improvement in ground VOR accuracy from 1.9° to 1.5°. The coverage cost, which is plotted on Figure 4.32, is approximately \$1.9 million which is consistent with the cost of \$1,959,900 derived in Section 4.2.2.1.

4.2.2.3 1982 CONUS Coverage Requirements

The parametric study was used to estimate the requirements and costs to implement full CONUS coverage at 18,000 feet under the conditions stipulated by the RNAV Task Force for the 1982 period (charted RNAV in 1982 was not recommended by the Task Force). The initial and improved radii corresponding to those conditions are 71.9 and 91.6 nm, respectively (Column H in Table 4.9). The results subsequently presented in this section are based on an initial and final radius of 70 and 90 nm, respectively, and are therefore slightly conservative. The final modifications are summarized in Table 4.10. A total of \$6,604,000 would be required to implement the 1982 system directly from the current system.

4.2.3 ρ-ρ Coverage Requirements

Coverage requirements for ρ - ρ (DME-DME) navigation were analyzed for both the 1977 and 1982 route widths and error budgets. A separate analysis for each was not required since ρ - ρ accuracies are such that the error budget which will support ± 4 nm route widths will also support ± 2.5 nm route widths (see Section 4.2.1.6).

Sets of DME/DME (ρ - ρ) VORTAC System requirements and associated implementation costs were identified so as to provide coverage for the NAFEC RNAV route structure and/or full CONUS coverage, at each of three altitudes, FL180, FL240 and FL300. The first step in this process was to determine the coverage gaps associated with each of these cases based upon the use of the current set of high altitude VORTACs. These results are shown in Figures 4.33, 4.34 and 4.35. It can be seen that virtually full ρ - ρ coverage is already available throughout the eastern and central portions of the country. The mountainous regions, however, present a major coverage problem. As a result of the dual coverage requirements, terrain influences, and the $\Delta\theta$ limitation, many coverage gaps exist. These are reduced markedly, however, at higher altitudes, with the most significant improvement occurring between FL180 and FL240.

The remainder of the $\rho\text{-}\rho$ study involved the iterative procedure of manually selecting new or upgraded stations which appeared to offer the potential of eliminating the coverage gaps and then subsequently testing these selections by executing the $\rho\text{-}\rho$ coverage program. The procedure required up to four iterations because the program frequently revealed that gaps which were thought to have been covered still remained due to terrain or Δ θ influences. Previously added stations were then removed, new ones added, and the process repeated.

At the conclusion of this process, DME requirements and the associated costs for the six situations addressed were identified and are summarized in Table 4.11.

For each situation, the emphasis was placed upon the upgrading of existing VORTAC stations. In fact, virtually all low altitude VORTACs within range of the coverage gaps were utilized. Their aggregate value in the reduction of the gaps was highly significant. There was only a minor difference between the station requirements for coverage at FL240 and FL300. The additional gaps at FL240 as compared to FL300 can be covered for the most part by the upgrading of VORTACs. Filling the gaps at FL240, however, virtually exhausted the capability to upgrade low altitude VORTACs and to add DMEs at VOR sites. Thus, a significant number of new stations were required to provide coverage at FL180. As a result, the associated costs are increased by 150 to 200 percent.

TABLE 4.7
FREQUENCY PROTECTION RESULTS 1977 CONDITIONS

Station ID		Type of Frequency Protection Violation	Example Fina Freque VOR	1	
BTM CKW CTB DYR ISN ISO LBF LOL MLP MOT MQT MSO PHP RHI RLG SDO NS1 NS2 NS3 NS4 NS5 NS6	111.6 112.2 114.4 116.8 116.3 109.6 117.4 116.5 117.8 117.1 109.0 112.8 108.4 109.2 113.8 114.3	53 59 91 115 110 33 121 112 125 118 27 75 21 29 85 90	1 None 1 1 1 1 1 2 1 2 1 1 1	117.5 115.0 114.4 117.2 113.3 108.0 115.7 117.0 115.9 117.7 114.1 112.4 117.9 113.6 115.4 112.7	122 97 91 119 80 17 104 117 106 124 88 71 126 83 101 74 49 86 85 94 114 78

- 1: Frequency change necessary in order to upgrade station.
- 2: Frequency change necessary to provide vacant frequencies for the upgrading and/or addition of other stations.

TABLE 4.8

SUMMARY OF PARAMETRIC RESULTS
GROUND NAVAID REQUIREMENTS TO PROVIDE FULL CONUS
COVERAGE AT 18,000 FT.

		No.of Low A Converted t Ops W	o High Alt. ith	No.of High Alt.Sta.'s Upgraded To	No.of New Added With	Implementatio Costs \$(000)	
		STD.VOR	DVOR	DVOR	STD.VOR	DVOR	
Unit	Costs -\$	7,500	168,000	160,500	279,900	372,000	
Initial Radius N. Mi.	Improved Radius N. Mi.			(d.1) (d.1) (d.2)			
130 110 110	N.A. 130 N.A.	6 10 10	0	1	0	0	45 515 635
90	130 26		2	4	2 2	0	1,733
90	110	27	3	2	2	1	1,960(1)
90 70 70	N.A. 130 110	24 64 75	5 6	15 13	9 1 4	1 2	2,699 4,380 5,521
70	90	75	7	10	5	5	6,604(2)
70 50	N.A. 130	77 84	18	37	32 0	1	9,534 9,965
50	110	103	30	34	0	2	12,014
50 50 50	90 70 N.A.	132 173 203	40 21	48 53	0 19 188	9 37	18,762 32,414 54,144

^{(1) 1977} full CONUS Coverage.

^{(2) 1982} full CONUS Coverage.

Table 4.9 Coverage Radii

0.25 0.25 0.1 1.0 1.0 Task Force Recommended Route Widths and Error Budgets 0.25 0.1 0.5 1.0 103.5 0.5 or 3% + 4 + splay 0.1 5.0 3.0 Airborne VOR Error - 8 A (deg) Ground VOR Errar - B G (deg) Flight Technical Error - F(nm) Airborne DME Error - $\rho_G(nm)$ Ground DME Error - p_G (nm) Computer Error - C (nm) Route Width - W (nm) Coverage Radius - R

A B C (1) D E F C Coverage Radii F C C C C C C C C C		_	-	T	_			-			
A B C (1) D E F C	*		Improved	2.5	.25	٦.	.25	1.0	1.0	1.0	91.6
A B C (1) D E F G		I	_	2.5	.25	7.	.25	1.0	0.1	1.5	71.9
A B C (1) D E F F F F F F F F F F F F F F F F F F			Improved	2.5	.25	-	.25	1.0	1.0	1.0	91.6
A B C (1) D E F F F C (1) D E F F F C (1) D E F F F F C (1) D E F F F F F F F F F F F F F F F F F F	:	0	Initial	2.5	.25	7.	.25	1.0	0.1	1.9	4.09
A B C (1) D E Initial Improved Initial Initial Improved Initial Improved Initial Improved Initial Improved Initial Initial Improved Initial Initial Initial Improved Initial Improved Initial Initial			mproved	4.0		-:		1.0	1.0	1.0	130.0*
A B C (1) D E E C (1)		4	Initial	4.0	.25	٦.	.25	1.0	1.0	1.5	122.5
A B C (1) Initial Improved Initial Improved 1.5	=		Improved	4.0		-		1.0	1.0	1.0	130.0*
A B C (1) Initial Improved Initial Improved 4.0 4.0 4.0 4.0 4.0 4.0	oge Rad	E		4.0	.25	-	.25	1.0	1.0	1.9	102.9
A B C (1) Initial Improved Initial Improved 4.0 4.0 4.0 4.0 4.0 4.0	of Cover		Improved	4.0	.25	٦.	5.	0.1	1.5	1.0	121.8
A B C (1) Initial Improved Initial Improved 4.0 4.0 4.0 4.0 4.0 4.0	xample	3	Initial	4.0	.25	-	3.	1.0	1.5	1.9	7.06
A B A A A A A A A A A A A A A A A A A A		(1)	Improved	4.0		-	3.	1.0	1.5	1.5	103.5
A B B 4.0 4.0 4.0 4.0 4.0 4.0 4.0 4.0 4.0 4.0		0		4.0	.25	-	5.	1.0	1.5	1.9	7.06
A		-	Improved	4.0	.5	-	5.	2.0	3.0	1.0	61.4
A 4.0 4.0 7.5 7.0 3.0 1.9		æ	Initial	4.0	5.	-	5.	2.0	3.0	1.9	54.7
- 6		4	Improved	4.0	5.	-	5.	2.0	3.0	1.5	57.9
> 0 0 0 0 1 0 0 0 0 0 0 0 0 0 0 0 0 0 0		1	Initial	4.0	3.	٦.	5.	2.0	3.0	1.9	54.7
				*	U	PG	ď	u	⊕∢	BO	œ

* frequency protection limit

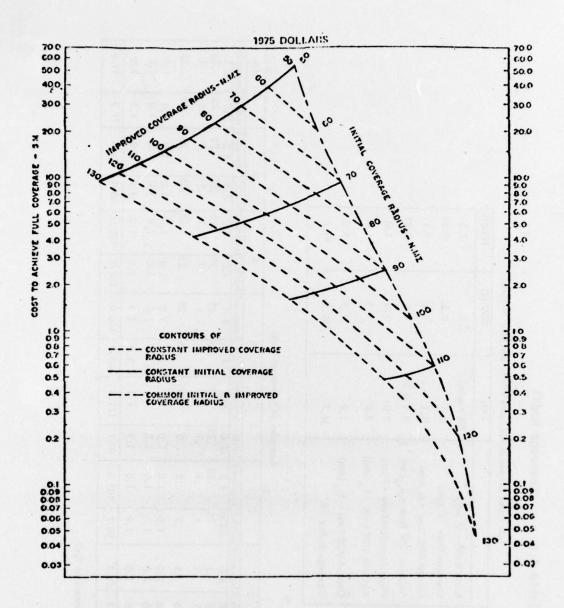


Figure 4.32 Non-Recurring Ground Station System Costs
Required to Provide Full CONUS Coverage
at 18,000 Feet

TABLE 4.10

VORTAC REQUIREMENTS TO PROVIDE

SUMMARY OF VORTAC REQUIREMENTS TO PROVIDE FULL CONUS COVERAGE AT FL 180 IN ACCORDANCE WITH 1982 TASK FORCE CONDITIONS

NEW V	OR STAT	IONS	(5)			NEW DV	OR ST	ATION	S (5)	
	001						002			
	003						004			
	007						005			
	009						006			
	010						800			
HIGH	ALTITUD	E VO	R STATIO	ONS UP	GRADEI	TO DVO	R (10)		
	DPR	GEG	LRD	MYL	RNO					
	FST	IMB	MLT	PHX	TBC					
TOW A	TOTOTOE	WOR	CTATION	IC LIDO	DADED	MO UTCH	AT MT	mine.	OPERATIONS	175
LOW A	LITTODE	VOR	STATIO	NS UPG	KADED	TO HIGH	ALITI	TODE	OPERATIONS	(/5
	ACH	COT	DMN	FSP	LBB	MMM	RLG	TBE		
	APN	CPR	DVL	HLV	LEB	MON	RSG	TKO		
	ATY	CTB	EAU	HMV	LFT	MOT	SDO	TWF		
	AXN	CTY	ECB	HON	LMT	MOT	SEG	TXC		
	BIS	CVG	EKN	HRS	LOZ	MSL	SFL	UBS		
	BJI	DDC	EMP	ISN	LVM	MTA	SHR			
	BKE	DGW	FJS	IWD	MAF	PGO	SLR			
	ĊHD	DHN	FLP	JKS	MAP	PIR	SUX			
	CNG	DHT	FMY	JMS	MLC	ROM	SWL			
	cos	DLL	FOT	LAR	MLU	RHI	TAS			
LOW A	LTITUDE	VOR	STATION	NS UPG	RADED	TO HIGH	ALTI	TUDE		
DVOR	OPERATI	ONS	(7)							
	DNW.	нів	MSO	ТРН						
	FBR	ISO	PTV							

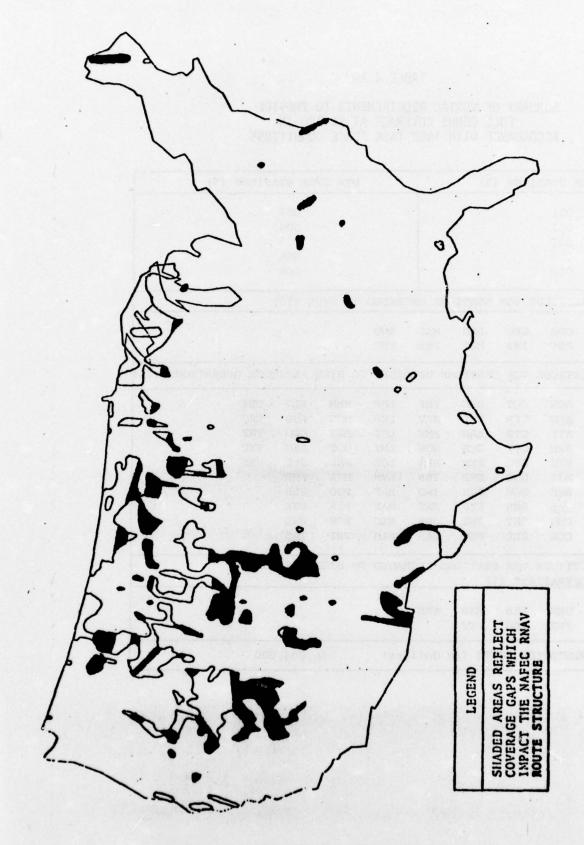


Figure 4.33 p-p Coverage Gaps at FL180 Considering Current High Altitude VORTACs Only ($\Delta\theta$ = 30°)



Figure 4.34 ρ - ρ Coverage Gaps at FL240 Considering Current High Altitude VORTACs Only (Δ θ = 30°)



p-p Coverage Gaps at FL300 Considering Current High Altitude VORTACs Only (A8 = 30°)

Figure 4.35

TABLE 4.11 P-P NAVAID SYSTEM REQUIREMENTS

	CONUS						NAFEC RNAV Route Structure						
Coverage Gap	FLI	80	FL240		FL300		FL180		FL240		FL300		
Options	No.	Total Cost (\$000)	No.	Total Cost (\$000)	No.	Total Cost (\$000)	No.	Total Cost (\$000)	No.	Total Cost (\$000)	No.	Tota] Cost (\$000)	
Convert low altitude VOR-DMEs or VORTACs to high altitude; oper.	51	383	50	375	42	315	41	308	36	270	28	210	
Add DMEs to low altitude VOR sites	11	950	9	778	9	778	9	778	9	778	7	605	
Add new DME sites	29	4466	8	1232	8	1232	27	4158	6	924	6	924	
TOTAL		5799		2385		2325		5244		1972		1739	

of 3 mm in the test of the contract of the con

4.3 TERMINAL AREA VORTAC REQUIREMENTS

This section presents the results of a detailed analysis of the effect of Area Navigation, implemented as envisioned by the RNAV Task Force [1], on the requirement for VORTAC stations and their improvements according to the needs of terminal area operations. The basic approach taken in this study has been to investigate four individual terminal areas (Denver, Philadelphia, New York, Chicago) with the primary intent being to eliminate as many existing VORTAC or VOR/DME stations as possible, while providing all services presently provided and meeting the accuracy and coverage requirements outlined by the Task Force for the 1977 and 1982 time periods. The considerations which were observed included high altitude coverage, low altitude coverage, terminal route coverage, route widths and accuracies according to the Task Force, uncompensated slant range error, DME saturation potential, VORTAC requirements for a continued VOR-oriented system, and existing VOR approach procedures. The resulting savings in VORTAC installations are evaluated in terms of savings in maintenance costs and in terms of potential savings in providing low altitude enroute coverage in other areas through station relocation. Other cost savings areas, such as the real estate which could be disposed of, is recognized. The results so determined are extrapolated over all of the high and medium density terminal areas.

4.3.1 Methodology

Each terminal area was studied individually, although because of their close proximity the Philadelphia and New York cases did interact. At each terminal a chart was drawn which shows the major airports and the forty-five mile terminal area boundary (forty-seven at New York). All airports for which instrument approach procedures exist within the terminal area were identified on the chart. All VORTAC and VOR/DME stations (low and high) within the area and within forty miles (frequency protection limit) of the boundary of the terminal area were also identified and on the chart. These charts formed the basis for the VORTAC requirements analysis, and are presented in Figures 4.36 through 4.38.

The first consideration evaluated was the requirement for high and low altitude area coverage according to Task Force requirements [1] within the terminal boundaries. Since this is not a high altitude area coverage study (see Section 4.2), consideration of the high altitude problem was limited to prohibiting removal of any high altitude VORTAC stations. Low altitude area coverage recommended by the Task Force is to support ±4 mile route widths in pre-1977 and 1977, and ± 2.5 mile route widths in 1982. In all three cases the service area of VORTAC in terms of its required accuracy [1] and the stated route width exceeds the forty-mile frequency protection limit for low altitude VORTAC stations. Therefore, forty miles was used as the range limit in this analysis. As a matter of fact, the accuracy service area using existing, unimproved VORTAC stations, but with the Task Force specified route widths and accuracies for other system error elements, would be 91.5 nm in 1977 and 60.8 nm in 1982, which are both greater than the forty mile frequency protection limit which in most cases takes precedence. For purposes of this study, the provision of low altitude RNAV coverage was limited to area where VOR route coverage exists in those cases where terrain problems were significant. The only area so affected in the four terminals studied was the mountainous region

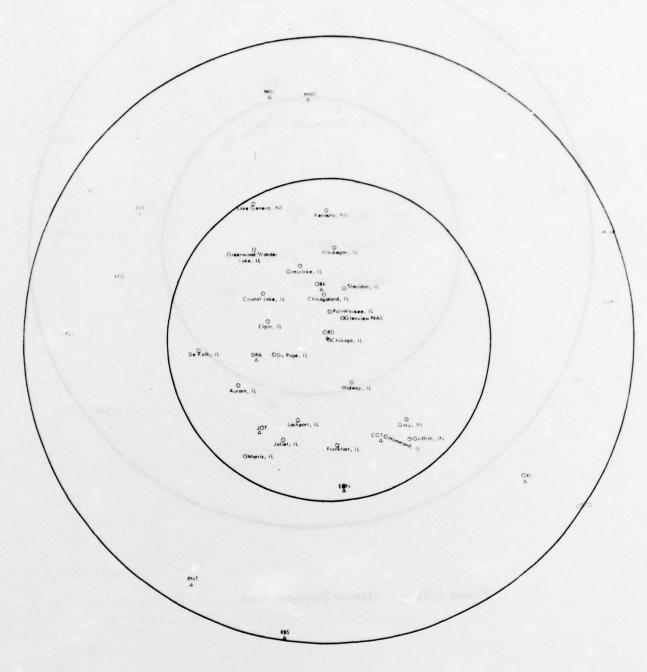


Figure 4.36 Chicago Terminal Area

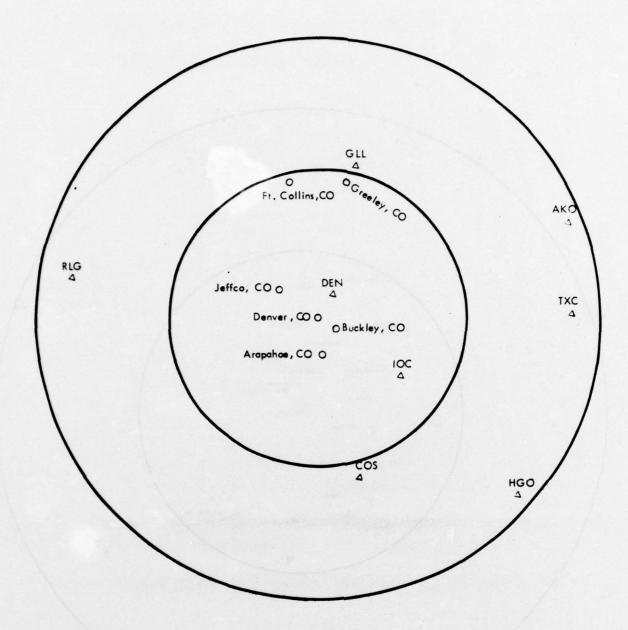


Figure 4.37 Denver Terminal Area

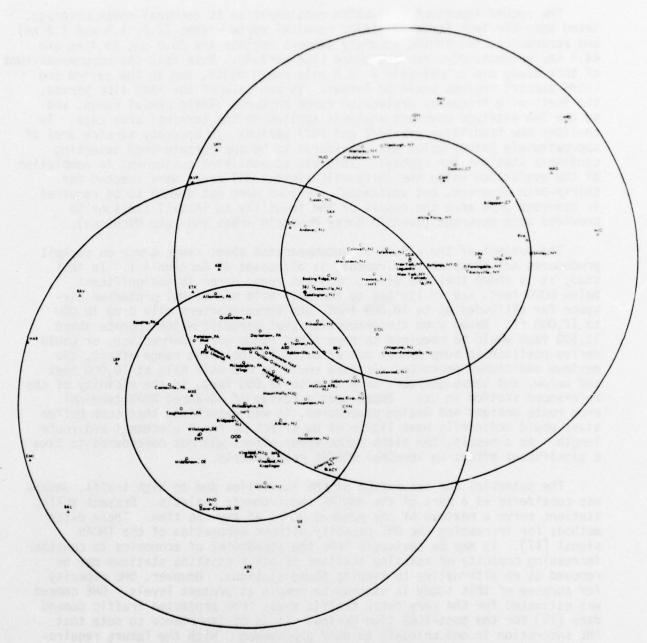


Figure 4.38 Philadelphia and New York
Terminal Areas

west of Denver. Stations within the terminal area boundary and outside of it were used for providing coverage, but only stations within the boundary were considered to be candidates for removal. Also any removal candidates were checked to insure that such removal would not affect enroute coverage outside of the terminal area.

The second important evaluation consideration is terminal route coverage. Based upon the Task Force specified terminal route widths (2.0, 1.5 and 1.5 nm) and accuracies, the VORTAC accuracy support regions are 26.8 nm, 29.4 nm and 44.1 nm, respectively, for the three time periods. Note that the recommendations of this study are to maintain a $\pm 2.0 \text{ mile}$ route width, and so the second and third support regions would be larger. In the case of the 1982 time period, the forty mile frequency protection range criteria limits useful range, and so the low altitude coverage analysis applies to the terminal area case. To consider the transition pre-1977 and 1977 periods, an accuracy service area of approximately thirty miles was considered to be appropriate when selecting candidate stations for removal. This was accomplished subsequent to completion of the evaluation using the forty-mile limit. All areas were checked for thirty-mile coverage, but additional stations were not deemed to be required if coverage gaps were the result of the inability to install stations to provided such coverage (certain Rocky Mountain areas and Lake Michigan).

The subject of the effect of uncompensated slant range error on cockpit procedures and airspace requirements is discussed in Section 4.1. In that study it is shown that the effect of slant range error is insignificant below 8000 feet, and is limited to one-half mile additional protected airspace for altitudes up to 10,000 feet, and three-quarters mile from 10,000 to 12,000 ft. Based upon the assumption that aircraft which operate above 12,500 feet would be required to have slant range error correction, or would derive position in some manner not affected by the slant range effect, the maximum additional protected airspace would be one-half mile at 10,000 feet and below, and three-quarters of a mile to 12,000 feet, in the vicinity of the referenced station in use. Based upon a review of advanced RNAV terminal area route designs and design procedures, it was determined that such buffer sizes would ordinarily have little or no effect on route placement and route length. As a result, the slant range error effect was not considered to have a significant effect on terminal VORTAC requirements.

The potential for saturation of DME facilities due to high traffic demand was considered as a part of the VORTAC requirements analysis. Present VORTAC stations serve a maximum of one hundred users at any one time. There exist methods for increasing the DME capacity without derogation of the TACAN signal [47]. It may be advisable from the standpoint of economics to consider increasing capacity of existing stations if other existing stations may be removed as an alternative to keeping those stations. However, DME capacity for purpose of this study is assumed to remain at present levels. DME demand was estimated for the very dense traffic areas from projected traffic demand data [21] for the post-1982 time period. It is of importance to note that DME saturation is not uniquely an RNAV phenomenon. With the future requirement for DME at several terminal areas being highly probable, DME saturation could occur in the near future if DME capacity improvements are not made even without the implementation of RNAV as the primary navigation system.

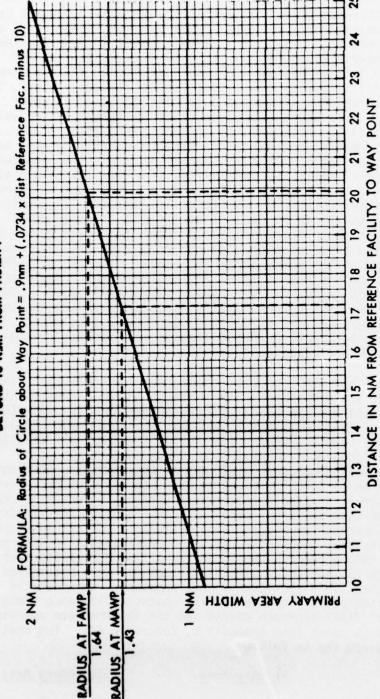
As a baseline for comparison with VORTAC station requirements given an RNAV-oriented environment, it is desirable to determine the requirement for VORTAC stations for terminal procedures presuming that RNAV would not be implemented. This analysis has been pursued in previous studies [13] and was based upon the presumption that, to a certain extent in busy terminal areas, VORTAC requirements would increase roughly in proportion to IFR traffic demand. This line of reasoning probably overstates the expected growth in VORTAC installations to some extent for two reasons. First, the requirement for DME capability at certain busy terminal areas is under active consideration. This would tend to increase the number of arrival, departure and holding fixes which could be supported by a given station. Of course this still does not provide the flexibility of routings and holding pattern orientations available with RNAV. Secondly, the typical approach taken by terminal control facilities in the face of a shortage of arrival fixes, for example, has been to use VORTAC stations further out from the airport(s). A major problem with this approach has been the lengthening of radar vector routings and an increase in workload on arrival control positions. In light of these factors it was decided that, even though station requirements would grow over the years, the number of new stations required presuming continuation of the VOR-oriented environment into the post-1982 time period probably would not be as great as the number which could be removed, given that RNAV is implemented. While this significantly understates the case for RNAV, the existing VORTAC structure was used for the baseline for present purposes due to the difficulty associated with accurately predicting the growth in VORTAC station requirements presuming continuation of the VOR environment.

In any case where a VORTAC station is being considered as a candidate for removal, the approach procedures which exist at nearby airports which depend upon that station must be taken into consideration. In order to insure that the removal of stations selected during this analysis would not degrade overall service in the terminal area, it was stipulated that all existing VOR and RNAV approach procedures depending on those stations would be replaced by RNAV approach procedures using one of the remaining stations. Also, in cases where an airport is served by only one approach procedure, and it is a VOR procedure dependent upon such a station, it would be replaced with both an RNAV approach and a VOR circling approach using the remaining stations. All such approach procedures were designed in detail in accordance with the procedure design and obstacle clearance requirements in Reference 48. In the case of each RNAV approach, the locations of the Missed Approach Point, Final Approach Fix and Intermediate Approach Fix were determined (lat-lon and rho-theta) for a straight-in RNAV approach. For obstacle clearance purposes, the primary and sencondary obstacle clearance areas were selected for the final approach course and the intermediate approach course (see Figure 4.39 taken from Appendix 2, Reference 49). The obstacle clearance areas required are as follows:

Route Segment	Primary Area	Secondary Area
Final Approach	250 ft	250 tapering to 0 ft
Intermediate Approach	500 ft	500 tapering to 0 ft
Initial Approach	1000 ft	500 tapering to 0 ft

AREA NAVIGATION

FINAL APPROACH COURSE PRIMARY AREA BEYOND 10 N.M. FROM FACILITY



EXAMPLE: Missed approach Way Point (MAWP) 17.2 NM from Reference Facility Radius (1/2 Primary area) = 1.43 NM

Final approach Way Point (FAWP) 20.1 NM from Reference Facility Radius (1/2 Primary area) = 1.64 NM

REFERENCE 17.2 NM MAWP FAMILY

Figure 4.39 RNAV OBSTACLE CLEARANCE AREA DIMENSIONS

Obstacle locations were determined from existing Jeppesen approach plates. Minimum altitudes for each approach segment were determined graphically. Also, recommended descent rate limits and the minimum 40:1 missed approach surface requirements were checked. The mimimums so determined were compared with the existing VOR minimums. In any cases where RNAV minimums were significantly higher than existing VOR minimums, the RNAV approach was discarded, and the VORTAC station which was the candidate for removal was retained. It was necessary to design three such RNAV approach procedures for Chicago, none for Denver, nineteen for Philadelphia and nine for New York. In most cases the resulting RNAV minimums were equivalent to or lower than the VOR minimums (usually when a circling approach was replaced). Note that approaches were designed for all airports affected by the removal of a VORTAC, whether within the terminal area boundary or outside of it.

In those cases where an RNAV approach could be successfully designed, but where the VOR procedure replaced was the only procedure available for that airport, VOR compatibility was maintained by designing a VOR circling approach using the reference facility selected for the RNAV approach for primary guidance. Design of VOR circling approaches is somewhat more involved than is the case for RNAV approaches. Primary and secondary obstacle clearance areas must be allocated as before (but of different dimensions) on the initial, intermediate and final approach courses. In addition, the circling area around the airport must be defined for obstacle clearance purposes, and its dimensions vary with approach category (A,B,C,D,E). Also the lowest standard circling minimums which may be applied are a function of approach category. Furthermore, the Final Approach Fix must be designated using a crossing radial from another VOR (or other radio navigation device). This fix must be provided within certain specified accuracy limits. The missed approach surface also is affected by the accuracy to which the Final Approach Fix is defined. It was necessary to design two such VOR approaches for Chicago, none for Denver, six for Philadelphia and six for New York. In most cases the resulting minimums were equivalent to those of the circling approaches replaced, but were higher than the straight-in approaches replaced.

4.3.2 Results

In Table 4.12 the results of these analyses are presented for each terminal area. In summary, approximately forty percent of all low and high altitude VORTACS within the terminal areas may be removed (only low altitude stations would be removed). Briefly, the results of each analysis are as follows:

CHICAGO (Figure 4.36) - Area and route coverage may be provided in the Chicago area by OBK, ORD, JOT and CGT within the terminal area, and ELX outside of it. The removal of DPA required designation of RNAV approaches at three airports, all of which were suitable replacements for existing VOR approaches. OBK and JOT were retained since they are high altitude facilities, with ORD and CGT filling in needed coverage. Terrain presented no problems and DME saturation would not be a problem with the complement of stations selected. Terminal route coverage (based on 1977 criteria) was provided everywhere except the extreme northeast sector, which is prevented by the existence of Lake Michigan.

DENVER (Figure 4.37) - Area and route coverage may be provided in the Denver area, except in the mountainous region west and northwest of Denver which presently has no low altitude coverage, by DEN (which is a high altitude facility) within the terminal area and TXC, COS and RLG exterior to it. The removal of IOC did not present any problems since it is not used for any instrument approach procedures. DME saturation is not a problem at Denver. Terminal route coverage gaps based on 1977 accuracy criteria (30 mile coverage radius) were not considered to be critical, since the mountainous terrain prevented provision of coverage.

PHILADELPHIA (Figure 4.38) - Complete area and route coverage may be provided at Philadelphia by RBV, CYN (which are high altitude facilities), OOD and PTW, all within the Philadelphia area. ENO was also included since the VOR approach procedure it supports (Dover, Delaware) could not be adequately replaced with an RNAV procedure. ACY was included in order to complete terminal route coverage (1977 criteria) in the southeastern sector. Terrain is not a problem at Philadelphia except that it had to be considered in selecting support reference facilities for certain of the RNAV approach procedures. With this complement of VORTACs, DME saturation is not a problem.

NEW YORK (Figure 4.38) - Complete area and route coverage may be provided at New York by JFK, SAX, RBV (which are high altitude facilities; note that RBV supports Philadelphia also), RVH and SBJ. LGA is also included in order to avoid DME saturation. Terrain is not generally a problem except in the selection of reference facilities for RNAV approach procedures. IGN, which is exterior to the terminal area, was also included in order to provide area and terminal route coverage in the northern part of the terminal area.

Table 4.12 Summary of VORTAC Requirements

Terminal Area	VORTACs Remaining	VORTACS Removable	Per Cent Removable
Chicago	OBK(H),ORD, JOT(H), CGT	DPA, EON	33%
Denver	DEN (H)	IOC	50%
Philadelphia	RBV(H), CYN(H), ACY, OOD, ENO, PTW	ARD, MIV, EWT, MXE	40%
New York	RVH, LGA, JFK(H), RBV(H), SBJ,SAX(H)	CMK, DPK, COL,	40%

/Note/ The parenthetical "H" means the station is a high altitude VORTAC

In order to demonstrate the advantages in terms of improved landing minimums which can often be gained from their use, the results of the RNAV approach procedure design efforts are presented in Table 4.13. It is evident from these results that there may be much to be gained from the establishment of RNAV procedures at other airports as well. Also, in order to document the results of the VOR circling approach procedure design effort, the landing minimums which resulted are presented in Table 4.14. Both the straight-in and circling minimums of the procedure replaced are listed. Note that the new circling minimums are on a par with, but usually slightly better than, the old circling minimums. While the specifications and requirements in [48] were strictly adhered to in the design process, it is possible that more conservative standards were applied in designing the published procedures, or that flight checking resulted in recommendations for slightly higher minimums. It should also be noted that, while the landing minimums are on a par with the existing approaches, the procedures themselves are less advantageous since they often require more severe maneuvers to align with the runway, and since they involve higher workload due to the necessity for a cross-radial used as the Final Approach Fix. Note that this is in contrast to the case of the RNAV procedures, which were for the most part more advantageous than the VOR procedures they replaced.

In order to reflect the economic impact of the savings which may be gained by removing (or relocating) the stations which are candidates for removal, it is necessary to first identify the types of savings which may be realized. The direct savings (those directly related to the removal of a station) include elimination of maintenance expense, prevention of routine equipment replacement or upgrading, prevention of need for eventual upgrading to DVOR status and release of real estate for other use. Routine maintenance and operating costs may be presumed to be 8% of new equipment installation cost [50], or \$48,500 annually for a dual VORTAC installation. The modernization of stations (conversion to solid state equipment, etc.) is not easily quantified in terms of cost, since data describing which stations require modernization is not available; however, older equipment in remaining stations could be replaced by removed new equipment, where available. The upgrading of status of stations to DVOR would, of course, not be necessary for those removed. This is an important consideration since many stations would require upgrading even if RNAV were to not be implemented. Conversion to doppler VOR costs \$160,500 for a dual station[44]. The potential value of real estate which could be disposed of is not estimated here, although it could be a significant factor since many of these sites are within or near metropolitan areas.

Upon review of the VORTAC requirements results determined above, which showed a range of savings from 33% to 50% at the four terminal areas studied, it is apparent that stations may be removed on a consistent basis. Review of the Denver case, however, reveals that it is a special case, since it would not be ordinarily expected that an entire terminal area could be supported with a single station (plus support from surrounding stations) which was the case in Denver due to the prohibitively high terrain to the west. Therefore, for purpose of extrapolating these results to terminal areas in general, the figure of 40% savings be used, but that it only be applied to terminal areas where at least four VORTAC stations already exist.

RNAV Approach Procedure Design Results Table 4.13

Ferminal Area	Airport	Procedure	Minimums	Procedure Replaced	Minimums Replaced
Philadelphia	Trenton, NJ(Mercer)	RNAV-6(RBV)	700(487)	VOR-A(ARD)	700(487)
	Trenton, NJ (Mercer	RNAV-24(RBV)	620(407)	VOR-24(ARD)	740(527)
	Trenton, NJ (Mercer)	RNAV-34(RBV)	620(420)	RNAV-24(ARD)	620(420)
	Trenton, NJ (Mercer)	RNAV-16(RBV)	620(407)	RNAV-16(ARD)	740(527)
	North Philodelphio	RNAV-33(RBV)	540(432)	RNAV-33(ARD)	460(352)
	North Philadelphia	RNAV-15(PTW)	700(580)	RNAV-15(PTW)	700(580)
	Contentille, PA(Chester)	RNAV-29(PTW)	980(318)	VOR-29(MXE)	1060(398)
	Vineland, NJ (Rudy's)	RNAV-26(ACY)	460(380)	VOR-A(MIV)	280(500)
	Vineland, NJ (Kroelinger)	RNAV-28(ACY)	540(453)	VOR-28(MIV)	520(433)
	Hammonton, NJ	RNAV-3(ACY)	320(251)	VOR-A(MIV)	(185)009
	Albion, NJ	RNAV-(OOD)	480(330)	VOR-4(MIV)	740(590)
	Milkille, NJ	RNAV-23(00D)	420(333)	VOR-23(MIV)	540(453)
	Toughkenamon, PA	RNAV-24(00D)	840(404)	VOR-24(MXE)	900(464)
	Philodelphia International	RNAV-98(00D)	520(497)	VOR-9R(EWT)	680(657)
	Philadelphia International	RNAV-27R(OOD)	540(529)	VOR/DME-27R(MXE)	540(529)
	Middletown, DE (Summit)	RNAV-17(00D)	520(449)	VOR-A(EWT)	(625)009
	Wilmington, DE	RNAV-9(00D)	480(401)	VOR-9(EWT)	420(341)
	Wilmington, DE	RNAV-32(00D)	480(405)	VOR-32(EWT)	540(465)
	Dover, DE (Cheswold)	RNAV-27(00D)	700(645)	VOR-27(ENO)	400(345)
Chicago	Aurora, IL	RNAV-27(JOT)	1140(434)	VOR-A(DPA)	1220(514)
	Chicago, IL(DuPage)	RNAV-10(ORD)	1180(424)	VOR-10(DPA)	1180(424)
	Elgin, IL	RNAV-36(ORD)	1300(510)	VOR-A(DPA)	1380(590)
New York	East Straudsburg, PA	RNAV-26(SAX)	1320(840)	VOR/DME-A(STW)	1480(1000)
	Andover, NJ	RNAV-3(SAX)	1240(657)	VOR-A(STW)	1480(897)
	Belmar-Farmingdale, NJ	RNAV-14(RBV)	500(343)	VOR-A(COL)	600(443)
	Neptune, NJ	RNAV-9(RBV)	640(540)	VOR-A(COL)	(095)099
	Danbury, CN	RNAV-8(PWL)	1340(883)	VOR-A(CMK)	1400(943)
	Newark, NJ	RNAV-11(JFK)	700(682)	RNAV-II (COL)	860(842)
	Amityville, NY(Zahn's)	RNAV-28(JFK)	420(366)	VOR-1(DPK)	(946)
	Farmingdale, NY	RNAV-19(JFK)	560(478)	VOR-A(DPK)	560(478)
	Rethrone NY	PNAV-33/IEK)	500/349)	VOD A /DDV	640/400

Table 4.14 VOR Circling Approach Minimums

			New Procedure			
		Approach	Circling	Procedure	Straight-in	Circling
Terminal Area	Airport	Category	Minimums	Replaced	Minimums	Minimums
Philadelphia	Coatesville, PA (Chester)	4	1020 (358)	VOR-29 (MXE)	1060(398)	1140 (478)
		8	1120 (458)		1060 (398)	1140(478)
		o	1120 (458)		1060 (398)	1200(538)
		Q	1220 (558)		1060 (398)	1280 (618)
	Vineland NJ (Rudy's)	٧	260 (480)	VOR-A (MIV)	-	280 (500)
	Vineland NJ (Kroelinger)	4	540 (447)	VOR-28 (MIV)	520(433)	580 (487)
	Harmonton, NJ	~	560 (491)	VOR-A (MIV)	-	600(531)
		8	560 (491)			600 (531)
		o	560 (491)			600(531)
		D	620(551)			620(551)
	Albion, MJ	٧	680 (530)	VOR-4 (MIV)	-	740(590)
	Toughkenamon, PA	٧	880 (444)	VOR-24 (MXE)	900 (464)	900 (464)
Chicago	Aurora, IL	V	1180 (474)	VOR-A (DPA)	1	1180 (474)
		8	1180 (474)			1180 (474)
		o	1180 (474)			1180 (474)
		Q	1260 (554)			1260(554)
	Elgin, IL	×	1360 (570)	VOR-36 (DPA)	1320(530)	1420 (630)
		8	1360 (570)		1320(530)	1420 (630)
		o	1360 (570)		1320 (530)	1420 (630)
		Q	1360 (570)		1320 (530)	1420 (630)
New York	East Stroudsburg, PA	Y	1260 (780)	VOR DME- (STW)	-	1480 (1000)
	Andover, NJ	Y	1340 (757)	VOR-A (STW)	+	1480 (897)
	Belmar-Farmingdale, NJ	4	540 (383)	VOR-A (COL)	-	600 (443)
		9	620 (463)			620(463)
		o	620 (463)			620 (463)
		Q	720 (563)			720 (563)
	Neptune, NJ	<	640 (540)	VOR-A (COL)	1	(095) 099
		8	100 (600)			(095) 099
		o	100 (600)			(095) 099
		Q	840 (740)			(095) (099)
	Danbury, CN	٧	1380 (923)	VOR-A (CMK)	1	1400 (943)
		8	1380 (923)			1400 (943)
		o	1380 (923)			1400 (943)
		Q	1380 (923)			1400 (943)
	Bethpage, NY	<	240 (408)	VOR-A (DPK)	1	540 (408)
	题 學 解 让 持 母 玩 民 司	a	(468)			(468)
		o a	600 (468)			640 (508)
		•	(890)00/			(895)00/

A survey of the high and medium density terminal areas has been performed in order to determine the number of low and high altitude VORTAC stations resident in each area (within 45 miles). The results are presented in Tables 4.15(high density) and 4.16 (medium density). Note that the totals of those terminals with four or more stations are presented separately, and totals 120 stations all together. Forty percent of these, 48 stations, could therefore be removed while maintaining full services and coverage. Note that this does not include the many low density terminal areas to which these criteria could also apply.

In summary, the annual maintenance savings which would accrue due to the removal of forty-eight stations would be \$2.33 million. This amount, of course, is exclusive of any ordinary modernization costs, such as would be required for conversion to solid state equipment, etc. The one-time cost savings which could be realized by not requiring the conversion to doppler VOR for the removed stations would be \$4.84 million.

The results presented here are quite conservative since the baseline used for comparison was the assumption that no additional VORTAC installations would be required for terminal area purposes if RNAV were not implemented. This would clearly not be the case. The NAS Plan [51] states that five new stations will be installed per year, although it does not present a

Table 4.15 Terminal VORTAC Stations - High Density Terminals

Terminal	Low	High	Total	>4
Atlanta	2	1	3	
Boston	2 2 3 4	3	3 5 5 6 5 2 3 4 7 2 2 10 5 9 5	5
Chicago	3	2 2	5	5 5 6 5
Cleveland	4	2	6	6
Dallas	4	1	5	5
Denver	1	1	2	
Detroit	2 2 4	1	3	
Houston	2	2	4	4 7
Los Angeles	4	3	7	7
Memphis.	1	1	2	
Miami	0	2 3 1 2 3 3	2	
New York	7	3	10	10 5 9
Phoenix	2	3	5	5
Philadelphia	8	1	9	9
Pittsburg	4	1	5	5
San Antonio	0	1		
San Diego	2	2	4	4 7
San Francisco	6	1	/	/
St. Louis	2 8 4 0 2 6 2		3 9	
Washington	7	2	9	9
TOTAL	63	34	97	81

Table 4.16 Terminal VORTAC Stations - Medium Density Terminals

Terminal	Low	High	Total	>4
Albany	3	1	4	4
Albuquerque	1	2	3	
Birmingham	4 7		5	5 8
Bradley	7	1	8	8
Buffalo	1 2 2 1 2	2 0	3	
Charlotte	2		2	
Cincinnati	2	0	2	
Columbus	1	1	2	
Dayton	2	ski fasedes 100	3	
Des Moines		Market Market 1988	2	100
El Paso	1	1	2	1.1250
Indianapolis	3	1	4	4
Jacksonville	0	2	2	la farant
Kansas City	3	de la lac	4	4
Knoxville	1	1	2	Est al
Las Vegas	3 0 3 1 0 2	2	2	
Louisville	2	1	3	150 01
Minneapolis	1	1	2	order o
Milwaukee	0	2	2	and the same
Nashville		1	2	Dishell -
New Orleans	3 4	1	4	4
Norfolk	4	1	5	5
Oklahoma City	. 0	2	2	e din
Omaha	0	2	3	
Orlando	0	2	2	la trans
Portland		2	2	
Providence	0 2 1	2 2 2 2 3 1	5	5
Raleigh-Durham	1	A STATE OF THE STATE OF	2	altio d
Rochester	2	e de la	3	
Sacramento	ī		3	L HENN
Salt Lake City	i	2 2 1	3	
Syracuse	2	Ī	3	Bit Hall
Tampa	2 0 1	3	43583222322424223222452322523333332	-
Tulsa	i	i	2	
TOTAL	54	47	101	39

breakdown as to purpose (terminal or enroute). The added stations are quite costly on an annual basis in comparison with the savings associated with the removal of stations, since the acquisition cost must be amortized over the useful life of the station. The total annual cost (maintenance plus amortization costs) per station [13] would be \$103,000 for a single transmitter station and \$128,000 for a dual installation. If RNAV could alleviate the requirement for some of these stations, equivalent savings would result.

4.4 PRELIMINARY ANALYSIS OF THE RNAV IMPACT ON ATC AUTOMATION

This section discusses the various ways in which the phased implementation of RNAV can affect plans for the continued development and implementation of Third Generation and Upgraded Third Generation ATC Automation. In those areas where a potential effect on terminal or enroute automated ATC capabilities is expected, further efforts have been made to quantify those effects on cost-sensitive factors.

4.4.1 Areas of Potential Impact

A total of six areas in ATC automation were determined to exhibit potential for impact in one or more of the three RNAV implementation phases. These phases have been considered both in the manner described by the Task Force [1] and as modified during this study, with emphasis on the latter. The terminal area ATC facets impacted include the definition and charting format of RNAV routes, the conflict prediction problem, and arrival aircraft metering and spacing (M&S). While the enroute areas with potential for impact also include the definition of RNAV routes and the conflict prediction problem, enroute flow control could also be affected. The types of ATC automation impacts considered include core storage requirements, software development, computer time usage, and other hardware requirements. It should be recognized that, while variations in three of these factors can directly impact cost, computer time utilization does not necessarily directly affect costs. Instead, the overall processing capacity of a given ATC facility installation is limited by the computing speed available and tasks required to be performed. When a new task or combination of tasks exceeds this capacity, a major upgrade in equipment may be required which could be very costly. However, where sufficient capacity is available, new tasks have no significant effect on costs.

The following section quantifies, where appropriate, the specific impact of RNAV in these six areas. For the most part the impacts were found to be minimal or even favorable. However, should preplanned direct operations be implemented extensively, a significant impact on computer time utilization could result which could have a cost impact at those ARTC Centers which are at or near the existing computer time capacity.

4.4.2 Quantification of Impact

4.4.2.1 RNAV Route Definition Impact

The introduction of RNAV routes or preplanned direct operations has an impact, particularly enroute, on ATC system computer resources. The ATC equipment so impacted are primarily the computer core storage requirements and and computer time utilization. Any needed software development would occur during the routine software modification process. Other ATC automation facilities would not be affected. The major effect of introducing an expanded RNAV route structure is to increase the number of routes in any given ARTCC and so to increase the number of routes which must be adapted to the center coordinates. As the total number of routes increases (particularly over the short term), core requirements will likewise increase. Computer time utilization is unaffected in this case. When preplanned direct route flight plans

become common, however, computer time utilization will be affected. The NAS Stage A computer software is presently capable of processing direct routes, and so additional software development is not required. Computer time utilization is affected by each pre-planned direct route filed since each center computer system affected must adapt the individual route to center coordinates. Core storage is not affected since the route description tables so created are identical to those created for conventional flight plans. The introduction of RNAV routes is not expected to significantly impact terminal ARTS system requirements, primarily since the ARTS system is not route-oriented.

A method has been developed [52] for determining the impact of implementing RNAV structures on ARTCC computer core requirements. Estimates were also made in that report of the effects of preplanned direct routes on computer time utilization. Since computer time utilization is not necessarily a direct system cost factor, it is not considered further here. Instead, the incremental core requirements due to the introduction of the RNAV route structure have been estimated. In all of these analyses, the results presented in Reference 52 have been reorganized in order to conform to the revised RNAV implementation plan presented in this report. The pertinent features of that plan as applied to this analysis are summarized below:

- Phase I: High Altitude VOR and RNAV structures as they presently exist. Very limited preplanned direct. Low Altitude Present VOR structure plus limited RNAV structure. Very limited preplanned direct.
- Phase II: High Altitude Extended RNAV structure and no VOR routes. Limited preplanned direct.

 Low Altitude Extended RNAV structure and realigned, reduced VOR structure complexity. Very limited preplanned direct.
- Phase III: High Altitude Same RNAV structure plus extensive preplanned direct.

 Low Altitude Same structures plus limited preplanned direct.

The Phase I analysis presumes that the high altitude RNAV structure will be of the same complexity as the present RNAV structure. Furthermore, a low altitude RNAV structure will be implemented. The extent of the low altitude RNAV structure in each ARTC center is assumed to be proportional to the number of high altitude RNAV routes. The proportionality factor in each case is taken to be 50% (to reflect limited implementation in Phase I) of the ratio of existing low altitude to high altitude VOR routes.

The Phase II analysis presumes an RNAV structure which is at least as complex (in each ARTC center) as the more complex of the existing VOR or RNAV structures. In addition, the complexity has been inflated 25% to account for limited parallel routes and additional city-pair coverage. That is to say, the high altitude RNAV structure is presumed to be 25% more complex than the existing RNAV or VOR structure whichever is more complex than the present existing VOR/RNAV combination. The Phase II low altitude RNAV structure is

Table 4.17 Additional RNAV Route Core Storage Required by Center, Phase I

\$ 40 000 000 000 000 000 000 000 000 000	Existing High Altitude RNAV Routes	Existing High Altitude VOR Routes	Existing Low Altitude VOR Routes	Present Route Core Storage Usage(K-Wards)	Estimated Low Altitude RNAV Routes	Additional Core Starage Required (K-Words)
NYC	25	31	20	24.5	28	4.8
×	30	90	39	18.9	20	3.4
ATL	31	23	69	23.7	47	8.0
ON	22	22	81	24.3	41	7.0
풍	4	29	92	28.5	58	6.6
808	6	37	47	18.3	9	1.0
30	23	33	8	27.1	29	4.9
MKC	23	26	63	21.7	28	4.8
HOU	20	22	45	16.8	20	3.4
న్ద	20	25	75	19.2	22	3.7
FTW	23	28	2	22.3	26	4.4
OAK	22	13	38	13.9	32	5.4
NEN	26	30	45	19.4	16	3.2
MIA	2	14	28	10.1	10	1.7
MEM	91	22	25	17.5	19	3.2
MSP	23	*	75	25.7	25	4.3
SEA	14	22	38	14.4	12	2.0
JAX	20	20	31	13.6	91	2.7
ABQ	20	32	53	20.4	17	2.9
SIC	16	25	19	20.4	23	3.9
AVERAGE	23	26	95	20.0	25	4.3

Table 4.18 Additional RNAV Route Core Storage Required by Center, Phase II

Additional Core Storage Required (K-Words)	4.5	4.9	0.5	-3.7	0.8	-2.7	-2.3	-3.9	-3.3	-3.6	4	1.5	4.	-1.7	-2.9	4.2	-2.6	-3.3	-3.7	-3.5	-2.9
Met Decrease In VOR Routes	77	25	89	75	78	89	77	67	15	09	2	38	59	32	38	8	47	40	99	65	62
Met Increases In RAAV Routes	28	47	101	87	126	\$	101	23	23	99	9/	20	22	36	2	95	25	36	23	82	ĸ
Deleted High Altitude VOR Routes	31	30	23	22	29	37	33	26	22	25	28	13	30	7	22	8	22	20	32	25	26
HgiH gnistix∃ VAM8 abutitlA Routes	25	30	31	22	4	٥	23	23	20	20	23	22	26	2	91	23	14	20	20	61	23
Hith motomits a VANA setucitly setucitly	39	38	39	28	55	46	4	33	28	31	35	28	38	18	28	43	28	25	40	31	35
Deleted Low Altitude VOR Routes	46	25	45	53	49	31	7	4	29	35	42	25	29	18	8	49	25	20	34	40	36
Estimated Low VANA Shittude RNAV Routes	70	39	8	81	115	47	83	63	45	7	2	2	45	28	25	75	38	31	23	19	09
Center	NYC	××	ATL	ONI	CHI	808	CLE	MKC	НОО	PCA	FTW	OAK	DEN	MIA	MEM	MSP	SEA	JAX	ABQ	SLC	AVERAGE

presumed to be fully implemented (i.e., the proportionality factor is taken to be 100% of the ratio of existing low altitude to high altitude VOR routes). Also, the VOR structure complexity is assumed to be reduced to 35% of its present state, reflecting the deletion of unnecessary routes and realignment of others. It has been further presumed that twenty percent of all flights (low and high, but mostly high) are, at least in part, preplanned direct (an average of 50% of each such flight plan being direct routings was selected).

The Phase III analysis presumes that route structures of the same complexity exist as in Phase II. However, the role of preplanned direct routings is expected to expand in both the low and high altitude environment.

Data concerning the present number of VOR and RNAV routes within each of the 20 ARTC centers within CONUS are taken from Reference 52. That reference also presents detailed estimates for the amount of computer core storage required (average) for the storage of VOR routes, RNAV routes (Phase I), and RNAV routes (Phases II and III). That reference presumed that the number of fixes required to define RNAV routes to the center computer would be reduced from the present level of twelve per route per center to eight as the Phase II and III periods are reached. This probably results since the VOR structure will have been removed, and so route segment interactions will be reduced. The per route core requirements are as follows:

Environment	Words/Route
VOR	200
RNAV Phase I	170
RNAV Phases II and III	130

Table 4.17 presents the existing high and low altitude route data for each center, and, in addition, the projected Phase I low altitude route structure complexity derived as explained earlier. That table presents both the present core storage usage plus the increment required by the interim low altitude RNAV route structure. On the average the core increase is 4.3 K-words (22%), while the largest increase would occur at the Chicago Center, 9.19 K-words (35%). To put these numbers in perspective, the standard NAS computer core provision is 655.36 K-words.

Table 4.18 presents the results of the Phase II core requirements analysis in which expanded high and low altitude RNAV route structures are projected, and where the high altitude VOR structure is eliminated and the low altitude structure is reduced to 35% of its present complexity. Note that the net core requirement impact is affected both by the changes in numbers of routes of each type and the number of words per route required as described in the preceding table. Table 4.18 indicates that the reorganization of route structures will actually diminish core requirements on average of 2.9 K-words (15%) compared to present usage. Since the only change from Phase II to Phase III is an increase in Preplanned direct filings, this table also applies to the Phase III situation. Additional computer time utilization, required for converting the preplanned direct route data to center coordinates, will increase as the number of preplanned direct flight plans increases as discussed in Reference 51.

4.4.2.2 Impact on Conflict Prediction Requirements

In the enroute environment, a reliable form of conflict prediction based on tracking data and flight plan prediction is a prerequisite to the unconstrained implementation of preplanned direct RNAV operations. Also, of course, some form of general area flight checking for navigation signal coverage will also be required. Plans for development of such a conflict prediction system include basic programs presently under development [52] plus eventual development of advanced programs, even to the point of automated conflict resolution message generation and delivery. Such advanced systems will make use of available tracking information and flight route data for purposes of projecting impending conflict situations. While the implementation of preplanned direct will mean more such routes at any given time, this should not affect the core requirements or even the computer time utilization of the enroute conflict prediction algorithm since the route data will already have been converted to center coordinates (when the flight plan was activated) and since the number of aircraft under examination will not be affected. When automated conflict resolution decision capabilities are implemented, they will be slightly more complex due to inclusion of RNAV conflict resolution techniques along with conventional techniques. However, this effect should be very minor.

Preplanned direct flight plans can be implemented to a certain extent prior to the advent of full scale operational conflict prediction techniques. However, such flight plans will be limited to areas where radar coverage exists. Therefore, where operational advantages exist, the use of preplanned direct flight plans where permissible should receive encouragement. While full scale enroute conflict prediction capability is a requirement for unconstrained preplanned direct, direct routings do not necessarily affect the complexity of the conflict prediction system.

In the terminal environment, the introduction of RNAV aircraft should have no direct effect on conflict detection algorithms (except where automated resolution decision capability is employed, as before). This is due to the probable use of tracking data only, not route data, in the conflict prediction process. Should route data be used, RNAV should improve the situation since the intended routes are well defined in advance, as opposed to radar vector routings.

4.4.2.3 Impact on Enroute Flow Control Program

The enroute flow control program is presently in a developmental stage. Its major purposes include the scheduling and rerouting of aircraft to balance capacity and demand in order to avoid serious delays and diversions. Capacity imbalances can occur due to shifts in demand, or to weather induced problems, or to control or navigation facility outages. The entire system is to consist of a central flow control facility and local flow control data collection and management facilities at each ARTC center. The only area of flow control activity which should be affected by RNAV operations is in the

rerouting function to be used to alleviate severe weather and facility outage effects. In performing this function, the central facility will be afforded more flexibility with the RNAV capabilities, particularly in being able to use the direct or random routing capability and parallel offset capability inherent with RNAV. However, the system will have to be able to deal with more data in order to be able to utilize this increased flexibility. This should not affect individual centers to a significant extent. The degree of effect upon the central facility is difficult to assess until the planned capabilities and functions of the system are more firmly established. However, the inclusion of RNAV should not cause a large perturbation in system complexity.

4.4.2.4 Impact on Terminal Metering and Spacing

The impact of RNAV operations on the core requirements and computer time utilization requirements of arrival metering and spacing concepts currently under consideration has been investigated in Reference 52. In that study the additional program code required for performing M & S with a mixed (RNAV and conventional) environment over that required presuming a conventional environment has been determined to require 0.45 K-words of additional core storage at each ARTS installation. This number is not considered to be very consequential since the projected core requirement for the entire M & S function is 16 K-words. It is particularly inconsequential in view of the benefits which can be made available through the use of 4D RNAV capabilities, as discussed in Section 3.4. In Reference 52 it has also been determined that the effect of RNAV operations on computer time utilization would be of no significance.

4.4.3 Economic Implications of RNAV Automation Impact

The primary RNAV impact determined above concerned computer system core requirements. The problem of determining the cost impact of increased core requirements is not necessarily straightforward. While cost per K-word data is available, determination of true cost is complicated by the fact that in some cases additional core memory may not be added at will, but may be limited by other hardware and program structure considerations. This is particularly true in the enroute environment where all NAS computers (9020A and 9020D models) are currently configured with the maximum planned active core complement (655,360 words). It is physically possible to add more core storage; however, existing software programs would have to be reconfigured in order to accommodate the modified core addressing scheme. Two methods for relaxing the enroute core constraint are presently under investigation: the modification of existing memory units to increase capacity (affecting 9020A sites only); and other hardware and software improvements which include provision of improved flight data displays which have integral data storage capability.

In view of all of the above factors, core storage availability is actually much more valuable than the basic equipment costs involved. When storage resources are in use at or near their limits, and additional capabilities such as RNAV are being evaluated, these planned capabilities must complete with each other for this available space, and so some form of cost figure should be attached to available storage space. The basic equipment costs are as follows:

Environment	System	Core Unit and Cost	Cost/K-Word
Enroute:	9020A	65 K-words, \$125,000	\$1900
Terminal:	9020D ARTS III	132 K-words, \$475,000 16 K-words, \$40,000	\$3600 \$2500

It should be noted that additional core storage capability is still available on both the present ARTS systems and the planned enhanced ARTS multi processor systems, and so the stated ARTS core cost is fairly realistic, while the NAS enroute costs should be considered as being extremely conservative. In particular, while the 9020D sytem has the highest per unit cost, there are presently no expansion plans for it, and so therefore its figure is the most conservative of all.

The enroute core requirement due to the implementation of RNAV is fortunately quite small through all three implementation phases. The mean Phase I impact of 4.3 K-words per center could be evaluated at a cost of \$12,900 per center, presuming a mean value for core storage of \$3000/K-word. In Phase II and likewise Phase III, the impact is a savings of \$8700 per site over present VOR structure requirements. Therefore, the cost of the core utilized would be \$21,600 less in Phase II than in Phase 1, and this capability could be used to accommodate other system growth requirements. As preplanned direct routes are implemented extensively in Phase III, this level of direct routings may not be feasible at some centers due to the increased computer time required for route adaptation. In contrast, the RNAV increment per ARTS III site due to inclusion of RNAV in the metering and spacing system is an inconsequential \$1100. This is particularly small in view of the benefits which RNAV, and 4D RNAV in particular, can provide to the M and S function.

4.5 IMPACT OF VNAV ON AIRSPACE CAPACITY

The RNAV operational concept described in Section 5 of this report includes the assignment of fixed altitudes and/or altitude restrictions as an integral part of the design of terminal area SIDs and STARs. It is possible to designate altitudes, or altitude restrictions, at a horizontal position, either by the explicit assignment of altitude at that point or implicitly by specifying a fixed gradient 3D route whose centerline passes through the geographical point of interest at the desired altitude. Fixed gradient 3D routes could be specified in either of two ways: (1) specified gradient or (2) specified altitudes. In the first case, the vertical profile of the 3D route is described by a 3 dimensional point in space from which a specified vertical gradient route emanates. In the second case, the gradient of the route is determined by drawing a straight line between two geographical points each of which has a specified altitude or altitude limit. The Task Force considered that 3D routes offered a potential for increasing airspace capacity.

This section presents an analysis of the potential of fixed gradient 3D routes as a design tool for the airspace planner to increase airspace capacity. 2D versus 3D vertical separation requirements are examined for crossing routes, and a comparison is made of the airspace required and the number of waypoints or altitude restriction points required for 2D and fixed gradient 3D routes. Finally, an analysis is presented of additional airspace requirements and operational constraints associated with climbing and descending parallel offset routes. They are examined for both fixed gradient 3D routes and procedural altitude control on 2D routes, and include the separation and operational problems associated with parent route or offset turnpoints in the vicinity of a crossing or intersecting route.

The results of this analysis support the conclusion that the use of fixed gradient 3D routes for procedural separation and/or the requiring of 3D capability for entry into certain airspace does not appear to offer sufficient payoff in either airspace capacity or operational utility to warrant consideration. A corollary of this conclusion is that the terminal area design approach recommended in Section 5 of this report need not be constrained by a requirement for fixed gradient 3D routes. Therefore, optimum arrival and departure routes may be designed on the basis of providing the shortest path length and most efficient altitude profiles for 3D equipped aircraft, and 2D equipped aircraft can also utilize these profiles. Additional economic benefits are available to 3D equipped aircraft through pilot selection of 3D gradients, as discussed in Section 3.1.

4.5.1 <u>Separation Requirements</u>

Vertical and horizontal separation have always been provided independentlyi.e., the joint probability distribution is not considered in assigning separation, and both the horizontal and vertical separation requirements are met in assigning altitude restrictions. The technique selected for VNAV separation [2] is the time-proven separation concept of utilizing 2 σ errors in the horizontal plane and 3 σ errors in the vertical plane. The along track error, reflected into the vertical plane, is not directly additive, but is considered a part of an error budget whose elements are combined in an RSS manner.

When the along track error is considered in this way, crossing routes may be described as "tubes" with rectangular cross sections whose dimensions describe the 2 σ horizontal, and 3 σ vertical, total system errors and the required vertical separation between the center lines of these routes may be derived geometrically by placing the routes such that edges of the "boxes" just touch. This technique was applied in the derivation of vertical separation requirements in Reference 2.

The consideration of along track error as a part of the vertical error budget is certainly an acceptable expedient. It does not, however, provide for a convenient method of computing the horizontal location of positions at which altitude restrictions may be applied to insure vertical separation of crossing RNAV (2D) routes, or for the crossing of a 2D route by a 3D route where an altitude restriction is necessary on the 2D route.

4.5.2 RNAV Separation (2D/2D)

Consider the crossing route situation depicted in Figure 4.40. The route widths L_1 and L_2 represent the total 2 σ cross track error of each route. The routes are both descending, as indicated by the arrows on the route centerline, and it is assumed that route #1 will cross over route #2. It is desired to establish two altitude restriction points to provide for procedural separation.

In determining the point along route #1 at which to place the restriction, the effective width of route #2 must be considered. It consists of two elements: 1) the width of route #2 at the crossing angle $\theta: \frac{L_2}{|\sin\theta|}$, and 2) the additional width of route #2 due to the width of route #1:

The longitudinal uncertainty of position on route #1 must also be taken into account. (This is the error which was accommodated in the VNAV case by making it a part of the vertical error budget). This along track error on route #1 may be considered as an additional element in the effective route width of route #2, but it should not be combined RSS, as it was in the VNAV case. In the VNAV case, the along track error on route #1 is but one element of the VNAV error on route #1. In this case (RNAV), it is the total error, along one coordinate of the horizontal plane for route #1, and is being combined with the other elements of the effective width of route #2 only for geometric convenience. The RTCA committee on Area Navigation, SC-116E, recommended the inclusion of along track error, reflected into the vertical, as part of the VNAV error budget [17] and it is included in this comparison of 2D and 3D operation to maintain consistency. The selection of an along

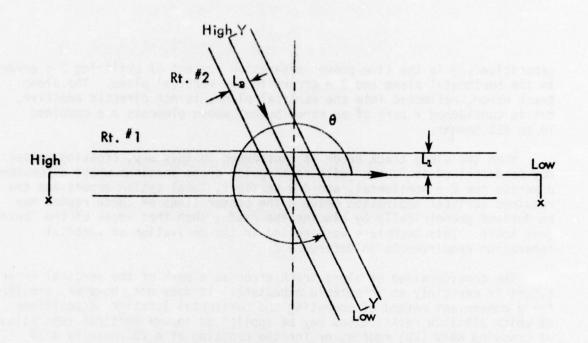


Figure 4.40a Plan View

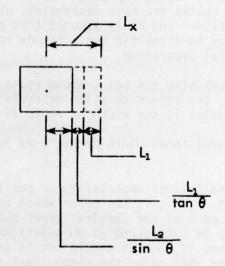


Figure 4.40b Vertical Cross-Section of Route #2 at Intersection

track error equal to cross track error is perhaps overly conservative and implies an error distribution which is not wholly representative of any existing RNAV system. Both the inclusion of the maximum along track error in the 2D case, and the RSS combination of the maximum along track error in the 3D case, combine to favor 3D in comparison with 2D separation in the following analysis. However, consideration of any specific postulated error distribution may be accomplished by modifying the along track error terms in the equations which are developed below.

Referring again to Figure 4.40, if $L_{\rm X}$ is defined as the effective width of route #2, at the intersection of route #1, then for purposes of route separation $L_{\rm X}$ is as shown in Table 4.19 for various values of the crossing angle θ .

Table 4.19 Effective Width of Crossing Route for Purposes of Altitude Separation

L _X = Effective Route Width
$\frac{L_1}{\tan \theta} + \frac{L_2}{\sin \theta} + L_1$
$-\frac{L_1}{\tan \theta} + \frac{L_2}{\sin \theta} + L_1$
$\frac{L_1}{\tan \theta} - \frac{L_2}{\sin \theta} + L_1$
$-\frac{L_1}{\tan \theta} - \frac{L_2}{\sin \theta} + L_1$

Since L_X is zero or positive for all values of θ , L_X may be expressed as:

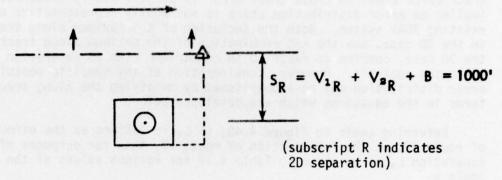
$$L_{x} = \left| \frac{L_{2}}{\sin \theta} + \left| \frac{L_{1}}{\tan \theta} \right| + L_{1}$$
 (4.2)

and similarly, by inspection:

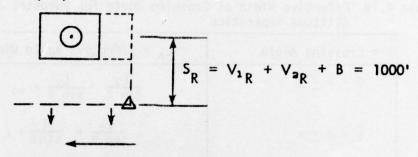
$$L_{y} = \left| \frac{L_{1}}{\sin \theta} + \left| \frac{L_{2}}{\tan \theta} \right| + L_{2}$$
 (4.3)

The location of altitude restriction points for crossing route separation with an example of altitude restrictions is given in Figure 4.41.

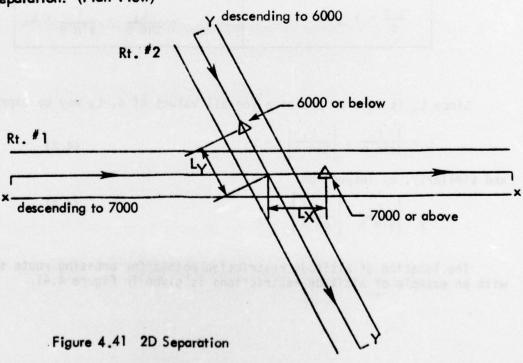
For 2D Separation: (x - x view)



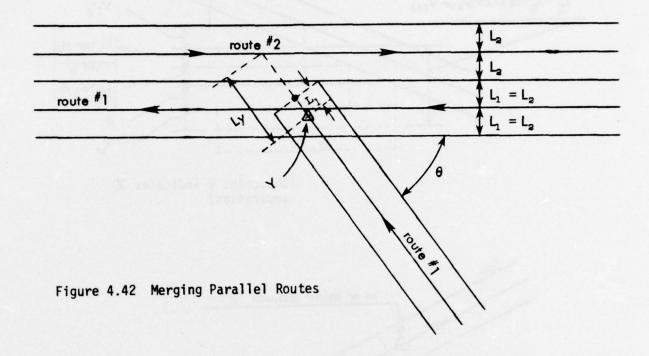
For 2D Separation: (y - y view)



For 2D Separation: (Plan View)



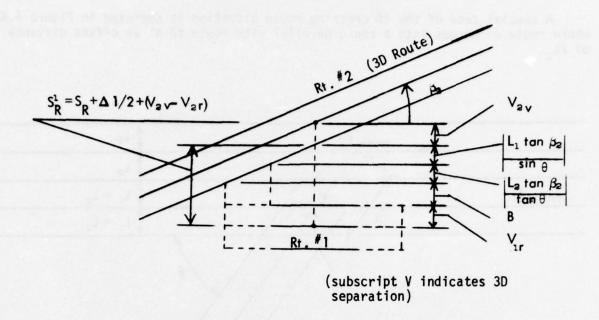
A special case of the 2D crossing route situation is depicted in Figure 4.42 where route #1 merges into a route parallel with route #2 at an offset distance of $2L_2$.



If route #1 were to continue across route #2 at the angle θ , altitude separation would have to be provided starting at waypoint Y. In the merging case, however, route #1 can be at the same altitude as route #2 since the contribution of along track error on route #1 to parallel offset distance requirements approaches zero as an aircraft makes the turn onto the parallel route, and with some type of turn anticipation the aircraft on route #1 will not violate route #2 airspace (see Section 5.3.3).

4.5.3 2D/3D Separation

Now consider a 3D route (#2) crossing a 2D route (#1) with θ defined such that the high end of route #2 would overlay the high end of route #1 (and the + axis of a right hand coordinate system) when θ = 0, as illustrated in Figure 4.43.



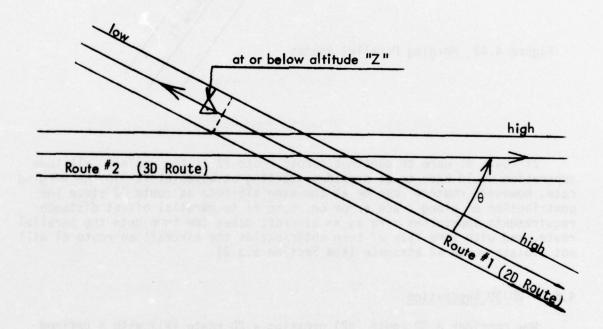


Figure 4.43 2D and 3D Crossing

The vertical separation between the center lines of intersecting 3D routes was derived in Reference 2 and may be expressed as:

$$S = V_1 + V_2 + B + L_1 \left| \frac{\tan \beta_2 - \cos \theta \tan \beta_1}{\sin \theta} \right| + L_2 \left| \frac{\tan \beta_1 - \cos \theta \tan \beta_2}{\sin \theta} \right| (4.4)$$

where $2V_1$ and $2V_2$ are the 3σ vertical dimensions of routes #1 and #2 respectively, B is a "buffer" of 300 ft., and,

$$V = \sqrt{(3\sigma_V)^2 + (3\sigma_{at} \tan \beta)^2}$$
 (4.5)

In the terminal area, the suggested value for $3\sigma_{V}$, the error due to altimetry, VNAV equipment, and flight technical error, is ± 350 ft. [17]. If a system error budget is assumed which results in a 2σ cross track error of ± 2 nm and which includes a 2σ cross track FTE of 1.0 nm, the 3σ along track error reflected into the vertical will be 274 ft. per degree of β .

Then,

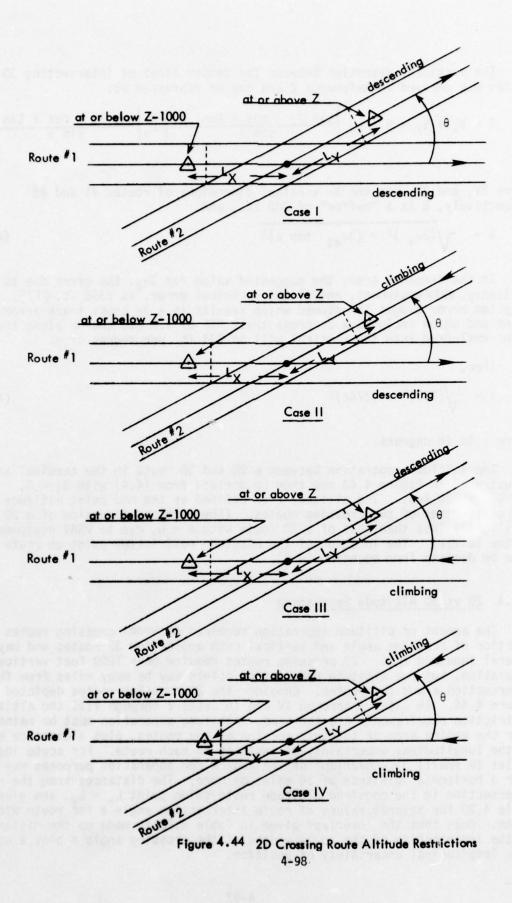
$$V = \sqrt{(350)^2 + (274\beta)^2} \tag{4.6}$$

where β is in degrees.

The vertical separation between a 2D and 3D route in the terminal area as illustrated in Figure 4.43 may then be derived from (4.4) with β_1 = 0, and V_{1R} = 320 feet. The altitude Z is defined as the mid point altitude between the centerlines of two crossing routes. (The vertical dimension of a 2D route is slightly less than that of a 3D route with β = 0, due to VNAV equipment error in the latter). The location of the altitude restriction point on route #1 may then be derived from equation (4.2).

4.5.4 2D vs 3D Altitude Separation

The amount of altitude separation required between crossing routes is a function of crossing angle and vertical path angles for 3D routes and may be several thousand feet. 2D crossing routes require only 1000 feet vertical separation, but the altitude restriction points may be many miles from the intersection along the routes. Consider the 2D crossing routes depicted in Figure 4.44. In cases I through IV and in cases V through VIII the altitude restriction requirements are the same. Altitude separation must be maintained over the entire area of the intersection of the routes, plus a distance equal to the longitudinal uncertainty of position on each route. For acute intersection angles (θ small) an "overlap" of the routes for separation purposes may exist over a horizontal distance of 30 miles or more. The distances from the route intersection to the required altitude restriction point $L_X = L_V$, are given in Table 4.20 for several values of route intersection angle θ for route widths of ±2 nm. Note that the "overlap" given in Table 4.20 is made up the distance due to the intersection of the route widths at the crossing angle θ plus a constant 2 nm longitudinal uncertainty of position.



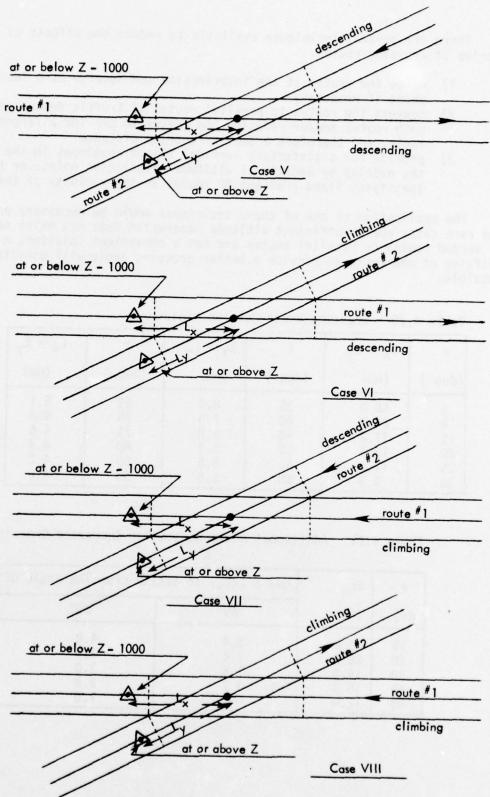


Figure 4.44 (cont.) 2D crossing route altitude restrictions
4-99

There are several techniques available to reduce the effects of the overlap of crossing routes:

 merge the routes at the intersection and demerge at a later point if necessary

 convert the routes to parallel routes if traffic demand requires both routes and/or redesign the routes to provide a larger crossing angle, and therefore a smaller overlap distance

3) provide for satisfactory vertical flight envelopes in the region of the overlap by additional altitude restriction points or by specifying fixed gradient 3D routes in the vicinity of the intersection

The application of one of these techniques would be necessary only in the rare cases where sufficient altitude separation does not exist naturally. If merged routes or parallel routes are not a convenient solution, a minor shifting of one route to provide a better crossing angle will usually be possible.

Table 4.20 Crossing Route Overlap Distance

θ	L _X = L _y	θ	L _X = L _y	θ	L _x = L _y
(deg.)	(nm)	(deg.)	(nm)	(deg.)	(nm)
5	48.0	35	8.3	65	5.1
10	24.9	40	7.5	70	4.8
15	17.2	45	6.8	75	4.6
20	13.3	50	6.3	80	4.4
25	11.0	55	5.8	85	4.2
30	9.5	60	5.5	90	4.0

Table 4.21 Additional Route Length for Increased Crossing Angles

θ (deg.)	2L _x (nm)	$\max \Delta$ (2L _X) to attain crossing angle of:	
		45°	60°
10	49.8	3.4	4.8
10 20 30	26.6	2.4	3.9
30	19.0	1.6	3.0
40	15.0	0.7	2.0
40 50	12.6	_	1.0

Table 4.21 gives the maximum additional route length which would have to be added to one route when redesigning that route to increase the crossing angle with another route to a more acceptable value (i.e., 45° - 60°). In the design of seven terminal areas [2], it was found that crossing angle was not a constraint. Minimum crossing angles were usually on the order of 30° between arrival and departure routes, and this angle did not produce vertical separation problems. Arrival routes to multiple airports, which have the potential for small crossing angles, usually are common and are demerged at the appropriate point. Conversely, departure routes are merged prior to the departure waypoints. An increase in route length to improve crossing angles would be a very rare occurrence, and would seldom involve addition of the maximum distance.

The remainder of this subsection addresses the vertical separation required between the centers of crossing fixed gradient 3D (VNAV) routes as compared with the vertical distance between the "effective" centerlines, in the vertical plane, of crossing 2D routes where the vertical path angle (VPA), or "fixed gradient", of the "effective" centerlines are determined by the constraints of altitude restriction points. The latter case is referred to as "2D vertical separation" for convenience in the discussion that follows.

The difference between 2D and 3D vertical separation is illustrated in Figure 4.45. S_V = the vertical separation between 3D routes at the intersection and S_R is defined as the vertical separation which would be required between two 3D routes if 2D (blocked altitude) separation were to be provided instead of 3D separation. Figure 4.45a illustrates VNAV (fixed gradient 3D) crossing routes. Figures 4.45b illustrates the blocked airspace and altitude restriction points for 2D separation. The distance of the altitude restriction points on each route from the intersection of the routes may be derived from Table 4.20. Vertical separation for 3D crossing routes is given by equation[4.4]. The vertical distance between the "effective" centerlines of crossing 2D routes, as illustrated in Figure 4.45b is given by:

$$S_R = \frac{S_{RA}}{2} + \frac{S_{RB}}{2} = L_x | \tan \beta_{RA} | + L_y | \tan \beta_{RB} | + 1000$$
 (4.7)

Figure 4.46 gives the vertical separation requirements for 3D intersecting routes, with a constant route width of ± 2 nm as a function of crossing angle and vertical path angles. The 2D "separation" as defined above is also given. It should be emphasized that 2D vertical separation can always be reduced to 1000 ft. by shifting one route horizontally to improve the crossing angle and/or allowing the effective maximum vertical path angle to increase on one side of the altitude restriction point and to decrease on the other side. Figure 4.46 applies to the case where it is desired to maintain a 3D ceiling or floor (or minimum or maximum average 2D vertical path angle) which is defined by a straight line between the waypoints defining the respective legs of the routes.

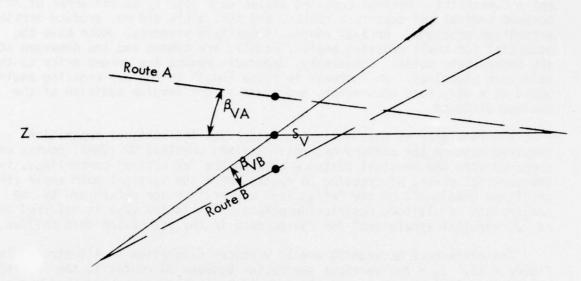


Figure 4.45a Vertical Separation of 3D Routes

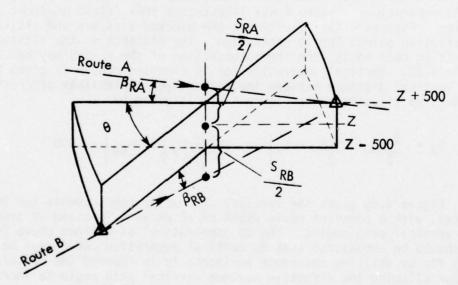
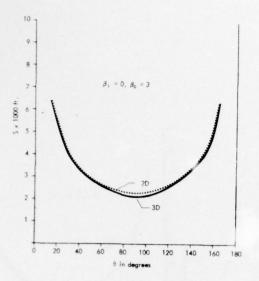


Figure 4.45b. 2D Separation

Figure 4.45 2D and 3D Vertical Separation



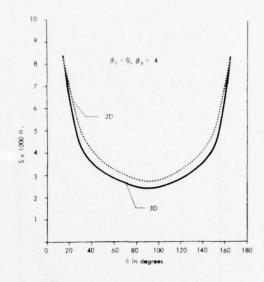
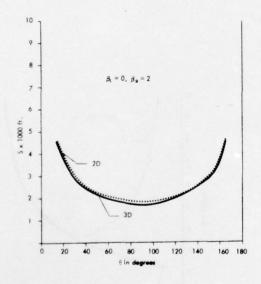


Figure 4.46a



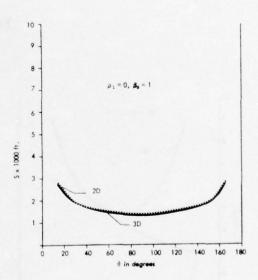
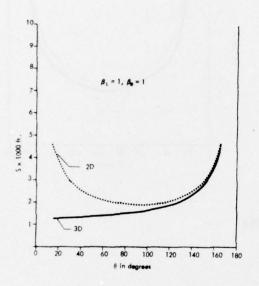


Figure 4.46 2D vs 3D Separation Requirements



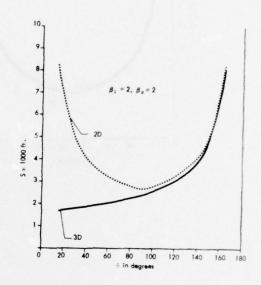
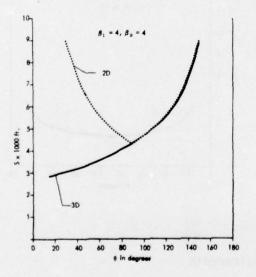


Figure 4.46b



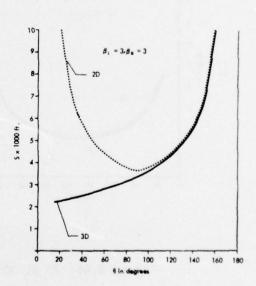


Figure 4.46 (continued) 2D vs 3D Separation Requirements

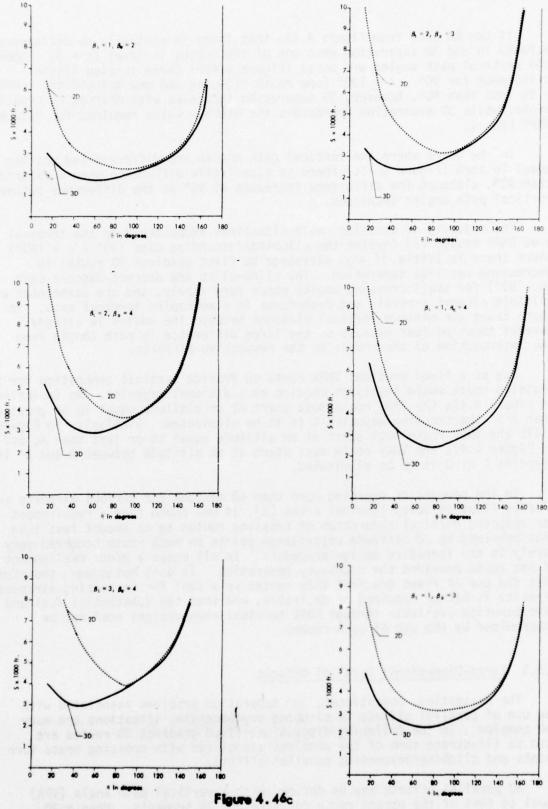


Figure (continued) 2D vs 3D Separation Requirements 4-105

It can be seen from Figure 4.46a that there is virtually no difference between 2D and 3D separation when one of the routes is level (β = 0). When the vertical path angles are equal (Figure 4.46b) there is also little difference for 90° < θ < 180° (one route climbing and one descending). When θ is less than 90°, however, 2D separation increases with decreasing crossing angle, while 3D separation approaches the minimum value required for "stacked" VNAV routes.

In the cases where the vertical path angles are different and neither is equal to zero (Figure 4.46c) there is also little difference when θ is greater than 90° , although the difference increases at 90° as the difference between vertical path angles increases.

The majority of crossing route situations encountered in the terminal area RNAV design [2] involve the climbing/descending case (90° < 0 < 180°) where there is little, if any, advantage to fixed gradient 3D routes in decreasing vertical separation. The climb-climb and descent-descent case (0 < 90°) for small crossing angles occur very rarely, and are associated with multiple airport arrivals and departures in a metroplex terminal area. In these cases the minimum vertical distance between the routes is already several thousand feet because of the large difference in path length from the intersection of the routes to the respective airports.

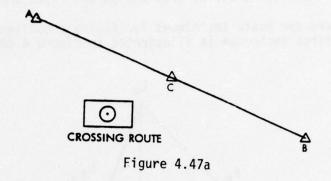
Use of a fixed gradient VNAV route to provide vertical separation for a crossing route would sometimes require an additional waypoint. As indicated in Figure 4.47a the VNAV route must start at an altitude equal to or greater than A if the crossing waypoint C is to be eliminated. Similarly, in Figure 4.47b the VNAV route must start at an altitude equal to or less than A, and in Figure 4.47c the VNAV route must start at an altitude between A and A' if waypoint C or D is to be eliminated.

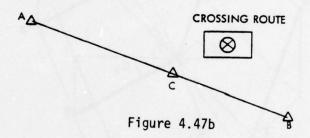
In the process of creating more than 40 designs for various airports and traffic flows in seven terminal areas [2], it was found that a requirement for reducing vertical separation of crossing routes to an amount less than that provided by 2D altitude restriction points on each route occurred very rarely in the iterative design procedure. In all cases a minor realignment of one route provided the necessary separation. It does not appear, therefore, that the use of fixed gradient VNAV routes as a tool for increasing airspace capacity is either required or desirable, and that the substantial fuel and time benefits available through RNAV terminal area designs need not be compromised by the use of such routes.

4.5.5 Three-Dimensional Parallel Offsets

The navigation, operational, and separation problems associated with the use of parallel offsets in climbing or descending situations are many and complex. In the following discussion, fixed gradient 3D routes are used to illustrate some of the problems associated with crossing route turn points and climbing/descending parallel offsets.

3D parallel offsets may be defined with a vertical path angle (VPA) equal to that of the parent route for straight 3D segments. When a 3D





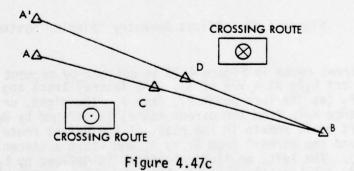


Figure 4.47 Waypoint Requirements for Crossing Fixed Gradient VNAV Routes

segment contains a turnpoint, offset routes with the same VPA as the parent route will experience an altitude discontinuity at the turn point. If this altitude discontinuity is to be avoided, the offset routes must be defined such that they emanate from a point on the bisector of turn angle, with the result that their VPA's differ from the parent route VPA.

There are two basic techniques for flying a constant 3D offset around a turn. The first technique is illustrated in Figure 4.48.

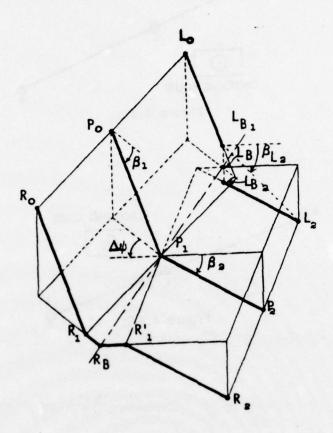


Figure 4.48 Offse Geometry "Simple" System

The parent route in Figure 4.48 is defined by segment PoP1 at a VPA of β_1 and by segment P1P2 at a VPA of β_2 . The lateral track angle from segment PoP1 to P1P2 (at the turn point P1) is Δ ψ . The right, or "outside" offset (at a distance R0P0 from the parent route) is defined by R0-R1-R8-R1-R2; i.e., the aircraft would remain in the plane of the parent route to point R1, fly level "around the corner" from R1 to R1 and start a descent at a VPA of β_2 at point R1. The left, or "inside" offset is defined by L0-L81-L2. When the aircraft reaches the vertical plane through the bisector R8L8 (which passes

through the intersection of the horizontal projections of segments L_0L_{B1} and $L_1L_{B2})$ the desired altitude changes instantaneously from L_{B1} to L_{B2} and a minimum VPA of β_{L2} is required to reach point L_2 . In other words, an offset on the inside of a turn results in a shorter distance available for the same altitude change, and consequently, a steeper vertical path angle. The VPA of the leg following a turn point for an "inside" offset is plotted in Figure 4.49 for β_1 = β_2 .

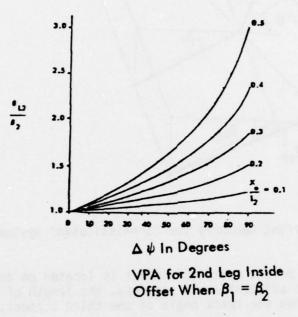


Figure 4.49 Inside Offset VPA

The significance of Figure 4.49 may be illustrated by an example: Consider an aircraft flying on a 6 mile left offset at a vertical path angle, β_1 of -3 degrees. The next leg of the parent route is 20 miles in length, has a VPA, β_2 , of -3 degrees and has a track 90 degrees to the left of the present leg.

 $\frac{\chi_0}{20}$ is then $\frac{6}{20}$ = 0.3; entering Figure 4.49 at $\Delta\Psi$ = 90° and χ_0/ℓ_2 = 0.3 gives $\frac{2}{\beta_{12}/\beta_2}$ of 1.85. Upon intercepting the 6 mile left offset of the next leg, the aircraft must attain an average VPA of at least 1.85 x 3 degrees, or 5.55 degrees. The second technique is illustrated in Figure 4.50.

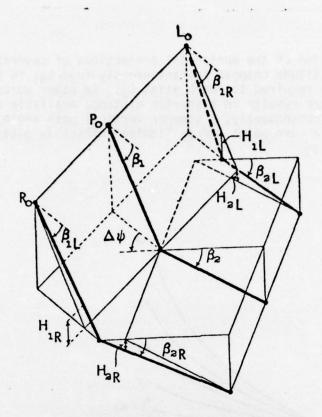


Figure 4.50 Offset Geometry for "Sophisticated" System

In this case, since the offset route turn point is located on the bisector of the angle between adjacent route segments, the length of the second offset is dependent upon the track angle of the third segment, as illustrated in Figure 4.51.

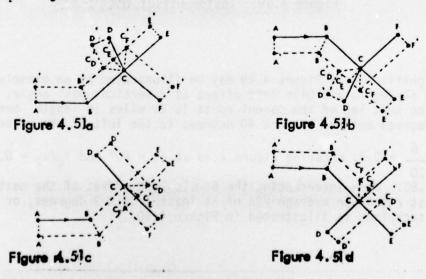


Figure 4.51 Offset Route Horizontal Plan Geometry

In all cases shown in Figure 4.50, the inside offset VPA will be larger than the parent VPA, but the increase required in the VPA will be less than the increase required in the "simple" system technique.

The crossing route vertical separation required for the "simple" system technique illustrated in Figure 4.48 is the same as that required for a straight segment with the following exceptions: 1) the separation above is with respect to leg 1, while the separation below is with respect to leg 2; 2) the effective route width (L) of segments 1 and 2 must include the offset distance.

The crossing route vertical separation required for the "sophisticated" system technique may be greater or less than that for the "simple" system technique, depending upon the geometry of the crossing route situation (see Reference 2 for a detailed discussion).

Figure 4.52 gives the crossing route vertical separation for a straight segment route, B, crossing at the turnpoint of another route, A, with a 6 nm "outside" offset. All conditions are as shown in Figure 4.53, $\beta_A = 3^\circ$; $\beta_B = 0^\circ$, and the length of leg 2 of route A is 25 nm. The geometry is as depicted in Figure 4.51c, configuration ABCE.

An example is illustrated on Figure 4.52. The specific geometry depicted in Figure 4.53 would require vertical separation between the route centerlines at the crossing point X of approximately 2400 feet for route B crossing above, and 2600 feet for route B crossing below route A.

If the "simple" system technique (Figure 4.48) were used by all aircraft on route A offset, their altitude at point X' (Figure 4.53) would be the same as at point X and no additional separation would be required above that required between route B and leg 1 of route A for route B crossing over route A. Likewise, the altitudes at point X and X' would be the same, and the separation required for route B to cross below route A would be the same, and the separation required for route B to cross below route A would be determined by the crossing angle between route B and leg 2 of route A.

4.5.6 Vertical Separation for Pilot-Selected 3D Routes

Figure 4.46 shows a comparison of the required separation between crossing VNAV routes and the vertical distance between the "effective" centerline of crossing 2D routes. The "effective" gradient of a 2D route may also be defined as a pilot-selected fixed gradient to or from the 2D altitude restriction point. It can be seen from Figure 4.46 that the actual separation of pilot-selected 3D gradient, as indicated by the dashed lines, when passing through the 2D altitude restriction points is always greater than, or equal to, the required vertical separation, as shown by the solid lines, for combinations of vertical path angles up to 4 degrees.

This section further analyzes the vertical separation provided between pilot-selected 3D routes which are based on altitude restriction points established for 2D separation.

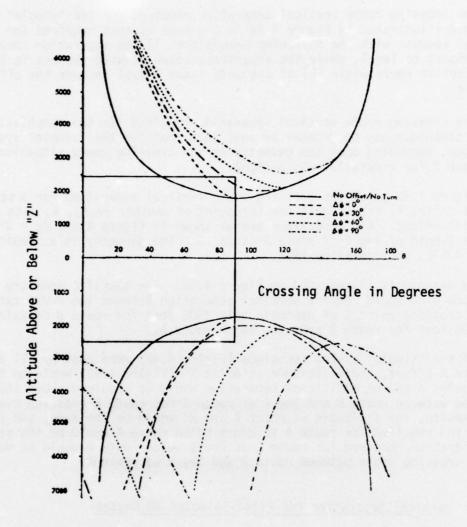


Figure 4.52 Vertical Separation for Turning Routes - "Sophisticated" System (6nm offset, 25nm legs)

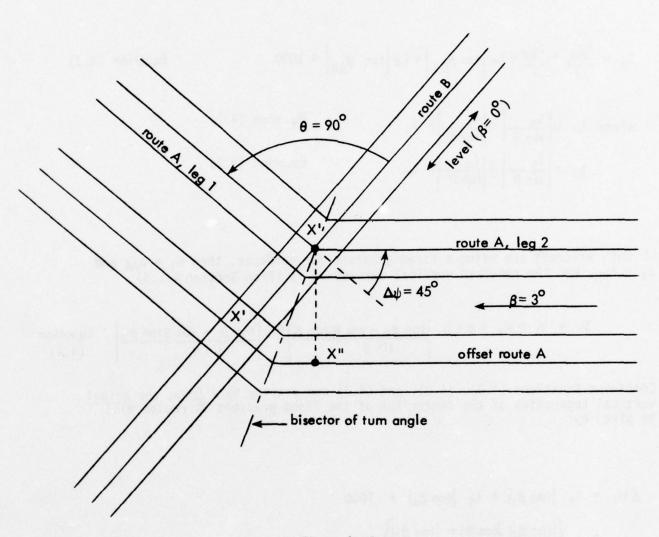


Figure 4.53
Projection of Crossing Routes in Horizontal Plane

In Section 4.5.4, it was shown that the vertical distance between the "effective" centerlines of crossing 2D routes (or of pilot-selected gradients), is given by the following equation:

$$S_{R} = \frac{S_{RA}}{2} + \frac{S_{RB}}{2} = L_{x} \left| \tan \beta_{RA} \right| + L_{y} \left| \tan \beta_{RB} \right| + 1000$$
 Equation (4.7)
$$\text{where } L_{x} = \left| \frac{L_{2}}{\sin \theta} \right| + \left| \frac{L_{1}}{\tan \theta} \right| + L_{1}$$
 Equation (4.2)
$$L_{y} = \left| \frac{L_{1}}{\sin \theta} \right| + \left| \frac{L_{2}}{\tan \theta} \right| + L_{2}$$
 Equation (4.3)

If both aircraft are using a fixed gradient for guidance, then β_1 = β_{RA} and β_2 = β_{RB} , and the required vertical separation is (from Section 4.5.3):

$$Sv = V_1 + V_2 + B + L_1 \left| \frac{\tan \beta_2 - \cos \theta \tan \beta_1}{\sin \theta} \right| + L_2 \left| \frac{\tan \beta_1 - \cos \theta \tan \beta_2}{\sin \theta} \right|$$
 Equation (4.4)

Combining Equations (4.2), (4.3), and (4.7) and setting $S_R = \Delta Hv$, the <u>actual</u> vertical separation of the centerline of the fixed gradient 3D routes will be given by:

$$\Delta Hv = L_1 |\tan \beta_1| + L_2 |\tan \beta_2| + 1000$$

$$+ L_1 \left\{ \frac{|\tan \beta_1| |\cos \theta| + |\tan \beta_2|}{|\sin \theta|} \right\}$$

$$+ L_3 \left\{ \frac{|\tan \beta_2| |\cos \theta| + |\tan \beta_1|}{|\sin \theta|} \right\}$$
Equation (4.8)

by definition, β_1 and β_2 are positive; therefore, for $0 \le \theta \le \frac{\pi}{2}$,

$$\frac{\left|\tan\beta_{2}\cos\theta\right|+\left|\tan\beta_{1}\right|}{\left|\sin\theta\right|}\geq\left|\frac{\tan\beta_{1}-\cos\theta\tan\beta_{2}}{\sin\theta}\right|$$
and
$$\frac{\left|\tan\beta_{1}\cos\theta\right|+\left|\tan\beta_{2}\right|}{\left|\sin\theta\right|}\geq\left|\frac{\tan\beta_{2}-\cos\theta\tan\beta_{1}}{\sin\theta}\right|$$
also,
$$for\frac{\pi}{2}\leq\theta\leq\pi,$$

$$\frac{\left|\tan\beta_{2}\cos\theta\right|+\left|\tan\beta_{1}\right|}{\left|\sin\theta\right|}=\left|\frac{\tan\beta_{1}-\cos\theta\tan\beta_{2}}{\sin\theta}\right|$$
and
$$\frac{\left|\tan\beta_{1}\cos\theta\right|+\left|\tan\beta_{2}\right|}{\left|\sin\theta\right|}=\left|\frac{\tan\beta_{1}-\cos\theta\tan\beta_{1}}{\sin\theta}\right|$$

By comparing equation (4.4) with equation (4.8), it can be seen that Δ Hv will always be equal to or larger than Sv if:

Figure 4.54 is a plot of ΔS_{min} vs β_1 and β_2 for the following conditions (reference Equation (4.6):

$$L_1 = L_2 = 2 \text{ nm}$$

$$V_1 = \sqrt{(350)^2 + (274 \beta_1)^2}$$

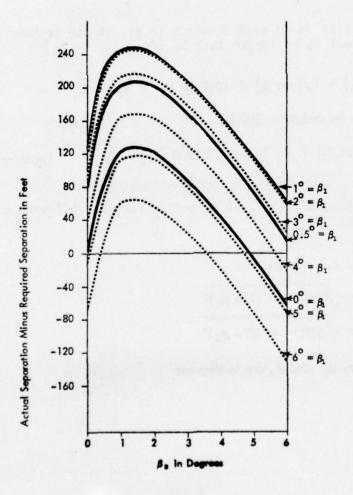
$$V_2 = \sqrt{(350)^2 + (274 \beta_2)^2}$$

where β_1 and β_2 are in degrees

The separation buffer shown in Figure 4.54 is the minimum buffer that will exist, independent of the crossing angle of the routes for $0 \le \theta \le \frac{\pi}{2}$, and is the actual buffer that will exist for $\frac{\pi}{2} \le \theta \le \pi$. If both routes are climbing or both routes are descending $(0 \le \theta \le \frac{\pi}{2})$ and $\theta \le 87^\circ$ the separation buffer will always be greater than zero for combinations of $0 \le \beta_1 \le 6^\circ$ and $0 \le \beta_2 \le 6^\circ$. If one route is climbing and the other route descending $(\frac{\pi}{2} \le \theta \le \pi)$, the altitude restriction points must be placed 0.2 nm further from the route intersection to insure vertical separation between pilot-selected combinations of fixed gradient routes up to 6 degrees.

Figure 4.54

Minimum Vertical Separation
Buffer - Actual Vertical
Separation vs Required Vertical
Separation for Pilot-Selected
3D Gradient to 2D Altitude
Restriction Point



The preliminary analyses of ATC system and user impact reported in References 11 and 12 have been updated and expanded as described in Sections 3 and 4 of this report. The results of these analyses as well as those of other supporting system analyses, described in Section 2.1, have a significant impact on the system design concept recommended by the RNAV Task Force. In Sections 5.1 and 5.2 below, the results of these studies are summarized and discussed in this context as a prelude to the development of a modified Task Force system design concept in Section 5.3 and the resulting implementation requirements described in Section 5.4. The impacts shown in this section apply to an all-RNAV environment. However, significant individual aircraft benefits are available to equipped aircraft as RNAV routes are implemented in the transition phases. The system concept which is defined includes a phased transition from VOR to RNAV, such as suggested by the Task Force, but the timing of the phases is dependent upon user demand rather than the fixed calendar periods recommended by the Task Force.

5.1 USER IMPACT SUMMARY

This section presents a summary of user benefits which will accrue through the use of 2D, 3D and 4D RNAV and 2D/3D approach procedures. The annual savings which can be expected to accrue to air carriers, business and other general aviation are listed, and the corresponding cost impact on both civil and military users of the National Airspace System is discussed.

5.1.1 Terminal Area Benefits

5.1.1.1 2D RNAV Benefits

In previous studies [9,12], it was determined that the dominant factors relative to their impact on user operations in the terminal area are route length and altitude restriction reductions. These benefits were attained by the development of a flexible RNAV design concept which represents a modification of the RNAV Task Force design concept. This terminal area design concept is summarized in Section 5.3.1.2 and described in detail in Reference 2. The fuel and time savings in an RNAV terminal area compared with current VOR/radar vector routes are derived in Section 3.1 for eight different types of aircraft for each traffic flow at nine major airports. The benefits at these airports were used to extrapolate the benefits to the sixty high are medium density airports including those identified in the Task Force Report [1]. The extrapolation was accomplished by applying a regression analysis to the nine airport data base for use in estimating the sixty airport benefits as a function of terminal area characteristics.

Table 5.1 summarizes fuel and time savings at 1984 traffic levels for 2D RNAV operations at the 60 major airports compared with VOR/vector operations. Estimates for business piston aircraft and other general aviation aircraft are based on average distance saved at six of the airports studied (New York was excluded because of the uniqueness of its characteristics).

Table 5.1 2D RNAV Fuel and Time Savings at 60 Major Airports 1984 Traffic Levels

	Fuel Savings (1000 lb)	Time Savings (1000 min)
Certificated Air Carrier Aircraft		
4 engine wide body 3 engine wide body 4 engine jet 3 engine jet 2 engine jet	140,697 226,941 59,423 301,662 267,252	394 951 288 2,172 1.930
Total Air Carrier	995,975	5,735
Business aircraft -		
Jets & Turboprops Piston Aircraft	2,167 1,638	91 1,023
Total Business Aircraft	3,805	1,114
Other General Aviation Aircraft	8,923	5,574
TOTAL	1,008,703	12,423

The average distance saved was computed to be 2.16 miles per operation, based on the data given in Table 3.1. From data given in Reference 33, it was estimated that the average speed of general aviation piston aircraft is 120 mph and their average fuel consumption is 0.8 lb per mile. The average fuel and time savings per operation (1.73 lb and 1.08 minutes) was then applied to general aviation itinerant operations in 1984. From NBAA data it was estimated that 14.75% of GA itinerant operations are business piston flights, in addition to the 4.85% business jet and turboprop operations. The remaining 80.4% are categorized as "other general aviation". Total general aviation itinerant operations at the sixty airports was estimated to be 6,419,000 [21]. Results of real time simulations of the New York terminal area [9,11] confirmed that the distance savings are available in an operational environment and are not reduced by the requirements for traffic sequencing. The results of the real time simulation described in reference [] further indicated a substantial reduction in arrival delays (an average of 4.7 minutes per arrival) which would result in time and fuel savings in addition to those shown in Table 5.1. The simulation also showed that an additional distance savings resulted, over and above that attributed to design efficiency. VOR and RNAV traffic was all flown over the RNAV routes in the simulation, with VOR traffic navigated by radar vectors and RNAV traffic self-navigated. On the average, the RNAV equipped aircraft flew 1.61 miles less than non-RNAV aircraft, and had a corresponding time savings of 1.02 minutes. These benefits are in addition to the fuel and time benefits (in the New York terminal area) which are realized by better altitude profiles and the average distance savings of 11.6 miles per operation which are due to the improved design efficiency of the RNAV design compared with current day VOR/vector routes.

From Tables 3.3 and 3.4, it can be seen that 190 million pounds of fuel and 914,000 minutes are projected to be saved over a total of 337,000 airline and business jet operations at JFK in 1984 through 2D RNAV due to improvement in the terminal area design by RNAV. Additional savings of 112 million pounds of fuel and 964,000 minutes can be expected to be saved through reduction in arrival delays and better adherence to designated terminal area routes. The 2D benefits at JFK then consist of those due to strategic design efficiency (which are included in the aggregate TMA benefits shown in Table 5.1) plus additional benefits of the same order of magnitude which result from improvement in tactical efficiency. It may be assumed that this improvement in tactical efficiency is related to both design complexity and traffic levels. Since the simulation was conducted only for the JFK design, no basis exists to extrapolate these additional savings to other terminal areas. It can be reasonably concluded, however, that overall tactical savings in the terminal area can be expected, and their magnitude is of the order of a significant percentage of the strategic savings.

5.1.1.2 3D Descent Benefits

Additional benefits available through the use of pilot-selected 3D descents at the 60 major airports were computed for certificated air carrier aircraft in Section 3.1.3, and are summarized in Table 5.2.

Table 5.2 Fuel and Time Savings Available From Air Carrier Pilot-Selected 3D Descents at 60 Major Airports at 1984 Traffic Levels

come tokentanist	Fuel Savings (1000 lb)	Time Savings (1000 min.)
4 engine wide body	23,000	54
3 engine wide body	69,900	300
4 engine jet	34,100	142
3 engine jet	168,500	1,150
2 engine jet	181,300	1,668
Total Air Carrier	476,800	3,314

The penalties in fuel and time which would be incurred in fixed gradient 3D climbs are given in Reference 12. No penalties for 3D climbs are anticipated since the terminal area design concept described in Section 5.3 provides for essentially unrestricted climbs through use of high performance climb envelopes. The benefits due to these envelopes are included in the 2D savings summarized in Section 5.1.1.1.

5.1.1.3 4D Benefits in M & S Terminals

Section 4 presents an analysis of the impact of 4D RNAV capability on arrival delays through a comparison of 4D RNAV time-control with presently planned metering and spacing automation improvements. Results of analytical studies and simulations have shown that 4D RNAV techniques may be employed in M & S terminals to significantly reduce arrival control error in comparison with the control error expected with metering and spacing alone. The analysis in Section 4 determines the runway capacity increase and corresponding reduction in delays which result from the reduction of interarrival control error with 4D RNAV in M & S terminals.

The time savings at the 25 major delay terminals in 1984 that can be realized with 4D navigation in addition to metering and spacing, which were derived in Section 3.4, were used to estimate the corresponding fuel savings. The average holding fuel flow for the five types of air carrier aircraft was derived on the basis of the 1984 projected fleet mix in Table 5.6 and applied to the total time savings. Total 4D savings in 25 M & S terminals are given in Table 5.3.

Table 5.3 Fuel and Time Savings through 4D in 25 Major M & S Terminals

	Fuel (1000 lb)	Time (1000 min.)	
Air Carrier Aircraft	980,370	9,044	

5.1.1.4 RNAV Approach Benefits

Area navigation has already found widespread use in instrument approach procedures. (More than 200 2D RNAV approaches have been requested by users and published to date.) RNAV approaches offer flexibility in selection of obstacle-free approach paths. An RNAV approach which is based on VOR/DME inputs cannot inherently have lower circling minimums than a VOR/DME approach, since it is subject to the same navigational errors (plus RNAV computer error), but it may be possible to establish a straight-in approach to a runway where only a circling approach is possible with VOR or VOR/DME procedures. The additional flexibility available from RNAV approaches is due to the capability to define approach paths which are not constrained to VOR radials or DME arcs. An analysis was made of 206 published RNAV approach procedures compared with other types of approaches published for the same runways. The differences in Minimum Descent Altitude (MDA) or Decision Height (DH) between the RNAV approaches and the corresponding VOR, VOR/DME, NDB, ASR, or ILS approaches to the same runways are summarized in Table 5.4 and are plotted for each comparison in Figure 5.1. The bars in Figure 5.1 represent a range of MDA differences on the horizontal areas and the number of the differences in that range on the left vertical axis represents an RNAV MDA lower than VOR/DME. The dashed lines on Figure 5.1 are the cumulative number of RNAV MDAs which are "X" better than VOR/DME, as read on the right vertical axis.

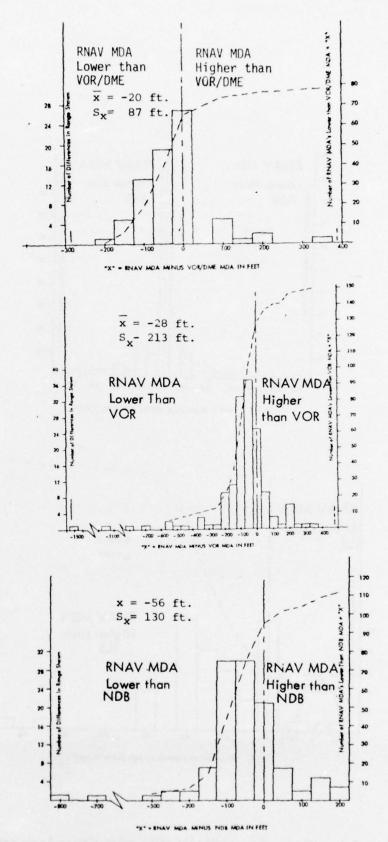
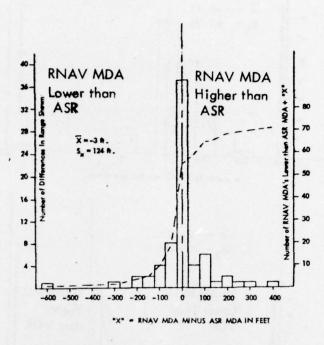


Figure 5.1 Comparison of RNAV MDA With Other Approach Minimums



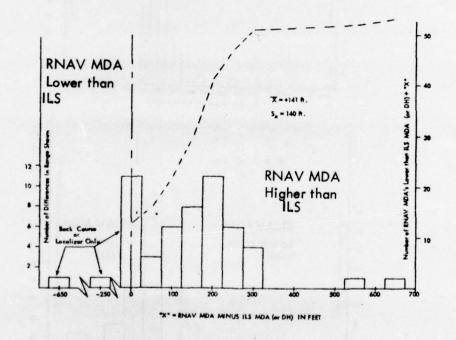
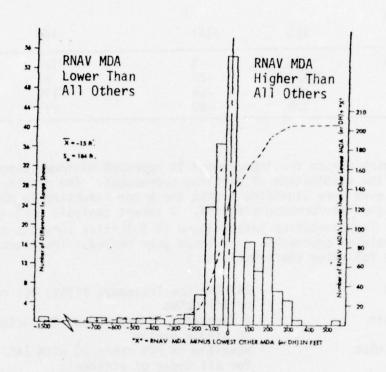


Figure 5.1 Comparison of RNAV MDA With Other Approach Minimums (continued)



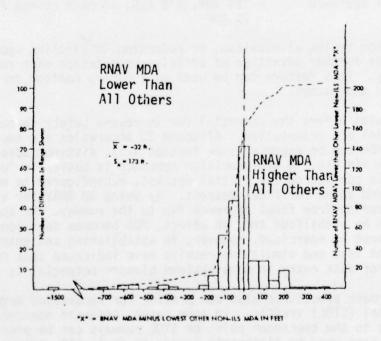


Figure 5.1 Comparison of RNAV MDA With Other Approach Minimums (continued)

Table 5.4 Difference in RNAV MDA or DH

	mean	difference (\bar{x})	standard deviation (S,	
RNAV MDA higher	ILS	+141	140	
RNAV MDA	ASR	-3	124	
lower	VOR/DME	-20	87	
	NDB	-56	130	
	VOR	-88	213	

The primary reason for improvement in approach minimums through use of RNAV lies in the elimination of circling approaches. The resultant increase in safety is even more significant than the minor reduction in MDA, particularly for higher performance aircraft. A recent analysis [54] indicated that National Transportation Safety Board (NTSB) files showed a total of 42 accidents involving approaches in a three year period. These accidents exhibited the following characteristics:

- pilots	- 24% Airline Transport Pilot, 40% commerical certificates
- aircraft	 6 turbojets, 1 turboprop, 25 piston twins, 10 single engine
- fatalities	 resulted in 52% compared with 14% for all types of accidents
- type of approach	- 76% VOR, 17% ILS, 5% back course ILS, 2% ADF

In addition to the elimination, or reduction, of circling approaches, RNAV offers the further advantage of positive orientation with respect to the landing runway. This feature can be used as a backup monitor for ILS, back course and ASR approaches.

3D RNAV also offers the potential for increased safety through more precise altitude control and orientation. Although 3D accuracies are such that no reduction in MDA can be expected (see Section 4.5), distinct advantages can be gained through its use. A non-precision approach is basically "unstabilized". The aircraft is configured for initial descent, reconfigured to maintain MDA, and reconfigured again for final descent. By using 3D RNAV for vertical guidance monitoring from final approach fix to the runway, the approach configuration can be stabilized and, in effect, MDA becomes decision height. Extreme care must be exercised, however, in establishing and entering 3D way-points. Flight test and simulation results have indicated that due to increased requirements for data entry in some systems blunder potential is increased.

RNAV approach procedures also can provide for separation between V/STOL and conventional (CTOL) traffic. In some cases, separate non-conflicting approach paths to the touchdown point on STOL runways can be provided. This technique has been used by Airtransit Canada in their STOL demonstration project

coordinated by the Canadian Ministry of Transport [55]. Routes between Ottawa and Montreal were structured to achieve minimum block times, consistent with minimum conflict with CTOL traffic. RNAV procedures are utilized to maintain segregation from CTOL traffic and to intercept MLS type approach facilities while satisfying obstacle clearance limits, and noise abatement requirements.

RNAV procedures have also been utilized to develop conflict-free helicopter approaches to the downtown-Manhattan heliport. The "copter RNAV" procedure provides an approach to a "point in space" where transition to helicopter special VFR procedures is accomplished. An example approach plate is given in Figure 5.2.

A rather obvious but significant benefit available from RNAV approach procedures is the distance saved in transitioning from terminal area arrival fix to initial approach fix. The flexibility available with RNAV allows placement of the initial approach fix along the arrival path. An example is given in Figure 5.3 for arrival to Alameda County airport near Albuquerque, New Mexico. An aircraft arriving from the north can accomplish a straight-in RNAV approach to runway 17 with 34 miles less distance traveled than if a VORTAC circling approach is required. No attempt was made to estimate the total distance saved annually, but many airports can be expected to exhibit similar characteristics.

VFR noise abatement procedures exist at many airports which cannot be flown under IFR conditions. RNAV offers the potential for IFR noise abatement procedures at some airports similar to those commonly associated with eventual MLS implementation. As an example (taken from reference 56), consider the current visual approach procedure to runway 13 at New York's LaGuardia Airport. The procedure involves a flight path over the Hudson River to intercept the ILS course 5 nm from the runway threshold (Figure 5.4). The procedure serves to concentrate the noise over the River during the initial phase of the approach. Relative to the straight-in ILS approach to runway 13, the noise abatement potential for this visual approach is significant. Along the ILS approach the aircraft is below 2000 feet (and the noise perceived on the ground is significant) along the last 7 nm of the final approach. For the visual (noise abatement) approach nearly 30% of these last 7 miles would be flown over water, therefore reducing the degree of noise exposure in comparison to the straight-in ILS approach. The visual approach is very nearly optimum in terms of the noise abatement potential on runway 18. Even with the broad coverage capability of MLS, any further reduction in the noise exposure would not be possible because of Harlem's proximity to the LaGuardia Airport.

The minima for this approach require a 3200 ft. ceiling and a visibility of 5 nm. If an RNAV approach duplicating the ground track of the visual procedure were established, then the minima could be reduced to those for the ILS approach (200 ft. and .5 nm) and the noise abatement procedure would thus apply under conditions when instrument flight rules prevail. An appropriate RNAV approach procedure, shown in Figure 5.5, involves 3 waypoints directing the flight path of arriving aircraft over the Hudson River to the interception of the ILS localizer and glideslope approximately 5 nm from the runway threshold. The broken lines either side of the nominal flight path

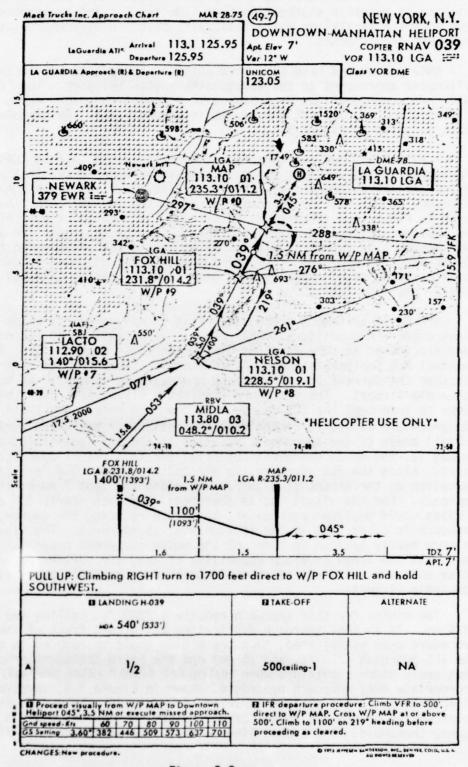
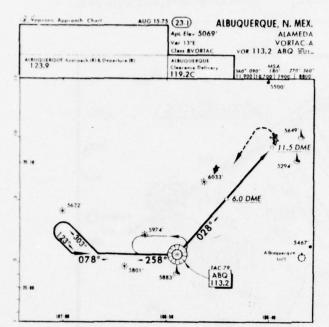
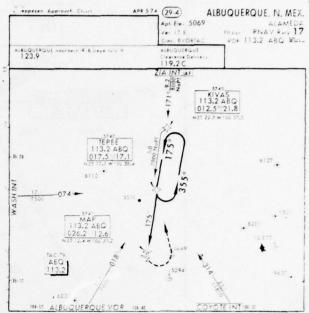
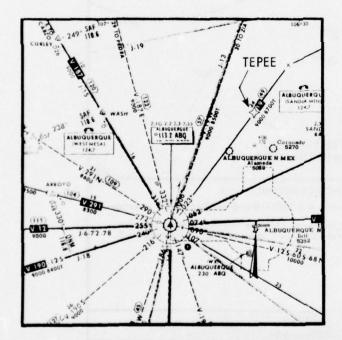


Figure 5.2

Copter RNAV Approach







VORTAC - A (FROM NORTH	H) (RWY 17)
ZIA - ABQ	30.0 nm
Procedure turn	8.0
ABQ - AIRPORT	11.5
CIRCLE TO LAND	2.0
TOTAL DISTANCE	53.5 nm
RNAV (FROM NORTH) (RW	Y 17)
ZIA - TEPEE	14.2 nm
TEPEE - AIRPORT	5.0
TOTAL DISTANCE	19.2 nm
DISTANCE SAVED	34.3 nm

Figure 5.3
RNAV Approach Distance Savings
5-11

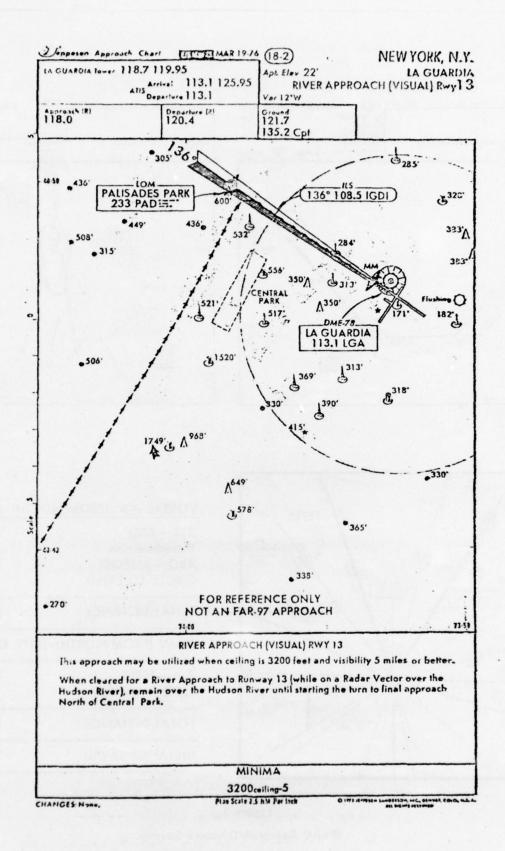


Figure 5.4 Hudson River Visual Approach to LGA-Runway 13

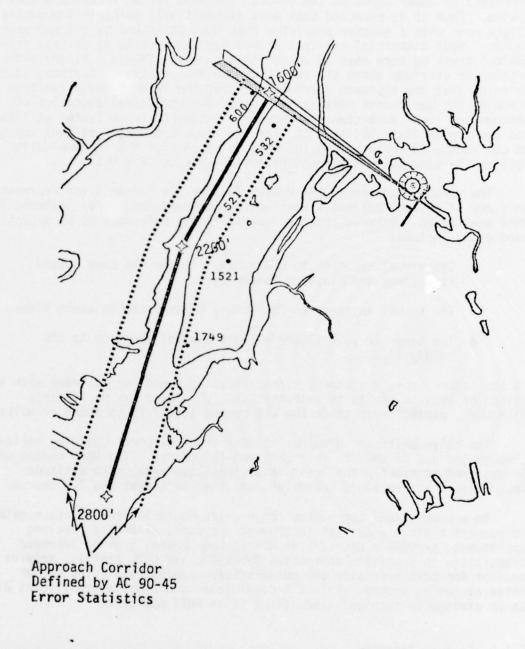


Figure 5.5 RNAV IFR Noise Abatement to LGA Runway 13

indicate the approach corridor defined by the maximum error statistics of AC 90-45A. Again it should be emphasized that the AC 90-45A statistics represent an upper bound on the errors permitted for an acceptable RNAV system. Thus it is expected that most aircraft will maintain the nominal flight path with a greater precision than that indicated by the approach corridor bounds. Most commercial carriers should not be expected to deviate from the nominal track by more than .3 nm and thus should maintain a flight path within the corridor above the course of the Hudson River. Further, it should be noted that the approach corridor defining the lateral deviation from track of the worst-case RNAV system avoids the structural obstacles of midtown and lower Manhattan, particularly the World Trade Center at 1749 ft. and the Empire State Building at 1521 ft. Thus the suggested RNAV approach to LGA runway 13 could be applied well in advance of MLS implementation to realize the same noise abatement potential expected for MLS.

The visual approach to LGA runway 13 over the Hudson River represents only one of several current visual noise abatement procedures suitable for RNAV adaptation. Other approaches identified by Reference 56 as potential candidates include:

- The visual approach to LGA runway 31 over the Long Island Expressway and Flushing Meadow Park.
- The Visual approach to PHL runway 9R over the Delaware River
- The Canarsie LDIN (lead-in light system) approach to JFK runway 13L.

In the latter case, the short, curved final approach leg achieved with the system of lead-in lights to intercept the ILS course may be feasible with RNAV, particularly since the ILS course itself is so short (2 miles).

The noise abatement problems at many other airports could be relieved through the use of RNAV to intercept the ILS course. The RNAV routes would be designed to overfly the least noise-sensitive area, with altitude restrictions set to avoid obstacles, and then intercept the ILS course.

In summary, RNAV approaches can provide obstacle-free approach paths to runways without a current instrument approach, eliminate circling approaches, provide a decrease in MDA in some cases, provide improved transitions to precision approaches for VSTOL and CTOL traffic, provide a monitor for both precision and non-precision approaches, allow use of noise abatement procedures in IFR conditions, and provide significant distance savings in terminal transitions to an RNAV approach.

5.1.2 Enroute Benefits

5.1.2.1 Route Length Effects

The analysis reported in Reference 12 was based on preliminary results of the NAFEC high altitude route structure analysis [4] and included flight mile fuel and time differences between charted RNAV, direct, and charted VOR

traffic due to route length differences and due to differences in number and types of conflicts and conflict resolution techniques. Subsequent efforts included the design and analysis of a 429 airport pair structure, expanded from the 185 airport pair structure upon which the preliminary analysis was based, and including consideration of weather routes and restricted areas in the high altitude enroute structure. The high altitude route analysis is described in Section 3.2 and results of the updated conflict analysis are discussed in Section 5.1.2.2.

Table 5.5 Average RNAV Route Length Savings over VOR in High Altitude Enroute Structure

	No. of 185	airport pairs in structure 429		
		no wind	including weather routes	including restricted areas
Average VOR flight length (nm) Average RNAV flight length (nm)		475.89		
Average Direct flight length (nm)				
RNAV structure advantage \langle over		2.35%	1.79%	1.61%
Direct structure advantage VOR	3.06%	2.69%	2.13%	1.95%

Table 5.5 gives the results of the flight mile analysis of the 429 airport pair high altitude route structure compared with the 185 airport pairstructures used in the previous analysis. Each route length was multiplied by the twenty-four hour traffic demand on that route as defined in the 1969 Peak Day sample. The sum of these flights was then, in each case, divided by the sum of the traffic demand between the selected airport pairs to yield the average expected route length to be flown by an aircraft at random. The calibration of the no-wind analysis to include weather routes and restricted areas is described in Section 3.2. Calibration of the direct structure benefits was accomplished in the same manner, yielding the savings indicated in Table 5.5. The traffic demand on the 429 airport pair structure represented about 67% of total traffic, compared with about 41% of the total on the 185 airport pair structure. The 429 airport pair structure represents a fairly accurate model of a total structure, however, since it will accommodate approximately 92% of the total traffic with minor extensions to include traffic from airports in close proximity to routes on the structure.

Similar results were obtained in the low altitude route length analysis described in Section 3.3 and summarized in Table 5.6. The direct structure benefit in Table 5.6 was derived by a comparison of VOR and great circle flight miles in Table 3.21.

Table 5.6 Average RNAV Route Length Savings over VOR in Low Altitude Enroute Structure

Average VOR flight length (nm)	160.76
Average RNAV flight length (nm)	156.96
Average Direct flight length (nm)	154.96
RNAV structure advantage over VOR	2.36%
Direct structure advantage over VOR	4.17%

The benefits shown in Tables 5.5 and 5.6 are applicable to an average route length to be flown by an aircraft at random in a fully developed RNAV route structure, and the high altitude results are somewhat conservative since the NAFEC structure used as a basis for the analysis is not an optimum structure. However, they are also representative of the route length savings available on typical RNAV routes as they are implemented. The direct structure savings are also available now, depending upon controller workload and radar monitoring. Wide application of preplanned direct flights will be dependent upon implementation of improvements in the enroute automation systems and on extensive flight checking for area NAVAID coverage.

The estimated savings in fuel and time which will result from shorter RNAV route lengths may be computed as follows:

	S	=	$a \cdot \Delta l \cdot n$ (5.1)
where,	S	=	annual savings in 1bs or minutes in 1984
	a	=	aircraft mix weighted average fuel in 1b per mile or time in minutes per mile
	Δ1	=	<pre>average miles saved per departure = % saved x weighted average stage length less 110 miles (terminal area distance [12])</pre>
	n	=	number of departures in 1984

The fleet mix of air carrier jet aircraft in 1984 was estimated from data provided by the Office of Aviation Policy of the FAA, as indicated in Table 5.7. The aircraft type weighted average values for "a" and for stage lengths in Equation (5.1) for air carrier aircraft were then derived from data in the Aircraft Cost and Performance Report [37] for CY1973. For air carrier aircraft, a = 25.2 lb/mile and 0.14 minutes/mile, and average stage length = 602 miles. Estimated total number of air carrier departures in 1984 is 6.3 million [21].

Table 5.7 Estimated Fleet Mix of Air Carrier Jet Aircraft in 1984

Aircraft Type	% of Total
4 engine wide body	5%
3 engine wide body	16%
4 engine regular body	4%
3 engine regular body	39%
2 engine regular body	36%

The 1972 mix of the business aircraft fleet was provided by the NBAA, as discussed in Section 5.1.1.1, and it was assumed that the 1984 mix would remain the same. Average fuel consumption was also obtained from NBAA data, and when combined with assumed average speeds of 450 mph for jets, 225 mph for turboprops and 120 mph for piston aircraft, resulted in estimates for "a" in equation (5.1) of 4.9 lb/mile and 0.13 minutes/mile for business jets. 3.3 lb/mile and 0.27 minutes/mile for business turboprops. Average stage lengths were also estimated from NBAA data as 514 miles for jets and 323 miles for turboprops. For all piston aircraft, "a" was estimated to be 0.8 lb/mile and 0.5 minutes per mile, with an average stage length of 226 miles. Total general aviation departures in 1984 were estimated from data in Reference 54 to be 6.5 million, of which 4.85% are business jet and turboprops, 14.75% are business piston aircraft, and 80.4% are other general aviation aircraft. Total estimated enroute fuel and time savings are summarized in Table 5.8. All air carrier and business jets and turboprops were assumed to fly in the high altitude structure and realize savings of 1.61% for charted RNAV and 1.95% for direct. The results are therefore conservative, since the portion of those flights which actually occur in the low altitude structure would save a larger percentage of route length. All piston aircraft were assumed to operate in the low altitude structure and realize savings of 2.36% for charted RNAV and 4.17% for preplanned direct.

Table 5.8 Total Fuel and Time Savings Over VOR in High and Low Altitude Enroute Structures in 1984

	Charted	RNAV	Preplanned Direct		
	fuel (1000 lb)	time (1000 min)	fuel (1000 1b)	time (1000 min)	
Air Carrier	1,097,320	6,108	1,329,000	7,415	
business jets & turboprops	9,008	420	10,907	510	
business piston	3,686	2,337	6,588	4,118	
Total business aircraft	12,694	2,757	17,495	4,628	
Other general aviation aircraft	22,300	13,900	39,400	24,600	
TOTAL	1,132,314	22,765	1,385,895	36,643	

5.1.2.2 Conflict Effects

The 185 and 429 airport pair charted high altitude RNAV route structures were subjected to fast time simulation to determine the effect of RNAV on system capacity and to generate data which could be used in a subsequent analysis to estimate the fuel and time impact of RNAV structure conflicts vs.

VOR structure conflicts, and to determine the impact on controller communications workload of various levels of VOR/RNAV traffic mixes. An analysis of the impact of conflicts on user fuel and time was performed utilizing the data from the initial fast time simulations of the 185 airport pair structure [12]. All conflicts which occurred during the simulation were solved in a post-simulation analysis through application of a conflict resolution logic model and the aggregate fuel and time used the resolution of the conflicts in a 100% RNAV traffic sample were compared with the aggregate fuel and time used in the resolution of the conflicts in a 100% VOR traffic sample. The results, when extrapolated to a national scale, indicated that a fuel benefit of 0.05% could be expected with 100% RNAV traffic compared with VOR. The difference in the time estimates was negligible. Since the conflict effects were an order of magnitude smaller than route length effects, they were not included in the computation of total annual benefits in both Reference 12 and this report.

5.1.2.3 Altitude Assignments.

The fuel savings due to conflict resolution differences included the impact of altitude restrictions during climb. Significantly fewer altitude restrictions were used in solving RNAV conflicts, primarily due to the convenience of the parallel offset maneuver. The availability of parallel routes, with little effect on conflict penalties, will both increase capacity and provide for more optimum altitude assignments in addition to the reduction in altitude restrictions during climbs. A rigorous analysis of the expected improvement in the distribution of altitude assignments in an RNAV environment is beyond the scope of this study. However, some insight may be gained into the magnitude of savings available by examination of several individual aircraft performance characteristics. Table 5.9 lists typical altitude related fuel penalties for several aircraft.

Table 5.9 Altitude Restriction Fuel Penalties

A/C Type	DC-8-63	DC-9-30	B-727	B-747
Flight Level For Zero Penalty	350	350	350	350
Mach No.	.80	.77	.80	.84
Gross Weight (x1000 lb.)	260	90	140	600
Flight Level	Pena 1	ty - Pounds	of Fuel	per Minute
100 130 150 200 250 270 290 310 330	116 104 93 74 39 28 17 9	58 49 33 27 25 17 12 7 0	90 85 79 51 26 20 13 5	183 171 159 140 83 50 18 0

It can be seen that a difference of one Eastbound or Westbound flight level can have a significant effect on fuel consumption (wind and other factors being constant).

5.1.3 Annual User Benefits

The fuel and time savings, comparing RNAV over VOR, for both terminal area and enroute operation were summarized in Sections 5.1.1 and 5.1.2 for three classes of users: air carrier, business and other general aviation operators. In this section estimates are made of the dollar value of those annual fuel and time savings.

5.1.3.1 Air Carrier Annual Benefits

Utilizing data from the Aircraft Cost and Performance Report [37] and the aircraft fleet mix derived in Section 5.1.2, the average cost of a minute of delay was estimated. The cost of fuel, oil, depreciation, and rentals was subtracted from total operating cost to obtain a flight-time sensitive (less fuel) cost of delay. The fleet mix weighted average was \$8.40 per minute for CY1973. This cost was then inflated by 14.5% to arrive at an estimate of \$9.62 per minute for average delay cost, less fuel, in 1975 dollars. The average 1975 airline jet fuel cost was estimated by inflating average fuel cost reported in the last quarter 1974 operating cost data for the 747, DC-10 and L1011 [57] by 10%. Fuel cost in 1975 was therefore conservatively estimated at \$.23 per gallon, or \$.036 per pound. These costs were applied to the terminal area 2D and 3D, and the enroute fuel and time savings. The air carrier estimates of cost of delay which were used in the 4D analysis of Section 3.4 were based on 1973 dollars, and included a mix of ground and airborne ATC induced delay. The fuel savings through 4D in M & S terminals were estimated in Section 5.1.1. Using an average 1973 fuel cost of \$.13 per gallon [57], the time portion of the total savings of \$9.46 per minute was estimated to be \$7.25 in 1973 dollars, or \$8.30 in 1975 dollars.

Total annual air carrier savings are summarized in Table 5.10 for 1984 traffic levels in 1975 dollars. The passenger time savings given in Table 5.10 are based on average number of revenue passengers for each aircraft type [58] and an assumed value of passenger time of \$12.50 per hour [59].

Table 5.10 Annual Air Carrier Jet Aircraft Savings RNAV over VOR at 1984 Traffic Levels (1975 dollars)

	Annual Savings in Millions of Dollars					
	Aircraft Fuel	Aircraft Time	Sub-Total	Passenger Time	Total	
Terminal Area - 2D Terminal Area-3D descent	35.9 17.2	55.2 31.9	91.1 49.1	93.5 45.8	184.6 94.9	
4D in M&S Terminal enroute	35.3 39.5	75.1 58.8	110.4 98.3	142.1 95.9	252.5 194.2	
Total	127.9	221.0	348.9	377.3	726.2	

The terminal area savings are larger than those estimated in the previous study [12] due to an improvement in the terminal area designs [2], coupled with the fact that the current estimate is based on 60 major airports compared with only 8 airports (6 terminal areas) in the previous study. The enroute savings are also much larger than those estimated in Reference 12 due to the following factors: 1) average total cost per mile increased from \$1.59 to \$2.23 (40%); 2) average enroute route length increased from 407 to 492 nm (21%); 3) average number of high altitude departures only for 1982, (3.6 million) were used in Reference 12, where total high and low altitude savings were estimated in this study (6.3 million air carrier departures in 1984), resulting in a further increase of 75%.

5.1.3.2 Business Aircraft Annual Benefits

The cost of jet fuel and aviation gasoline for business (and general aviation) operators was estimated from information obtained from a telephone survey of Fixed Base Operators in July, 1975. The estimated 1975 cost is \$.70 per gal. for jet fuel and \$.72 per gallon for aviation gasoline, or approximately \$.12 per pound for each. The cost of delay was conservatively estimated to consist only of maintenance cost. The average maintenance cost was derived from operating cost data provided by the NBAA. The average delay cost (less fuel) for all jet, turboprop, and piston business aircraft was estimated to be \$.15 per minute in 1975 dollars. Table 5.11 summarizes annual savings to business aircraft operators in 1984 through RNAV compared with VOR. Savings due to pilot-selected 3D descents in the terminal area can be substantial for individual jet aircraft but the total savings were not estimated, since less than 5% of business aircraft are jets. The passenger time savings given in Table 5.11 are based on average passenger loads for various business aircraft from NBAA data, and on an assumed value of passenger time of \$12.50 per hour [59]. Although some companies utilize a "value added" concept to compute values of executive time which can range as high as \$40 per minute, no attempt was made to quantify these additional passenger time benefits.

Table 5.11 Annual Business Aircraft Savings RNAV over VOR at 1984 Traffic Levels (1975 dollars)

STATE OF STA	Annual Savings in Millions of Dollars					
	Aircraft . Fuel	Äircraft Time	Sub-Total	Pass. Time	Total	
Terminal Area - 2D	. 46	.17	.63	.87	1.50	
Enroute	1.52	.42	1.94	2.17	4.11	
Total	1.98	.59	2.57	3.04	5.61	

5.1.3.3 Other General Aviation Annual Benefits

Average overhaul costs for general aviation aircraft were estimated from a 1972 Aircraft Price Digest [60], updated by a telephone survey of Fixed Base Operators. The estimates ranged from \$.14 minute of flight time for light twins to \$.02 per minute for single engine, four place aircraft. The fleet weighted average overhaul cost for general aviation aircraft in 1975 dollars was estimated to be \$.03 per minute. Routine maintenance and calendar inspections were not included. The fuel cost was estimated at \$.12 per pound as in the case of business aircraft.

A summary of annual savings for non-business general aviation aircraft is given in Table 5.12. A passenger time value was not considered appropriate for pleasure flying and was not computed.

Table 5.12 Annual Non-Business General Aviation Aircraft Savings RNAV over VOR at 1984 Traffic Levels (1975 dollars)

	Annual Savings in Millions of Dollars			
	Fuel	Time	Total	
Terminal Area - 2D Enroute	1.07 2.68	.17 .42	1.24 3.10	
Total	3.75	.59	4.34	

5.1.3.4 Total Annual User Benefits

Table 5.13 summarizes total annual user benefits for air carrier, business, and other general aviation aircraft.

Table 5.13 Total Annual User Savings RNAV over VOR at 1984 Traffic Levels (1975 dollars)

	Annual Savings in Millions of Dollars					
	Aircraft Fuel	Aircraft Time	Sub-Total	Pass. Time	Total	
Air Carrier	127.9	221.0	348.9	377.3	726.2	
Business	2.0	0.6	2.6	3.0	5.6	
Other Gen.Aviation	3.7	0.6	4.3		4.3	
Total	133.6	222.2	355.8	380.3	736.1	

The savings summarized in Table 5.13 are predicated on a charted RNAV structure enroute. Although the savings available in a preplanned direct enroute environment are 21% greater than in a charted RNAV structure, the total RNAV savings in 1984, including terminal and enroute, are only 6% greater for the total RNAV environment which utilizes all preplanned direct in lieu of charted RNAV enroute.

5.1.4 User Equipment Impact

The levels of RNAV capability and the airborne equipment costs necessary to obtain each level were analyzed in Section 3.5. Systems which meet the minimum operational characteristics [7] will range in price from \$3000 to \$100,000. However, the 2D RNAV savings desribed in Section 5.1.3 are available to any aircraft equipped with the most basic system. This section discusses the cost of equipping compared with the benefits to be realized for each user group.

5.1.4.1 Air Carrier Equipment Impact

The savings available over VOR to the estimated 3100 U.S. air carrier jet aircraft in 1984 are \$203 million per year for 2D RNAV (terminal plus enroute) and an additional savings of \$49 million per year for 3D descents and \$110 million for 4D in M & S terminals. The 2D savings alone will support an average amortization of airborne equipment cost of \$65,000 per year per aircraft. Addition of 3D and 4D capability, which are relatively small increments of the basic cost of an airline quality RNAV system, will increase the available payback per year per aircraft by \$16,000 for 3D and by \$36,000 for 4D.

5.1.4.2 Business Aircraft Equipment Impact

The total annual business aircraft savings of \$2.57 million derived in Section 5.1.3.2 is the maximum possible savings, based on all aircraft being equipped with RNAV. If only those aircraft are equipped which would be required to have RNAV, for operation in the high altitude structure and at high and medium density hub airports (see Section 3.5), then only 11,000 of the projected 67,000 business aircraft in 1984 will accrue benefits, and the total savings will be reduced proportionately. The total cost to equip these 11,000 aircraft with RNAV is \$66.4 million, or an average of \$6000 per aircraft. Based on the data presented in Table 5.10, an average business aircraft can expect to save approximately 3% of annual operating cost (fuel and maintenance) with RNAV. (This estimate was derived by inflating the 1.61% and 2.36% enroute distance savings to include terminal area savings.) Since annual dollar savings are dependent upon annual operating cost and independent of RNAV equipment cost, payback will be much faster for higher performance aircraft. Estimated annual savings for an average business aircraft in each of four categories is given below:

Estimated annual savings due to 2D RNAV (fuel and maintenance only)

Type Aircraft

jet
turboprop
piston multi-engine
piston single-engine

It can be seen that a reasonable payback period can be projected on the basis of fuel and maintenance cost savings alone for jets and turboprops. However, the value of passenger time must be considered to obtain a reasonable payback period for piston aircraft. Piston business aircraft flew an average of 257 hours in 1973 according to NBAA data. Projected additional savings would average 8 aircraft hours at an average passenger load factor of 3+. Even conservative estimates of the value of executive time would result in substantial additional savings and provide a reasonable payback for the cost of RNAV equipment.

5.1.4.3 Non-Business General Aviation Equipment Impact

As in the case of business aircraft, the total annual general aviation savings of \$4.46 million per year represent the maximum available if all aircraft were RNAV equipped. With the TCA concept described in Section 3.5, only 7,500 of the estimated 145,000 non-business general aviation aircraft in 1984 would be required to have RNAV. The cost to equip these 7,500 aircraft is \$37.3 million, or an average of \$5,000 per aircraft. It should be noted that more than 4000 of these aircraft do not have DME, and that the DME cost is included in the average.

As was discussed in Section 5.1.4.2, a reasonable payback period for RNAV equipment for small piston aircraft cannot be projected on savings in fuel and maintenance cost alone. However, as evidenced by the current popularity of RNAV equipment among general aviation users (over 4500 equipped), there are other benefits available which are difficult to quantify. Among these benefits are its use as a VFR pilotage aid, elimination of circling approaches, and access to airports which do not currently have a VOR approach. The tendency of general aviation operators to purchase navigation equipment to obtain additional capability even when not required is evidenced by the fact that more than 25,000 general aviation aircraft are equipped with DME. It is therefore reasonable to anticipate that many more than the 5% of non-business general aviation aircraft which would be required to have RNAV under the TCA concept would actually equip.

5.1.4.4 Impact on Military Aircraft

The Department of Defense operates more than 27,000 aircraft, making them one of the largest of the users of the National Airspace System. In the DOD comments on the Task Force Report [61], it was pointed out that any program to implement RNAV on a mandatory basis must consider the extensive

effort and expenditures required to equip military aircraft. A very limited number of military aircraft have a navigation capability that would be directly usable for the area navigation concept recommended by the Task Force, and a minimum of seven years lead time would be required, after approval of the RNAV concept, to include the necessary funds in the Federal Budget, to procure equipment, and to complete all aircraft modifications.

The impact of RNAV implementation on military operations and equipment requirements will be largely dependent upon the implementation concept and timing. The questions which military planners must answer include the following:

- 1) How many aircraft (and what type) have a requirement to fly in the national airspace system?
 - a) High altitude structure (FL180-FL450)
 - b) Low altitude structure
 - c) High density terminal areas
 - d) Medium density terminal areas
- 2) What is current and planned navigation capability of each type?
- 3) Which aircraft would be required to have RNAV capability? When?
- 4) What aircraft could attain RNAV capability through modification of existing or planned navigation equipment?
 - a) Software modification only
 - b) Minor mechanical modification
 - c) Major modification
- 5) What are flight test requirements?
 - a) for certification to current standards (AC90-45A)
 - for certification of other than VOR/DME or TACAN based navigation equipment
- 6) What equipment or procedural requirements are assumed which, if relaxed, would have a major impact on either 4) or 5) above? What relaxation of requirements would be necessary?
- 7) What are overall retrofit requirements?
 - a) Schedule
 - b) Training and support requirements
 - c) Funding requirements
- 8) What benefits to military operations could be realized through the use of RNAV?

The potential impact on military operations and equipment requirements caused by the implementation of RNAV varies widely among the services. In the following paragraphs, a brief assessment is made of current navigation capability of each military service, plus the U.S. Coast Guard, as it applies to Area Navigation.

The Air Force flies more than eighty aircraft types, of which approximately twenty types are equipped with inertial systems and twenty types are equipped with doppler navigation systems. With the exception of the VIP fleet and the Military Airlift Command transport aircraft, very few of these systems have either the necessary long range accuracy or the capability for VORTAC update to accommodate RNAV operations. The majority of Air Force tactical aircraft are equipped with both VOR and Tacan. However, they are extremely space limited and would require retrofit with newly developed dual-purpose equipment to accommodate both primary mission requirements and navigation capability in an RNAV environment. Table 5.14 presents a summary of Air Force aircraft navigation capability.

In many ways, Army aviation operations in the National Airspace System are similar to those of general aviation. Aircraft types range from light utility helicopters to medium size turboprops. The aircraft are broadly dispersed geographically, and flights range from a variety of tactical missions to VFR and IFR operation in the low altitude structure. Only four percent of the approximately 10,000 Army aircraft even fly in the high altitude structure (OV-1, UX and a few U-21s). It is estimated, based on conversations with Army personnel, that a like percentage (3-5%) of the total inventory will ever operate in Group 1 and Group II TCAs. A listing of Army aircraft and current and planned navigation capability is given in Table 5.15.

Of the estimated 375 aircraft which would be required to have RNAV capability because of operation in the high altitude structure (OV-1 and UX), the 225 OV-1 aircraft have dead reckoning navigation systems which could be adapted for RNAV use. The 120 UX aircraft are equipped with a Tacan, and therefore would need only an RNAV computer. If the other aircraft were required to have RNAV capability, primarily due to operation in high and medium density terminals, an additional 1500, plus the UTTAS and AAH, could possibly utilize Loran as an RNAV system. Of the remaining aircraft, only the U-8 and U-21 (361 aircraft) will have VOR/DME or Tacan and therefore require only an RNAV computer. The remaining 8000 aircraft would require a DME as well, and 2000 of those would also require VOR (or Tacan).

Fifty percent of the approximately 7000 Navy aircraft are equipped with Tacan which is suitable for conversion to RNAV capability and all of the long range Navy aircraft are equipped, or will be equipped with Loran-C. The P3B, P3C, A7D, A7E, S3 and F14 fleets are equipped with inertial navigation systems which are adaptable to RNAV operation. The remaining aircraft could not be equipped for RNAV operation without a massive equipment development and retrofit program. Aircraft such as the A4, F4 and A6 are extremely space limited and RNAV capability could not be provided through addition of separate RNAV equipment.

Table 5.14 USAF Aircraft Navigation Equipment

				1		Iner	ial	
Aircroft	Tacan	VOR	DME	C/D	Отеда	Туре	Accuracy	Doppler
A-7D	×			1		ASN-90	.74	APN-190
A-10	×	×						7.11.
A-37	×	×						
A-IE	×	x						
8-1	×			1		ASN-101	.10	APN-185
B -52B	×	×						
B-52C	×	×						APN-89
B-52D/E/F	×	×						APN-108
B-52G/H	×	×				LN-15A	.50	APN-89A
FB-111	×					AJN-16	.50	APN-185
B-57	×	×						
C-5A	×	×		×		NIS-105	.80	Nortronics
C-7A	×	×						
C-9A	×	×	×					
C-47	×							
KC-97	×	×		1				
VC/C-118A	×	×						
C-119	×	×						
C-123	×	×						
C-124	×	×						APN-147
C-130	×	×		some	planned			APN-147
C-131	×	×						
C-135	×	×				planned	<1.0	
C-140	×	×						NC-103
VC-137	×			×		LTN-51	1.0	
C-141						planned	<1.0	APN-147
EC-135	×							APN-81
EC-137	×							
EC-121	×	×		some				APN-153
EB-66 F-4	×	×				10	2.0	APN-82
F-5E	×	×		planned		LN-12	3.0	
F-15	×					LN-33 LN-31	1.0	
F-100/101/	×	1				F14-21		
102/106	×	1						
F-105	×			some		ASN-100		
F-111A/E	×			some		LN-14	2.0	
F-111D/F	^					LN-16	0.5	
F-104	×					LN-12	3.0	
RF-4C	-	1		1		- X	0.0	
RF-101	×					^		
RF-5A	×							
T-29	×	×		1				
T-33	×	x						
1-37	-	×	×	1				
T-38	×		-	1				
T-39	×	1						
T-41		×						
T-43		×		1	MATERIAL PROPERTY.	19.00		
U-3		×		1				
U-4		×					4 1111	
U-6		×						
U-10		×					15 111	
U-17		×						
H-3	×	×						APN-175
H-53		×		-				APN-175
H-1	×	×						
0-2	×	×						
OV-10	×	×						

Table 5.15 Army Aircraft Navigation Capability

Туре			Current			Planned
	Tacon	VOR	Doppler	INS	Loran	
OV-1	X	X	X	X		
T-41						
T-42		X				
U-1	76	X				
U-6		X				
U-8		X				+ DME
U-10		X X X X X				
U-21	X	X				
AH-1	State Inc.					+ Doppler
CH-34		X X				
CH-47		X	- 300			+ Loran
CH-54		X				
OH-6					1	+ VOR (50%)
OH-13						
OH-23						
OH-58	Landar I	v				+ VOR (50%)
TH-13		X				
TH-55 UH-1		_				
UTTAS		X			V	+ Loran (10%
AAH	19 V 19	^			X	
UX	×	×			×	

The impact on the Coast Guard would be nominal since all USCG aircraft have relatively sophisticated multi-sensor navigation systems designed for logistics and search and rescue missions.

The current and planned Coast Guard aircraft fleet and navigation capability is listed in Table 5.16.

Table 5.16 USCG Aircraft

Aircraft Type	VOR	DME	TACAN	INS	Loran A	Loran C	Nav. Comp.
HU-16E	X	Х	X		X		
HH-52A	X		X				
HH-3F	X		X		X	X	AYN-1
HC-130B	X	X	X	ARINC-561	X	X	ARINC-161
HC-130H	X	X	X	ARINC-561	X		AYN-1
VC-4	X	X					
VC-11	X	X		ARINC-561			ARINC-561
MRS	X	X		X			Mk-2

The HU-16E (Grumman Albatross) fleet is in the process of being phased out and will not be affected by an RNAV requirement. The fleet of 94 HH-52A helicopters are equipped with the necessary sensors but would require the addition of RNAV equipment. The HH-3F helicopters are currently equipped with the AYN-1 navigation computer which can be adapted for RNAV use through development of appropriate procedures. The HC-130 B/H aircraft are equipped with ARINC-561 inertial systems which can be adapted for DME update. The single VC-4 (Gulfstream I) will require the addition of RNAV equipment, and the VC-11 (Gulfstream II) is equipped with ARINC-561 inertial. The new Medium Range Search aircraft (MRS) fleet will be equipped with ARINC-582 area navigation equipment. It is anticipated that the use of RNAV procedures will allow establishment of discrete tactical arrival and departure routes in busy terminal areas which will enhance the quick response of search and rescue missions with procedural separation from other traffic.

Many tactical aircraft are only occasional users of the National Airspace System and therefore would accrue the benefits of RNAV at a slower rate than the constant users. However, the military constitutes in total a very large segment of the users of the National Airspace System, and are in general the only users whose primary mission requirement is not the ability to navigate within a specific airways structure. The preplanned direct operational concept described in Section 5.4, which allows the use of VORTACs as waypoints, could form the basis for a transitional implementation requirement for these military aircraft with a secondary mission requirement to navigate in the NAS, with an ultimate requirement that all military aircraft conform to the same equipment and procedural standards as other users of the NAS when user demand is sufficient to warrant proceding to the second and third implementation phases. The ability to navigate on a "VORTAC direct" basis during the transition period in both the high and low altitude RNAV structures would then allow military aircraft access to the National Airspace System with no reduction in operational flexibility under that currently available in the VOR airways structure. Normal operation could therefore continue during the seven year period which DOD has estimated as being required to obtain full RNAV capability. The Air Force, Navy and Coast Guard have a requirement for virtually all their aircraft to fly in the National Airspace System. The Army requirement is much less severe - probably less than 5% of their aircraft would be required to have RNAV equipment.

5.1.5 Transition Period User Benefits

The user benefits described in Sections 5.1.1, 5.1.2, and 5.1.3 are applicable to an all-RNAV environment in 1984. The realization of benefits by individual users, however, is not dependent upon an all-RNAV environment. The implementation concept described in Section 5.4 is based on providing enroute and terminal RNAV routes on a gradual basis in accordance with user demand. An example of a potential RNAV implementation scenario is given in Table 5.17. If it is assumed that 2D and 4D terminal area benefits are available to equipped aircraft as RNAV is implemented in the terminal areas, that 3D and enroute benefits are available as aircraft are equipped with RNAV on a schedule which is planned to correspond with the availability of routes, the annual air carrier transition period savings, derived from an analysis described in Reference 14, are as indicated in Table 5.17.

SYSTEMS CONTROL INC PELO ALTO CALIF
IMPLEMENTATION OF AREA NAVIGATION IN THE NATIONAL AIRSPACE SYST--ETC(U)
DEC 76 W H CLARK, E H BOLZ, H L SOLOMON DOT-FA72WA-3098
FAA-RD-76-196 AD-A039 225 UNCLASSIFIED 4 95 AD AO39225

Table 5.17a Sample RNAV Implementation Scenario

Termin	al Area	RNAV Des	ign Impl	ementati	on by Ca	Lendar Year:
1982	1983	1984	1985	1986	1987	1988
EWR*	BOS*	TPA*	CVG*	BUF	ORF	ВНМ
LGA*	DFW*	CLE*	MEM	CLT	OKC	PVD
JFK*	DEN*	SEA*	PHX	IND	DAY	DSM
ORD*	FLL*	DTW*	SAN	SLC	OMA	ELP
LAX*	IAD	MSY*	MCO	MKE	JAX	RDU
ATL*	MIA*	IAH	BAL*	BDL	SAT	TYS
	SFO*	MCI	PDX	CMH	SDF	ALB
	DCA*	PHL*			ROC	TUL
		PIT*			SYR	SMF
		MSP*			ABQ	
		STL*			BNA	
		LAS*				

*4D M&S Implemented starting in 1985

Fleet mix and percent of aircraft equipped with RNAV

	19	982	19	983	19	984	19	985	19	986	19	987	19	988
	*	*	*	%	8	8	*	8	8	8	*	8	8	*
	of	eqp		eqp		eqp		-				-		eqp
	flt		flt		flt		flt		flt		flt		flt	
4Eng W.B.	4.1	33	4.3	67	4.6	100	4.8	100	5.2	100	5.6	100	6.1	100
3Eng W.B.	12.5	35	13.2	68	13.8	100	14.4	100	16.7	100	18.9	100	20.9	100
4 Eng.	10.3	0	9.4	22	8.6	44	7.8	66	7.1	73	6.4	80	5.7	86
3 Eng.	46.2	0	48.1	33	49.9	67	51.4	100	50.4	100	49.5	100	48.6	100
2 Eng.	26.8	0	24.9	30	23.1	61	21.6	91	20.6	93	19.6	95	18.7	96

2D and 4D Terminal Area benefits assumed to be available to equipped aircraft as RNAV is implemented in the terminal areas. 3D and enroute benefits assumed to be available as aircraft are equipped.

Table 5.17b

Annual Air Carrier Transition Period Savings
-in millions of 1976 dollars -

calendar vear	aircraft fuel	aircraft time	sub-total	passenger time	total
1982	6.3	6.0	12.3	9.6	21.9
1983	27.4	40.8	68.2	65.8	134.0
1984	53.6	72.7	126.3	117.5	243.8
1985	105.6	185.7	291.3	300.6	591.9
1986	111.7	209.6	321.3	341.3	662.6
1987	117.6	219.5	337.1	359.8	696.9
1988	123.1	252.8	375.9	406.2	782.1

5.2 SYSTEM IMPACT SUMMARY

The results of several analyses concerning the potential impact of RNAV implementation on the ATC System are reported in Section 4 of this report. In this section, the results of these studies, combined with the results of other related studies, are summarized as they relate to certain key areas of ATC System impact: VORTAC requirements, controller productivity and training, ATC automation, charting and flight inspection, airspace capacity and route development. This section is intended to summarize the overall impact of RNAV on the ATC system in these areas and it is recognized that the specific requirements and impact may differ in individual areas and cases.

5.2.1 VORTAC Station Coverage Requirements

The analysis of VORTAC system implementation costs described in Section 4.2 and 4.3 was intended as a preliminary overall estimate of the VORTAC costs required to support RNAV in the high altitude structure and the savings available through reduction of the number of VORTACs required to support navigation in the high altitude enroute area and in the terminal areas. It was not intended to supplant the requirement for the detailed analysis which should be conducted based on the development of an optimum enroute structure for implementation which is coordinated with detailed terminal area designs for the appropriate high and medium density terminal areas. The analysis is conservative in that the costs were determined relative to the existing VORTAC system, and no estimates were made of additional VORTAC requirements to support an expanded enroute VOR airways system which would meet the same traffic demand. The analysis of terminal area requirements was based on the removal of existing VORTACs. The enroute analysis did not include consideration of the removal of high altitude VORTACs which provide completely redundant coverage. It can be expected that a number of high altitude VORTACs can be eliminated when a detailed analysis is made based on the optimum route structure which is to be implemented, resulting in an annual maintenance savings of approximately \$48,500 for each station removed. Elimination of only 12 VORTACs would provide first year maintenance savings equal to the one time cost of providing high altitude charted RNAV coverage in the 1977 period, and would pay for full CONUS coverage for 1977 in less than four years.

The estimated implementation cost to provide VORTAC coverage for the high altitude enroute structure is given in Table 5.18 for the Task Force recommended route widths for both charted and preplanned direct RNAV. The implementation cost to provide DME/DME coverage is also given. Part of all of the cost in Table 5.18 can be expected to be offset by annual maintenance savings from removal of redundant high altitude VORTACs, which were not considered in this study.

It was determined that the support of RNAV in high and medium density terminal areas would require a minimum of 48 fewer VORTACs than a VOR structure, for an annual maintenance savings of \$2.3 million. One-time cost savings in in the terminal area could also include modernization costs of the 48 removed VORTACs.

Table 5.18 VORTAC Implementation Impact

	COST	S
	VOR/DME	DME/DME
High Altitude Charted RNAV, ±4 nm route width	\$597,300	\$5,244,000
High Altitude Pre-planned Direct RNAV, ±4 nm route width	\$1,959,900	\$5,799,000
High Altitude Pre-planned Direct RNAV, ±2.5 nm route width	\$6,604,000	\$5,799,000

	Annual Main	tenance Savings
	Number of VORTACs	Dollars
Enroute Terminal Area	not determined 48	not determined \$2,328,000

Table 5.19 is taken from the National Aviation System Plan [51], and gives the enroute Navigation Aids F & E Program Plan for 1976-85. It can be seen that the VORTAC implementation costs estimated for the RNAV high altitude structure are nominal when compared with normal planned expenditures for improvement and expansion of the VOR system, which average in excess of \$20 million per year.

5.2.2 Controller Impact

5.2.2.1 Controller Productivity

A major concern of the Task Force was the possibility that air traffic controllers might experience an unacceptable increase in workload during the transition period of mixed VOR/RNAV operations prior to the advent of a predominently RNAV environment. As reported in Reference 12, it was determined that controllers are capable of operating in this mixed environment with reduced workload and no reduction in system capacity, and that significant reductions in controller communications time can be expected as the percent of RNAV traffic increases. These conclusions were supported by analyses based on data from both real-time [9] and fast-time [10] simulations. Additional fast-time simulations utilizing a much larger and more representative RNAV structure indicate similar trends in conflict reduction which were the basis of the previous analyses. Another previous analysis [13] concluded that

Table 5.19

Enroute Navigation Aids F & E Program Plan

						(Amoun	t in r	PLAN (Amount in millions of dollars)	dolla	Ê						Total
Program		1976		1977		1978		1979		1980	4	1976-80	1	1981-85	21	1976-85
	No.	Amt.	No.	Amt.	No.	Amt.	No.	Amt.	No.	Amt.	No.	Amt.	No.	Amt.	No.	Amt.
VORTAC System		6.6		14.8		26.3		24.9		24.9		96.5		95.2		191.7
TVOR/VORTAC/VOR-DME	1 "	1		2.5		2.5		2.5		2.5		171	+	12.5		23.6
Relocate		1.2	0	2.0	. 60	1.0	. "	1.0	9	1.0	18	6.2	8	8.0	38	14.2
	-			12		100		100		100		60.2		109	-	130 1
Add DME to VOR/TVOR.	8	1.2	\$. 64	\$	2.8	8	1.4	8	1.4	140	9.6	100	14.0	240	23.6
Low and Medium Frequency NAVAIDS	- ;	7		2.3		s.		z.		9.		3.9		6.0		9.9
Improve	Ļ	7.		2.3		3.		3.		3.		3.9		6.0		9.6
Total	Ļ	5.7		17.1		26.8		25.4		25.4		100.4		101.2		201.6

overall controller productivity increases of 10% in the terminal and 14% enroute could be expected in an all-RNAV environment compared with an all VOR airways environment. The data that supported this analysis was based on simulations which pre-dated the Task Force Report. Analysis of a subsequent terminal area real time simulation [11] indicated that RNAV equipped arrivals experienced a significant reduction in holding delays along with an increase in operation rates, which tends to verify the correlation between reduced workload and increased productivity. Although other aspects of enroute controller productivity were not further analyzed, the 30% increase in communications productivity both enroute and in the terminal area as reported in References 11 and 12 which were based on post-Task Force simulations, tend to support the earlier conclusions. In the terminal area, a substantial reduction in the use of radar vectors was also observed and navigation was returned to the cockpit through use of parallel offsets and direct to waypoint instructions.

5.2.2.2 Controller Training

One of the conclusions drawn from the real-time simulation of the New York terminal area [9] was that a thorough and extensive training program is necessary before controllers can use RNAV procedures in a way that will result in maximum benefit to the system and the users. This conclusion was supported by the difference in attitude of the controllers before and after the experiment and by the improvement in performance as the experiment progressed.

Several specific RNAV functions were introduced and recommended phraseology was presented for each. These functions are briefly described below.

- Parallel Offset the capability to fly an uncharted route parallel to an established (charted) route. This capability requires input to the RNAV equipment, specifying direction and amount of offset. The aircraft will navigate to and from the offset on a 45° angle unless otherwise specified.
- 2) Next Leg Offset the capability to intercept and fly a specified parallel offset on the next leg (segment) of the RNAV route.
- 3) Delay Fan a technique using RNAV to effect a delaying maneuver in lieu of radar vectoring.
- 4) Resume Navigation a means of returning an RNAV equipped aircraft to its charted route, after it has been radar vectored.
- 5) Cancel Offset an instruction which returns an aircraft from an offset to its established route.
- 6) Direct to Waypoint an instruction which causes the aircraft to navigate directly from present position to a specified waypoint.

The training phase of the simulation required approximately 40 hours of simulation time during which five controller teams were trained in the use of RNAV maneuvers. Simulation was continued until their performance in the lab and the data measures taken during each run indicated that the training curve had leveled off.

A further measure of the effect of training and experience was the result of a compilation of answers to a controller questionnaire prior to training, after training, and after the experiment itself. The questions concerned what the reaction of controllers to implementation in the real-world would be, whether RNAV would really work and should be required, if RNAV would make the controllers job easier, if RNAV would improve capacity, etc. In almost all cases the average responses progressed from negative in the pre-training period to slightly positive at the post-training point to more positive after completion of the experiment. Similar attitudes have been encountered in flight test situations with controller participation.

Controller training must include not only the use of RNAV procedures and phraseology, but the recognition and understanding of RNAV equipment capability and the pilot workload associated with various maneuvers. Although the anticipated gradual implementation of RNAV (see Section 5.4) should allow the development of skill in the use of RNAV procedures over a period of time, a dedicated training program will be required to insure that the proper level of controller familiarity will be available to insure an orderly evolution of an RNAV environment. From discussions with the FAA Office of Personnel and Training, it was concluded that an initial period of dedicated RNAV training at the Academy in Oklahoma City would be required in order to provide each radar terminal with core personnel trained in terminal area RNAV procedures. Subsequent instruction would then be provided through routine on-the-job training. Training in enroute RNAV procedures can be accomplished through individual study based on Air Traffic Training Refresher Units published periodically as part of the National Air Traffic Training Program. Two such refresher units on RNAV have been published to date, but they would have to be modified and expanded to reflect the RNAV implementation plan which is adopted.

It is estimated that development of a 40 hour RNAV course at the academy would cost \$6633, based on the current cost of developing similar programs, and that annual course maintenance cost would be \$8072. If it is assumed that two controllers from each of the 194 TRACONs and TRACABs estimated to be in commission in 1985 [14] would be required to attend one 40 hour course, then twenty-two classes of 18 controllers each would be required to accomplish the initial training at an average per diem rate of \$33 and average travel of \$250, total initial training cost including course development costs would be \$218,691.

5.2.2.3 Controller Comments

An appraisal of the use of RNAV and VNAV in terminal area ATC operations was made by a group of five field controllers who participated in the real time simulation activity at NAFEC [11]. Their remarks, which are given in Appendix H, tend to confirm the results and conclusions presented in this section and other sections of this report.

5.2.3 ATC Automation

The impact of the implementation of RNAV on ATC automation was analyzed in Section 4.4. Six areas in ATC automation were determined to exhibit potential for impact in one or more of the RNAV implementation phases. The terminal area ATC facets impacted include the communication and definition of RNAV routes to the ARTS system, the conflict prediction problem, and arrival aircraft metering and spacing (M & S). The enroute areas impacted also include the definition of RNAV routes and the conflict prediction problem, as well as enroute flow control.

The primary impact concerns computer system core requirements. The enroute core and terminal area requirement due to the implementation of RNAV will incur an additional cost of \$258,000 to support a mixed VOR/RNAV environment. However, the core requirements for an all RNAV environment are \$174,000 less than the current requirements.

5.2.4 Airspace and Airport Capacity

Several studies have been conducted which relate to the impact of RNAV on airspace capacity. Route width effects were analyzed in a route width/capacity study [5], in the design of a high altitude enroute charted RNAV structure [4], and in the design of terminal area RNAV structures [2]. Fast time simulation was utilized as a design tool for the high altitude enroute charted RNAV structure. The basic simulation outputs were route segment loading and identification and categorization of conflicts. Two levels of RNAV structure were analyzed. The first was a 185 airport pair structure and the second a 429 airport pair structure which was evolved later in the design process. The traffic sample used in the analysis was based on the 1969 peak day IFR data [23]. The 185 airport pair structure provided RNAV routes for 41% of the traffic sample, and the 429 airport pair structure accommodated 67% of the traffic sample.

Figure 5.6 is a summary of common route, non-common route, and total conflicts as a function of percent of traffic on RNAV routes. The conflict count from the 429 airport pair simulation is normalized to the 185 airport pair traffic level. It can be seen from Figure 5.6 that the total number of conflicts decreased with increase in percent RNAV traffic, which is an indication of increased capacity. The common route types include overtake and head-on conflicts and the non-common route types include intersection, merge, diverge, and proximity conflicts. (The conflict types are described in detail in Reference 12). The change in the number of common and noncommon route conflicts with change in percent RNAV mix is due to the fact that the VOR and RNAV structures are completely independent but overlaid upon one another. As traffic is diverted from the VOR to the RNAV structure, (say 25% RNAV), fewer common route conflicts would occur on the VOR structure due to reduced density and the difference would not be made up by common route conflicts on the RNAV structure because of even lower density, since number of conflicts vary roughly as the square of density [12]. On the other hand the number of non-common route conflicts can be expected to increase because the two route structures intersect profusely.

The total number of conflicts is less for 100% RNAV traffic than for 100% VOR because the RNAV is a more ordered structure. Increased use of parallel routes would drop the common route conflict count even more, at the expense of increasing intersection type conflicts. The latter are usually more cheaply resolved from a time and fuel viewpoint, however. They also require less controller time to solve, although they may require more intense surveillance to insure timely detection of a developing conflict. The level of non-common conflicts is higher on the 429 airport pair structure since it is a more complex structure than the 185 airport pair structure, and therefore has more intersecting routes.

The impact of VNAV on airspace capacity was analyzed in Section 4.5 and it was concluded that, with rare exceptions, the use of fixed gradient VNAV routes as an airspace design tool is not productive from a capacity viewpoint. Furthermore the 2D/3D real time simulation described in Reference 11 indicated that there are no advantages apparent in an airspace based upon either fixed VNAV gradients or "stacked" VNAV routes and that there are disadvantages to both the controller and the user. It was also found that fixed gradient VNAV departures were significantly penalized in terms of time to transit the terminal area compared with either radar vectored flights or RNAV equipped aircraft. Utilization of pilot selected VNAV gradients had no impact on system operation, but provided significant benefit to the user. It should be emphasized however, that airspace design should accommodate the pilot-selected use of VNAV to the extent practical because of the operational and economic benefits available.

The terminal area RNAV design concept was tested in real time simulations at NAFEC [9,11]. The primary objectives of the first simulation were to evaluate controller workload and to determine if capacity would decrease in a mixed VOR/RNAV environment. The results indicated that operation rates remained essentially constant (with a constant demand traffic sample) while controller workload decreased.

The simulation described in Reference 11 was structured to measure any changes in capacity (either increase or decrease) which occurred with the introduction of various mixes of VOR and RNAV traffic. It was found that RNAV arrivals experienced a significant decrease in holding delays, a decrease in time in system and an increase in operation rates, thus indicating an increase in capacity. No changes in departure delays or operations rate were observed.

In the 4D payoff analysis in Section 3.4, it was concluded that the use of 4D RNAV in M & S terminals can have a large impact on terminal area airspace capacity. Estimates of hourly IFR arrival capacity which were developed for each of eight major airports in 1984, are itemized in Table 5.20.

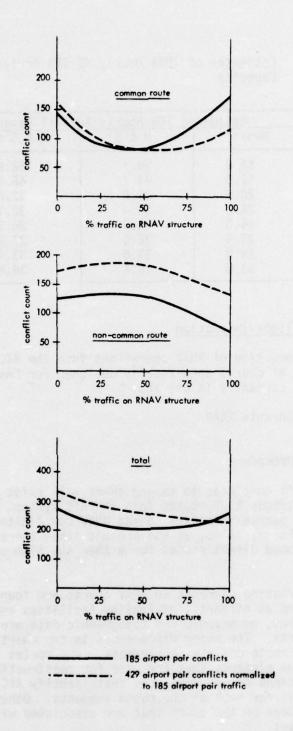


Figure 5.6 Conflict Count vs % RNAV Mix

Table 5.20 Estimates of 1984 Hourly 4D IFR Arrival Capacity

Airport	Estimat	ed IFR Hourly Ar	rival Capacity
	Manual	M & S	M & S + 4D
ORD	55.0	59.7	62.9
JFK	42.0	44.7	46.4
LGA	28.5	31.0	32.8
EWR	28.0	30.4	32.1
PHL	24.0	25.8	26.9
MIA	33.0	35.5	37.2
DEN	29.0	31.6	33.4
SF0	33.0	36.4	38.8

5.2.5 Charting and Flight Inspection

The limited introduction of RNAV operations into the ATC system has produced a new series of charts specifically designed for RNAV users. Those RNAV charts which are currently in use are:

High Altitude Enroute RNAV

RNAV SID

RNAV STAR

RNAV Approach Procedure

As the use of RNAV continues to expand there will exist a requirement to introduce more low altitude RNAV routes into the ATC system. As these routes are developed it will become necessary to add RNAV Low Altitude Enroute Charts to the list of RNAV charts. Also, at the present time, there is no satisfactory chart for RNAV preplanned direct routes for either the high or low altitude route structure.

Many of the information elements on RNAV charts are found also on current VOR charts. Such items as airports, navigation facilities and frequencies, communication facilities, geographic and topographic data are all common to both RNAV and VOR charts. The major differences in the charts can be attributed to the routes and the route definition elements. VOR routes are located overlying VOR radials. The elements that are used for constructing the routes are VORs and VOR intersections or DME distances that identify ATC reporting points, and omni-bearing values for each of the route segments. Other information elements which may appear on the chart that are associated with the VOR route segments are as follows:

- Minimum Enroute Altitude (MEA)
- Minimum Crossing Altitude (MCA)
- Minimum Obstruction Clearance Altitude (MOCA)
- Maximum Authorized Altitude (MAA)
- Distance Between VOR Stations
- Route Segment Distance
- Approach Minimums and Profile View

All of these listed elements are necessary on RNAV charts as well as VOR charts. The major differences between the VOR and the RNAV charts are in the route defining elements. In the RNAV structure the waypoint and its associated name becomes the major route definition element. The waypoint itself should be designated by the two techniques described in Section 5.3.4. One other, and usually lesser used, route defining element may be found on RNAV charts. This is the RNAV intersection. The RNAV intersection is formed by either a distance and bearing from a waypoint or the intersection of the bearings from each of two waypoints. An additional information element that is needed to define the route is the bearing to or from the waypoint. Once the route has been defined it is sometimes necessary to identify waypoint change points along the route. These points are defined by a distance from the waypoint and they are placed on the chart when the waypoint change must come at a point other than the midpoint of a route segment. This can be caused by a number of reasons such as facility coverage, unusable signals, frequency protection, etc. In developing an RNAV route, considerations of route width relative to the reference facility are different than VOR route width criteria. However, this has a considerable impact upon ATC procedures but not upon charting.

The use of preplanned direct (PPD) RNAV routes poses some major charting problems. Preplanned direct routes will be used primarily in the enroute phases of flight so only that aspect of PPD routes will be considered in this discussion. At the present time, through the provisions of Advisory Circular 90 - 63[62], it is possible to file an RNAV PPD flight plan. However, due to the uncertainties of VORTAC coverage along the route and the RNAV route width variations discussed in Advisory Circular 90-45A, the PPD route must be entirely within ATC radar coverage. Consequently, the filed altitudes, waypoints and route segments between waypoints must be carefully chosen to assure radar coverage. It is assumed that this requirement will continue to be applied until increased automation and flow control are realized, and area flight checking is accomplished.

Flight checking is currently accomplished as a part of the procedure for the implementation of new routes. An additional flight checking requirement will develop in the implementation of RNAV routes in accordance with the route development and implementation process described in Section 5.4. The evolution to an all-RNAV environment will involve both the implementation of the total optimum RNAV structure on a gradual, individual route basis, and the implementation of an increasing use of preplanned direct routes. The flight checking process must therefore be structured to accomplish two objectives: (1) the implementation of individual routes on a demand basis, and (2) eventual flight checking of the entire CONUS on an area basis to define acceptable VORTAC coverage for random preplanned direct routes.

Although specific requirements exist for RNAV charting and flight inspection, the gradual implementation of RNAV routes should cause an impact little more than that of current continuing requirements.

5.2.6 System Impact Summary

A summary of the impact of RNAV on the ATC System is given in Table 5.21. Costs and savings for the period from 1982 to 2000 were computed in a separate study [14], and are based on the assumed implementation scenario described in Table 5.17. Savings and costs are also presented in terms of their present (discounted) value, based on guidelines promulgated by the Office of Management and Budget [63]. The ATC system benefit to cost ratio for the time period of 1982-2000 is computed to be 9.9.

-	Table 5.21 ATC System I	ATC System Impact Summary	Sa	Savings	Costs	ts
			3	-		
			Total	Value	Total	Present Value
VORTAC	Terminal Area (Maintenance)	Additional VORTAC requirements are nominal compared with F&E planned budget. \$2.3 million per year sayings	\$36.1M \$8.0M	\$8.0M		
	Enroute (Implementation and maintenance)	due to reduced number in terminal areas. Additional savings available in removal of high altitude VORTACs should offset new station costs and result in overall savings.	100 100 100 100 100 100 100 100 100 100		\$2.4M	\$0.8M
CONTROLLER	Terminal Area	Controller workload will decrease and productivity will increase through use of RNAV procedures both enroute and in the terminal area. Controllers and pilots	\$26.4M \$8.0M	\$8.0M		
	Enroute	can function efficiently in a mixed VOR/ RNAV environment, and productivity increases will be realized during the transition period from VOR to RNAV.		\$422.0M\$120.7M		
	Controller Training	Controllers must be trained in use of RNAV procedures to insure orderly evolution of RNAV and realization of user and system benefits			9926-923	
IMPLEMENTATION	ATC Automation	Initial cost of \$258,000 would be reduced to a savings of \$174,000 in all-RNAV environment, allowing use of increased core capacity for other system growth features.			\$19.8M	\$19.8M \$13.0M
250 2007 50 11 70 20	Charting and Flight Inspection	Charting and flight inspection will be a continuing routine requirement as routes are implemented from the master route structure.		20125		
685, 373 867 - Pas 10 - Mille	Route Development	An optimum, coordinated master route structure must be designed as a first step toward RNAV implementation.				
Milie Tour to College	Airspace Capacity	Capacity will be increased in terminal and enroute structures through use of 2D RNAV and use of 4D in M&S terminals in the transition period as well as in an all RNAV environment.			10 13 600	
TOTAL			\$484.5M	\$484.5M \$136.7M \$22.2M \$13.8M	\$22.2M	\$13.8%

5.3 SYSTEM DESIGN CONCEPT

The system design concept described in this section, and the implementation concept described in Section 5.4, are intended to expand, clarify, and modify the concept proposed by the RNAV Task Force, taking into consideration the impact of alternate concepts on the user and the system as summarized in Section 5.1 and 5.2.

5.3.1 Airspace Design

The achievement of the benefits described in the previous sections are dependent upon the implementation and airspace design concept which is essentially based on the Task Force concept, but which has been modified in several ways based on operational and economic assessments of that concept. This sub-section summarizes this airspace design concept in the areas of enroute design, terminal area design, route width requirements, and the application of VNAV, and is intended to complement the discussion on route development in Section 5.4.4.

5.3.1.1 RNAV Enroute Structure Design

An analysis of Task Force concepts, and the development of necessary modifications thereto is described in Reference 4 and 12. The methodology used in the analysis is summarized in Section 2.2.2. The enroute design concepts listed below consist of a summary of the results of the analysis of Reference 4. The overriding principle in RNAV enroute structure design is that an optimum total structure should be developed and used as a basis for the implementation of individual routes.

Terminal Arrival and Departure Waypoints

Terminal arrival and departure waypoints should be aligned in accordance with traffic flow demands. This requires that the design of enroute and terminal structures be closely coordinated.

One-Way Routes

One-way routes should be used where their application will reduce traffic congestion and not as a general rule as proposed by the Task Force. The hemisphere rule for altitude separation should be adhered to except when local conditions indicate an operational advantage may be gained by its suspension. One-way route segments are an effective way to accommodate extended climbs and descents to and from enroute altitude.

Multiple Routes

The Task Force concept of providing for two offset routes between parallel route centerlines is unnecessarily restrictive and results in route

mile penalties. Charted routes should be established to accommodate traffic demand in the most effective manner, considering parallel offsets as a tactical maneuver used in conjunction with all other factors which impact network design effectiveness.

Route Merging and Bending

The first step in RNAV route structure design should be to examine traffic flows to determine both the possibilities for merging and the requirements for multiple routes to handle traffic demand. The appropriate combination of merging, bending, and demerging should be derived from a consideration of minimizing user cost consistent with system requirements. The criterion for minimizing user cost should be flight miles, i.e., route mileage multiplied by projected route utilization.

Intersection and Merge/Demerge Angles

Whenever feasible, intersection and merge/demerge angles should be greater than 10 degrees, particularly in areas of high traffic flow, where the high intersection occupancy time would cause excessive controller workload.

Other Factors

The development of an optimum total enroute structure requires consideration, in addition to the foregoing factors, of such characteristics as intersection densities, waypoint densities, waypoint/intersection proximity and excessive route turning. The Task Force concept of great circle routes between airport pairs should be expanded to include charted alternate weather routes whose segments are great circle routes.

Preplanned Direct Routes

The preplanned direct concept described in Section 5.4.1 places flight planning responsibility on the user and requires the use of existing way-points, or VORTACs, within the charted RNAV structure.

Route Widths

The ± 2.5 nm route widths recommended by the Task Force are not required, and traffic demand can be met with constant ± 4.0 nm route widths.

5.3.1.2 Terminal Area Design

Based on the terminal area design analysis described in Section 2.1.3 and 5.1.1, a set of terminal area design guidelines was developed [2]. The following paragraphs present a summary of the guidelines contained in Reference 2.

Arrival and Departure Sectors - The application of the basic design should incorporate consideration of traffic distribution demand as well as runway orientation constraints. Similarly, the number and size of departure and arrival sectors should be varied as necessary to accommodate traffic demand, rather than being constrained to equal size octants, as recommended by the Task Force. Metroplex areas require even further modification of the Task Force Design. In these areas it may be necessary to abandon the alternating arrival and departure sector concept in favor of parallel traffic corridors in order to minimize undesirable altitude restrictions and excessive route lengths. The concept of using arrival and departure sectors ("wagon wheel") in designing terminal area routes should be used as a basic design tool for most medium and high density terminal areas. The size and orientation of these corridors should be based primarily on the direction and density of traffic flow. The rigid specification of the location, size and orientation of the corridors, such as the octant concept suggested by the RNAV Task Force, should not be used since it can create operational penalties such as increased distance flown and additional altitude restrictions in complex terminal areas. In metroplex areas particularly, arrival and departure fixes should be located on the basis of enroute traffic flows, and traffic should proceed to a charted conventional downwind, base and final approach sequence in the most direct manner possible without interference from adjacent terminal routes. Altitude restrictions should be determined from typical aircraft altitude vs along track distance profiles.

Low Altitude Waypoints - The location of low altitude arrival waypoints in the Task Force RNAV terminal area design model (25 nm from the airport) is in general agreement with the location of current radar vector/VOR feeder fixes at most airports. Although the location of low altitude departure waypoints 15 nm from the airport is somewhat closer than usually used today, this closer location does not present any serious problems in developing the terminal route structure. The downwind, base and final approach leg used in the Task Force design within 25 nm of the airport is consistent with current airport traffic patterns, and should be charted. If an operational advantage can be gained by eliminating the downwind, or downwind and base, legs of an approach, then these route segments should be eliminated on an ad hoc basis, through the use of appropriate RNAV instructions.

Low Altitude Routes - The low altitude traffic does not need to be constrained by the octant concept. In most terminal designs it is possible for departing low altitude traffic to proceed direct to their enroute route structure once they pass the low altitude departure fix. Similarly, arrivals can proceed from their point of origin to the low altitude arrival fix without interfering with high altitude traffic flow.

Terminal Area Dimensions - The use of a nominal 45 nm circle centered at the primary airport is a satisfactory conceptual aid in defining the extent of a non-metroplex terminal area. The 45 nm circle does not necessarily have any operational significance however. Consequently in actual terminal area design, operational considerations concerning the location of terminal enroute boundaries should take precedence over the use of the conceptual

45 nm boundary. The concept of defining the metroplex terminal area by encircling each airport with a 45 nm ring and then encircling all the 45 nm rings with one large ring, as recommended by the the Task Force, is not feasible since in congested regions such as New York since the size of the large ring would be excessive. It is necessary in these metroplex areas to limit the extent of the terminal area by considering the location of all major airports within the metroplex and the proximity of adjacent terminal areas, and establishing a ring slightly larger than 45 miles which will accommodate the necessary arrival and departure waypoints.

Vertical Design - A vertical gradient of 300 ft/mile for arrivals, consistent with the Task Force recommendations is acceptable for all aircraft. Above 10,000 ft., descent gradients of 400 ft/mile may be used for high speed descents by high performance aircraft. The Task Force recommended departure gradient of 400 ft/mile is not acceptable for all aircraft at all altitudes encountered in the terminal area. An altitude envelope based upon aircraft performance is more desirable than a single departure gradient value. The following envelope which is based upon aircraft performance was used in several of the terminal designs in Reference 2 and should be used as a general guideline.

Altitude	Gra	dient
0-10000 Ft. 300	-550	ft/mile
10000-18000 Ft. 150	-350	ft/mile
18000-25000 Ft. 100	-200	ft/mile

Fixed gradient VNAV routes should not be utilized, but the 2D design should incorporate vertical separation for crossing routes which is sufficient to allow pilot selection of VNAV descents. Vertical departure envelopes for higher performance aircraft, whose minimum gradient is higher than those listed above, should be utilized when shorter departure routes will result.

Route Locations - Most terminal areas are currently structured in a wagon wheel or spoke type flow pattern with fixed arrival and departure points. However, most spokes are not of equal size nor are they spaced in an octant pattern. In Phase 1 all RNAV routes within the terminal maneuvering area (inside of the low altitude arrival and departure fixes) should overlie the basic radar vector routes. Outside of the terminal maneuvering area, RNAV routes should be established in areas where user benefits will accrue. User benefits include shorter route lengths and a minimum of restrictions during climb and descent. The Phase 2 2D terminal designs must be able to accommodate both conventional radar vector/VOR and 2D RNAV traffic. This represents a considerable constraint upon the design. The flow patterns for this transitional time period should be based upon the Phase 3 terminal design, which should be created prior to the development of the Phase 2 design. The location of feeder fixes and departure fixes should be near those created in the Phase 3 design but moved as necessary to accommodate VOR traffic. Traffic in the terminal maneuvering area should generally conform to the Phase 3 design.

The recommended terminal design procedure begins with the development of the horizontal projection (or ground tracks) of the arrival and departure routes to all major airports in the terminal area. The first task in developing the horizontal design is to define the center and the lateral extent of the terminal area design. For most terminals which have a single major airport, the center of the major terminal airport can be used as the center of the terminal complex. Occasionally in a metroplex area this technique is not satisfactory.

Once the center of the terminal area has been determined, the terminal area, for design purposes, is defined by a radius of approximately 45 nm. For metroplex areas it is sometimes necessary to adjust this radius value to accommodate all airports in the area, but without infringing upon adjacent terminals or metroplex areas. Within the terminal complex two areas are defined. One is the terminal transition area and the second is the terminal maneuvering area. Generally, the area within 15-20 nm of the airport contains the terminal maneuvering area while outside of this distance is the terminal transition area. Routes in the terminal transition area remain fixed no matter which active runways are being used at the airports within the terminal area. Conversely, the routes in the terminal maneuvering area do change as the active runway in use changes. Once the terminal maneuvering area and the terminal transition area have been defined the traffic flows can be used to establish the arrival and departure routes to both primary and satellite airports. It is often desirable to locate an arrival-departure sector boundary along a major traffic flow direction in order to keep arrival and departure routes as short as possible.

Gradients for the 2D routes should be based upon the performance characteristics of several aircraft types under varying conditions. The descent gradient is nominally a constant at 300 feet/mile for descent operation below 10,000 feet. Above 10,000 feet the 300 ft per mile gradient approximates the fuel optimum standard descent and 400 ft per mile gradient approximates the time optimum high speed descent. Descent profiles should accommodate this range. The climb profiles vary widely depending on aircraft type, ambient temperature, aircraft weight and climb airspeed. The vertical route design is accomplished by utilizing a vertical profile plot, for each proposed route, of altitude versus along track distance, with arrival routes being plotted first, at a constant 300 feet/mile. Departure routes are designed which will accommodate the lowest performing aircraft. Additional departure routes are then designed which will either allow unrestricted climb or provide shorter departure routes for aircraft with a specified minimum performance capability.

5.3.1.3 Route Width Requirements

The route widths required for the implementation of routes in the terminal and enroute strutures are given in Table 5.22.

Table 5.22 Route Width Requirements*

	Terminal	Enroute	
Phase 1 ±2.0 nm a,b		in accordance with 7110.18[38]	
Phase 2 ±2.0 nm a,b		±4.0 nm	
Phase 3 ±2.0 nm a,b		±4.0 nm b	

- * does not include additional airspace for slant range error, which is discussed in Section 5.3.6
- a-or ±4.0 nm depending upon distances from VORTAC[38]
- b-change from Task Force recommendation

5.3.1.4 VNAV Operational Concept

As envisioned by the RNAV Task Force, VNAV, or 3D RNAV, would provide the following additional features over RNAV:

- the capability for an aircraft to climb or descend to a desired altitude, reaching that altitude at a pre-selected point in space.
- development of routes within terminal airspace with vertical profiles to provide the airspace planner a tool for optimizing use of airspace.
- enhance safety in instrument approaches by providing vertical guidance to noninstrumented runways and by permitting the programming of selectable climb/descent profiles.

The Task Force envisioned wide spread use of VNAV both enroute and in terminal areas, with 3D capability required in the high altitude enroute airspace in the post-1982 time period. A summary of the Task Force recommended VNAV concept is given in Table 5.23.

Implicit in the Task Force VNAV concept is the expectation that the establishment of 3D routes and procedures will be benefical to both the ATC system and the users. Potential benefits identified by the Task Force and comments on those benefits are discussed below:

- Increased flexibility for the controller due to ability for aircraft to arrive at an assigned altitude at a specified point in space
- Economic benefit to the user in the ability to optimize time and fuel through adherence to a preplanned vertical profile

The economic benefit to the user was found to be dependent upon an airspace design with sufficient flexibility in the vertical plane to allow a variety of climb and descent profiles [2,12] rather than imposing a sub-optimum fixed gradient profile.

3) Improved airspace utilization through specified vertical paths.

Airspace utilization using fixed gradient VNAV was analyzed in Section 4.5 and it was concluded that the opportunity to improve airspace utilization with fixed gradient VNAV routes occurs so seldom in the terminal area route design process that it does not warrant serious consideration.

A concept of "stacked" VNAV routes for intercept of parallel precision approach paths was analyzed in the simulation study described in Reference 11. It was found that this concept is not desirable from either an operational or an airspace capacity viewpoint. Therefore a flexible VNAV concept is recommended which will allow aircraft so equipped to obtain economic benefits through ad hoc selection of 3D gradients, but which allows 2D equipped aircraft to fly in the same airspace.

Departure vertical envelopes should be provided to accommodate all aircraft using the terminal area. Additionally, departure envelopes which will provide shorter routes for the higher performance aircraft should be provided. These routes could be used by both 2D and 3D equipped aircraft, and would be defined in the vertical plane by a "floor" of altitude restrictions (and an occasional altitude "ceiling") to insure procedural separation from crossing routes. Any aircraft capable of the required minimum climb performance could utilize these departure routes. There would be no penalty imposed on the lower performance aircraft since the design would be optimized for minimum departure route lengths, coupled with economical descent profiles, which accommodate all aircraft. The design is therefore optimized for both route length and altitude profiles for both 2D and 3D equipped aircraft.

Pilot selection of 3D descents within the constraints of crossing altitude restrictions would allow user optimization of either fuel consumption or time. The use of 3D descents, with the descent angle and speed selected by the pilot, could improve either fuel or time performance as compared with the normal 2D procedural descent procedures. Altitude restrictions at intersections of routes would be specified to accommodate the separation requirements of 3D route crossings. As discussed in Section 4.5.4, 3D crossings generally require more vertical separation then 2D, although in some cases the 2D case requires more.

A summary of the recommended VNAV operational concept is given in Table 5.24.

5.3.2 RNAV Airspace Definition

The Task Force recommended that RNAV be the system of navigation in the high altitude enroute structure and in selected high and medium density terminals in Phase 2 and 3, and this recommendation is also included in the implementation concept presented in this report, even though the timing of the implementation phases would be determined by user demand. The list of terminals

was based on traffic forecasts and is given in the Task Force Report [1]. The terminal area design concept described in Section 5.3.1.2 results in terminal areas with an approximate radius of 45 nm. In order to prevent an undue penalty to low cost general aviation aircraft which may operate at other than the primary airports within the terminal area (see Section 3.5), terminal areas should be defined in a manner similar to that in which terminal control areas (TCA) are now defined. In that manner, RNAV equipment would be required only for operation at primary airports, and a few airports in close proximity to the primary airports. A low altitude path to other airports would then be available without the requirement for RNAV.

5.3.3 Turn Anticipation

The issue of turn anticipation is critical because of its effects on the complexity of RNAV equipment and on the cockpit procedures necessary to reliably provide for anticipation stems from protected airspace considerations. In order to compensate for overshoots, current procedures provide an additional 2.0 nm protected airspace for 10 nm past the turn point [38]. It would be desirable from an airspace viewpoint to require all aircraft to anticipate turns to such a degree that only minimal overshoot would occur and no additional protected airspace would be required.

Overshoots at turn points in low and high altitude enroute operations are, for the most part, insignificant, since most enroute turn angles are small. Overshoots at turn points in terminal area operations can be critical, however, due to the fact that most turns are of sufficient magnitude to cause large overshoots and that the airspace is crowded.

During the General Aviation Flight Test program described in Reference 7, data was recorded on turn overshoots without anticipation and with a method of procedural anticipation. Simulation experiments reported in Reference 8 also included turns with both procedural and automatic anticipation.

The results of these tests indicate that the use of some method of turn anticipation will obviate the need for additional protected airspace and that a procedural technique, in lieu of automatic, is an acceptable minimum requirement. AC90-45A [19] suggests a method of procedural turn anticipation wherein the turn is anticipated by an amount equal to one nautical mile for each 100 knots of ground speed. Tests have concluded that the technique is acceptable, but additional tests are required to determine the full range of acceptable techniques.

5.3.4 Waypoint Standards

The RNAV concept involves both charted and pre-planned direct routes whose segments or, in some case the entire route, are great circle. A variety of navigation sensors, ranging from VORTAC to self-contained to long range aids will be utilized. RNAV systems will range in complexity from single rho/theta analog computers to sophisticated digital multi-sensor dead reckoning systems. The charting of waypoints is necessary for the optimum use of these routes by pilots with the total range of airborne equipment.

Table 5.23 Task Force VNAV Concept

es ou buone	Phase 1	Phase 2	Phase 3			
High Altitude Enroute	no 3D routes established	establish vertical profiles to accommodate A/C equipped with 3D	3D capability required for enroute trans- ition			
Low Altitude Enroute	no 3D routes established (may be assigned to aircraft equipped with 3D)					
High Density Terminals		establish 3D arrival procedures with fixed descent gradients to all airports in terminal area	Preferential treatment to aircraft equipped with 3D			
Medium Density Terminals	establish 3D departure pro- cedures to meet user needs	establish 3D arrival pro- cedures to meet user needs	preferential treatment to air- craft equipped with 3D			
Low Density Terminals		establish 3D arrival procedures	s to meet user needs			
Approaches - All Airports	establish 3D approaches to the extent practicable					

Table 5.24 VNAV Operational Concept

	Phase 1	Phase 2	Phase 3			
High Altitude Enroute	no 3D routes established	establish procedures to allow pilot selec- tion of 3D descent for A/C equipped with 3D	establish pro- cedures to allow pilot selection of 3D descent for A/C equipped with 3D			
Low Altitude Enroute	no 3D routes established					
High Density Terminals	establish departure envelopes in which optimum climb pro- files may be selected to meet user needs	establish procedures to allow selection of 3D descents for A/C equipped with 3D				
Approaches - All Airports	establish 3D approac	ches to the extent practic	able			

An analysis of waypoint location, waypoint designation, facility selection, parallel offset considerations, waypoint charting, and route definition was conducted as described in Section 2.1.1. The results of the analysis indicated that the most acceptable system is a dual designation system consisting of the radial/distance and pronounceable five alpha techniques.

Radial/Distance System

5

The radial/distance system is a full 360° radial grid system with 1 nm radial waypoint granularity. For example, 27213 would completely designate the waypoint located on the 272° radial at 13 nm from the VORTAC. This system combines all the advantages of the clock grid and the cardinal radial grid with the added advantage of increased resolution. This designation technique provides 46,800 potential waypoints within the coverage area of each "L" facility (40 nm radius). This resolution offers two additional advantages. First, the charting and utilization of waypoints can be done with varying resolutions depending upon the specific requirements. Second, as can be seen from the preceding figures, the density of waypoints is proportionately larger in the proximity of a VORTAC, which is typically coincident with terminal area operations.

The Radial/Distance system would also incorporate double designation requirements for waypoints requiring greater accuracy than the 1°/1 nm capability of the basic grid. For example MAP 13/93.3,0.3 would designate the missed approach waypoint located on the 93.3 radial, 0.3 nm from the VORTAC. This system is inherently advantageous to the low capability RNAV user since the bearing and distance from a VORTAC, which is the basic input to the RNAV computer, are included in the charted waypoint information block and are communicated verbally by the controller in the case of impromptu waypoints.

Pronounceable 5 Alpha Waypoint Designators

The Pronounceable 5 Alpha technique names all RNAV waypoints with a five letter combination which can be pronounced in spoken English for communication purposes, charted for geographic orientation purposes and utilized in both the airborne and ground system computers. For example:

Fenner	=	FENER	Mesquite	=	MESKY
Camerron	=	CAMON	Cabin Creek	=	CABIN
Chapin	=	CHAIN	Ruskin	=	USKIN

Assuming that two character positions of the available 5 character slots are reserved for vowel sounds results in more than enough 5 letter combinations to handle not only the existing waypoint code requirements but includes a growth capability on the order of 70 times the current requirements. At the present time it is the FAA's stated policy [62] to assign five-character pronounceable words to all waypoints. This includes renaming all existing route waypoints, SIDs/STARs and IAPs other than the waypoints which correspond to facility locations, which will continue to be referred to by their appropriate 3 letter designators (e.g., Robbinsville-RBV, etc.). As planned, these waypoint names will be the only code used throughout the system for communications,

navigation computer inputs and depiction on aeronautical charts. The following is a reproduction of the current FAA implementation plan stated in Reference 62 for this technique.

- All new waypoint assignments for routes, SIDs, STARs and RNAV IAPs will be in five-letter name combinations.
- 2) Nonredundancy must be maintained in the RNAV system only. Other route intersections or facility identifiers will retain current names and codes, i.e., three-letter or number, two letters.
- 3) Where a waypoint exists with a name and either a five-or three-character code (except facility idents), such names will be revised to five letters and other codes deleted.
- 4) Waypoints with less than five-letter names will continue to be used until a need exists for such name assignments for alternate use. However, only the three or four letters will be used and other codings shall be cancelled.
- 5) Initial change will be in the existing three, four and five letter names which are nonredundant within the RNAV route system by cancellation of existing code assignments on a planned effective date.
- 6) Other name changes and code deletions to be controlled with charted amendment cycles (28 day interval).
- IAP waypoint revisions to be started following revision of enroute system.

Impromptu Routes

Impromptu routes (enroute) should normally be communicated using the Pronounceable 5 Alpha waypoint designator. Three distinct cases arise regarding impromptu route definition using waypoint designators. The recommended techniques for the three cases are: 1) impromptu routes based on charted waypoints should be communicated using the Pronounceable 5 Alpha designator, 2) impromptu routes based on uncharted waypoints should be communicated using the Radial/Distance designator for station referenced systems, and 3) impromptu routes using uncharted waypoints should not be used by aircraft whose RNAV system does not have the capability to accept rho/theta inputs unless a capability for converting from rho/theta to an acceptable input format is maintained.

5.3.5 Parallel Offsets

The term "Parallel Offsets" should not be confused with "parallel routes". Parallel routes are defined by their own waypoints, while a parallel offset is defined by a perpendicular distance from a parent route, and does not contain its own waypoints. Parallel routes are established in order to provide for traffic demand, while parallel offsets are to be used tactically to solve

potential traffic conflicts or to provide conflict-free climb or descent paths in an impromptu manner. There are four issues involved in the use of parallel offsets: 1) granularity requirements, 2) paralleling in turns, 3) constant radius turns and 4) climbing and descending offsets. Paralleling in turns includes both the issue of turn anticipation (Section 5.3.3) and the issue of method of determining the offset turn point. The issue of constant radius turns is applicable to both parallel offsets and parallel routes. The following conclusions have been drawn from the results of simulations and flight tests [7,8,9].

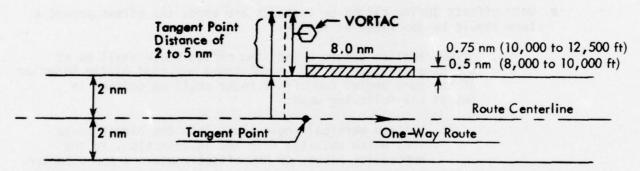
- Area navigation systems should have the capability to fly parallel offsets in increments of 1 nm out to 20 nm. The displaced needle CDI method of flying parallel offsets is not acceptable due to linearity and scale limitations.
- Paralleling in turns is a required RNAV function. Constant radius turns are not required to provide separation in turns since high speed differentials encountered enroute are accompanied by gentle turns, while the larger turns required in the terminal area involve lower speeds and lower speed differentials. Procedural paralleling in turns is an acceptable method. The amount by which a turn should be anticipated should be increased by the amount of the offset for an inside offset, and decreased by the amount of the offset for an outside offset.
- When offsets during climbs or descents are used, the offset around a turn should be described as follows:
 - a) offset outside a turn point: the offset route shall be at the same altitude as the parent route waypoint at the bisector of the turn angle; the offset route shall be defined in one of the following ways:
 - a fixed vertical angle (less than the parent route VPA) which emanates from the intersection, in the horizontal plane, of the offset route and the bisector.
 - 2) it may remain in the plane of the parent route (same VPA) until abeam the waypoint, level off at waypoint past the turn, and then continue in the parent route plane.
 - b) offset inside a turn point: the offset route shall be defined in one of the following ways:
 - a fixed vertical path angle (greater than the parent route VPA) which emanates from the intersection, in the horizontal plane, of the offset route and the bisector.
 - 2) a vertical path angle the same as the parent route VPA prior to the intersection, in the horizontal plane of the offset route and the bisector, and a larger VPA after the turn which allows interception of the parent route VPA prior to the next waypoint.

The airspace requirements for 3D offsets is discussed in Section 4.5.4.

5.3.6 Slant Range Correction

Slant range correction is required for all 3D RNAV systems and for 2D systems operating above 12,500 ft. MSL except when less than 2,500 ft AGL. The resolution of the station altitude input for slant range correction should be 1000 ft.

In the low altitude enroute structure, with route widths ±4 nm, no additional protected airspace is required to compensate for the effects of slant range error. In the terminal area, between 8000 and 12,500 feet, within ±8 nm along track from the tangent point, additional protected airspace is required on the VORTAC side of the route for tangent point distance of 2.0 to 5.0 nm. The additional protected airspace should extend for 8.0 nm along track beyond the VORTAC for one way routes and for 8.0 nm along track on either side of the VORTAC for two way routes, as illustrated in Figure 5.7. Segregation of route width extensions by altitude bands is possible in an RNAV terminal area, since altitudes are assigned to SID/STAR waypoints. A detailed analysis of the effect of slant range error is given in Section 4.1. Analysis of seven terminal area designs [2] determined that the use of additional airspace due to slant range error did not inconvenience the airspace planner and did not impact benefits.



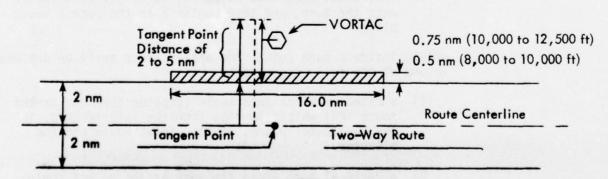


Figure 5.7 Protected Airspace for Slant Range Error in Terminal Area

5.3.7 Equipment Functional Requirements

The RNAV Task Force recommended a set of minimum functional requirements which in general exceed the capability of many RNAV systems currently in production and the minimum functional requirements for 2D and 3D RNAV systems are the subject of continuing simulation, flight test, and analysis. The following listing of functional requirements is representative of the system design concept discussed in this Section and the implementation concept described in Section 5.4. They are extracted from Reference 7 and modified as appropriate by the results of additional analysis reported herein. For controllers to effectively use RNAV, all operations based on like commands should result in like maneuvering of the aircraft over the ground.

5.3.7.1 2D Equipment Functional Requirements

- Input Data. The area navigation equipments must accept inputs from navigation sensors, descriptions of reference facility locations (where applicable) and waypoint locations. In some types of equipment the desired track bearing may also be an input quantity.
- 2) <u>Position Determination</u>. The RNAV equipment must determine the aircraft position. Position may be determined either in a station referenced (bearing and distance) coordinate frame or in a geographically referenced (lat/lon) coordinate frame.
- Track Determination. In systems not requiring the input of desired track angle the desired track must be determined within the RNAV equipment.
- 4) Cross Track Deviation (or distance). The RNAV system must relate the present position to the desired track in order to derive the cross track deviation.
- 5) Output Data. The RNAV equipment must provide the distance to the next selected waypoint and the cross track deviation (or distance) as primary RNAV outputs to the display devices.
- 6) Display Types. The RNAV display may be either linear or angular. However, the flight technical error, which varies with display type and with scale factor, must be properly estimated and included in the accuracy evaluation.
- 7) Other Sensors. Navigation sensors allowed in the context of "other sensors" specifically includes DME/DME operation and multisensor configurations.
- 8) Waypoint Storage. A minimum storage capability for more than one waypoint is required for high and medium density terminal area operations. Single waypoint systems are acceptable for enroute operations.

- 9) <u>Turn Anticipation</u>. Turn anticipation is required, and may be accomplished procedurally or by derivation of commanded track in the RNAV computer.
- 10) Parallel Offsets. Area navigation systems must have the capability to fly parallel offsets in increments of 1 nm out to 20 nm using a centered needle indication.
- Slant Range Correction. Slant range correction is required for all systems operating above 12,500 ft MSL except when less than 2,500 ft AGL.

5.3.7.2 3D Equipment Functional Requirements

Vertical navigation is an adjunct to 2D RNAV capability and the following 3D capabilities must be implemented in addition to the requirements of Section 5.3.7.1 in order to attain 3D capability.

- Input Data. The VNAV equipment must accept input information describing the desired vertical profile. It must provide slope line selection of a gradient angle to a preset altitude at a waypoint or, alternatively, it must be possible to set altitudes associated with waypoints to create the desired slope line reference. In addition, the equipment must provide for leaving or reaching an altitude at a predetermined distance along track to or from the waypoint.
- 2) Desired Altitude. The VNAV equipment must determine the desired altitude that corresponds to the position estimated within the 2D RNAV computations.
- 3) Output Data. The VNAV equipment must provide vertical track deviation as an output to the navigation guidance display.
- 4) <u>Display Types</u>. VNAV displays may be either linear or angular. However, the vertical flight technical error, which varies with display type and scale factor, must be properly estimated and included in the vertical accuracy estimate.
- 5) <u>Slant Range Correction</u>. Automatic correction of the slant range effect must be provided for in all 3D RNAV systems.
- 6) <u>Vertical Maneuver Anticipation</u>. When vertical guidance equipment is utilized, vertical maneuver anticipation of some type will be required. The anticipation requirement may be able to be accomplished procedurally in combination with an altitude alert signal.

5.3.8 Accuracy Tolerances

The Task Force Report [1] and AC90-45A [19] list specific error budgets for area navigation systems which will insure the adequacy of the route width requirements. Flight test results [7,8] have indicated that the RSS method of combining errors, as suggested in References 1 and 19, results in errors larger

than those actually experienced due to correlation between various system error elements and between flight technical error and the system error elements. Further analysis is planned for existing and further flight test data which will explicitly define the correlation among the various error elements, and allow a more accurate specification of system error budgets for both ρ/θ and ρ/ρ navigation. FTE will not be considered a part of the resulting error budget for purposes of compliance testing, but will be considered by the airspace planner when combined with system error in a manner which recognizes the correlation between FTE and system error.

5.4 RNAV IMPLEMENTATION REQUIREMENTS

The economic and operational benefits of RNAV to both the user and the ATC system have been identified, and many of the benefits are available now. The enroute fuel and time savings from reduced route lengths will become available as routes are implemented in response to user requirements. Terminal area benefits to the user are dependent upon shorter path lengths, more optimum descent profiles, and conflict-free departure paths with unrestricted climb profiles. The terminal area benefits require the design and implementation of new routes which, in the initial implementation phase, will be implemented as practicable in conjunction with the implementation of enroute RNAV routes. Approach benefits are available immediately in the elimination of circling approaches and access to noninstrumented runways.

If benefits cited in this report are to be realized, a carefully coordinated time-phased systems approach must be employed. A three phase implementation program similar to the one recommended by the task force is required to insure an orderly transition to an RNAV environment. An evolutionary concept is required which will achieve the system design concept with minimum impact on system and user, but which is based on the attainment of the ultimate concept and benefits.

It is particularly important that the design of RNAV structures for high altitude, low altitude, terminal and transition areas be closely coordinated. These designs should be created for the Phase 3 system and the implementation of routes in the transition phases should be based on the total Phase 3 system design to the extent practicable. Flight checking of routes should proceed in accordance with user requirements, but provisions should be made for the development of flight checked area coverage charts which will define areas and NAVAID coverage within which pre-planned direct routes may be eventually utilized without the requirement for continuous radar monitoring.

Investigative work is still required, particularly in the areas of controller display requirements, avionics standards and equipment MOCs, system accuracy determination and compliance criteria, and the development of optimum integrated enroute and terminal route structures. The route structure development to date was for the purpose of assessing RNAV impact and payoff, and although representative of an optimum structure, from that viewpoint, requires considerable expansion and coordination. Task Force recommended research and development efforts which have been completed to date include controller workload in a mixed VOR/RNAV environment, assessment of system and user payoff, development of terminal area guidelines, development of representative terminal area structures for payoff assessment, and development of recommended waypoint designation standards. These studies, as well as partially completed studies in the areas of avionics standards and system accuracy determination, have produced results which allow definition of an RNAV implementation concept in enough detail to allow it to proceed concurrently with other ongoing efforts. A summary of RNAV implementation requirements is given in Table 5.25, and discussed in the following paragraphs.

5.4.1 High Altitude Enroute

An immediate effort should be undertaken to develop and coordinate an optimum high altitude charted RNAV structure, based on the concept described in Section 5.3.1.1, and coordinated with low altitude, terminal and transition designs. Route development guidelines and tools developed by NAFEC should be employed in this process. Existing high altitude jet RNAV routes should be realigned to conform with the optimum total design or eliminated as appropriate. Additional routes should be implemented to meet user requirements as practicable, depending upon fulfillment of flight checking and charting requirements.

Phase 1

In Phase 1, VOR routes should be realigned as required as RNAV routes are implemented from the optimum route structure. Limited pre-planned direct flights will be accommodated, with radar monitoring as required. Flight planning responsibility for pre-planned direct flights will rest with the user, and waypoint selection will be in accordance with the concept described in Section 5.3.4. Pilot selection of 3D profiles will be allowed where not in conflict with ATC specified descent requirements, and tactical use of parallel offsets should be initiated. Limited resectorization should be accomplished to optimize control of traffic.

Phase 2

In Phase 2, 2D RNAV utilizing the charted high altitude structure will be the system, and all VOR routes will be deleted. Procedures should be developed to allow pilot selection of 3D descent and climb profiles. The evolution of the use of pre-planned direct will continue, and complete resectorization will be accomplished for optimization of routes.

Phase 3

Phase 3 is essentially unchanged from Phase 2 except that increased use of preplanned direct will be possible without radar monitoring. The Phase 3 RNAV concept calls for 2D charted RNAV as the system in the high altitude structure with pre-planned direct accommodated according to user requirements. It is anticipated that pre-planned direct flights will use existing waypoints, or VORTACs as waypoints, unless an operational advantage can be gained in using uncharted waypoints. All uncharted waypoints will be communicated by the 5 number rho/theta designator. Non-station-referenced systems that do not have the capability to input rho/theta waypoints should not use impromptu uncharted waypoints unless the capability of converting from rho/theta to an acceptable input format is maintained. Flight planning of pre-planned direct flights will be the responsibility of the user.

5.4.2 Low Altitude Enroute

An immediate effort should be undertaken to develop and coordinate with users an optimum low altitude charted RNAV structure, based on the concept

described in Section 5.3.1.1 and coordinated with the high altitude, terminal and transition designs. Route development guidelines and tools developed by NAFEC should be employed in this process.

Phase 1

Low altitude charted RNAV routes should be implemented from the optimum route structure in response to user requirements. The routes should be based on the optimum total design to the extent practicable and VOR routes realigned as required. All routes should be public use routes unless an operational advantage can be gained. Limited pre-planned direct flights should be accommodated, with radar monitoring as required. Flight planning of preplanned direct flights will be the responsibility of the user and waypoints will be selected in accordance with the concept described in Section 5.3.4. Charted waypoints, or VORTACs as waypoints, should be utilized for pre-planned direct flights. Limited resectorization should be accomplished as required. Tactical use of uncharted parallel offsets should be initiated. Pilot selection of 3D descents and climbs may be utilized where not in conflict with ATC specified descent requirements.

Phase 2

Phase 2 is a continuation of the evolution to wider use of charted RNAV routes and radar monitored pre-planned direct initiated in Phase 1. As RNAV routes exceed the number of VOR routes, unnecessary VOR routes should be deleted, the remaining VOR routes realigned to conform to RNAV routes to the extent practicable, and complete resectorization accomplished to optimize routes. Procedures should be established to allow pilot selection of 3D descents and climbs.

Phase 3

In Phase 3, 2D charted RNAV with a scaled down VOR airway structure will be the navigation system. As automation improvements are implemented and flight checking of areas is accomplished, pre-planned direct flights will be accommodated without the requirement for radar monitoring. It is anticipated that pre-planned direct flights will use existing waypoints, or VORTACs as waypoints, unless an operational advantage can be obtained. All uncharted waypoints will be communicated by the rho/theta designator as in the high altitude structure.

5.4.3 Terminal Area

Terminal area routes should be implemented as quickly as practicable in conjunction with high and low altiude RNAV routes and to overlie radar vector routes. 2D SIDs/STARs should extend from the enroute arrival or departure waypoint to the final approach waypoint or airport departure waypoints. Terminal area designs should be developed as described in Section 5.3.1.2, and routes implemented from these designs to the extent practicable. Route development guidelines and tools developed by NAFEC should be employed in this process. 2D/3D approaches should be designed for all runways, including ILS, to the extent practicable, consistent with IFR requirements.

Phase 1

Design 2D RNAV routes at all high and medium density terminals, coordinating designs with high and low altitude enroute designs. Implement 2D RNAV routes as required in conjunction with high and low altitude enroute implementation. Design 2D RNAV routes for low density and non-radar terminals as required for implementation. Realign VOR/vector paths to accommodate RNAV routes as practicable, or realign RNAV routes if necessary. Pilot selection of 3D descents may be utilized where not in conflict with ATC specified descent requirements. Establish RNAV procedures compatible with M & S terminals when need exists.

Phase 2

Automated metering/sequencing is assumed at selected high density terminals. Design 2D RNAV routes at all low density and non-radar terminals. Implement 2D RNAV routes as required in conjunction with high and low altitude enroute implementation. Realign VOR/vector routes as necessary to conform to 2D RNAV routes. 2D RNAV is the navigation system in high density terminals. Establish procedures to allow pilot selection of 3D descent and climb for A/C equipped with 3D.

Phase 3

Automated metering/sequencing is assumed at high and selected medium density terminals and conflict detection is assumed at all radar terminals. 2D RNAV is the system in medium density terminals. 4D may be required in M & S terminals.

5.4.4 RNAV Route Structure Development

The development of the RNAV route structure will have a direct and immediate impact on several services within the FAA as well as agencies and user groups outside of the ATC system. This impact is expected to start in the very near future, since the first major step required in order to implement RNAV in such a manner as to achieve benefits to the users and the ATC system, is to develop high altitude, low altitude and terminal area route structures for the Phase III implementation period. Development of these structures should be initiated immediately for the following reasons:

- The objective of the Phase III period is to have an optimum master route structure design (for all environments) implemented so that maximum benefits to all parties may be derived. The best and most direct way to accomplish this objective is to design the optimum structure first, and then implement individual routes as demand arises and as flight checking facilities permit.
- Knowledge of the form of the final objective route structure will allow for the orderly formulation and implementation of RNAV operational plans by all parties involved (users and system).
- Route structure development is a long lead-time operation by the very nature of the task, since extensive coordination between users and FAA headquarters and field personnel is required throughout the process. Since the Phase III structure is to be implemented (in part, at least) as the Phase II structure, the final result is needed in a relatively short time.

The impacts expected on the system and other groups as a result of the entire route implementation process will occur both during the route structure development phase, and as the structures are implemented. However, the types of impacts involved will be significantly different in each case as hypothesized below.

Route Structure Development Phase Impacts

<u>Air Traffic Service - FAA Headquarters</u>: Headquarters personnel should be responsible for the entire route design effort for all three environments (high altitude, low altitude and terminal area), and should interact with all other groups involved.

Air Traffic Service - Field Offices: Enroute Center and TRACON personnel will be involved intimately in the route structure development process in order that local airspace allocation, air traffic pattern and demand and restricted noise sensitive area problems are considered. They will also assist in the coordination between SRDS and Headquarters ATS in enroute/terminal interface which will insure a final structure which optimizes user economic benefits.

Systems Research and Development Service: The major role of SRDS is to support ATS in resolving technical issues arising out of the route structure development problem, including the application of enroute and terminal area design techniques and tools, and user benefit optimization techniques, which have been developed. Such issues are also expected to include determination of any additional automation needs for the new route structures, Navaid station requirements and route coverage problems, and avionics standards development.

National Aviation Facilities Experimental Center: NAFEC would be involved in supporting ATS and SRD3 in resolving technical issues, as above, including the application of route design tools which have been developed.

Flight Standards Service: AFS personnel should be relied upon to provide Navaid station coverage data for route structure support planning.

Military Organizations and User Groups (Air Carrier and General Aviation): These groups should provide consultation and inputs concerning routing reqirements and suitability of candidate structure designs.

Route Structure Implementation Phase Impacts

Air Traffic Service - FAA Headquarters: Headquarters should be responsible for the piece-by-piece implementation of the RNAV structures as user needs and capabilities demand, and as AFS flight checking progress allows. In the process VOR routes will be deleted or reoriented as user capabilities allow and as need arises for improved airspace organization. All required charting data will be provided to the appropriate government and private agencies. Required video map data shall also be compiled.

<u>Air Traffic Service - Field Offices</u>: TRACON and center personnel will coordinate implementation procedures with Headquarters. Region personnel will receive and process user requests and comments.

SRDS & NAFEC: These services will provide technical support to resolve local implementation problems, and will support automation improvements necessitated by the RNAV implementation process.

Flight Standards Service: Flight Standards will be responsible for flight checking all routes and instrument procedures before actual implementation. Additionally, they will carry out the orderly area coverage flight checking required for implementing RNAV direct flight plans.

Military Organizations and User Groups: Since the implementation schedule is intended to be influenced heavily by user needs, close contacts with these groups will be maintained throughout the route structure design and implementation phases.

Table 5.25

AREA NAVIGATION IMPLEMENTATION CONCEPT HIGH ALTITUDE ENROUTE (FL180-450)

Phase 1	Phase 2	Phone 3
A.@Design on optimum charted RNAV structure to accommodate all the high altitude traffic. Implement routes from the structure in response to user requirements. All routes should be public use routes. Enroute structure design should be cardinated with terminal designs.	A. 2D RNAV utilizing the charted high altitude structure will be the system.	A. *some as phase 2.
B. Whadign VOIL jet routes as required as RNAV routes are implemented.	B. Delete all VOR rautes.	B. same as phase 2.
C. Wilmited pre-planned direct flights accommodated, with radar surveillance. All pre-planned direct flights utilize charted waypoints, VORIACs, or radial/distance uncharted waypoints.	C. & Pre-planned direct where RNAV routes not designated, with radar surveillance. All pre-planned direct flights utilize charted way-points, VORTACs, or radial/distance uncharted waypoints.	C. *When necessary automation improvements are made, pre-planned direct RNAV routes will supplement the charted RNAN structure. All pre-planned direct flights utilize charted waypoints, VORIACs or radial/distance uncharted waypoints. Radar surviellance no langer required when route is flight when route is flight
D. Establish 2D SIDy/STARs for route continuity as necessary. Pilot selection of 3D descent may be utilized where not in conflict with 2D procedural descent requirements.	D. * Establish procedures to allow pilot selection of 3D descent for A/C equipped with 3D.	D. ' same as phase 2.
E. GGreat circle routes as practical-flat earth computation acceptable.	E. & All routes great circle unless user/system operational advantage exists - flat earth computation acceptable.	E. Osame as phase 2.
F. Route widths in accordance with 7110.18 - slant range correction required.	F. & Route widths ± 4 nm (constant) - waypoints as in section 5.3.4.	F. Brame as phase 2.
G. Designate parallel routes when necessary to meet traffic demands.	G. same as phase 1.	G. * same as phase 1.
H. & Parallel offsets (uncharted) will be utilized tactically.	H. Grame as phase 1.	H. & same as phase I.
1. Limited resectorization	. Complete resectorization for optimization of routes.	I. some as phase Z.
Bexpansion or clarific	Dexpansion or clarification of lask Force Concept	*modification of last Force Concept

Table 5.25 (continued)

AREA NAVIGATION IMPLEMENTATION CONCEPT

LOW ALTITUDE ENROUTE

	Phase 1	Phase 2	Phase 3
. Design an accommod ment route requirement for terminals.	A. \(\Partial \text{Design} \) an optimum charted RNAV structure to accommodate all the low altitude traffic. Implement routes from the structure in response to user requirements. All routes should be public use routes. Enroute structure design should be coordinated with terminals.	A. Gsame as phase 1.	A. * 2D charted RNAV with scaled down VOR airway structure will be the navigation system.
. Officer of routes are	B. @Realign VOR airways where required as RNAV routes are implemented.	B. Delete unnecessary VOR airways and realign remaining airways to conform to RNAV routes to extent practicable.	B.* same as phase 2.
with rada utilize ch	C. OLimited pre-planned direct flights accommodated with radar surveillance. All pre-planned flights utilize charted waypoints, VORIACs, or radial/distance uncharted waypoints.	C. 4 same as phase 1.	C.* same as phase 1.
. Great cir	Great circle routes as practicable - flat earth computation acceptable.	D. same as phase 1.	D. same as phase 1.
. * Route wid revised (n	E. *Route widths in accordance with 7110.18 as revised (ref. section 5.3.6). Slant range correction required above 12,500 ft. MSL	E. * same as phase I	E. * same as phase 1.
. A Parallel off tactically .	F. @ Parallel offsets (uncharted)may be utilized tactically.	F. Osame as phase 1.	F. 🕀 same as phase 1.
1	Limited resectorization.	G. Complete resectorization.	G. same as phase 2.
.* Pilot selection where not in requirements.	H.* Pilot selection of 3D descent may be utilized where not in conflict with 2D procedural descent requirements.	H.* Establish procedures to allow pilot selection of 3D descent for A/C equipped with 3D.	H.* same as phase 2.
G. Limited re H.* Pilot select where not requirement	esectorization. ction of 3D descent may be utilized in conflict with 2D procedural descent nts.	G. Complete resectorization. H.* Establish procedures to allow pilot selection of 3D descent for A/C equipped with 3D.	G. same

Table 5.25 (continued)

AREA NAVIGATION IMPLEMENTATION CONCEPT TERMINAL

	Phase 1	Phose 2	Phase 3
3 2	Assumes ARTS Capability at all high and selected medium density terminals.	Assumes metering/sequencing at selected high density terminals.	Assumes metering/sequencing at high and selected medium density terminals and conflict detection at old radar terminals.
i	A. *Terminal Design Concept (ref. section 5.3.1.2) implemented to extent practicable at all radar terminats.	A. *Terminal Design Concept (ref. section 5.3.1.2) implemented at all radar terminals.	A. same as phuse 2.
aci	Design 2D INAV routes at all high and medium density terminals, coordinating designs with high and low altitude enroute designs. Implement 2D RNAV routes as required in conjunction with high and low altitude enroute implementation. Design 2D RNAV routes for low density and non-radar terminals as required for implementation. Realign VOR/vector paths to accommodate RNAV routes as practicable, or realign RNAV routes if necessary.	B. Design 2D RNAV routes at all low density and non-radar terminals. Implement 2D RNAV routes as required in conjunction with high and low altitude enroute implementation. Realign VOR vector routes as necessary to conform to 2D RNAV routes. 2D RNAV is the navigation system in high density terminals. Establish climb envelopes to provide shorter departure routes for higher performance aircraft.	B. same as phase 2, plus 2D RNAV is the navigation system in medium density terminals.
نا	Pilot salection of 3D descents may be utilized where not in conflict with 2D procedural descent requirements.	C. Establish procedures to allow pilot selection of 3D descent for A/C equipped with 3D.	C. same as phase 2.
	D. Establish 4D procedures when need exists.	D. 4D capability may be required at M & S terminals.	D. same as phase 2.
	E. Whoute widths in accordance with 7110.18, as revised (see section 5.3.6). Slant Range correction required above 12,500 MSL.	E. * same as phase 1.	E. * same as phase 1.
	Establish 2D/3D approaches for all airports to the extent practicable, consistent with IFR requirements.	F. same as phase 1.	F. same as phase I.
	##	Chexpansion or clarification of Task Force Concept *modification of Task Force Concept	

CONCLUSIONS

6.0

The results obtained from economic and operational impact analyses, and from various supporting system studies, indicate that the advantages of area navigation to both the users and the ATC system are sufficient to warrant implementation of the area navigation concept described in this report. This overall conclusion is supported by the specific conclusions listed below:

User Payoff

- Substantial economic benefits are available to RNAV equipped users operating in both a high and low altitude 2D charted RNAV structure, and additional benefits are available through the use of preplanned direct routes. These benefits are not dependent upon full implementation of RNAV structures, but may be realized on individual routes as they are implemented. However, these benefits are dependent upon development of a master route structure from which routes are implemented as required.
- Substantial economic benefits are also available to RNAV equipped users operating in an RNAV terminal area environment. While the total savings described in this report are dependent upon full implementation of RNAV routes in the terminal area when a large enough percentage of the aircraft are equipped to warrant changing to an RNAV terminal area design, substantial benefits are available as individual terminal routes are implemented in conjunction with low and high altitude enroute segments.
- The use of pilot-selected 3D descents can result in substantial user savings in both fuel and time, particularly in RNAV terminal and transition areas. These savings are also available now through the use of VNAV in the VOR structure, and are not dependent upon full implementation of RNAV. The use of fixed gradient VNAV routes in a terminal area design imposes user penalties and does not increase airspace capacity. The benefits available in 2D RNAV terminal areas need not be compromised by the use of suboptimal 3D climbs or descents.
- Increased arrival operation rates and reduced arrival delays can be expected in a mixed RNAV/VOR environment, as well as in an all-RNAV environment.
- When properly implemented, the terminal area design concept described in this report will provide significant economic benefits to departing aircraft by providing minimum length departure routes with few climb restrictions.
- 2D and 3D approach procedures will increase safety through elimination of circling approaches, and will provide economic benefits through reduction of distance traveled during transition to approaches.
- The use of 4D time control navigation can provide a significant reduction in arrival delays in M & S terminals, and will provide substantial economic benefits to the user.

- RNAV can directly reduce departure ground delays induced by enroute and transition traffic congestion through the ability to utilize parallel routes. This same ability will allow more optimum altitude assignments.
- The cost of acquiring and maintaining the RNAV equipment required by the operational concept described in this report is nominal compared with the overall benefits available to all classes of user.

System Impact

- Contrary to the concerns expressed by the RNAV Task Force, controllers will be able to function efficiently in a mixed VOR/RNAV environment with reduced workload and with an increase in system capacity.
- A continuing controller training program will be necessary to provide a level of controller familiarity with RNAV procedures which will insure the timely evolution of an all-RNAV environment.
- Lack of slant range correction in RNAV equipment below 12,500 feet will not have an adverse effect on terminal area or low altitude enroute route design.
- The use of pilot selected VNAV gradients will provide significant user benefits while having no impact on ATC system operation.
- The implementation of RNAV will allow substantial savings in VORTAC costs in the terminal area, and the cost to provide VORTAC coverage in the high altitude enroute structure is nominal when compared with the projected F & E budget for Navigation Aids in support of the VOR airways structure. These costs can probably be offset entirely by maintenance savings incurred through the removal of redundant high altitude VORTACs.
- The impact of RNAV on ATC automation is nominal during the transition phase of a mixed VOR/RNAV environment, and the enroute and terminal computer core requirements for an all RNAV structure are less than those for the current VOR structure. This additional capability could then be used for other system growth requirements.
- The increase in airspace capacity necessary to accommodate expected traffic demand in the 1980's can be provided by an RNAV structure with enroute widths of a constant +4 nm and terminal route widths of a constant ± 2 nm or ± 4 nm depending upon distance from the VORTAC.
- The use of fixed gradient VNAV routes imposes user penalties and generally does not increase airspace capacity. Terminal area routes should be designed to allow pilot selection of 3D climbs and descents.

Operational Concept

- The realization of the benefits available to both the system and the users is dependent upon implementation of the operational concept as described in this report, which is in turn dependent upon early development of a total optimum RNAV structure.
- A critical first step in the RNAV implementation process is the development of optimum, complete master high altitude and low altitude charted RNAV structures. These structures must be closely coordinated with the development of terminal area structures for the appropriate high and medium density terminals. Individual routes should be implemented from the master structures.
- Although further research and development efforts are necessary in the areas of avionics standards, system accuracy, and impact on other elements of the upgraded third generation system, the implementation of the RNAV concept can proceed in parallel with these efforts.
- The system design concept that should be implemented is a modification of the Task Force Concept, which consists of a more flexible terminal area structure, which is optimized for traffic flow and user economics, and charted high and low altitude structures. Preplanned direct flights should not be as widespread as envisioned by the Task Force, and, when utilized, should be based primarily on use of charted waypoints and Navaids.

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APPENDIX A

TERMINAL AREA AIRCRAFT PERFORMANCE DATA

The data tables used in the analysis of terminal area altitude restrictions are contained herein. The high speed climb and descent tables list the distance required to climb from sea level to the indicated altitude, or vice versa for descent. The data is taken from Reference 12, and includes the 250 KIAS speed limit below 10,000 feet. The descent data below 10,000 feet was modified for the aircraft when needed, increasing the descent rate to 400 feet per mile (reflecting the more realistic procedure of descending at a lower speed and higher descent rate in the terminal area). Time and fuel data were not used.

The altitude restriction penalty tables list the time and fuel penalties during climb and descent. In each case they represent the difference between the cruise speed and fuel consumption rates at the restriction altitude and at a nominal cruise altitude and weight. The conditions used are listed in the table below:

Table A.1 HIGH ALTITUDE CRUISE CONFIGURATIONS

		For Climb Case	2		For Descent	Case
Aircraft	Mach	Weight	Altitude	Mach	Weight	Altitude
DC-9	0.77	110 K1b	FL 270	0.77	90 K1b	FL 350
B-727	0.80	160 K1b	FL 270	0.80	140 K1b	FL 350
DC-8	0.80	300 K1b	FL 310	0.80	220 K1b	FL 390
B-747	0.84	650 K1b	FL 310	0.84	550 K1b	FL 390
DC-10						FL 390
F.28						FL 390
Lear 25						FL 410
FH 227						FL 200

Both sets of tables are also provided for the long range descent case for the DC-9, B-727, BC-8 and DC-10.

	ALTITUDE	TIME (Hours)	DISTANCE (nm)	FUEL (Pounds)	AIRCRAFT
CLIMB					
	0.	0.1.11	0.0	0.	DC-9
	1000.	.0004	1.66	88.	DC-9
	200 .	.0128	3.52	176.	DC-9
	3001.	.0192	4.98	264.	DC-9
	400 .	.0256	6.4	352.	UC-9
	50	. 321	8.30	440.	UC-9
	600 .	,0392	10.32	528,	DC-9
	701.	.0464	12.34	616.	UC-9
	800 .	. 0536	14.56	704.	DC-9
	900 .	0618	16.38	792.	DC-9
	100/0.	.0680	18.40	850.	UC-9
	1100 .	.0922	27.08	1162.	UC-9
	1200 .	.1014	30.56	1264.	UC-9
	13000.	•1106	34.04	1366.	DC-9
	1400 .	.1198	37.52	1468.	DC-9
	1500	. 1290	41.00	1570.	DC-9
	1600 .	.1406	45.78	1690.	DC-9
	1700 .	.152	50.56	1810.	DC-9
	1800	.1638	55.34	1930.	DC-9
	1900 .	•1754	60.12	2050.	DC-9
	20000.	•1870	64.90	2170.	DC-9
DESCENT					
	0.	0.00	0.00	0.	DC-9
	10 .	•0096	2.50	18.	UC-9
	200 •	•0192	5.07	36.	DC-9
	3000 · 400 · .	•028	7.50 10.3	54. 72.	DC-9
	5000.	.0384	12.50	90.	DC-9
	60 .	.0572	15.02	104.	DC-9
	70 .	.0634	17.54	1:8.	UC-9
	800 .	.0756	20.6	132.	DC-9
	90 10 .	.0848	22.58	146.	UC-9
	100 10.	.0940	25.10	160.	DC-9
	1100 .	•1144	31.74	186.	DC-9
	120	1 98	3 . 78	193.	DC-9
	130 1 .	.1252	35.82	191.	DC-9
	140 10.	.1306	37.86	206.	DC-9
	15000.	•1360	39.90	213.	UC-9
	1600 .	.1410	41.98	218.	DC-9
	170 .	.1460	44.06	223.	UC-9
	180"	.1510	46.14	229.	UC-9
	190 16.	.1560	48.24	234.	DC-9
	200 .	.1610	50.30	239.	DC-9

	ALTITUDE	TIME (Hours)	DISTANCE (nm)	FUEL (Pounds)	AIRCRAFT
CLIMB					
	0.	0.000	0.0:	0.	8727
	101.	. 1 6 1	1,00	121.	8/21
	2000.	.0120	3.20	242.	8/2/
	30 10.	.0181	4,80	303.	3727
	400	.024	6.40	48	8721
	5001.	. U.S	8.00	605.	13721
	601.	.0379	9.86	728.	8727
	70 .	.04+0	11.72	851.	8/2/
	80.1 .	.0510	13.58	974.	B727
	901.	• 0580	15.4	1097.	8121
	1000.	•0650	17.30	12 0.	8727
	1100 .	.0774	21.92	1436.	B727
	120) .	.0898	20.54	1652.	3727
	1300.	.1122	51.16	186 .	8727
	14000.	.1146	35.78	2084.	8727
	1500	.1270	40.40	2300.	8/2/
	1607).	.1372	44.92	2461.	3727
	17000.	.1474	49.44	262 .	B727
	18000.	.1576	53.96	2783.	8/2/
	19000.	•1678	58.48	2944 .	B727
	20) .	•1780	63.0	3105.	B721
DESCENT					
	0.	0.0.0	0.0	0.	8727
	100 .	.0112	2.50	32.	8727
	200 .	.0224	5.00	64,	3727
	300 .	.0336	7.50	96.	8727
	40 7 .	•0448	10.00	128.	13727
	50 1 .	.0560	12.50	160.	H15
	60 F	07-1	15.26	182%	3727
	8001.	.0764	18.02	204.	B727
	900 •	•0860 •0968	20.78	2'0.	8727
	100 .		23.54	248.	6727
		,107	20.30	270.	13/27
	120 0.	1370	30.48	292.	3/2/
		•1274	34.6	.514.	8727
	13000	.1376	38.84	500.	8727
	14011 •	.1478	43.02	358.	8727
	1500	•1580	47.20	380.	8/2/
	160 11.	.16!	49.14	390,	B727
	1700	• 1652	51.08	400.	8/2/
	180 .	1700	53.02	410.	8727
	1900 •	.1724	54.90	420.	8/21
	200	•1760	56.90	430.	H127

	ALTITUDE	TIME (Hours)	DISTANCE (nm)	FUEL (Pounds)	AIRCRAFT
CLIMB					
	0.	0.7.0	0.20	0.	DC-8
	1000.	,0.64	1,76	2.8.	DC-8
	20 1 .	.0128	3,52	456.	DC-8
	30 1	.0192	5.28	684.	UC-8
	40 .	.0256	7.04	912.	UC-8
	500 .	.0320	8.00	1140.	DC-8
	600 .	.0398	10.91	1366.	UC-8
	700 .	.0476	13.00	1592.	DC-8
	800 .	.05 14	15.10	1818.	DC-8
	901	.0632	17.20	2044.	DC-8
	1000 .	.0710	19.30	2270.	UC-8
	11000.	.0912	25.76	2824,	UC-8
	1200 .	.0904	28.62	3058.	UC-8
	150 .	.1076	31.48	5292.	UC-8
	1400	.1158_	34.34	3526.	UC-8
	150).	.1240	37.20	3760.	UC-8
	1600 .	.1344	41.34	4016.	DC-8
	1700 .	.1448	45.48	4272.	DC-8
	180 1).	.1552	49.62	4528.	DC-8
	19000.	.1650	53.70	4784.	UC-8
	200	.1760	57.90	5040.	DC-8
DESCENT	200		37.		000
	0.	0.000	0.00	0.	DC-8
	100	.0130	2.50	59.	DC-8
	2000.	.0260	5.00	117.	DC-8
	300 .	,039)	7,50	170.	DC-8
	4000.	.0521	10.1.	234.	DC-8
	500 .	.0650	12.50	293.	DC-8
	607 •	.0760	15.74	5.59.	UC-8
	700 .	.08.2	18.91	385.	DC-8
	8070.	.0998	2 . 2	430.	DC-8
	901 .	.11 4	25,40	476.	UC-8
	10 .	.1250	28.70	52.	DC-8
	11000.	•1514	38.60	630.	DC-8
	1200 .	.1568	40.80	652.	UC-8
	130 11.	.162?	43.00	073.	DC-8
	1400 .	.1670	45,20	695.	UC-8
	1500	.1730	47.40	716.	DC-8
	160 .	.1780	49.8	734.	DC-8
	1700 .	.1846	52.36	752.	DC-8
	18090.	.1904	54.84	771.	DC-8
	19001.	.1962	57.32	789.	UC-8
	20 10 1.	.20	59.80	807.	nc-a

	ALTITUDE	TIME (Hours)	DISTANCE (nm)	FUEL (Pounds)	AIRCRAFT
CLIMB					
	0.	J.	U.U.	0	13/4/
	10 .	.0056	1.40	201,	8747
	200 .	,01(2	2,80	520.	13/4/
	300 .	,0168	4.20	780.	8/4.
	400	.02.4	5.00	104 -	B/4/
	50) .	.0280	7.0	130	8/4/
	60 10.	.036	9,40	1700.	8747
	70 1	.0452	11.80	210 .	B/47
	80 10.	.0538	14.20	250 .	8/47
	90 .	, 9624	10.00	290 1.	8/4/
	10 01.	.071	19.0	350 1	8747
	110.10.	.1032	31.20	4890.	13/4/
	1200 .	.1144	35,40	5380.	8747
	130 .	,1256	39,60	5870.	8747
	14001.	.1368	43.80	65 0.	8747
	15000.	.1480	48.00	6850.	8747
	16000.	.1614	54.20	7410.	8747
	1700 .	.1748	60,40	7970.	B/4/
	18000.	.1882	65.40	8530.	13747
	19000.	.2016	72.80	90 -0.	8747
	200 .	,2150	79.00	9050.	B747
DESCENT				, , , , ,	
	0.	0.07	0.0	0.	8747
	100 .	.0 186	2.20	34.	8747
	2091.	.0172	4.40	68.	8747
	3000.	. 1258	6.60	102.	8747
	4000.	.034	8.80	156.	B747
	5000.	.0430	11.00	170.	3/4/
	6000.	• 0540	14.18	204.	8747
	70 : .	. 1650	17.36	238.	8747
	8000.	,0760	20,54	27.	13747
	900 .	.0879	23.72	306.	8/4/
	1.000	.0980	26,90	340.	8747
	1.00%	.1308	37.80	440.	8/4/
	120 9.	• 1.36	40.00	462.	13/4/
	1300 .	.1424	43.40	478.	13/47
	1400 .	•1482	46.20	494.	8747
	1500 .	•1540	49.0	510.	8/4/
	16000	.1572	51.50	522.	8741
	1700 .	.1604	54.00	534.	8747
	1800 .	.163	56,50	540.	13741
	190 .	.1608	59.00	5.18.	8747
	2011 .	•170	61.50	570.	13/4/

	ALTITUDE	TIME (Hours)	DISTANCE (nm)	FUEL (Pounds)	AIRCRAFT
CLIMB					
	0.	0.000	0.00	0.	UC10
	100 .	•0056	1.50	220.	DC10
	2000.	.0112	3.00	440.	DC10
	300 .	.0168	4.50	600.	UC10
	4000.	.0224	5.00	880.	UC10
	50.50	.0280	7.50	1100.	DC10
	600 .	.0348	9.30	1326.	UC10
	700 .	.0416	11.10	1552.	UC10
	800.	•0484_	12.90	1778.	DC10
	9011.	.0552	14.70	20.14.	UC10
	1000.	.0627	16.50	2230.	DC10
	1100.	.088	26.20	308 .	UCIU
	1200 .	. 1960_	29.40	3540.	UC10
	13000.	.1044	32.60	3600.	DC10
	14000.	.1122	35.80	3860.	UC10
	15000.	•1200	39.00	4120.	UC10
	1600 .	•1308	43.70	4440.	DC10
	1700 •	•1416	48.40	4760.	UC10
	18000.	•1524	53.10	5080.	DC10
	19000.	.1632	57.80	5400.	DC10
	2000 .	.1740	62.50	5720.	UC10
DESCENT					
	0.	0.0	0.00	0.	UC10
	1000.	.0132	2.50	81.	UC1U
	2003.	.0264	5.00	162.	DC10
	3017.	.0396	7.50	241.	DC10
	4000.	.0528	10.07	325.	DCTO
	5000.	.0650	12.50	406.	UC10
	60 1 .	.0782	15.82	460.	UC10
	700 .	, 1904	19.14	520.	UC10
	8000.	•1026	22.46	586.	UC10
	90:) .	.1148	25.78	640 •	UC10
	10000.	.1270	29.10	706.	DC-10
+	1100 .	•1590	40.41	848.	UC10
	1200 .	.1650	42.8	876.	DC10
	130	.1710	45.32	905.	DC10
	140 .	•1770	47.76	953.	DC10
	1500	•1830	50.20	961.	DC10
	16001.	.1850	52.74	982.	DC10
	170 10.	.1930	55.28	1003.	UC10
	1800 •	.1980	57.82	1024•	UC10
	190.10.	.2031	60.36	1045.	UC10
	2000.	.2080	62.90	1000.	DC10

	ALTITUDE	TIME (Hours)	DISTANCE (nm)	FUEL (Pounds)	AIRCRAFT
CLIMB					
	0.	0.0.00	0.10	0.	F.28
	100	.0047	1,14	45.	F.28
	200 .	• 0 95	2,28	90.	F.28
	300	.0141	5.42	135.	F.28
	400 .	.0186	4,56	180.	F.28
	500 .	.0233	5.70	2.5.	F.28
	600 .	.0280	7.04	208.	F.28
	70 -	. 1527	8,36	51).	F.28
	800 .	.0373	9.72	354.	F.28
	9004.	.0420	11.06	397,	F.28
	1000 .	.0467	12.40	4+0.	F.28
	11000.	.0520	14.08	486.	F.28
	1200).	.0573	15.76	532.	F.28
	13000.	.0627	17.44	578.	F.28
	1400 .	.0680	19.12	624,	F.28
	1500	.0733	20.80	670.	F,28
	16000.	.0800	23.20	720 •	F.28
	17000.	.0867	25.60	770.	F.28
	18001.	•0935	28.00	820.	F•28
	1900	.1000	30.43	870.	F.28
DESCENT	2001 .	.1067	32.80	920.	F.28
DESCEIVI	0.	0.0000	0.00	U.	F.28
	100%	.0090	2.00	19.	F.28
	2007.	,0180	4.00	38.	F.28
	300 .	.0270	6.00	57.	F.28
	4001.	.0360	8.0	76.	F.28
	5000.	.0450	10.70	95.	F . 28
	600	. 3543	12.60	113.	F.28
	70 00 •	.0637	15.20	131.	F.28
	807 .	.073	17.80	149.	F.28
	900	.0824	20.40	107.	F.28
	10.0.	.0917	23,0	185.	F.28
	1100 .	.10	20.00	215.	F.28
	1200	.1103	30.20	247.	F.28
	13000.	.1197	35.80	2/8.	F.28
	1400 .	.1290	37.40	309.	F.28
	15000	.1383	41.07	340.	F.28
	160 .	.1476	44.80	570.	F.28
	1700 .	.1570	48.60	402.	F.28
	1800 .	.1663	52.40	430.	F.28
	190 10.	.1757	56,20	460.	F.28
	2000	•1850	60.00	490.	F.28

	ALTITUDE	TIME (Hours)	DISTANCE (nm)	FUEL (Pounds)	AIRCRAFT
CLIMB					
	0.	0.0 00	0.00	0.	LEAR
	10().	.0133	1.20	16.	LEAR
	200 .	.0067	2.40	32.	LEAR
	300	.0100	3.60	48.	LEAR
	4000.	.0134	4.80	64.	LEAR
	500 .	.0167	6.00	80.	LEAR
	6011.	.0197	6,80	94.	LEAR
	70	, 1227	/.60	108.	LEAR
	800: .	.0257	8.40	122.	LEAR
	9000.	.0287	9.20	136.	LEAK
	1000 .	.0317	10.00	150.	LEAR
	11000.	. 0360	11.71	160.	LEAR
	1200 .	• 0434	13.42	181.	LEAR
	1300 .	• 0447	15.13	197.	LEAR
	14000.	.0490	16.84	212.	LEAR
	15000.	.0534	18.55	228.	LEAR
	16000.	.0577	20.26	243.	LEAR
	17000.	.0620	21.97	259.	LEAR
	18000.	.0663	23.08	274.	LEAR
	19000.	.0707	25.39	290.	LEAR
	20001.	.0750	27.10	305.	LEAR
DESCENT		•			
	0.	0.0000	0.00	0.	LEAR_
	1000.	.0073	2.40	11.	LEAR
	2000.	.0147	4.80	22.	LEAR
	300 .	.0220	7.20	33.	LEAR
	400).	.0294	9.60	44.	LEAR
	5000.	.0367	12.0	57.	LEAK
	6000.	.0440	14.40	65.	LEAR
	700 .	.0513	16.80	76.	LEAR
	800 •	• 0587	19.20	86.	LEAR
	90 .	•0660	21.60	97.	LEAR
	100	.0733	24.00	107.	LEAR
	11000.	.08.6	26.81	110.	LEAR
	1200 .	.0880	29.62	125.	LEAR
	1300 .	.0953	32.43	154.	LEAR
	140.1	.1327	35.24	143.	LEAR
	150 .	.1100	38.05	153.	LEAR
	16000.	.1175	40.86	162.	LEAR
	1700 .	.1247	43.67	171.	LEAR
	1800 .	.1320	46.48	180•	LEAR
	1900 .	.1394	49.29	189.	LEAR
	2001 .	.1467	52.10	198.	LEAR

	ALTITUDE	TIME (Hours)	DISTANCE (nm)	FUEL (Pounds)	AIRCRAFT
CLIMB					
	0.	0.30	0.1.	0.	F2. 7
	10).	.0115	1.83	28•	F2 7
	2000.	.0230	3.65	50.	F227
	301.	.0344	5.48	84.	F2 7
	400 .	. 0459	7.30	112.	F227
	5000.	.0594	9.43	159.	F2 1
	600 .	.0730	11.50	165.	F2 7
	700 .	. 0865	13.68	192.	F227
	8000.	.101	15.80	218.	F221
	9000.	.1154	18.35	248.	F2 1
	10 1710.	.1309	20.90	279.	+21
	1100 .	.1463	23.45	309.	F2.7.
	12070.	.1617	26.00	359.	F2.7
	13000.	.1809	29.50	374.	F2 7
	140 11.	.201	33.00	410.	F2_7
	150) .	,2192	36.50	445.	F2.7
	1600 .	.2385	40.00	480.	F2_7
	17000.	.2646	41.80	524.	F2_1
	1800 .	.2908	49.60	569.	F227
	19000.	.3171	54.40	613.	F2.7
	2000.	.34.10	59.20	657.	F2.7
DESCENT					
	0.	0.0000	0.02	0.	F227
	100%	. 2067	1.90	12.	F2.17
	200 .	.0133	3.80	24.	F2:7
	30.11.	.0203	5.70	36.	F2 7
	4000.	.0266	7.60	48.	F2 7
	50	.035	9,50	60.	F2 7
	6070.	.0416	11.90	74.	F2 7
	700 •	.050	14.50	8	F221
	80	.0583	10.70	102.	F227
	9000.	.06.7	19.10	116.	F2 /
	10/10	• 0750	21.50	130.	F227
	11007	. 983	23.84	143.	F2 7
	1200	.0910	26.18	156.	F227
	130	.09-10	28.53	168.	F2 7
	1400 .	.107	30.86	181.	F22/
	15000.	•1150	33.20	194.	F227
	16070.	.1233	35.76	205.	F2:/
	17000.	.1317	38.32	216.	F22/
	1800 .	.1400	40.88	228.	F2.7
	19000.	.1484	43.44	239.	F2_/
	2000.	.1567	46.00	250.	F2 17

	ALTITUDE	△-TIME (Hours)	△-FUEL (Pounds)	AIRCRAFT
CLIMB				
	0.	.0 1826	10.529	DC-9
	100	.001772	9.947	DC-9
	2001.	.0117:8	9.364	DC-9
	3001.	,001605	8.702	UC-9
	400 .	.051611	8.19"	UC-9
	50011.	.001557	. 7.617	DC-9
	600	.0.15.3	7.118	UC-9
	70	.0 1449	6,019	DC-9
	800 .	1394	6,120	UC-9
	9000.	.001340	5.621	DC-9
	1000	.0 11286	5.12	UC-9
	1100	.0 285	5.344	UC-9
	120.0.	.0 13250	4.976	DC-9
	13000.	.01 216	4.009	UC-9
	14000.	.0 181	4.241	DC-9
	15000.	.01.146	3.873	DC-9
	1600 .	.00 114	3.589	DC-9
	17007.	.000.82	3.305	DC-9
	1800).	.000.51	5.022	DC-9
	19000.	.00 018	2.738	DC-9
	-2001 .	0 3614	2.454	UC-9
DESCENT				
	0.	.001748	11.708	DC-9
	100 .	.001694	11.158	DC-9
	2000.	.001640	10.608	DC-9
	3000.	.0.1587	10.057	DC-9
	4001.	.0 1533	9.507	DC-9
	5000.	.001479	8.957	DC-9
	60 1.	, 1425	8.481	UC-9
	700 .	.0.1371	8.005	UC-9
	800 .	.0 1516	7.528	DC-9
	90.) .	.0 1262	7.052	DC-9
	1000 .	.0 1208	6.570	UC-9
	1.100.	.00 207	8,090	DC-9
	120 0.	172	7.65/	DC-9
	13001.	.000138	7.2 5	DC-9
	1400 .	. 10 1 3	6.792	UC-19
	1500 .	.0. 68	6.359	UC-9
	1601 .	.01 56	6.068	UC-9
	1700 .	.011 4	5.7/7	UC-9
	1800 .	0 // 28	5,485	DC-9
	190 10.	010 60	5.194	DC-9
	2010.	01 92	4.903	DC-9

	ALTITUDE	△-TIME (Hours)	△-FUEL (Pounds)	AIRCRAFT	
CLIMB					
	0.	.0 1908	15.795	8/2/	
	10,	.) 1854	14.971	8727	
	200 .	.0018	14,147	13/27	•
	300 .	.0/174/	13, 32	872	
	40 .	.0 1693	12.49	13/27	
	5000.	.001639	11.6/5	8727	
	60 ;	.0 (158)	10.965	13721	
	700 :	.0 1531	10.255	8/27	
	8000.	.0 1470	9.515	6727	
	90 10 .	.00142	8.036	8727	
	10	.0 1368	8.126	B727	
	1100 .	,01 75	8.997	B727	
	12000.	.01 14	8.575	8/2/	
	13000.	.00 0 7	8.152	B727	
	14000.	00 27	7.730	B727	
	1500 .	010 60	7.307	B727	
	1600 .	010 59	6.53	8727	
	17000.	01 158	5.99	8727	
	1800 .	-,00 057	5.340	B/27	
	1900-1.	-,00 150	4.692	872,	
	2000-	-,001 55	4,038	8727	
DESCENT					
	0.	.0 1831	16.536	8727	
	100 .	.0 1777	15.742	8727	Comment
	20 .	.0.1723	14.9.8	8727	0
	3000.	.011679	14.154	13727	()
	401 .	.J 16L	15.500	B727	
	5070.	•0 1562	12,500	H721	Lookers
	60 / •	.0 1508	11.873	8/2/	00
	70 .	.0 1454	11.180	B727	-
	800 .	.001397	10.487	B727	
	90 1.1.	. 101345	9.794	B727	CHARLES
	100).	.0.129	9.101	B727	Total Control
	1(0)0.	0 . 4	11,451	B727	-
	120	-,00 37	11.008	8727	Commission
	13000.	031 71	10.564	B727	1
	14000	00.104	10.121	B727	1
	1500 .	0 10137	9.671	8727	lender
	1607 .	0.0136	8.9.7	8727	00
	17000.	-,00-135	8.318	8727	
	1800 •	-,03 135	7.038	8727	
	190 10.	00 134	6.959	B727	
	20 .	0133	6.279	B727	

		ALTITUDE	△-TIME (Hours)	△-FUEL (Pounds)	AIRCRAFT
CI	IMB				
		0.	0 1040	27 000	
		100 .	.0 1868	27.084	UC-8
CJ		200 •		27.117	UC-8
177		300 .	.0 1707	27.151	DC-8
		40.		27.184	DC-8
-		50 .	•0 (1653	27.218	UC-8
Mary Company			.001590	27.251	UC-8
REST CONTRACTOR		000 .	0.101545	24.983	DC-8
Magazza		7030.	.0 1491	2 - 715	DC-8
Marie A county		80 .	.0 11436	20.406	UC-8
-		900	.0.1382	18.178	UC-8
-		1000.	.001328	15.910	DC-8
		120		13.671	DC-8
Existend .		13000.	.01/262	12.79	DC-8
		1400 .	.00 177	11.927	DC-8
-		15000.	.0 0135	11.054	DC-8
(16000.	.000.93	10.182	00-8
Becquies		17000.	.00 51	9.664 9.146	DC-8
		18000.	.000 9	8.629	DC-8
2.1		1900).	000/33	8.111	DC-8
		2007 .	000 75	7.593	UC-8
DE	ESCENT		. 000 773	7.535	00-8
		0.	.0/1819	27.489	DC-8
		100 .	.) 1765	26.274	DC-8
		200 .	.0 1711	25.06	DC-8
		500%	.001658	23.845	DC-8
		400	.011604	24.631	DC-8
		5001.	.001550	21.416	DC-8
		600 .	•071496	20,471	DC-8
		70 .	.0 14/12	19.526	UC-8
		80 /4.	.0 1387	18.582	DC-8
		900 .	.001333	17.63/	DC-8
		101 .	.0 1279	16.692	DC-8
		110 0.	.00 250	18.030	DC-8
		120	.0 /0213	18.054	DC-8
		1300 .	.0 (0171	17.2/	DC-8
		140 .	.0.10129	16.501	00-8
		150 .	.000.87	15.724	DC-8
		1600 .	.011045	15.182	UC-8
		1700.	.01113	14.639	DC-8
		1800 .	01 39	14.097	DC-8
		1900 .	000.81	13.554	UC-8
		20 00.	0.0123	13.012	UC-8

	ALTITUDE	△-TIME (Hours)	Δ-FUEL (Pounds)	AIRCRAFT
CLIMB				
	0.	.0"1972	62,416	B747
	1001.	.0 1918	59.782	1374
	200 .	.0 1864	57.148	13/4/
	30%,	.3 181	54.514	15/47
	40 .	.0 .1757	51.8.0	13/47
	50 0.	.0 1703	49.246	8747
	00 1.	.0 :1649	40.984	8747
	70 0.	.031595	44.72	B747
	80	,001540	42.460	B747
	90	. 1486	40.198	8747
	10 000.	.0 1432	57.936	13747
	1100.	.0 11227	21.164	8747
	120 U.	.05 194	20.162	8747
	130 0.	.011161	19.160	B747
	140).	.0 / (128	18.150	B74/
	150 .	•00:195	17.156	B741
	160 % .	.01 (6	16.758	8/4/
	1/00 .	.00 025	16,360	B747
	18000.	03 839	15,962	B747
	1900 .	0.1 .44	15,004	8747
DECCENIE	2000 .	1) (11.79	15.160	B747
DESCENT				
	0.	•0 (1925	42.390	B747
	1000.	•0 11871	40.306	B747
	2010.	.001818	58,202	8747
	30) .	.001764	36,228	B747
	4000.	.00171	34.174	1374
	50 .	.091657	32.120	B741
	600 .	. 1 1603	10.5	13747
	70 .	•0 -1549	28.550	B741
	80	.0.1494	26.704	8747
	904 •	00 14 10	24.979	8/4/
	100 0.	.0 1386	23.194	B747
	1.000.	.09 (181	26.741	13747
	120 0.	.0.0148	25.7.7	13/4/
	130	. 0. 0115	24.7/2	13747
	140	.01 081	23.788	8741
	15000.	.0 10 148	22.803	B74 /
	160 .	.01 013	22.524	B747
	1/00 .	-,00 021	22.245	. B747
	1800 .	010 150	21.967	8747
	1900 .	000 90	21.68	8/4/
	200 0.	0 ₹ 125	21.409	B74/

	ALTITUDE	Δ-TIME (Hours)	△-FUEL (Pounds)	AIRCRAFT
CLIMB				
	0.	.031934	27.306	UC10
	100	.0.1830	26.150	UC10
CO	200).	.001826	25.006	UC10
	30 .	.071773	23.856	UC10
	4000.	.0 11719	22.106	DC10
	500).	.001605	21.506	DC10
Mary State of the	6000.	.001611	20.490	DC10
100 (d)	7000.	.0 (15)7	19.424	UC10
No.	800 .	.0115 2	18.358	UC10
BASE CALCULAR	901 .	.011418	17.292	UC10
	100	.001394	16.226	UC10
	1100 .	.00.463	13.630	UC10
	12000.	.03 427	12.846	DC10
and the same of th	1300 .	.0 1392	12.063	DC10
1 4 4	1400 .	.000356	11.279	DC10
(15010.	.00.321	10.496	DC10
0	1600	.000288	9.826	DC10
- Control	1700 .	.00 1255	9.157	UC10
-	18000.	.000222	8.437	DC10
	19000.	.000189	7.818	DC10
	200	.010156	7.148	DC10
DESCENT				
	0.	.001895	25.858	DC10
	100 .	.001841	24.841	DCTO
	2000.	.001787	23.825	DC10
	300 1.	.0 1734	22.806	DC10
	4000.	.001680	21.788	DC10
	500 .	·0 1162n	20.7/1	DC10
	60 10 .	.0.1572	19.847	DC10
	700 .	.001,518	18.924	UC10
	8090.	.0 11463	18.00	UC10
	90.0.	• 0014119	17.077	DC10
	10 .	.001355	16.153	DC10
	11000.	.01.1423	17.094	DC10
	12000.	.00 388	16.382	DC10
	130	. 1 352	15.6.9	UC10
	14006.	.01 317	1.4.957	DC10
	15000.	•000281	14.245	UC10
	160) .	.000248	13.635	UC10
	170	.00 .215	13.024	DC10
	1800 .	.0 183	12.414	UC10
	190 .	.000150	11.803	DC10
	20 0 .	.000117	11.193	UC10

	ALTITUDE	△-TIME (Hours)	△-FUEL (Pounds)	AIRCRAFT
CLIMB				
	0.	.001630	7.391	F.28
	100).	.0/1576_	6.854	F.28
	2000.	.001523	6.317	F.28
	3001.	.001469	5.780	F.28
	400 .	.011416	5.243	F.28
	500 .	.0:1362	4.700	F • 28
	6000.	.0 1308	4.396	F.28
	7000.	.0 1254	4.087	F.28
	8000.	. 331199	3.777	F.28
	9000.	.011:45	3.468	F.28
	1000 .	.001 91	3.158	F.28
	14000.	.0.9311	3.566	F.28
	120	.0 10274	3,519	F.28
	1300 .	.01 236	3.112	F.28
	1400 .	· du 199	2.835	F.28
	15000.	.00 1162	2.658	F • 28
	16000.	.011126	2,501	F.28
	17000.	.009	2.413	F.28
	1800.	.00.755	2.356	F.28
	1900.	.000 19	2.28	F.28
	200 .	00 17	2.12	F.28
DESCENT				
22002111	0.	.001608	8.627	F•28
	100 .	.0 11504	8,115	F.28
	200	.0115	7.602	F.28
	3000.	.0 1447	7.090	F.28
	4090.	•071393	0.577	F.28
	5000.	.0 11359	6.065	F.28
	6070.	•0 11285	5.078	F • 28
	/000.	.0 1231	5.291	F.28
	800	.0 1176	4.915	F.28
	9000.	. J 1:22	4.518	F.28
	10000.	.0 1068	4.131	F.28
	1(00.	•01-288	5.357	F.28
	12000.	•070251	5.103	F.28
	13000.	.000213	4.850	F.28
	14000.	.0.0176	4.596	F.28
	15000.	.0 139	4.342	F,28
	16000.	.0 1 5	4.194	F.28
	17000.		4.046	F.28
	180 0.		3.898	F.28
		001 03	3.750	F • 28
	20 0.	0 39	3.602	F.28

	ALTITUDE	△-TIME (Hours)	Δ-FUEL (Pounds)	AIRCRAFT
CLIMB				
	0.	.001695	3.832	LEAR
	100 .	.0.1642	3.598	LEAR
	2010.	1588	3.364	LEAR
	30 .	.001535	3.129	LEAR
	400 .	.001481	2.895	LEAR
	50 10.	.001427	2.661	LEAR
	60 .	.0 1373	2.526	LEAR
	7070.	.0 1319	2.392	LEAR
	8099.	.0 1264	2.257	LEAR
	9000.	.001210	2.123	LEAR
	100.0.	•001156	1.988	LEAR
	11000.	.00.376	2.164	LEAR
	12000.	.00.339	2.066	LEAR
	13007.	.000301	1.967	LEAR
	1400.	.0 1 264	1.869	LEAR
	15000.	.0 1227	. 1.7/1	LEAR
	16000.	.000191	1.724	LEAR
	1700).	.00.156	1.677	LEAR
	1800 .	.00 120	1.630	LEAR
	19000.	• 0 85	1.583	LEAR
	200 10.	.01 49	1.536	LEAR
DESCENT				
	0.	.0:1690	3.962	LEAR
	1000.	.0 1642	3.738	LEAR
	2000.	.001588	3.513	LEAR
	3000.	,0 1535	3.289	LEAR
	4000.	• U01481	3.064	LEAR
	50 10 .	.001427	2.840	LEAR
	6040.	.0 1373	2.673	LEAR
	70	.0 1319	2.506	LEAR
	8007.	. 1 1264	2.339	LEAR
	9000	.001210	2.172	LEAR
	106	1 56	2.015	LEAR
	1:000.	•0 1376	2.539	LEAR
	12000.	.0 0339	2.428	LEAR
	13000.	.0 13 1	2.317	LEAR
	1400 .	204	2.206	LEAR
	1500 .	.00 227	2.095	LEAR
	1600 .	.00 191	2.031	LEAR
	1700	.010156	1.967	LEAR
	180 .	.00 12	1.903	LEAR
	190	.0. 185	1.839	LEAR
	20000.	.000 149	1.775	LEAR

	ALTITUDE	△-TIME (Hours)	△-FUEL (Pounds)	AIRCRAFT
CLIMB				
	0.	.00 086	3.187	F227
	1009.	. 00 085	3,168	F2./
	2000.	.0 084	5.149	F221
	300 .	.0 84	3.130	F2 /
	4000.	.) 83	3.111	F2_/
	500 .	.010 182	3.093	F2.17
	60 10 .	.000081	3.074	F227
	7000.	.0.0.81	3.050	F2.7
	800).	.0 (280	3.036	F221
	9000.	.00.079	3.017	F2'7
	100 0.	.000078	2.998	F227
	11000.	.00 077	2.979	F2.17
	12060.	.0 / 077	2.960	F2:17
	13000 •	• 000 1.76	2.941	F2 1
	14000.	• 0 75	2.921	F2.1/
	15000.	.004074	2.904	F227 F227
	1700).	.001074	2.885	
	18000.	.01172	2.866	F227 F227
	19000.	.00 071		F227
	200 .	.) 70	2.828 2.809	F2:7
DECCENIT	20	• / _ /	2.009	- FG-1
DESCENT	0.	.0.44165	4.118	F217
	1000.	.000156	3.944	F227
	2000.	.00 1146	3.771	F227
	3000.	.0:137	3.597	F227
	4017.	.000127	3.424	F221
	50 10.	.01 1 8	3,250	F2 /
	00 111.	.0 (1.6	3.0.5	F221
	7006	. 0 94	2.759	F2.7
	800 .	.00 82	2.514	F227
	90 10 .	.1 7	2, 68	F2 /
	100 .	.00 (58	2,023	F2.7
	11000.	.0 19 149	1,807	F2:7
	1200 .	.0.0140	1.591	F227
	1300 .	. 11132	1,376	F227
	14000.	.000.23	1.160	F2:17
	1500	.) 14	,941	F2 7
	10000.	.000011	• 75	F2 7
	170.0.	.00 8	•560	F227
	18000.	.01 . 6	.378	F227
	1900 .	.011013	.189	F227
	2010.	0.0 0.0	0.	F227

Chours Charles Charl						
DESCENT 0. 0.0096 2.50 18. 0C-9 2000. 0192 5.00 36. 0C-9 3000. 0288 7.50 54. 0C-9 3000. 0288 7.50 54. 0C-9 400 0.384 10.01 72. 0C-9 5030. 0480 12.50 90. 0C-9 600 0.572 15.04 104. 0C-9 700 0.654 17.58 118. 0C-9 8000. 0756 20.12 132. 0C-9 903. 0848 22.60 146. 0C-9 1004. 0940 25.20 160. 0C-9 1100 1028 27.80 171. 0C-9 1101 1126 30.43 182. 0C-9 1404 1292 35.60 205. 0C-9 1404 1292 35.60 205. 0C-9 1500 1500 43.48 232. 0C-9 1600 40.84 23.01 194. 0C-9 1600 1580 38.20 216. 0C-9 1701 1540 43.48 232. 0C-9 1801 1603 1460 40.84 239. 0C-9 1900 1700 1540 45.48 232. 0C-9 1900 1700 1700 48.76 247. 0C-9 2010 1780 51.40 255. 0C-9 1900 0.016 5.00 60. 8727 2000 0.0216 5.00 60. 8727 5000 0.0540 12.50 151. 8727 6000 0.0540 12.50 151. 8727 6000 0.0552 21.14 22.5 8727 8000 0.956 24.02 246. 8727 1000 10852 21.14 22. 8727 8000 0.956 24.02 246. 8727 1100 100 100 26.9 8727 1100 100 100 26.9 8727 1100 100 100 26.9 8727 1100 100 26.9 8727		ALTITUDE	TIME	DISTANCE	FUEL	AIRCRAFT
0. 0.0033 0.30 0.0059 1030096 2.50 18. DC-9 20000192 5.00 36. DC-9 30300288 7.50 54. DC-9 40000384 10.01 72. DC-9 50300480 12.50 90. DC-9 60000572 15.04 104 DC-9 7000654 17.58 118. DC-9 80000756 20.12 132. DC-9 90300848 22.66 146. DC-9 100320940 25.20 160 DC-9 110011028 27.80 171. DC-9 12011146 30.43 182. DC-9 13001204 35.01 194. DC-9 15001380 38.20 216. DC-9 15001380 38.20 216. DC-9 15001380 38.20 216. DC-9 17011540 43.48 232. DC-9 18001620 40.84 224. DC-9 19001700 48.76 247. DC-9 19001700 48.76 247. DC-9 20101780 51.40 255. DC-9 19001700 48.76 247. DC-9 20100216 5.00 60. B727 30030324 7.50 91. B727 50000540 12.50 151. B727 50000540 12.50 151. B727 50000644 15.38 175. B727 70010748 18.26 199. B727 80000852 21.14 223. B727 90000956 24.02 246. B727 10000956 24.02 246. B727 10000956 24.02 246. B727			(Hours)	(nm)	(Pounds)	
0. 0.0033 0.30 0.0059 1030096 2.50 18. DC-9 20000192 5.00 36. DC-9 30300288 7.50 54. DC-9 40000384 10.01 72. DC-9 50300480 12.50 90. DC-9 60000572 15.04 104 DC-9 7000654 17.58 118. DC-9 80000756 20.12 132. DC-9 90300848 22.66 146. DC-9 100320940 25.20 160 DC-9 110011028 27.80 171. DC-9 12011146 30.43 182. DC-9 13001204 35.01 194. DC-9 15001380 38.20 216. DC-9 15001380 38.20 216. DC-9 15001380 38.20 216. DC-9 17011540 43.48 232. DC-9 18001620 40.84 224. DC-9 19001700 48.76 247. DC-9 19001700 48.76 247. DC-9 20101780 51.40 255. DC-9 19001700 48.76 247. DC-9 20100216 5.00 60. B727 30030324 7.50 91. B727 50000540 12.50 151. B727 50000540 12.50 151. B727 50000644 15.38 175. B727 70010748 18.26 199. B727 80000852 21.14 223. B727 90000956 24.02 246. B727 10000956 24.02 246. B727 10000956 24.02 246. B727	DESCENIT					
10)	DESCEIVI	0.	0.0001	0.30	0.	DC-9
2000. 0192 5.00 36. 0C-9 3030. 0288 7.50 54. 0C-9 4000. 0384 10.00 72. 0C-9 5030. 0480 12.50 90. 0C-9 6000. 0572 15.04 104. 0C-9 703. 0664 17.58 118. 0C-9 8000. 0756 20.12 132. 0C-9 9030. 0848 22.66 146. 0C-9 10002. 0940 25.20 160. 0C-9 1100. 1028 27.80 171. 0C-9 1203. 1116 30.43 182. 0C-9 1404. 1292 35.60 205. 0C-9 1404. 1292 35.60 205. 0C-9 1500. 1380 38.20 216. 0C-9 1600. 1460 40.84 224. 0C-9 1700. 1540 43.48 232. 0C-9 1800. 1620 46.12 239. 0C-9 1900. 1700 48.76 247. 0C-9 2030. 1780 51.40 255. 0C-9 DESCENT DESCENT 0. 0.0 0 0.0 0.0 0.0 0. B727 2000. 0216 5.00 60. B727 2000. 0216 5.00 60. B727 5000. 0540 12.50 91. B727 6000. 0644 15.38 175. B727 7001. 0748 18.26 199. B727 8000. 0852 21.14 222. B727 9000. 0956 24.02 246. B727 1000. 1050 26.90 270. B727			THE RESERVE AND ADDRESS OF THE PARTY OF THE		and the second s	
3000. 0288 7.50 54. 0C-9 4004. 0384 10.01 72. DC-9 5030. 0480 12.50 90. DC-9 6004. 0572 15.04 104. DC-9 700. 0654 17.58 118. DC-9 8000. 0756 20.12 132. DC-9 9030. 0848 22.66 146. DC-9 10042. 0940 25.20 160. DC-9 1100. 1028 27.80 171. DC-9 1201. 1116 30.40 182. DC-9 13004. 1204 35.00 194. DC-9 1404. 1292 35.60 205. DC-9 1500. 1380 38.20 216. DC-9 1500. 1380 38.20 216. DC-9 1600. 1460 40.84 224. DC-9 1700. 1540 43.48 232. DC-9 1800. 1700 48.76 247. DC-9 2010. 1780 51.40 255. DC-9 DESCENT 0. 0.01 0.010 25.0 30. B727 2000. 0216 5.00 60. B727 2000. 0324 7.50 91. B727 6000. 0540 12.50 151. B727 6000. 0540 12.50 151. B727 7041. 0748 18.26 199. B727 8000. 0852 21.14 222. B727 8000. 0852 21.14 222. B727 9000. 0956 24.02 246. B727 1000. 1060 26.90 270. B727 1156 29.84 288 B727 12000. 1252 32.78 306. B727						
## 100						
12.50						
15.00						
700						
8000. 0756 20.12 132, UC-9 9000. 0848 22.66 146. UC-9 10000. 0.940 25.20 160. UC-9 11.00 . 1028 27.80 171. UC-9 120111.16 30.40 182. UC-9 130001204 35.01 194. UC-9 140011292 35.60 205. UC-9 150001380 38.20 216. UC-9 150001380 38.20 216. UC-9 160001460 40.84 224. UC-9 170011540 43.48 232. UC-9 180011700 48.76 247. UC-9 200101780 51.40 255. UC-9 DESCENT 0. 0.01 0.00 0.00 0. B727 20000216 5.00 60. B727 20000216 5.00 60. B727 40010432 10.00 121. B727 50000540 12.50 151. B727 60000540 12.50 151. B727 60000644 15.38 175. B727 70010748 18.26 199. B727 800010852 21.14 222 B727 90000956 24.02 246. B727 100011156 29.84 288 B727 120001252 32.78 306. B727						
9030						
100 1. 1028 27.80 171 171 171 171 172 173 174 17						
1100						
1201						
13000						
140-1 . 129-2 35.60 205. DC-9 1500 . 1380 38.20 216. DC-9 1600 . 1460 40.84 224. DC-9 1700 . 1540 43.48 232. DC-9 1800 . 1620 40.12 239. DC-9 1900 . 1700 48.76 247. DC-9 2000 . 1780 51.40 255. DC-9 DESCENT O. 0.0 - 0.00 0 0 0 8727 2000 . 0216 5.00 60. B727 2000 . 0216 5.00 60. B727 400 . 0432 10.00 121. B727 400 . 0432 10.00 121. B727 5000 . 0540 12.50 151. B727 6000 . 0644 15.38 175. B727 7001 . 0748 18.26 199. B727 8000 . 0852 21.14 222 B727 8000 . 0956 24.02 246. B727 1000 . 1060 26.90 270. B727 11000 . 1156 29.84 288. B727 12000 . 1252 32.78 306. B727						
1500						
1600 . 1460						
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1900 1 1700 48.76 247 DC-9 2010 1 1780 51.40 255 DC-9 DESCENT 0. 0.0 0.0 0.0 0.0 0. 8727 1000 1018 2.50 30 8727 2000 10216 5.00 60 8727 3000 10324 7.50 91 8727 400 10432 10.00 121 8727 5000 10540 12.50 151 8727 6000 10644 15.38 175 8727 7001 10748 18.26 199 8727 8000 10852 21.14 222 8727 9000 10956 24.02 246 8727 1000 1156 29.84 288 8727 12000 1252 32.78 306 8727						
DESCENT 0. 0.01. 0.00 0.00 0. 8727 1000. 0108 2.50 30. 8727 2000. 0216 5.00 60. 8727 3000. 0324 7.50 91. 8727 407. 0432 10.00 121. 8727 5000. 0540 12.50 151. 8727 6000. 0644 15.38 175. 8727 7001. 0748 18.26 199. 8727 8009. 0852 21.14 222. 8727 9000. 0956 24.02 246. 8727 1000. 1156 29.84 288. 8727 12000. 1252 32.78 306. 8727						
0. 0.00 0.00 0.8727 1000. .0108 2.50 30. 8727 2000. .0216 5.00 60. 8727 3000. .0324 7.50 91. 8727 400. .0432 10.00 121. 8727 5000. .0540 12.50 151. 8727 6000. .0644 15.38 175. 8727 7001. .0748 18.26 199. 8727 8000. .0852 21.14 222. 8727 9000. .0956 24.02 246. 8727 100.0. .1060 26.90 270. 8727 1103. .1156 29.84 288. 8727 12000. .1252 32.78 306. 8727						DC-9
1000	DESCENT					
2000. .0216 5.00 60. B727 3000. .0324 7.50 91. B727 400. .0432 10.00 121. B727 5000. .0540 12.50 151. B727 6000. .0644 15.38 175. B727 7001. .0748 18.26 199. B727 8000. .0852 21.14 222. B727 9000. .0956 24.02 246. B727 1000. .1060 26.90 270. B727 11000. .1156 29.84 288. B727 12000. .1252 32.78 306. B727		0.	0.01	0.00	0.	8727
2000. .0216 5.00 60. B727 3000. .0324 7.50 91. B727 400. .0432 10.00 121. B727 5000. .0540 12.50 151. B727 6000. .0644 15.38 175. B727 7001. .0748 18.26 199. B727 8000. .0852 21.14 222. B727 9000. .0956 24.02 246. B727 1000. .1060 26.90 270. B727 11000. .1156 29.84 288. B727 12000. .1252 32.78 306. B727		1000.	.0108	2.50	30.	8/27
400 .0432 10.00 121 8727 5000 .0540 12.50 151 8727 6000 .0644 15.38 175 8727 7001 .0748 18.26 199 8727 8000 .0852 21.14 222 8727 9000 .0956 24.02 246 8727 1000 .1060 26.90 270 8727 11034 .1156 29.84 288 8727 12000 .1252 32.78 306 8727		2000.	.0216		60.	8721
5000. .0540 12.50 151. B727 6000. .0644 15.38 175. B727 7001. .0748 18.26 199. B727 8000. .0852 21.14 222. B727 9000. .0956 24.02 246. B727 1000. .1060 26.90 270. B727 11030. .1156 29.84 288. B727 12000. .1252 32.78 306. B727		3000.	.0324	7.50	91.	8/2/
6000. .0644 15.38 175. B727 7001. .0748 18.26 199. B727 8000. .0852 21.14 222. B727 9000. .0956 24.02 246. B727 100.0. .1060 26.90 270. B727 1103. .1156 29.84 288. B727 12000. .1252 32.78 306. B727		407 .	.0432	10.00	121.	8721
70010748 18.26 199. B727 80000852 21.14 222. B727 90000956 24.02 246. B727 10001060 26.90 270. B727 11001156 29.84 288. B727 120001252 32.78 306. B727		5000.	.0540	12.50	151.	8727
8000. .0852 21.14 222. 8727 9000. .0956 24.02 246. 8727 1000. .1060 26.90 270. 8727 1100. .1156 29.84 288. 8727 12000. .1252 32.78 306. 8727		0000.	.0644	15,38	175.	B721
8000. .0852 21.14 222. 8727 9000. .0956 24.02 246. 8727 1000. .1060 26.90 270. 8727 1100. .1156 29.84 288. 8727 12000. .1252 32.78 306. 8727		7001.	.0748	18.26	199.	B727
9000 • .0956 24.02 246 • B727 100 0 • .1060 26.90 270 • B727 1100 • .1156 29.84 288 • B727 12000 • .1252 32.78 306 • B727			.0852			872₹
100 0. .1060 26.90 270. B727 1100 . .1156 29.84 288. B727 12000. .1252 32.78 306. B727		9000.	.0956	THE RESIDENCE OF THE PARTY OF T		
1100 • • • • • • • • • • • • • • • • • •		100			270.	B721
12000. .1252 32.78 306. B727						
						B727
13000 1348 35.12 324. 5121		13000.	.1348	35.72	324.	8727
1400 • • 14+ 38,00 542• 8727						
150001540 41.60 360. B727						
1600 · . 1616 44.46 374. B/2/						
1700 ·						
180001768 50.18 401. B727						
1900 • ,1844 53.04 414. 8727						
20 1 • • • • • • • • • • • • • • • • • • •						

	ALTITUDE	TIME (Hours)	DISTANCE (nm)	FUEL (Pounds)	AIRCRAFT
DESCENT					
DESCEIVI	0.	0.0	0.0	0.	UC-H
	100 .	.0130	2.50	59.	DC-8
	20 .	.0260	5.01	118.	DC-8
	3000 .	.0390	7.50	170.	UC-8
	400 .	.0520	10.01	255.	DC-H
	500	.0650	12.50	294.	UC-8
	6070.	,0700	15.72	340.	UC-8
	700 •	.0842	18.94	385.	UC-8
	8000.	,0998	2'.10	431.	OC-8
	900.	-1114	25.38	470.	UC-8
	10000.	.1230	28.60	522.	UC-8
	1100 .	.1338	31.80	560.	UC-8
	1200%	.1446	35.16	598.	DC-8
	1300 .	.1554	38.44	635.	UC-8
	1400 .	•1602	41.72	673.	DC-8
	15000.	.1770	45.00	711.	UC-8
	16000.	.1830	48.50	746.	UC-8
	17001.	.1990	52.00	781.	DC-8
	18000.	.210*	55.50	815.	DC-8
	1900 .	.2210	59.0	850.	UC-8
	200	.2320	62.50	8-5.	UC-8
DESCENT					
	0.	0.00.00	0.30	0.	UC10
	1000.	.0140	2.50	82.	DC10
	200).	•0280	5.00	163.	DC10_
	3000.	.0420	7.50	245.	DC10
	4000.	.0560	10.0	326.	DC10
	500 •	.0701	12.50	408.	UC10
	0000.	.0810	15.90	467.	UC10
	70 11 •	.0932	19.50	526.	DC10
	8001.	.1048	22.70	584•	UC10
	900 .	.1164	26.10	643.	DCIU
	1000 .	.1280	29.50	702.	DC10
	1:07	.1390	32.86	749.	UC10
	12000	.1500	36,22	796.	DC10
	13000.	.1610	39.58	842.	UCIU
	1400 .	.1720	42.94	839.	DC10
	15000.	.1830	46.30	936.	DC10
	1600 .	.1928	49.36	972.	UC10
	1700 .	.2026	52.4	1008.	DC10
	1800 .	,2124	50,48	1043.	DC10
	1900 .	.22:	58.04	1079.	DC10
	200 0.	.2320	61.60	1115.	UCIO

	ALTITUDE	△-TIME (Hours)	△-FUEL (Pounds)	AIRCRAFT
DESCENT				
	0.	.001630	11.323	UC-9
	100 .	.0 11576	10.7/0	UC-9
	200 .	.0 1523	10,218	DC-9
	3011 .	.0 1469	9.6.5	DC-9
	40) .	.001416	9.1.3	DC-9
	50 .	.0 11362	8,500	DC-9
	600 .	. 101308	8.076	UC-9
	700 .	.011254	7.591	DC-9
	8000.	.011199	7.107	UC-9
	901 .	.0 1145	6.022	UC-9
	1001 .	.0 1091	6.138	UC-9
	11007.	.01142	5.725	UC-9
	12000.	.00 993	5.308	UC-9
	13000.	.00 944	4.891	UC-9
	1400 .	.0.0895	4.479	. UC-9
	1500 1.	.000846	4.064	UC-9
	16000.	.0 0798	3.736	DC-9
	17001.	.000750	3.408	UC-9
	18000.	.000773.	3.081	UC-9
	1900 .	•000655	2.753	UC-9
	200000	.000617	2.425	DC-9
	20	•00,0		
DESCENT				
	0.	.011775	19.658	8721
	10: •	.0 1721	18.7/3	B727
	20 .	.001668	17.8	8721
	300 .	.001614	17.003	B727
	4000.	.001561	16.118	8727
	5000.	.001507	15.233	8727
	6000	.001453	14.510	8/27
	7011.	.001399	13.788	B727
	800 .	.001344	13.065	B727
	90011.	.001290	12.343	8727
	10000.	.001236	11.620	B727
	11000.	.001187	10.935	B727
	120 .	.001138	10.251	B727
	1300 .	•0 1 89	9.560	8727
	1400 .	.001040	8.382	B727
	15000.	.) 991	8,197	B727
	1600 .	.09 940	7.602	B727
	17000.	.0 0834	7.126	8727
	1800 .	.090837	6.591	872/
	1900 .	.0 785	6.055	B72/
	2000	.0 0734	5.520	872/
	20.0		0.720	C. ('2.)

	ALTITUDE	△-TIME (Hours)	△-FUEL (Pounds)	AIRCRAFT
DESCENT				
	0.	.001763	27.778	DC-8
	1000.	.001709	26.550	DC-8
	2000.	.001655	2 .322	UC-8
	3001.	.001602	24.093	DC-8
	4000.	.001548	22.865	DC-8
	5000.	.001494	21.637	UC-8
	600 .	.001440	20.678	DC-8
	70	.0 11386	19,718	DC-8
	8000.	.0 1531	18.759	DC-8
	900 .	.001277	17.799	DC-8
	100 .	.001223	16.840	UC-8
	110:0.	.001174	15.972	UC-8
	12000	.0 11125	15.104	UC-8
	1300 .	.001176	14.236	DC-8
	1400 .	.001 27	13,568	UC-8
	15000.	•000978	12.500	DC-8
	1600 .	.01 927	11.728	1)C-13
	1700 .	,9 875	10.957	DC-8
	18001.	.00 824	10.185	DC-8
	19000.	.0 772	9.414	UC-8
	200 .	•00 ·721	8.642	UC-8
DESCENT				
	0.	.001817	26.647	DC10
	10 .	.0 1763	25.630	DCIU
	201 .	.0 1719	24.612	DCIO
	30 .	.0 1656	23.595	DC10
	400 .	.001602	22.57/	DC10
	500 .	.011548	21.560	DC10
	6000.	.001494	20.636	UC10
	70) .	.001440	19.713	DC10
	800 .	.001385	18.789	UC10
	9000.	.001351	17.865	DC10
	100 0.	.001277	16.942	DC10
	11000.	.001228	16.144	DC10
	1200 .	.011179	15.345	UC10
	13000.	.001130	14.547	DC10
	1400 .	.0 1081	13,748	DC10
	1500 .	.001 32	12.950	DC10
	16000.	•0 984	12.200	DC10
	17000.	•010936	11.450	DCTO
	1800 .	•00 889	10.700	DC10
	19000.	.00 841	9.950	DC10
	2000).	.09 793	9.200	0010

APPENDIX B UNITED AIR LINES FLIGHT PLANNING PROGRAM OUTPUT FORMAT AND KEY

ALTERNATE WEATHER ROUTE STUDY EXAMPLE OUTPUT

SCIL	U					
acho	WCOUA C.	30049 126	2 04	1		
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	015	STANCE			DIS	STANCE
KUUT	E WIND	GROUND		ROUTE	WIND	GRUUND
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160	1777	2018		080	1821	2061
130	1779	2014		180	1823	2580
020	1784	2014		046	1027	2076
170	1707	2034		05C	1838	2584
100	1768	2016		090	1060	2124
140	1700	2023		190	1866	2125
110	1799	2024		03C	1871	2147
120	1799	2027		200	1873	2141
Unt	1806	2542		010	1082	2161

03:56/00:01	54160 150	100 • X X X	RT:15 (P)
000 2014NN	3370/3911 2029/	/3370 CLB DO C50 2579'	791 LAX
KCA 203	37 83C -01 473	3 24109 106 579 27 108	603
167 365	37 630 -05 468	3 24070 076 544 39 87	596
168 250	37 630 -05 469	25567 066 535 28 61	535
580 638	37 835 -02 472	2 28090 090 562 71 152	303
182 271	37 830 -01 473	3 30087 071 544 30 62	321.
POU 187	37 030 04 479	32084 039 510 21 45	276
-uAS 100)	34064-031 20 26	250
33 3414 03:	55 585 3094	м83	
27 3484 03:	54 655 3094	m83	
27 3526 63:	55 697 3094	m83	
25 3569 03:	54 740 3094	NBS	

UASIX H/UC10/A 64/3 LAX P1200 370
FAA/DIRECT LAX 1AD 15
SCI NY 397

UNITED

Flight Plan Forecast

71. FLIGHT PLAN FORECASTS are provided from an Automatic Flight Planning and Monitoring computer system (AFPAM). The following message format is a typical Flight Plan Forecast for an advance section of Flight 29 from JFK to SFO.

NOTE: The Captain and Dispatcher are responsible for validating all flight plans including Flight Plan Forecasts and for assuring they select the same forecast.

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3)												RT: 5	
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6)	-SFO	120					25021-	-021		22	52		450
	a	ь	C	d		C	f						
7)	39	5333	05:24	11	26	000	1134						
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9)	HON	W.F.O.					OF CON						

FLIGHT PREPARATION Flight Planning

FLIGHT OPERATIONS MANUAL 3/22/74 Page 60.7

- 1) Address of message.
- Radio Communications number Published flight number. Omitted 2) if both radio and published number
 - are identical. Flight date
 - Plane number
 - Selcal code
 - Dispatch Sector controlling departue station
- ORDDD Sector controlling enroute portion of flight while traversing ORDDD area Appears only in long range flight plans not originating or terminating in ORDDD area Dispatch Sector controlling
- arrival station
- i Preferred Diversion Station(s)
 Will appear when Planning Alert Message in effect for arrival station.
- Releasing Dispatch Sector R: Flight Plan Forecast number.
- Planned flight time 3) a.

 - Planned High time
 Planned total burnout 113, 100 lbs.
 Combined FAA reserve and contingency fuel 23,000 lbs.
 Holding/detouring fuel 15,200 lbs.
 Alternate/diversion fuel 4,800 lbs. If PDS (line 2) has not been Assigned to the flight, this is the computer-generated fuel required for the most distant alternate. If PDS has been assigned, this is the computer-generated fuel required for the diversion station(s) or the required alternate(s) whichever is farther. "*" indicates the figure was entered manually and not computer-generated.
 - g. Alternates OAK, SJC. If no alternate, "NA". If a fuel value appears in "f" and "NA" in "g", this is a reduced mach flight plan. The fuel value is the difference between standard mach burnout and reduced mach burnout and is added to the total fuel.
 - h. Amount of fuel, if any, being ferried for the next flight.
 - i. Computer-stored route number
 - (P) indicates plan computed in accordance with Company fuel policy; omitted if not. "SPD" in this space indicates a B747 fifth-pod plan. "YDI" indicates B-727 yaw-damper-inop. plan.
- 4) a. Pounds of fuel/min. gained over Base Plan. If blank, this is a reduced mach flight plan with burnout less than the Base Plan. "000" indicates this is the Base Plan.
 - Total route mileage 2307 nautical miles.
 - c. Planned takeoff gross weight 536,800 lbs.
 - d. Operationally allowable takeoff gross weight 677,100 lbs. This is the least of:
 - 1) Max structural T/O gross weight, or
 - 2) Max allowable landing weight plus burnout, or
 - 3) Max, allowable T/O weight (performance or runway limited) as determined by conditions expected to prevait at time of T/O, i.e. runway, wind, temp., etc. This is entered by Dispatch when they expect allowable T/O weight will be less than 1) or 2) above. When it is put in, the basis for its derivation, i.e. wind, runway, temp., flap setting, etc. is noted in the remarks section.

NOTE: Actual allowable T/O weight must be checked just prior to T/O - Page 118. para. 16.

Planned landing weight - 423,700 lbs.

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FLIGHT PREPARATION Flight Planning

UNITED

- f. Maximum allowable landing gross weight which was entered in the FPF request 564,000 lbs.
- Deviation from standard temperature used in computing climb performance -12^o above standard.
- Headwind (indicated by "-") or tailwind component used in computing climb performance - 15 knots headwind.
- Zero fuel weight 378,700 lbs.
- Fuel required for release 156,100 lbs. If no fuel is being ferried, the fuel required for release is the figure in the next column(k).
- k. Total fuel to be carried if load and operating conditions permit. This will be the fuel required for release if no fuel is being ferried. This is the fuel weight used in the computation of the flight plan.
- 1. Departure station at which fuel in k. is boarded.
- 5) a. Flight plan check points. Reporting point when preceded by a minus sign "-". An "R" suffix indicates an overwater reclear point.
 - b. Segment mileage in nautical miles.
 - c. Flight level in thousands of feet at check point. AFPAM produces optimum altitudes for center stored routes which may differ from stored altitudes. The Captain will compare the plan altitude with the center stored-altitude (pages 75-100.18), and if different, request the plan altitude from ATC.
 - d. Indicated mach number at check point,
 - e. Deviation from standard temperature at check point.
 - f. True air speed in knots at check point.
 - g. Wind-direction in tens of degrees (30=300°) and speed in knots (05 8) at check point.
 - h. Headwind or tailwind component at check point,
 - i. Ground speed at check point.
 - j. Segment time in minutes.
 - k. Segment burnout in hundreds of pounds.
 - 1. Fuel remaining in hundreds of pounds at check point.
 - Wind vector and component used in determining descent performance data.

מבוווט

- Summarized flight plans for other altitudes over the route. When the
 policy FPF is a reduced mach plan, a summary plan is included at
 standard mach.
 - a. Highest flight level in thousands of feet (usually POD altitude).
 - b. Planned takeoff gross weight.
 - c. Total flight time enroute.
 - d. Total burnout.
 - e. Numbers in this space indicate amount of burnout in pounds for each minute of time gained based on flight time in Base Plan. "OOO" indicates this is the Base Plan. "XXX" indicates disregard this summary plan as flight time and fuel are greater than the Base plan. Amounts greater than 3094 appear as "3094".
 - f. Mach number used to compute summary plan.
- NOTE: An "X" in a summary line between "e" and "f" indicates that, due to the higher burnout at summary altitude, either the maximum ATOG or the maximum fuel capacity (or both) are exceeded.
- Data required by ARTC.
 - a. Flight identification.
 - b. FAA aircraft type and equipment codes.
 - c. True air speed during first cruise segment.
 - d. Departure station.
 - e. Planned departure time (GMT)
 - f. Flight level planned for first segment in hundreds of feet.
 - g. Planned route of flight. (If preferred, "PREFERRED" added; if center-stored, "C/S" added.)
- Remarks Three lines are reserved for Dispatchers' and/or Meteorologists' comments pertinent to the operation of the flight.

Gross Weight and Fuel Work Sheet (UF 1699)

- 72. PREPARE a Gross Weight and Fuel Work Sheet for all flights at flight planning offices, except as noted below. A copy of the form should accompany each flight and be filed with the flight papers.
 - A. Dispatchers and Captains may use it as the work sheet for computing fuel requirements for load planning purposes.
 - B. 'When the Captain agrees with the Dispatcher's fuel computation, or vice versa, he uses appropriate words such as agree, same, etc. without duplicating the previous computation.

EXCEPTION: When the Captain accepts the Flight Plan Forecast, it is not necessary to prepare the Gross Weight and Fuel Work Sheet.

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FLIGHT PREPARATION Flight Planning

APPENDIX C

AIRPORT PAIR ROUTE STRUCTURE

PERFORMANCE STATISTICS

AND

MINIMUM ROUTE STRUCTURE WIND

MILE PLOTS

07/03/75 19100107 ARSTEP	7 ARSTER	000373G57	155	1/5000		2	300							DA1E 070375	7057	2	2	PAGE	•	
						ROUTE	ROUTE STRUCTURE COMPARISONS	UNE	CUMPAR	ISONS										
				PAFF	RRED	ROUTE	1 GROUP		IND AN	PREFERRED ROUTE: GROUND, WIND AND TOTAL BENEFIT	BEN	EFIT								
					6	IA BA	DATA BASED ON RUNS	RUNS		1 THROUGH 39	6									
						IRPOR	ATRPOST PAIR	=		040 OEN										
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ROUTE STRUCTURE COMPARISONS
ALTERNATE DOUTE RENEFITE PREFERRED ROUTE. ADDITIONAL HOUTES AND TOTAL

DATA HASED ON RUNS I THROUGH 39 AIRPORT PAIR 11 OND DEN

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VOR ■	2.6 (.30) + -3.0 (34) =4 (05)	3.9	3.1 (.35) + -3.2 (37) =2 (02)	3.9 (000	1.9 (.4 +6 (0 = 5.3 (.3	+ -1.6 (18)
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FAA	5 (05) + 5 (03) #	+ 1050.	+ #		-3.1 (35) + 3.2 (37) + -2 (02) =	.10)	-3.1 (35) .6 (.19) + -1.4 (16) =
S I S I	+ .5 (.05) + = .9 (10) =	+ # (000 · · · · · · · · · · · · · · · · ·	+ 11 (20.1)		403	+ 2.9 (.00) + 2.9 (.33) +	+ 1.9 (.22) + 1.9 (.22) +
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		13 13814	THEN POUTER GRUUND	PREFERENCE POUTER GROUND. MIND AND TOTAL BENEFIT	NEF11			
			NATA HASED ON R	RUUS 1 THROUGH 39				
,			ATRPORT PAIR	ZE DEN ORD				
	7 4 H 7	SCI	FAA	SCI/FAA NAFEC	* O.*	PRE-PLANNED	300	*
UAL PNAV	000.30.	2.0 (.28) + -11 (02) = 1.0 (.21)	3.5 (.48) ((11.1 (15) 1.1 = 1 (.35)	3.5 (.48) + -1.1 (15) + = 2.4 (.33) =	5.2 (.71) 3.3 (.45) 8.5 (1.10)	000	* N . S	
SCIZNAFEC	-2.0 (28) + 1 (.02) = -1.9 (27)	000 - 0	1.4 (-20) + -1.0 (-14)5 (.06)	1.4 (.20) + -1.0 (14) + * .5 (.06) =	3.2 (3.4 (6.6 (2.00	3.5	200
FAAZNAFEC	-3.5 (48) + 1.1 (.15) = -2.4 (33)	1.00	(00.) 0. + (+ # (000)	1.7 (.24) n.4 (.60) 6.1 (.84)	-3.5		666
SCIVFANAFEC	-3.5 (-,48) + 1.1 (-,15) = -2.4 (-,33)	+ 1.0 (20)	(00.) 0. + (+ H (000.	1.7 6 6.1 6	+ - 3 - 5 - 5 - 5 - 5 - 5 - 5 - 5 - 5 - 5		35.6
V007A	+ -3.3 (45) = -8.5 (-1.18)	- 5.2 (44) + - 5.4 (47) = -6.6 (91)) + -1.7 (24) + -1.4 (60) = -6.1 (84)	-1.7 (24) + -4.4 (60) + = -6.1 (84) =	0000	-5.2	000	6000
PAF-PLAINED	(30.) 0	2.0 (.28) + -1 (02) = 1.9 (.27)	3.5 (.48) 1 + -1.1 (15) 2 = 2.4 (.33)	3.5 (.48) + -1.1 (15) + = 2.4 (.33) =	5.2 (.71) 3.3 (.45) 8.5 (1.16)		9.35	.453
VDR/C	-5.2 (12)	+ -5.8 (47) = -6.6 (91)) + -4.7 (24)) + -4.4 (60)) = -6.1 (84)	+ -4.4 (60) +	000	+ 13.3 (12)	000	666

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ROUTE STRUCTURE COMPARISONS
LITERWATE ROUTE BENEFIT: PREFERRED ROUTE: ADDITIONAL HOUTES AND TOTAL

	ğu	8.5 (1.16) -1.2 (17) 7.3 (1.00)	-1.2 (17)	1.0 (.94)	1.0 (15) 1.0 (15)		8.5 (1.16) 2 (02) 8.3 (1.14)	0000
	PRE-PLANNED		-1.9 (14) + -2.9 (41) #	1.2 (15) +	-2.4 (33) 1.2 (16) + -1.2 (17) =	-8.5 (-1.18) + 1.0 (.14) + -7.5 (-1.04) H	+	-8.5 (-1.18) +
HOUTES AND TUTA	, v	-2.1 (28) + 6.4 (.88) =	-2.1 (-20) -2.1 (-20) 4.5 (-62)	6.1 (.84)	6.1 (.84) .1 (.02) + 6.2 (.86) =	+ W	-1.0 (-1.17) -1.5 (1.02) +	(51) 8
ALTERNATE ROUTE RENETITI PREFERRED ROUTE. ADDITIONAL HOUTES AND TUTAL DATA BASED ON HUNS 1 THROUGH 34 AJRPORT MAIR 21 DEM UND	. SCI/FAA NAFEC	2.4 (.33) + .2.2 (30) + .2.2 (.03) =	+ -2.2 (30) + = -1.7 (24) =	(000.)	+ # (000.	-6.1 (85) +1 (02) + -6.2 (86) =	+ -1.2 (-16) +	+ -1.0 (85) -7.1 (98) +
MFFITT PREFERHED RO NATA HASED ON HUNS AJRPORT MAIR 21	FAA	2.4 (.33) + -2.2 (30) = .7 (.03)	5 (30)	0000	6666	-6.1 (02)	2.4 (.33) + -1.2 (16) # 1.2 (.17)	-6.1 (85) -1.0 (13) -7.1 (98)
RNATE BOUTE BENE	SCI NAFEC	1.9 (.00)	666	2.2 (.24)	2.2 (.30)	2.1 (.28)	1.0 (.14)	1.2 (.17)
AL TE	JANA A	(00.) 0.	1.9 (37)	-2.4 (33) -2.2 (35) 2 (93)	-2.4 (33)	- 8.5 (-1.18) - 2.1 (-24) - 6.4 (89)	1.0 (.10)	11.2 (1.17)
		UAI RNAV	SCIZNAFEC	FAA/WAFEC	SCIVERINAFEC	Vue/A	PRE-PLANNED	7/00/

01/03/15 19100	19100107 ARSTEP 000373G57	175000	N	000					3	ATE.	DATE 070575	Σ.	•	P A G E	-	1
		ROUTES	HOUTE STRUCTURE COMPARISONS STRUCTURE RENEFIT: GROUND, FIND AND TOTAL	RENEFITE GRUUND, FIND	RUUND.	SONS	101	14								
			DATA BASED ON RUNS	NO RUNS		1 THRUUGH 39										
			AIRPURT PAIR	IR 21	DEN	DEN ORD										
	UAL	NAFEC	FAA		SCI/FAA NAFEC	AA		V08	~	4	PRE-PLANNED	NNED			20	
UAL BNAV	_	_	3.5 (3.5 (.48)	2	.2 (.72)		0.	(000		5.2	_	.72)
	+ (00.) 0. +	1.9 (.27)	+ -3.3 (.03) #	-3.3	.03)	+ #	1.2 6	.88)	+ #	000	33	+ =	7.3		1.00)
SCITANFEC		(00.) 0.	1.4 (.203	1.0	.203	3	.2 ((44)	•	0.0	283		3.2	_	.43)
	+ (50.) (1. #		= -1.7	- (44) +	-3.2	74)	+ 11	1.3	.18)	+ #	0.0	13)	+ #	2.5		.30)
FAAZNAFEC	-3.5 (48)	5.2 (20)	00.	(00)	000	(00)		1.7 (.24)	٠.	2.5		٠	1.7		.24)
	(03)	1.7 (0.	. 000.	0.	.00)	9	1 2.9	. 96)		-1.2	17)		7.1	_	. 973
SCINEAMAFEC	-3.5 (48)	-	0.		0.			1.7 (.24)	•	-3.5	48)		1.7	_	.24)
	=2 (03) =	1.7 (.24)	о о о • н		000	.000	+ 4	6.2 (. 86)	+ #	-1.2		+ 11	7.2		.975
VOR/A	-5.2 (-1.7 (24)	-1.7	243		0.	.003	•	-5.2	723		•	J	.00.
	E -6.4 (80) =	-4.5 (67)	6.2	1.86)	-6.2		+ 11	00.	.000	+ #	7.5	(-1.04)	+ 11	ω «.		.12)
PPF-PLAURED		2.0 (.28)	3.5	(8)	3.5	.48)	200	5.5 (127.		0.0	(00.		5.5		.72)
	1.0 (.14) =		= 1.2 (1.2 (- 11	25	1.023	· u		(00.	- 11			11.
7/60/	-5.2 (72)	-5.2 (30)	-1.7	243	-5.4	24)		0.0	.000	•	-5.2 (72)	٠	0.0		.000
	-	.5.A	= -7.1 (- (86) =	-7.1	98)		.8	12)			(-1.15)	u			000

	UAL 9NAV		2	PREFERENCE OF THE PREFERENCE O	A P P K	ROUTE STRUCT D POUTE: GROUN DATA BASED UN ATRPORT PAIR FAA HAFEC	GROUND GROUND ED ON R	STRUCTURE COMPARISONS GROUND, WIND AND TOT ED UN RUNS I THROUGH PAIR 3: ORD LA AA SCI/FAA FEC NAFEC	4 ×	39		, a		d - 224 4	PRE-PLANNED			900	
UAL BNAV	066	(00.	4.5.7	661	+ 11	446	. 265	1.2.8	(19)	+ "	16.4		36)	2.3.9		236	7.0		. 36)
SCIJMAFEC	4 2.8 4	591.	+ 11	666	• •	7.2	423		0000	**	7.4 19.1 26.5		1.11)	000	6000	366	19.1		1.54)
FAAZNAFEC	+ n	26)	-4.8		+ 4	000	666	-4.8 +-7.2 =-12.0	2 (+ 11	2.6		15)	-4.8	(28) (43)	223	2.6		703
9CT/FA/WAFEC	4.6	(6).	+ 11	666	+ #	7.2	.42)	•••		+ "	19.1	4.1.	1.543	• 11	366	366	19.1		1.15.1
V0P/A	-7.0 (+-16.4 (=-23.4 (-		+-19.1	(-1.13) (-1.57)		-2.6		-7.4 +-19.1 =-26.5	(-1.13) (-1.57)	+ #	000		666	+-19.1	(-1.13) (-1.57)	325	••••	000	6000
DOK-PLANED	3.8.2	50.	+ 11	66.	+ u	7.2	.42)		(000.	+ #	7.4		1.113	• • •		666	19.1		1.545
Y08/C	-7.0 ()	1-13-1	(-1.13)		-2.6		-7.4	(-1.13)	+	0.0		(00)	1.7-	(4.1.4	93			(00)

PAGE 15			ED VOK		-19) 23.4 (1.38)	. 11	.000) 26.5 (1.57)		9	+	41) = 12.4 (.7	-	23) = 15.3 (66)		.75) + 5.2 (.31)	. 5.2 (.00) 26.5 (1,57)	= 19.2		(00.) 0. + (44)
DATE 070375	TOTAL		PRE-PLANNED			2.7 (-	300 200	-3.9 (-	· ·	0.01		= -3.9 (-26.5 (-1.59)		-	000	0.	-26.5 (-1.59)	7.3 (.44)
	NAL ROUTES AND TOTAL		VOR	21 6 7 7 20		= 11.4 (.68)	*-16.4 (97)	. 10.2 (.60)	14.5 (F 7.2 (45)			= 10.2 (.60)	J.	000.	24.5.1		= 14.1 (.84)	(00.)	-5.2 (31)
2 300 ROUTE STRUCTURE COMPARISONS	REVEFITE PREFERRED ROUTE. ADDITIONAL DATA HASED ON RUNS I THROUGH 39	31 ORD LAX	SCI/FAA NAFEC	-3.2 (19)	+ 4.4 (.26)			_	+ 9-1 (72)		_	(000) 00 +	(00.) 0.	+ 16.4 (-1.59)	.:	-	+ 3.9 (.23)	3.4 (.23)	+ 11.2 (-67)	(26)
ROUTE STRUCTO	CATA HASED ON R	ATRPORT PAIR	FAA	8.8	- 4.7 (28)	-	7.9.1		(00.) 0. +	J	_	# 2.9 (54)		7.3	z -7.2 (43)		+ -5.2 (31)		· ·	=-12.4 (74)
	ALTERNATE BOUTE R		STI	i.	= 1.2 (.07)	_	(00.) 0. =			= -() (16)	0.0	(00.) 0. =	-26.5 (-1.59)	+ 16.4 (.98)	(10) 2.01		5.9 (.23)	-26.5 (-1.50)	(19.	()
			0 140 2	(00.) 0		3.2 (* -1.2 (07)	-	(82.) (-28)		1 -1.4 (26)	2	٠	=-11.4 (68)		(01.)	J	-23.4 (-1.40)	(11,)	
				DAI RHAV		SCIINAFEC		FAAZWAFEC		SCTIFAINABEC	,		WD8/A		PRE-PLANNED			VURIC		

101 51/50/10	100100	19100107 ARSTEP	20111000	1657	5	000373			~1	300								٥	DATE 07	070375		a	PAGE	9	
								Route		STRUCTURE		CUMPARISONS	180	رن 2											
						PRE	FERRE	TOOM O	-	PREFERRED ROUTE: GROUND. MIND AND TOTAL BENEFIT	. NIN	A O	10 10	DIAL	BEN	EF11									
								DATA B	ASE	BASED ON R	RUNS	1	ROUL	THROUGH 39											
								AIRPORT PAIR	2	PAIR	1 11	LAX		080											
		3 A A	>		2	SCI			FAA	E C		SC	SCI/FAA NAFEC				× 0		PRE	PRE-PLANNED	ED		,	300	
UAL RNAV		000	(00.	+ #	10.01		.41)	1-1-0-0-1	9.5	.053	+ 11	0.1.		.05)	+ H	13.4		.213 .213	+ u	0 0 0			13.4		. 213
SCIINAFEC		-5.7 (423	+ 11	200		.000	-6.7 + -3.6 =-10.3	7 9 5		+ 11	+ -3.6		493 263 753	+ 11	-1.3		. 093	+ -3.6	222	49) 26) 75)		1.7		25.0.0
FAA/WAFEC		0 9 9 9	05)	+ 11	6.7 5.6 10.3		.48) .75)	+ H	000	.000	+ 11	000		6000	+ 11	2.3		1.04)		000	6000	+ "	2.5		1.04)
SCTIFAINAFEC		0 0 %	51	+ #	5.6		48) 26) 75)	+ "	000	.000	+ n	000		.000	+ 11	111.4		1.04)	+ "	000	(000	+ 11	2.3	-:-	1.04)
NUP/A		13.4	2.5.1	+ 11	1.3		56)	-14.4 + -2.3 =-16.7	725	(-1.05)			(-1.05) (-1.22)	(-1.05)	+ 11	000		6000	+ -2.3		(-1.05)	+ +	000		.000.
PRF-FLANKED	. "	- H		+ 11	6.7 5.6 10.3			+ 11	550	.000	+ 11	000		(00)	+ #	14.4 2.3		1.043	+ n	000	000	+ 11	2.3		1.043
YOBYC	* "	13.9 0		+ 11	1.3			-14.4		(-1.95) (-1.17) (-1.22)	+ 11		(-1.05) (-1.22) (-1.22)	(-1.05)	+ 11	000		.000	+ -2.3 =-16.7		(-1.05)	+ 11	000		6000

	900	5 (62)	3 (01)	7 (1.21)	7 (1.21)	7 (.12)	7 (1.21) 7 (49) 0	(000.
		10.4 10.5 1.9	9.9	1.6.7	+ 10 7.9 7.9	+ 1.7	16.7	
T01AL	PRE-PLANNED	3 (02) 1.8 (15) 2.1 (15)	-10.3 (76) + 6.6 (.49) # -3.7 (27)	+-1.2 (09)	+ -1.2 (09)	-16.7 (-1.22) + 8.4 (.61) = -8.3 (61)	000000000000000000000000000000000000000	+ 6.7 (-1.22)
ROUTES AND	× ×	16.4 (1.19) +-10.2 (74) = 6.2 (45)	+ -1.7 (13) = 4.6 (.34)	+ -9.6 (70) = 7.1 (.52)	16.7 (1.22) + -9.6 (70) = 7.1 (52)	0000	16.7 (1.22) + -8.4 (61) = 8.3 (61)	.00 (000)
STRUCTURE COMPARISONS REFERRED ROUTE, ADDITIONAL ED ON RUNS 1 THROUGH 39 PAIR 41 LAX ORO	SCI/FAA NAPEC	+ 1.3 (- 1.02)	-10.3 (75) + 7.9 (.58) = -2.4 (18)	6000	0000	-16.7 (-1.22) + 9.6 (.70) = -7.1 (52)	+ 1.2 (.09)	+ 7.9 (-1.22)
ROUTE STRUCTURE OR RENEFITE PREFERED REDATA BASED ON RUNS ATROOT PAIR 41	N F F F E C	1.3 (-, 02)	+ 7.9 (.58) = -2.4 (18)	(000.000.000.0000.0000.0000.0000.0000.0000	000.	-16.7 (-1.22) + 9.6 (.70) = -7.1 (52)	+ 1.2 (.09)	+ 7.9 (-1.22)
ALTERNATE ROUTE REM	SOL	8 - 1 - 6 - 6 1 1 1 1 1 1 1 1 1 1 1 1 1 1	(600.	10.3 (58) + -7.9 (58) = 2.4 (18)	10.3 (.75) + -7.9 (58) = 2.4 (.18)	+ 1.7 (.13) = -4.6 (34)	10.3 (.75) + -6.6 (49) = 3.7 (.27)	(47)
74	WAL	(00°) 0° · · · · · · · · · · · · · · · · · ·	+ 8.4 (.62) = -1.6 (11)		.3 (.02)	-16.4 (-1.20) + 10.2 (.74) = -6.2 (45)	+ 1.4 (.02) = 2.1 (.13)	18.5 (-1.20)
		UAI PNAV	SCIZMAFEC	FAA/NAFEC	SCITFATNAFEC	4/00A	DER-PLANNED	VCR/C

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	⊃ <u>2</u>	12			Z.	PRE STI	ROUTE STRUCTURE COMPARISONS PREFERRED ROUTE: GROUND: WIND AND TOTAL BENEFIT DATA HASED ON RUNS I THROUGH 39 AIRPORT PAIR 5: EWR ORD CI FAA SCI/FAA FEC NAFEC NAPRC	A TAG	ROUTE STRUC ROUTE: GROU ATA HASED ON ATRPORT PAIR FAA	ASED GREEN RT PAI	ROUTE STRUCTURE COMPARISONS D ROUTE: GROUND: WIND AND TOT DATA HASED ON RUNS I THROUGH AIRPURT PAIR 5: EMR OR	E CC	DHPARIS ND AND 1 THRO ENR SCI/FAA NAFEC	ARISON AND TO THROUGH EWR C	HPARISONS D AND TOTAL 1 THROUGH 39 ENR ORB SCI/FAA NAFEC	Z L		0 <			RE-P	PRE-PLANNED	9			30	
IIAS PNAY	+ 11		600	+ 11	545		193	• "	3.5		.43)	• •	1.5	-	199	+ +	-00			+ #	-00				-04		
SCIANAFEC	-2.2	111	666	+ 11	000		600	+ 4	200			+ n	000	555	666	+ #	1.6	-	.451	+ 11	 	100	3.69	•••	4.6		199
FAA/MAFEC			36)	+ 11	2.6.4		(51)	+ 11	000		000	+ #	200	-	123	+ #	5.5		.513	+ •	5.5		513		5.5		51)
SCTIFAINAFEC	.3.5			+ 11	000		(00)	+ 4	N. 0. 2			+ 11	000		6000	+ 11	7.3		.45)	+ #	2.4.	-			. a . a . 1 . 4 . 4 . 4		. 193
VuR/A			35.	- 11	5.3		643	+ 11	5.5		76)	+ #	3.3		64)	+ 11			6666	+ u			2000	+ 11	000		666
FRE-PLANNED	- 6 9			+ 11	5.5		643) 643	+ 4	5.7		.52)	+ #	3.3		64)	+ 11	000		6000	+ 11	200		(00)	+ #	000		6000
7790	1 - 6 - 1 - 1 - 1 - 1 - 1 - 1		183		2.0		64)	+ 1	5.5		.52)		3.3		.45)				.000	+	0.0		(00)				.000

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01 AUSTED COUT/1857 0005/1
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ALTERNATE REUTE PENEFIT: FREFERRED HOUTE. ADDITIONAL HOUTES AND TUTAL DATA BASED ON PUNS I THROUGH 39 ROUTE STRUCTURE CUMPARISONS

EWR DRD

51

AIRPURT FAIR

	VANA	SPI	FAANAFEC	SCI/FAA NAFEC	, 0 × 0 ×	PRE-PLANNED	300
UAL PARY	(00.)	\$.5 (.49) + .8 (.12) = 4.4 (.61)	3.2 (.43) + 1.9 (.26) ± 5.0 (.69)	3.5 (.49) + .8 (.12) = 4.4 (.61)	5.7	.67) 4.9 (.68) .11) + -5.4 (75) .78) =5 (07)	+ 3.0 (.41) = 7.9 (1.08)
SCIZMAFEC	-3.5 (49)	(00.) 0. +	+ 1.0 (.14) = .6 (.09)	+ 4		.19) 1.4 (.19) 01) + -6.2 (87) .18) = -4.9 (68)	+ 2.2 (.30)
FALINAFEC	-3.2 (4) 1.9 (25) 5.0 (70)	. 4 (. 05) + -1.0 (- 14) 6 (- 09)		1.0 (14)	+ -1.1 (15)	.24) 1.7 (.24) .15) + -7.3 (-1.01) .09) = -5.5 (77)	# + S.1.1
SCIZFAZNAFEC	-3.5 (49)	000.	+ (-0 (-14)	000	+ -1.4 (19) + -1.1 (01) -1.3 (.18)	19) 1.4 (.19) (01) + -6.2 (87) 18) = -4.9 (68)	+ 11
W00/4	+ -4.9 (68) + -5.7 (11)	-1.4 (19) 1.3 (18)	-1.7 (24) + 1.1 (15) =6 (09)	+ 1.4 (19)	000	.00) + -6.2 (86)	5.5
PRE-PLANNED	+ 5.4 (68)		+ 7.3 (1.00)	+ 6.2 (-86) = 4.9 (.67)) 2.9 0.5 C	.00) .85) + .00(.00) .85) + .00(.00)	* * * * * * * * * * * * * * * * * * *
Vn9/C	+ -3.9 (68) = -7.9 (-1.10)	+ -2.2 (19) = -5.5 (48)	+ -1.1 (16)		+ -2.2 (31) + -8.4 (-1.17) 31) + -8.4 (-1.17)	0000

u7/e3/75 19100101 ARSTEP	101 ARSTEP 000373657		8/8000	2	300	0						TAU	DATE 070375	521	۵.	PAGE	2	
				ROUTE	STRUC	10.0	GHO	ROUTE STRUCTURE COMPARISONS										
			POUTE ST	RUCTURE	RENE	E	SROUNG	STRUCTURE RENEFIT: GROUND, WIND AND TUTAL	DNA	TUTAL								
				DATA BASED ON RUNS	SED OF	RUN		1 THROUGH 39	39									
				AIRPOR	AIRPORT PAIR	2.		ENR URD	0									
	ANA	o d	SFI		FAA		00 Z	SCI/FAA NAFEC			YOK -		PRE-PLANNED	NNED		>	0 U	
UAL RNAV	(00.) 0. +	2.5	300	4 H		36)	200		303	1.02	(81.93)	255	- 45	(20.0)	• •	100	1.24)	
SCIVNAFEC	-2.2 (30) + -2.2 (31) # -4.4 (61)	***	600.	+ 11		02)		000		1.6	-	653	13.3		• •	N. 6 . 4	(84°.	
FAA/WAFEC	-2.5 (37) + -2.4 (33) E -5.0 (70)	+ 11	(20)	• •		666	2000	(20.0)	556	7.7	555	222	7.1.2	(52)	• •	7.92	52	
SCIVEANAFEC	+ -2.2 (30)	• "	666.	+ 11		02)	+ 11	000	6000	1.3	:	45)	11.6	(22)	+ 11		. 653	
VORZA	1.1 (15) + -6.8 (94) = -5.7 (79)	3.3		3.7		61)			.453 .633 +	96.5		333	16.20		• •	0.2.2	6	
PRE-PLANED	1.1 (5)	+ 1.6 # 0.9	(52:)	+ 3 - 5 - 5		52) 24)	4.9 4.9		45) +	6.2		522	000		• •	0 3 3	1.15)	
VURIC	1.1 (.15) + -9.0 (-1.25) E -7.9 (-1.10)	5.3 = -6.8 = -5.5	45)	3.7		523	2.6.5	3 (9 41)	55.6	2.5.5		622	0 4	(-1.17)	+ "		666	

10100111 ST30110	ARSTEW 000313651	657 000 573	00 5			UA1E, 070375	PAGE 25
			HOUTE STRUCTURE COMPARISONS	COMPARISOMS			
		BUEFFRUE	BREFFRRED POUTE: GROUND, WIND AND TOTAL BENEFIT	MIND AND TOTAL	RENEFIT		
			DATA HASED ON RUNS I THROUGH 39	S I THROUGH	*		
			AJRPOPT PAIR 61	. ORD ENR			
	PAL	SFI	FAA NAFEC	SCI/FAA NAFEC	× 0 ×	PRE-PIANNED	*0
UAL BUAV	000000000000000000000000000000000000000	7 (-113) + -1.0 (-17) + -3 (04)	000000000000000000000000000000000000000	.0 (.00)	.00 (00)	1) + -1.1 (19) + -1.5 (09) + -1	0.00
SCITNAFEC		00000	+ (1.2 (.21)	1.2 (.21)	-7 (-13) + 1.2 (-21) 5 (-08)	1) + 1.8 (32) + 1.6 (.27) + 1.8 (.05) =	1.2 (.21)
FAAZNAFEC	+2 (04)	1.2 (21)	6666	(000.	00000) + -1.1 (19) + .3 (.09) + .3 (.13) =	000.
SCIVEANAFEC	(04)	+ -1.2 (21)	6000		6000	()	(000.
V087A	+ - 2 (- 04)	+ -1.2 (21)	600000000000000000000000000000000000000	(000.) 0	000.) + -1.1 (19)) +8 (13) =8	66600
CHANTTO-LOU	1.1 () ()	+ -(-6 (27) =3 (05)	+ 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	1,1 (.19)	1.1 (.19)	+ II (000.)	1.1 (.19)
V097C	0 (.00)	(15.) 5.1-+	(00.) 0. +	(00.) 0	(00.) 0. +	1 + -1:1 (-:19) +	(00.) 0.

	¥ 0		3 (000)	6 (.28)	6 (.28)	000.	8 (.15) 5 (.21) 0 (.34)	(00.) 0
		•••	-			• • • •	8.0.0	
		+ 11	••	• •	+ "	+ 11	• •	•
	ED				0673	213	0000	
	PRE-PLANNED							
	RE-I	-1.3	5.1.	633	6 4 4	2.0	000	F. 5.
	-	+ #	+ 11		+ n		+ 4	17
		6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6	6000	28)	28)	6666	213	.000
	α	••••		• • •	• • •			•••
	×0×	Non	non	099	000	000	820	00
							2.0.2	
DATA BASED UN RUNS I THROUGH 39 AIRPORT PAIR 61 ORD EMR		+ "	+ 11	+ "	+ 11	+ 11	+ 11	+
1 THROUGH 39	*	283	283	6666	000	283	073	28)
THROU	SCI/FAA NAFEC							
1 0	NA NA	29.11	29-	000	000	11.6	0 = =	-1.6
		+ 11			+ 11	+ "	+ 11	
		28)	363	600	6000	99	. 073	600
PAIN	EC	.28)	28)			28)		26)
100	FAA	200	500	000	000	099	E 2 4	0.9
DATA BASED ON RUNS ATREORT PAIR 61		5.11	24:		•	0.11	4 2 2	7
40		• •		+ 11	+ "	+ #	+ 11	•
		600	6000	28)	283	.00)	.05) .21) .26)	.00)
	STI				:			
AL TERNATE	2	? ? ?	300	5.01	2.5-	200		
150		+ #	+ 11	+ #	+ 11	+'#	+ 4	
		600	.000	243	.28)	.043	. 5031	.003
	۷>	• • •				1.1	• • •	: .
	RNA	000	MOM	54.1	1.6	200	.5 .1	5.0.
				1	1	•		
				+ "	+ 11	+ 11	**	
					0			
		>	EC	EC	CET/FA/WAFEC		OHE-PLANED	
		VAL BNAV	SCIZNAFEC	FAA/NAFEC	FA	<	PLA	U
		7	13	**	13	V08/V	u o	V09/C

	77								
	-	SOI	ATRPORT DAIR 6	BCI/FAA NAFEC	> a <	PRE	PRE-PLANNED	30	
UAL PNAV			-1.0 (24)	+ -1.4 (24)	900	.003	1 (- 19) 7 (- 11)	0.00	999
SCIJNAFEC	+ 1.0 (15)	(00.	+4 (07)	+	+ 1.27	.13) + 1.8 .21) + .4	.4 (.06) +	5.5.7	
FAA/NAFEC	. 1.4 (.24)	.7 (.13) 4 (.07) = 1.1 (.20)	(000	0000	+ #	28) + 7	.7 (13)	0.00	.28)
SCIVEALNAFEC	.0 (.00)	+ .4 C .073	000000000000000000000000000000000000000	(000.)	+ 11	.00) .28) + .7 .28) =4	.1.1 (19) .7 (.11) +	0.4.4	. 28)
VURIA	+ 2 (04)	+ -1.2 (21)	+ -1.6 (28) + -1.6 (28)	+ -1.6 (28) + -1.6 (28)	• • •	.000	9 (-15)	000	666
PRE-PLANNED	1.1 (.19) + .5 (.11) * 1.7 (.30)	1.8 (.32)	+ 1.1 (19) 7 (13)	+ 1.1 (19) + (13) (1:1	159		2.00	555
37404	+ - 2 (- 04)	15. 1.2 (21)	+-1.6 (28)	+-1.0 (28)	••••	.00)	9 (-119)	000	0000

	ğu	+-13.9 (60) 12.9 (50)	-12.9 (50) 10.2 (44)	-12.2 (52) + 35.7 (1.54) = 23.6 (1.01)	+ 35.7 (1.54) = 23.6 (1.01)
	PRE-PLANNED	-4.4 (19) +-19.7 (85) a-24.1 (-1.04)	14.9 (.65) 14.9 (.65) 2.8 (.12) -2.7 (-12) +-48.6 (-2.11) +-48.6 (-2.11) +-12.9 (-50) +-18.7 (81) =-33.7 (-1.47) =-33.	.0 (.00) -12.2 (52) -17.6 (76) -12.2 (52) .0 (.00) + 35.7 (1.54) + 30.0 (1.30) + 35.7 (1.54) .0 (.00) = 23.6 (1.01) = 12.4 (.54) = 23.6 (1.01)	.0 (.00) -12.2 (52) -17.6 (76) -12.2 (52) .0 (.00) + 35.7 (1.54) + 30.0 (1.30) + 35.7 (1.54) .0 (.00) = 23.6 (1.01) = 12.4 (.54) = 23.6 (1.01)
3E NE F 17	* <	1.0 (.04) +-13.9 (60) =-12.9 (50)	2.8 (.12) +-12.9 (56) =-10.2 (44)	-12.2 (52) + 35.7 (1.54) = 23.6 (1.01)	-12.2 (52) + 35.7 (1.54) = 23.6 (1.01)
PREFERRED RUITEI GROUND, MIND AND TUTAL BENEFIT DATA BASED ON RUNS 1 THROUGH 39	SCI/FAA NAFEC	13.1 (.57) 1.0 (.04) -4.4 (19) +-49.6 (-2.16) +-13.9 (60) +-19.7 (85) =-36.5 (-1.59) =-12.9 (55) =-24.1 (-1.04)	14.9 (.65) +-48.6 (-2.11) =-33.7 (-1.47)	(000.) 0	(000. 000. 000. + H
RUUTE STRUCTURE CUMPARISONS REUTE! GROUND, MIND AND TUTA ATA BASED ON RUNS 1 THROUGH ATRPURT PAIR 7: IAD LA	FAA	13.1 (.57) +-49.6 (-2.16) =-36.5 (-1.59)	14.9 (.65) +-48.6 (-7.11) =-33.7 (-1.47)		(000 000 000 000 000 000 000 000 000 00
PREFERRED	SOI	+ -1.0 (04) = -2.7 (12)	000000000000000000000000000000000000000	-14.9 (-,64) + 48.6 (2.08) = 33.7 (1.45)	-14.9 (64) + 48.6 (2.08) = 35.7 (1.45)
	7472	(00.) 0 #	1.8 (.04) + 1.0 (.04) = 2.7 (.12)	-13.1 (56) + 49.6 (2.12) = 36.5 (1.56)	-13.1 (56) + 49.6 (2.12) = 36.5 (1.50)
		UAI PNAV	SCI/NAFEC	FAA/NAFEC	SCIJFAINAFEC

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PAGE

UATE 070375

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000573

19100107 ARSTEP 000373657

07/03/75

C-20

6000 KN4 0000

000

255 255 255 200 200 200 200 200

22.11

17.6 =-12.4 =-15.7 =-23.6

17.6 = 12.0 = 12.2 = 18.7 = 18.7

21.3 21.3 12.9 10.9

. 6693 1. 633 1. 633 1. 633 5603 5533

1 + H + H + H + H

PREPLANED

22.11

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.5.4

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12.2 +-15.7 E-23.6

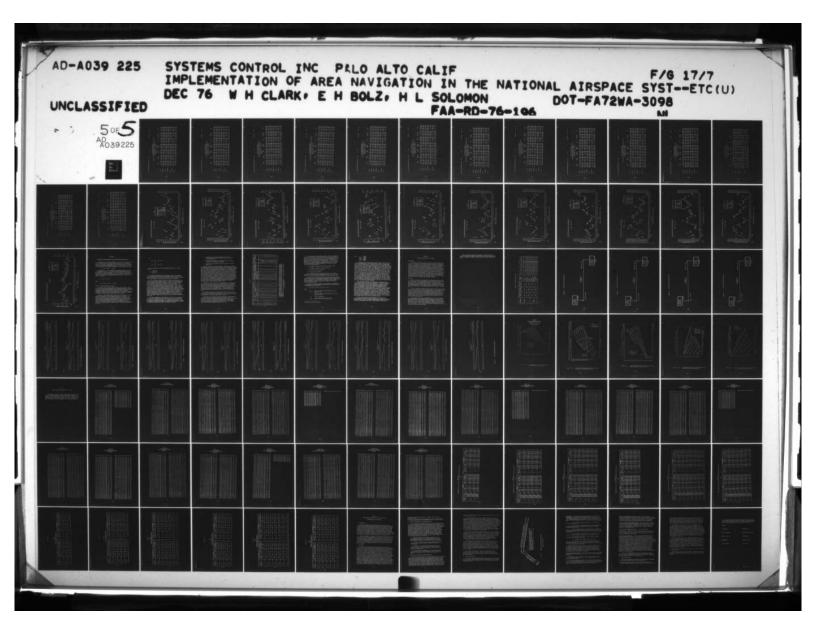
(-1.55) (-1.02)

12.2

6.53

PAGE 29	V 08	1.0 (.04)	2.8 (.12) + 17.7 (.77) = 20.4 (.88)	+ 34.3 (1.49) = 22.2 (.96)	+ 34.3 (1.49) = 22.2 (.96)	+ 7.2 (.31)	5.4 (.23) + 30.3 (1.31) = 35.7 (1.55)	0000
DATE 070375	PRE-PLANNED	-4,4 (-,19) +-8,1 (-,36) =-12,5 (-,55)	+-12.6 (56) =-15.3 (67)	+ 4.0 (77)	-17.6 (77) + 4.0 (18) 13.6 (60)	-5.4 (24) 23.1 (-1.02) 28.5 (-1.25)	0000	-5.4 (24) +-30.3 (-1.33) =-35.7 (-1.57)
TOTAL	× 0 ×	1.0 (.04) + 15.0 (.65) * 16.0 (.69)	+ 10.5 (.12) = 13.2 (.57)	-12.2 (53) + 27.1 (1.18) = 14.9 (.65)	-12.2 (53) + 27.1 (1.18) = 14.9 (.65)	6666 6666 6666 6666 6666 6666 6666 6666 6666	+ 23.1 (1.00) = 28.5 (1.24)	+ -7.2 (31)
STRUCTURE COMPARISONS RENEFIT: GRUUND. WIND AND ED ON RUNS I THROUGH 39 PAIR 7: IAD LAX	SCI/FAA NAFEC	13.1 (.57) +-12.1 (53) = 1.1 (.05)	+-16.6 (73) = -1.7 (08)	00000	6000 0000 0000 0000 0000 0000 0000 000	12.2 (.53) +-27.1 (-1.18) x-14.9 (65)	+ -4.0 (17) = 13.6 (.59)	12.2 (.53) 34.3 (-1.50) 22.2 (97)
ROUTE STRUCTURE OF STRUCTURE OF STRUCTURE RENEFIT: GRAND BASED ON RUNS ATHPURT PAIR 7:	FAA	+-12-1 (53) 13-1 (53)	14.9 (.65) +-16.6 (73) * -1.7 (08)	000000000000000000000000000000000000000	+ 1000	12.2 (.53) +-27.1 (-1.18) =-14.9 (65)	+ -4.0 (-17) = 13.6 (-59)	12.2 (.53) +-34.3 (-1.50) =-22.2 (97)
75.16.51	STI	1.8 (08) 4.5 (20) 7.8 (20)	++	-14.9 (65) + 16.6 (.73) = 1.7 (.08)	-14.9 (65) + 16.6 (.73) = 1.7 (.08)	+-10.5 (40) =-13.2 (58)	+ 12.6 (.55)	-2.8 (17) +-17.7 (17) =-20.4 (89)
7 ARSTEP 040173G57	PNAV	(00.) 0	1.8 (20) + -4.5 (20)	-13.1 (57) + 12.1 (13) 1.1 (05)	+ 12-1 (57) + 12-1 (53) = -1-1 (05)	+-15.0 (66) =-15.0 (56)	# 8.1 (.19) # 12.5 (.55)	-1.0 (04) 22.2 (97) =-23.2 (-1.02)
07/03/75 19100101 4PSTEP		UAL RHAV	SCIZNAFEC	FAAZMAFEC	SCTIFAINAFFC	VOOLA	PRE-PLANNED	VERIC

	JAU.	PHEFERRET SAME	RUUTE: GROUNE DATA BASED ON F AIRPORT PAIR	PHEFERRED RUNTE: GROUND, MIND AND TOTAL B DATA BASED ON RUNS 1 THROUGH 39 AIRPORT PAIR 81 LAX IAD EFF	BENEF11	PRE-PLANNED	0
UAS RNAV	(00.) 0.	26)	+ 11	+ -4.7 (26)	11.9 (.67) + -1.9 (28) = 6.9 (.39)	0000	+ -4.9 (28)
SCIVAFEC	+ 4.7 (05) = 3.8 (25)	(00.) 0. +	+ 3.2 (.18) + 9.7 (.54)	+ .0 (.003 + .0 (.003 * .0 (.003	11.0 (.62) +3 (02) = 10.8 (.60)	+ 4.7 (.26) = 3.8 (.22)	11.0 (.62) +3 (02) = 10.8 (.60)
FAA/MAFEC	+ 1.5 (A)	+ -5.2 (17)	(00.) 0. +	+ -3.2 (18) + -9.7 (55)	1.5 (.25) + -3.5 (19) = 1.1 (.00)	+ 1.5 (41) -7.3 (41) -5.9 (33)	4.5 (19) + -3.5 (19)
TOT /FAZNAFEC	+ 4.7 (-26)	+ =	6.5 (.37) + 3.2 (.18) = 9.7 (.54)	+ =	+ 11.0 (02) 10.8 (02)	+ 4.7 (.26)	11.0 (.62) +3 (02) = 10.8 (.60)
4780.	+ 4.9 (67)	+ 11.0 (62) + .3 (.02) =-10.9 (61)	+ 3.5 (25) + 3.5 (19) + -1.1 (06)	-11.0 (62) + 3 (62) =-10.8 (61)	0000	+ 4.9 (67) = -6.9 (39)	000 000 000 + n
OBNORTH A	(00.) 0. *	+ -1.7 (26) = -3.8 (22)	+ -1.5 (08) = 5.9 (.33)	+ -4.7 (26) = -3.8 (22)	+ -0.9 (28)	(000.	11.9 (24)



						ROUTE	STRUCTU	ROUTE STRUCTURE COMPARISONS	ARISO	N.										
			LTERNAT	ALTERNATE ROUTE BEWEFIT! PREFERRED HOUTE. ADDITIONAL HOUTES AND TOTAL	BENE	F171 P	REFERRE	D MOUTE	. ADD	ITION	1 ×	OUTE	ONT S	TOTAL						
					AC	TA BAS	ED ON 8	DATA BASED ON RUNS I THROUGH 39	THROU	GH 39										
						IRPURT	AIRPORT PAIR		LAX 140	140	-									
appropriate and the second	חאר	1		108			FAA	•	SCI/FAA			*	VOR	PRE	PRE-PLANNED	8		1	VOR	
	3	*		NAFEC		2	MAFEC		NAFEC										u	
UAL BNAY	6.	(00.)		Ċ	8	5.0	(88,)		-	.22.		6.9	(88,)		•	(00.)				3
	•••	(00.	4 - 2 - A	ü				1.4					. 44	• •	5.3		• •			.44)
SCI/NAFEC	3.6	(66.)		0 0 0	6	9.7	(55.)			.00	-	10.8	(09.	-	3.6			10.8		609
	4 -1.4	(04)		0.	. (00.	-2.5	(14)		0	.000	+		(50)	٠	1.9-	(38)	•	5	:	03)
	* 2.4	(11.)		(00.) 0.	6	1.5	(0".)			.000		10.2	(75.	•	-5.0		•	10.2		.57
FAA/WAFEC	-5.9	(33)	i	-	53	•	(00.		Ċ.	55)			(90.		.5.9	(33)	-	=		.00
		(90.	+ :		()	•	600.	+ 2.5	_:	14)		2.0	===	• •	2.00 +		+ •	2.5		===
		1200	•	4					•			:						:		:
SCI/FA/NAFEC	3.8	(22.)		0. 10.	(00.	4.7	(55.)		٠.	.000	-	0.0	600		3.8	(55.)	-	10.0		.603
	. 2.4	(+1.			• • • • • • • • • • • • • • • • • • • •	7.2	100	- "		000	- "	10.5	.57	• •	-5.0			10.2		.57)
VOR/A	.6.0	(39)	-10.8	_	7	-	(06)	-10.6	_	61)		•	(00.		6.9-	(39)		•		.00
		(05)	- "	.5 (-05)	5.5	-3.1	35	-10.2		58)		•••	66	- 4	13.1		• •			600
PRE-PLANED	0.	(00.		i.	ຄ	5.0	.33)			22)		6.0	.39		0.0	(00.				
	2.3	. 30	- н		. (91.	10.1	15.	2.9		16)		13.1	(#7.					13.1		74)
J/00A	6.0	(39)	٦.	(16) 8.0	25	-1.1	60	-10.0				00	600		6.9		•	• •		.000
	a -7.8	(144)	-						.:	.58)			(00.		13.1	(74)				00

SCI

		12			, 0, 4	PREFE SCI NAFEC	N C	BOUTE STRUCTURE C B ROUTE: GROUND: MI DATA BASED ON RUNS AIRPORT PAIR: 9: FAA	TES GRED ORT PA	GROUD ON PAIR	ND. H	ROUTE STRUCTURE COMPARISONS ROUTE: GROUND, WIND AND TOT TA BASED ON RUNS 1 THROUGH IRPORT PAIR 9: HIA CLI FAA 8CI/FAA NAFEC NAFEC	AND TO THROUG MIA C	PREFERRED ROUTES GROUND, WIND AND TOTAL BENEFIT DATA BASED ON RUNS 1 THROUGH 39 AIRPORT PAIR 91 HIA CLE CI FAA 8CI/FAA	9 6		× 00 ×		E	PRE-PLANNED	G C C C C C C C C C C C C C C C C C C C			ğu	
UAL BWAY	• •		665	+ "	000		500	* "	3.50	na.	25.57	44		.500	••	0 W =			+ =	000	666	555	0 N 3	- Ju	
SCI/NAFEC	44 1		565.	+ 11	***	777	666	~ m	3.3	200	255		000	886	••	212	400	36.4	1	000	66.	222	414		35.4
FAA/NAFEC	3.2.5			+ 11	77.5			+ #	000	000	866	1.2.7			• •	- N. C.		33.	+ •	5.5		588	-3.3		333
SCT/FA/NAFEC	441	1.1	503	+ 4	200		666	~ m		o.v.m.	273		000	666		6				000		222	414 64w		863
VOR/A	+ #	3.1	1.001	- "	9.4-			-1.0	3.3		886	+ # 9 9 9	0.0m	44)	+ #	••••	400	566		8.50	(-1.00)	222	••••		305
PRE-PLANNED VOR/C		3	666	• • •			.50)	w.m =	504 0		573	44		666	**	8 N. C			• •	000	666	222	6.24		

			ALTE	RNATE	F 90	ROUTE STRUCTURE COMPARISONS ALTERNATE ROUTE PENETITI PREFERRED ROUTE, ADDITIONAL ROUTES AND TOTAL DATA BASED ON RUNS I THROUGH 39	ENEF	IT:	PREF	RUUTE STRUCTURE COMPARISONS NEFIT: PREFERRED ROUTE+ ADDIT OATA BASED ON RUNS I THROUGH	RO RO	UTE.	A POCH	UTE. ADDITION	1 .	HON		AND 1	OTA							
							4	ATRPURT PAIR	1	8	•		HIA CLE	STE							7					
	Ruak	45			SCI	U		12 1	NAFEC		No.	SC.	SCI/FAA NAFBC				40×			PRE-PLANNED	LAVANE	0			20	
UAL RNAV	•••	566	988	***		655	+ #	3.4		# 6 k		*;;	-::	333	+ 11	# 0 F		594	• "	07.7.		651.	••			404
SCIZNAFEC	3	555	322	••••	000	666		3.4	100	353		900	J	566	• •	A. 4.		.013	••	177		61:	••	2.3		45.
FAA/NAFEC		.343	555	214				000		6000	••	-3.3		34)	+ •		000	365	* "	-3.4				200		565
SCI/FA/NAFEC	?		522	••••	000	966	••	3.3		355	++			666	+ "	A. 4.	500	.45.	••	044		666	• "			6.00
V08/A		. 000	385			C. C	+ 11			566	+ #	4.4			+ #	000	000	999	+ #	4-4		355	+ #			866
PPF-PLANNED		555	255	0.00	000	661.	+ 11	3.4		35.	••	000		697	• •	1.3		61.	- 4	000		6000	+ 11	4.0 W.0		353
VnP/C	4 4		265				+ 11			565	+ 11	E. 4.			+ 1	000		363	• •	-4.3						600

				570 55	104	ROUTE STRUCTURE CUMPARISONS ROUTE STRUCTURE RENEFIT: GROUND, WIND AND TOTAL DATA BASED ON RUNS 1 THROUGH 39 AIRPORT PAIR 93 H34 CLE	RUCT PATA	URE	ROUTE STRUCTURE CUMPARISONS RUCTURE RENEFIT'S GROUND, WIND NATA BASED ON RUNS I THROUGH AIRPORT PAIR 93 MJA CL	TURE IT 6 RUNS	CUMP GROUN	THROUGH	UND, WIND AN	9	01 AL	100 50					per est		
	12	44			Sr.1 NAFEC			* \$	FAA		6.2	GCI/PAA NAPEC				84		PRE	PRE-PLANNED	A		9 u	
UAL BHAY	505		666	44.				3.00		233	4.0		.500	• •	0.00		1.00)			555	4.5.5	900	1.00.1
SCI/NAFEC	5.0	88.	335		000	966	+=	764	353	28)		900	566		415	100	.053	3.3.3	1-1	.345	+ =	984	
FAA/NAFEC	-5.6		585	1.2.6		.28)		000		888			283	+ =	4.3.3		569	-5.6		57)	4.4	200	3.5
SCI/FA/NAFEC	.0.5	65	623	• •	999	866		7.8	353	226		000	888	+ 11	604	Li	. 651	**************************************			***	en a	.05.
VUR/A	• N. e	(35.	535	1 4.9		.503		3.3		225	9.24		503	+ 11			366		300			000	999
PRE-PLANNED	1:1	:::			0.0	.349	+ 11	5.6	.573	533	4.5 1.6		.503	+ 4	6.0				000	666	6.0.0		
7/80/	. 5.5	.563	535	4 4	ili	.50	+ 11	3.3		888	4 4		45)	+ #	000		888	- 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1	3	393	• •	000	

				PREFE	286.0	ROUTE STRUCTURE COMPARISONS PHEFERRED ROUTE: GRUUND, WIND AND TOTAL BENEFIT	STRUC	TURE NO.	COMPA	R1501	. S. T. D. T. A. L. D. T. D. T	E NE	E								
					4	DATA BASED ON RUNS	FD ON	20.00		1 THROUGH 39	2 .	P			10						
No.		12		SCI		ATRPORT PAIR 101 FAA NAFEC	FAA NAFEC	2	. 2	CLE MIA SCI/PAA NAFEC	1	Territor.	1	84	E	74-39	PRE-PLANNED			5 0	
HAL BNAV	•••	6666		5 (70)	585	***		686	955			• "	5.61	2.033	••		666		2.01		2.03
SCITHAFEC	***		••	6666	388	•••		356			866		5.5	2.75	+ •	2.5	(45)		5.5		1.713
FAA/WAFEC		50	••	(10) 6	355		000	666	***		336	-~	5.0	1.71)	••	940			5.0		.533
SCI/FA/WAFEC	4.00	3.00		600	966		200	336	••••		666		5.5	1.71		20.0		••	5.5		1.711
V64/4		(50.1-5	24.5	2 (-1.75)		21.15.0	(-1.75) (-2.29)		22.5		-1.753			338	• •	2.7	-9.5 (-1.05) + -9.7 (-1.05) 8-19.2 (-2.08)	••	•••		366
POE-PLANVED			440	5 (27)			r 703		***				5-2	2.043	••	000	6666	••	2.61		1.01)
3/304		(-1.05)	110.5	5 (-1.75) 7 (-7.35)		1.00	(-1.75) (-2.24)		-10.5		-1.753			666		2.7.5	-9.5 (-1.05) -9.7 (-1.05) -19.2 (-2.08)	••			388

							ROUT	E STRUC	TURE	ROUTE STRUCTURE COMPARISONS	RISONS	-		-	-	-	-				
			7	TERNAT	IE ROU	7E 9E	VEFITI	ALTERNATE ROUTE GENEFIT: PREFERRED HOUTE. ADDITIONAL	RED H	OUTE	ADDITE	DNAL	KOUT	KOUTES AND	D TUTAL	1					
						1122	DATA B	DATA BASED ON RUNS	RUNS		I THROUGH 39	30	1				*				
							AIRPO	AIRPORT PAIR	=		CLE NIA								-		4
		141			36.1			Z.						80						90	
		NAN			NAFEC			NAFEC		NAPEC	MC			4		PRE-F	PRE-PLANNED				
UAL BNAY		2 00	(00.	-2.5	-	27	:	-	6	-2.5	(15.27)		19.2	-	2.04)	•		600		-	2.04)
	••	00	.000			.45)	2.1	3 (•••03)	35	4.1.	35.	• •	19.2		2.043	9.7		52)	19.5		2.04)
SCIVAFEC	2	5 6	.273		,	.00		9 9	17.0		(00.)		21.7	1 2.	303	2.5		273	21.7	-	2.30)
	• •	9:	6	٠.		(00	+ 1.6		151	•	666		9.1.		2.47	-3.2			23.1		
												-									
FAA/WAFEC		55.3			000			000	666	44.5		**	21.3		2.26)	1.6	-:-	.50)	21.13		2.243
SCI/FA/NAFEC		50	.273		00	666	•	• •	150	•••	666	•	21.7		2.30)	2.5	_:		21.7		2.30)
		1.1 6	(04.		000	.000	. 2.0	_	. (22)	••	(000)		23.3	-	2.47)		:	. (00	23.3	-	2.47)
VOR/A	-10	3	2.083			(-2,36)	-21.1	•		-21.7	ů.	1	•		1	-19.2	٠.	(60	i	-	.00.
	-10.5		.2.043	2-23.5		(-2.54)	1-21.3	3 (-2,31)		8-23.3	(-2.54)	• •	•••		666	B-24.1	(-2.62)	353	• •		38
PRE-PLANNED		c. 4	.00.		-		-1.8	6 (20)	56	-2.5	(75)	•	10.2		2,04)		-	.000	19.2		2.04)
			.523			.08)	= 2.7		. 293	.7	.083	-	24.1		55)				24.1		2.55)
5 3/auA	-19.2	- 2 -	2.08)			(-2.36)	-21.1			-21.7		-	••		.000	-19.2	-	(6)		00	.00.
	-19.5		000	2.7.		101.01		2000			101		•	•		0.	1 36.	170	•		•

07/03/75 19100107 ARSTEP	101 ARSIEP UNUITEST	111000 151	2 300			UATE 070375	PAGE	20
				AUGUST MOUNTAINS				
		POUTE ST	PUCTINE BENEFIT	STRUCTURE BENEFIT GROUND, MIND AND TOTAL	D TOTAL			
			DATA BASED ON RUNS	JNS 1 THROUGH 39				
			AIRPOHT PAIR 108	101 - CLE - NIA				
	RNAV	Sr.1 NAFEC	FAA	GCI/FAA NAFEC	o «	PRE-PLANNED		40A
UAL RNAV	666.	4 - 6 - 1 (98)	# -6-7 (23)	4.0 (+ 9.5 C 1 1 2 C 2 C 2 C 2 C 2 C 2 C 2 C 2 C 2	1.01) . 0 (.00) 1.03) + .4.8 (52) 2.04) = .4.8 (52)		(1.03)
SCT/NAFE"	. 6.1 (0000	+ 1.4 (.16)	0000	5.5 (+ 17.8 (1 # 23.3 (2	29) + 3.3 (44) 2.47) =7 (08)	5.5	(1.89) (2.47)
FAAZNAFEC	+ 6.7 (50) + 6.7 (.72) - 2.1 (.23)		0000	+ -1.4 (16)	5.0 (+ 16.4 (1 = 21.3 (2	.53) -4.6 (50) 1.74) + 1.9 (.20) 2.20) = -2.7 (30)	5.0	(1.74)
SCI/FA/NAFEC	+ 9.1 (.88)	0000	+ 1.4 (.16)	0000	+ 17.8 (1	2.47) -4.0 (44) 2.47)7 (08)	5.5 + 17.6	(1.89)
V08/V	-9.5 (-1.03) + -9.7 (-1.05) =-19.2 (-2.08)	-5.5 (60) +-17.8 (-1.94) =-23.3 (-2.54)	-5.0 (54) +-16.4 (-1.78) =-21.3 (-2.31)	-5.5 (60) +-17.8 (-1.94) #-23.3 (-2.54)	900	.00) +-14.5 (-1.04)	•••	366
PRE-PLANNED	+ 4.8 (.52) = 4.8 (.52)	4.9 (141) + -5.3 (16) = .7 (08)	4.6 (20) + -1.9 (20) = 2.7 (29)	4.0 (36) 4 -3.3 (36) 8 .7 (.08)	+ 14.5 (1	1.51) . 0 (.00) 1.54) + .0 (.00) 2.55) = .0 (.00)	9.5	(1.54)
VuR/C	-9.5 (-1.03) + -9.7 (-1.05) =-19.2 (-7.08)	-5.5 (60) +-17.8 (-1.94) =-25.5 (-2.54)	-5.0 (54) +-16.4 (-1.78) E-21.3 (-2.31)	-5.5 (60) +-17.8 (-1.94) =-23.3 (-2.54)	•••	.00) -9.5 (-1,04) .00) +-14.5 (-1,58) .00) =-24.1 (-2,62)	+ #	666

07/03/75 19100107 ARSTEP	107 ARSTEP 000373G57	657 000373	300		3	DATE 070375	PAGE 39
			ROUTE STRUCTUR	ROUTE STRUCTURE COMPARISONS			
	•	AL TERNATE ROUTE BI	ENEFIT: PREFERRED	ROUTE RENEFIT: PREFERRED ROUTE. ADDITIONAL KOUTES AND	AL ROUTES AND TUTAL	1	
			DATA BASED ON RUNS	JNS 1 THROUGH 39			
			ATRPORT PAIR 111	111 JFK SEA			
	RILAY	SCI NAFEC	FAA	BCI/FAA NAFBC	× 0×	PRE-PLANNED	, O & O & O
HAL BNAV	(00.) 0	6.7 (.30) + -3.8 (17) = 2.8 (.13)	+ 5.7 (09) + 5.7 (.25) = 3.7 (.16)	+ -3.6 (-17) = 2.6 (-13)	30.8 (1.36) +-12.5 (55) = 18.4 (.81)	.0 (.00) + .3.3 (15) = -3.3 (15)	30.8 (1.3e) +-12.5 (55) = 18.4 (.81)
SCIZNAFEC	+ 3.8 (.17) # -2.8 (13)	(00°) 0° + n	-8.7 (39) + 9.6 (.42) = .8 (.04)	(00.) 0	24.2 (1.06) + -A.6 (38) = 15.5 (.68)	+ .5 (.02) = .5 (.02) = -6.2 (27)	24.2 (1.06) + -8.6 (38) = 15.5 (.68)
FAA/NAFEC	2.1 (.no) + -5.7 (25) = -3.7 (16)	+ -9.6 (42)	00000	+ -9.6 (12) =8 (04)	32.9 (1.45) +-18.2 (80) = 14.7 (.65)	+ -9.1 (40) = -7.0 (31)	32.9 (1.45) +-18.2 (80) = 14.7 (.65)
SCITFALNAFEC	+ 3.8 (-17) = -2.8 (-13)	+ #	+ 9.6 (.42)	60000	24.2 (1.06) + -8.6 (38) = 15.5 (.68)	+ .5 (30) + .5 (.02) = -6.2 (27)	24.2 (1.06) + -8.6 (38) = 15.5 (.68)
V0R/A	-30.4 (-1.37) + 12.5 (-55) 18.4 (62)	-24.2 (-1.07) + 8.6 (-38) =-15.5 (69)	-12.9 (-1.46) + 15.2 (.81) =-14.7 (65)	-24.2 (-1.07) + 8.6 (.38) =-15.5 (69)	0000	-30.8 (-1.37) + 0.2 (.41) =-21.7 (97)	6000
DRE-PLANNED	+ 3.3 (.15) = 153	+5 (02) = 6.2 (27)	+ 4.1 (.40)	+5 (02)	30.8 (1.36) + -9.2 (40) = 21.7 (96)	0000	30.8 (1.36) + -9.1 (40) = 21.7 (.96)
7/80/	-30.8 (-1.37) + 12.5 (.55) =-18.4 (82)	-24.2 (-1.07) + 9.6 (.38) =-15.5 (69)	-32.9 (-1.46) + 18.2 (.81) =-14.7 (65)	-24.2 (-1.07) + 8.6 (.38) =-15.5 (69)	600.	-30.8 (-1.37) + 9.1 (.41) =-21.7 (97)	60000

	UAL VANS			S I I NAME OF STATE	11 11 11 11 11 11 11 11 11 11 11 11 11	, 00 ta	ROUTE STRUCTURE BENEF. ATA BASED ON AIRPORT PAIR FAA	E STREE BENIET PASED WAFEC	ROUTE STRUCTURE C RUCTURE BENEFIT: GR DATA BASED ON RUNS AIRPORT PAIR 111: FAA	TI GRE	OUND,	ROUTE STRUCTURE COMPARISONS ROUTE STRUCTURE BENEFIT: GROUND, WIND AND TUTAL DATA BASED ON RUNS I THROUGH 39 AIRPORT PAIR 11: JFK. SEA AIRPORT PAIR 11: JFK. SEA CI FAA REC NAFEC NAFEC	30	5	4 VOR		R4	PRE-PLANNED	6		Š	750 STO 1
IAL RNAV	200	666		2.12	.050 .070 .071	+ #	3.1.5		5691	••	2.12	29.	953	+ 11	18.4	56.	* * * * * * * * * * * * * * * * * * *			20.9	6 M 3	53.
SCIZNAFEC	-1.2 (-1.7 (-		+ u	900	666	+ "	336		659	+ =		300	866	+ 19	19.7		+ -5.0			19.7	200	
FAA/WAFEC	-1.6		- 4	446	.02)	• •	000		866	+ =	440	(02)	225	+ #	19.3	.203	-1.6		073	19.3	200	
SCIJFAZNAFEC	1.1.7		+ 11	000	666	**	4.4.0		0.020	+•	•••		666	+ 12	19.7		+ -5.0			19.7	200	
VOR/A	-20.9 (-2.6 (19.7	225	1.67	• "	-19.3		20)		-19.7	26.193	555		900	666	-20.9				000	666
PRE-PLANED	# 3.4 C	.153	+ #	5.0 6	.05)	+ "	5.4		31.)	+ 11	5.0	000	223	2 - 2	20.9 (.033	••	000	6000	20.9		26.6
2/80/	2.50.9		19.7		193	+ 11	-19.3 C		65)		-19.7	(19)	555		000	6666	-20.9			+ 11	000	366

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DATA BASED ON PUNS I THROUGH 39

ATRPUPT PAIR 121

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	HINA	SCI	NAFEC	SCI/FAA NAFEC	× 0 ×	PRE-PLANNED	80
UAL RNAV	(00.) 0. +	3.0 (.15) 12.7 (13) -3. (.62)	+ (1 +	1.4 (.07)	30.4 (1.51) + -5.9 (30) # 20.5 (1.22)	(000.	
SCITARFEC	-3.0 (15) + 2.7 (13) =3 (62)	000000000000000000000000000000000000000	+ 2.2 (.11) = .5 (.03)	-1.6 (08) + 2.2 (.11) - 5 (.03)	27.4 (1.36) + -3.3 (16) = 24.1 (1.20)	-3.0 (15) + 2.7 (.15) =3 (02)	
FAA/WAFEC	-1.4 (9.7) + .5 (03)	+ -2.2 (11)	200	(000.	29.0 (1.44) + -5.4 (27) = 23.6 (1.18)	+ .5 (.03) = .8 (04)	29.0 (1.44) + -5.4 (27) = 23.6 (1.14)
SCTIFAINAFEC	-1.4 (67)	1.6 (.08)	000 000 + H	000.	29.0 (1.44) + -5.4 (27) = 23.6 (1.18)	+ (07)	+ -5.4 (27) = 23.6 (1.18)
VU9/A	-30.4 (-1.53) + 5.9 (.30) =-24.5 (-1.23)	-27.4 (-1.38) + 5.3 (.16) =-24.1 (-1.23)	-29.0 (-1.46) + 5.4 (.27) =-23.6 (-1.19)	-29.0 (-1.46) + 5.4 (.27) =-23.6 (-1.19)	(000.	-30.4 (-1.53) + 5.9 (.30) =-24.5 (-1.23)	00000
PRF-PLANNED	(00.) 0. +	3.0 (.15) + -7.7 (13) = .3 (.02)	+5 (03)	+5 (03)	\$0.4 (1.51) + -5.9 (30) = 24.5 (1.22)	(000.) 00. + H	30.4 (1
2/88/6	-30.4 (-1.53) + 5.9 (.30)	+ 5.5 (-1.38)	+ 5.4 (-1.46) -29.0 (-1.46) + 5.4 (-27) + 5.4 (-27)	+ 5.4 (-1.46)	0000	-30.4 (-1.53) + 5.9 (.30)	0000

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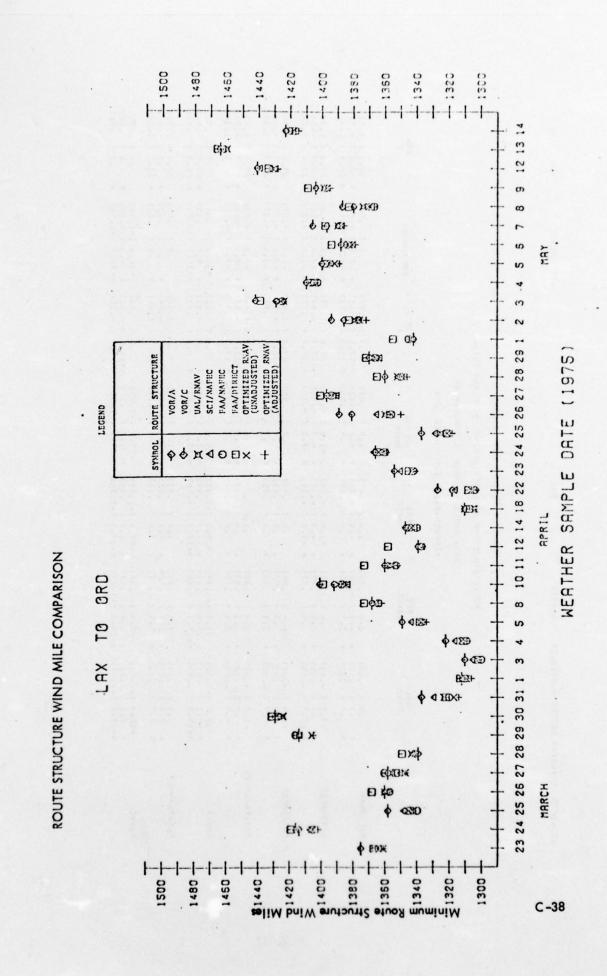
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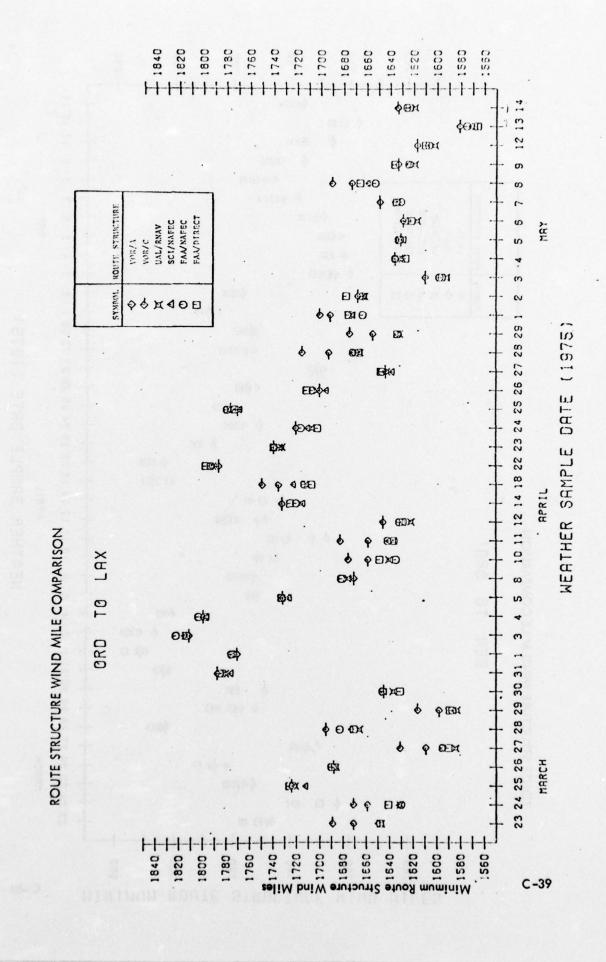
HOUTE STRUCTURE COMPANISONS
ALTERNATE ROUTE RENEETT PREFERRED ROUTE. ADDITIONAL HOUTES AND TOTAL

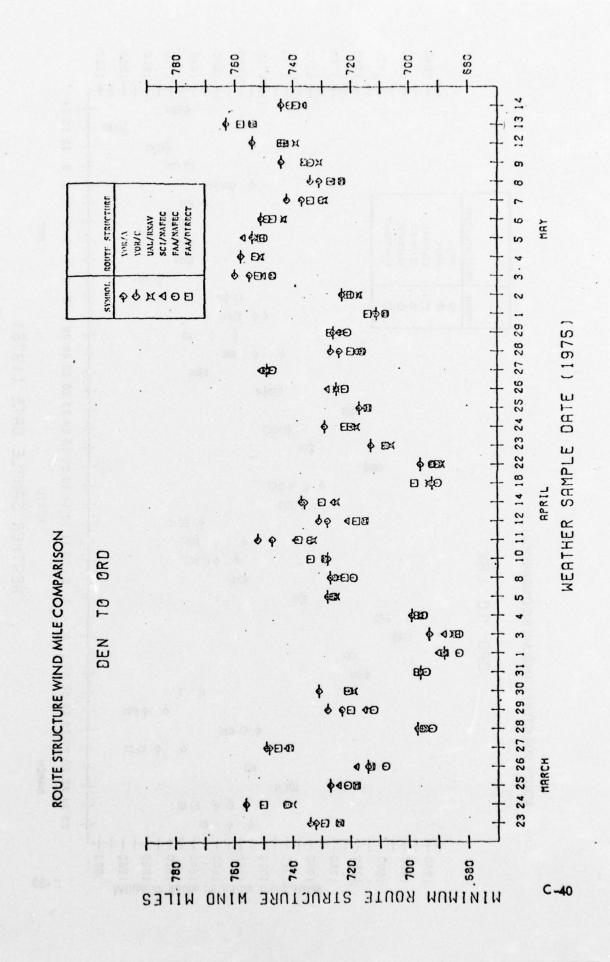
DATA BASED ON RUNS I THROUGH 39 AIMPORT MAIR 121 SEA JFK

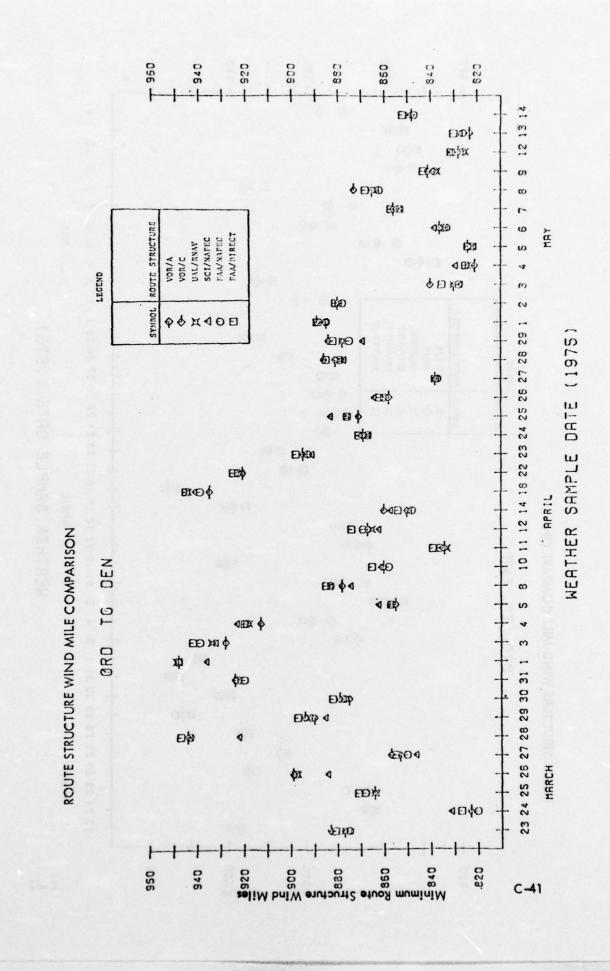
	UAL	SCI NAFEC	FAA	SCI/FAA NAFEC	a <	PRE-PLANNED	g o
UAI BHAV	(60.) 6.	.3 (.02) + 2.6 (.13) - 2.9 (.13)	# + H	.1 (.003)	+ 11	24.5 (1.23) .0 (.00) -6.2 (31) + .2.5 (13) 16.3 (.92) = .2.5 (13)	24.5 (1.22) 5.8 (.19)
SCIZNAFEC	+ -2.6 (13) = -2.9 (15)	00. 00. + 00. + 00. + 00.	+ -2.5 (13) + -2.5 (13)	+ -2.5 (13)	24.1 (1.21) + -8.8 (44) = 15.4 (.77)	+-5.1 (26)	24.1 (1.20) + 1.2 (.06) # 25.4 (1.26)
FAAZMAFEC	#	+ 2.5 (03)	(000.) 0	(000.)	23.6 (1.18) + -6.3 (31) * 17.3 (87)	2.6 (13) + -3.4 (17) =	23.6 (1.18) + 3.7 (.19) = 27.3 (1.36)
SCIZEAZAAFEC		+ 2.5 (03)	000000000000000000000000000000000000000	000.	+ 2.5 (13) + .0 (.00) + .0 (.00) + -6.3 (31) + -2.6 (13) = 2.0 (.10) = .0 (.00) + -6.3 (31) + -2.6 (13)	+ -2.6 (13) + -3.4 (13)	23.6 (1.18) 3.7 (.19) 27.3 (1.36)
75574	-24.5 (-1.24) + 6.2 (-31) =-18.3 (92)	-24.1 (-1.22) + 8.8 (.41) =-15.4 (78)	-23.6 (-1.19) + 6.3 (-32) =-17.3 (68)	-23.6 (-1.19) + 6.3 (.32) =-17.3 (88)	6000		10.0 (.50)
PRE-PLANNED	2.5 (.13)	$\begin{array}{cccccccccccccccccccccccccccccccccccc$	+ 2.6 (.13) = 3.4 (.17)	+ 2.6 (.13) = 3.4 (.17)	+ -3.7 (19) = 20.7 (1.04)	(000	24.5 (1.22) + 6.3 (.31) # 50.7 (1.55)
3/a0A	-24.5 (-1.24) 1 -3.8 (-19) =-28.3 (-1.43)	000,) 0, (95,1-) 2,2-2 (00,) 0, (91,1-) 23.6 (-1,19) .0 (-1,1-) 24.5 (-1,1-) 1.1-2 (12,1-) 2,5-2 (00,) 2,1-3 (12,1-) 2,1-3 (12,1-) 3,1-3 (1	-23.6 (-1.19) + -3.7 (19) =-27.3 (-1.38)	-23.6 (-1.19) + -3.7 (19) =-27.3 (-1.38)	-10.0 (50)	-24.5 (-1.24) + -6.3 (32) =-30.7 (-1.56)	000

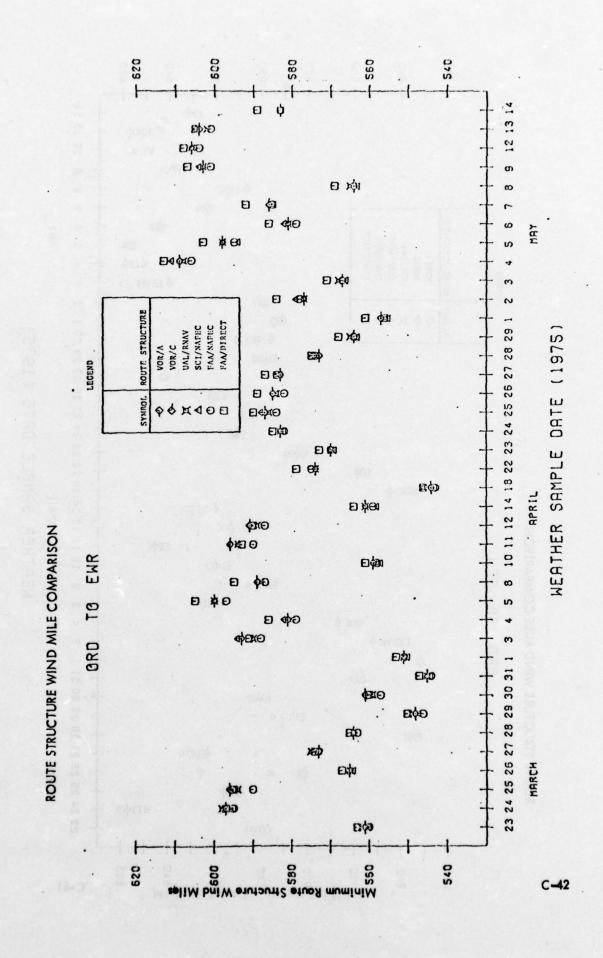
7,03775 19100107 ARSTE	107 ARSTEP 000373G57	57 000373	2 300		•	DATE 0/0575	PAGE	2 2
			ROUTE STRUCTURE COMPARISONS	RE COMPARISONS				
		ROUTES	ROUTE STRUCTORE HENEFILL GROUNDS WIND AND TOTAL	GROUND WIND AN	ם ומדאר			
			AIRPORT PAIR 121	TOTAL SEA JFK				
7403 (160 1	UAL	NAPEC	FAA	SCI/FAA NAFEC	×08	PRE-PLANNED	807	
IAI RNAV	(000) 00 · · · · · · · · · · · · · · · ·	3.0 (.15) +1 (01) = 2.7 (.15)	1.4 (07)	1.0 (.07)	30.4 (1.52) +-12.1 (61) = 18.3 (92)	.0 (.00) + -2.5 (13) # -2.5 (13)	30.4	1.52
SCIZNAFEC	-3.0 (15) + .1 (.01) E -2.9 (15)	(000.) 0	+3 (08) = -2.0 (10)	+ -1.6 (08) +3 (02) # -2.0 (10)	+-12.0 (60) = 15.4 (.77)	+ -2.4 (12) = -5.4 (12)	27.4 (1.37)
TAANAAFEC	-1.4 (07) + -4 (02) =9 (05)	+ 1.6 (.08) + .3 (.02) = 2.9 (.10)	(000.	0000	+-11.7 (59) = 17.3 (.87)	+ -1.4 (07) + -2.0 (10) # -3.4 (17)	29.0 C	1.45)
SCTAFA/NAFEC	+ .4 (07) + .4 (02)	+ 1.6 (0.08) + 3 (0.02) + 0.5 = 0.23	00000	000	+-11.7 (59) = 17.3 (-87)	+ -2.0 (10)	29.0 C	1.45)
4/an/	-39.4 (-1.54) + 12.1 (-61) =-[8.5 (92)	-27.4 (-1.39) + 17.0 (.61) =-15.4 (79)	-29.0 (-1.47) + 11.7 (.59) =-17.3 (88)	+ 11.7 (.59)	666 666 666 666 666 666 666 666 666 66	-30.4 (-1.54) + 9.7 (.49) E-20.7 (-1.05)	0000	500
PRE-PLANNED	+ 2.5 (.09) = 2.5 (.13)	5.0 (.15) = 5.4 (.12)	+ 2.0 (.10) = 5.4 (.17)	+ 2.0 (.10) = 3.4 (.17)	30.4 (1.52) + -9.7 (48) # 20.7 (1.04)	000000000000000000000000000000000000000	30.0 C	1.523
7/89/0	-30.4 (-1.54) + 2.1 (.11) =-28.3 (-1.43)	+ 2.0 (-1.38) + 2.0 (.10) =-25.4 (-1.28)	-29.0 (-1.47) + 1.7 (.09) =-27.3 (-1.38)	-29.0 (-1.47) + 1.7 (.09) =-27.3 (-1.38)	+-10.0 (50)	-30.4 (-1.54) +3 (02) =-30.7 (-1.56)	• • •	666

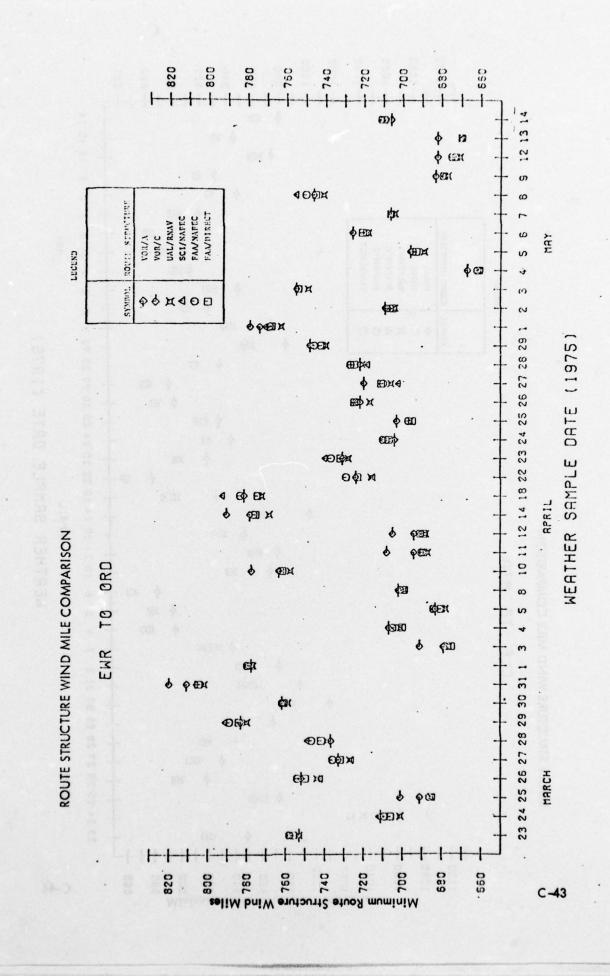


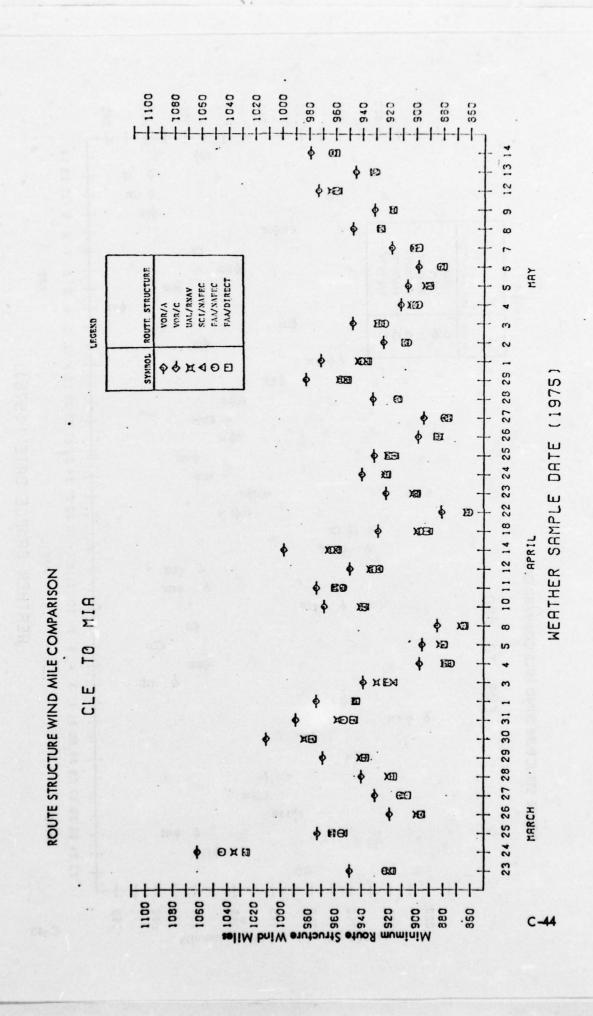


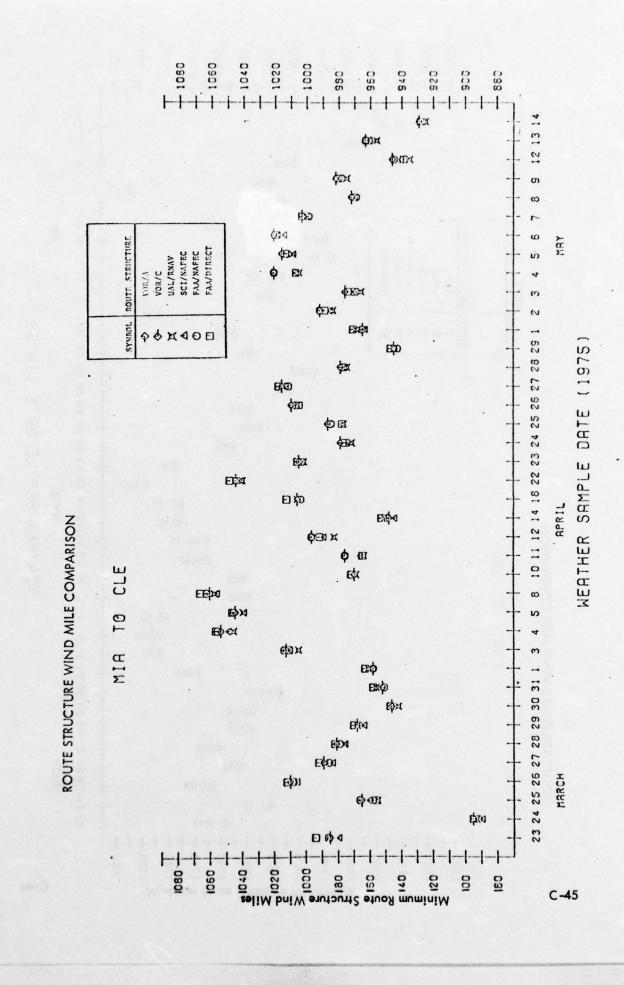


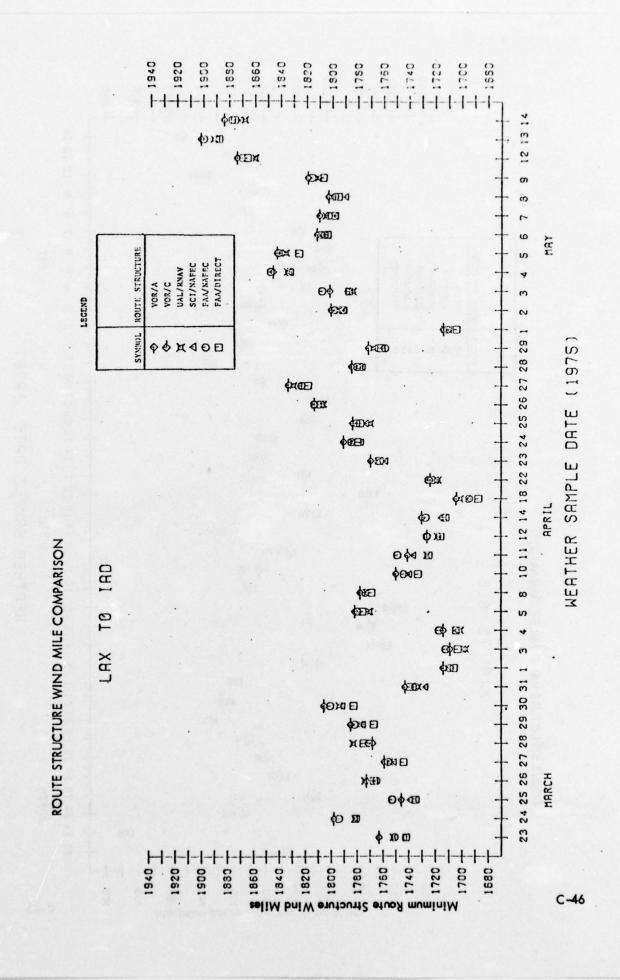


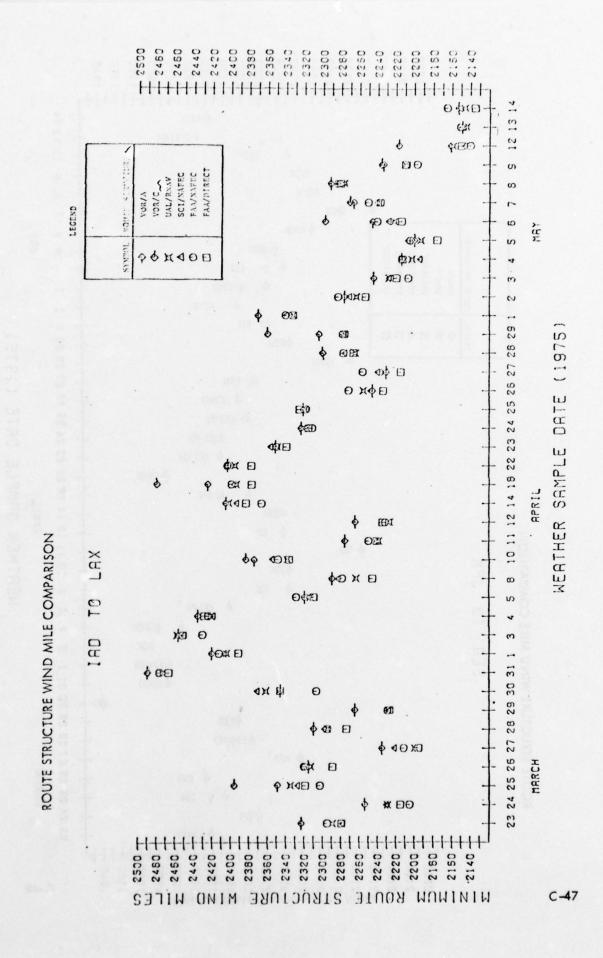


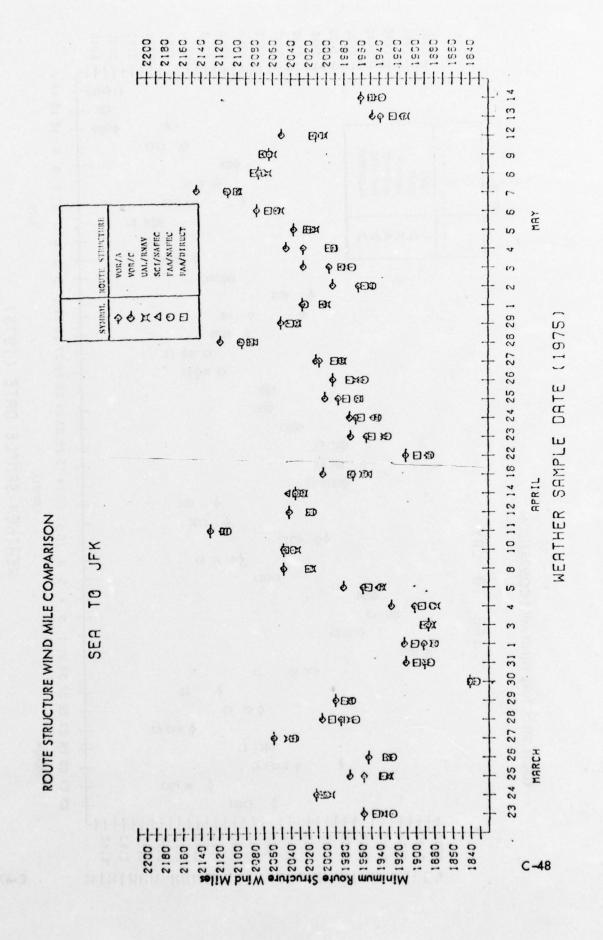


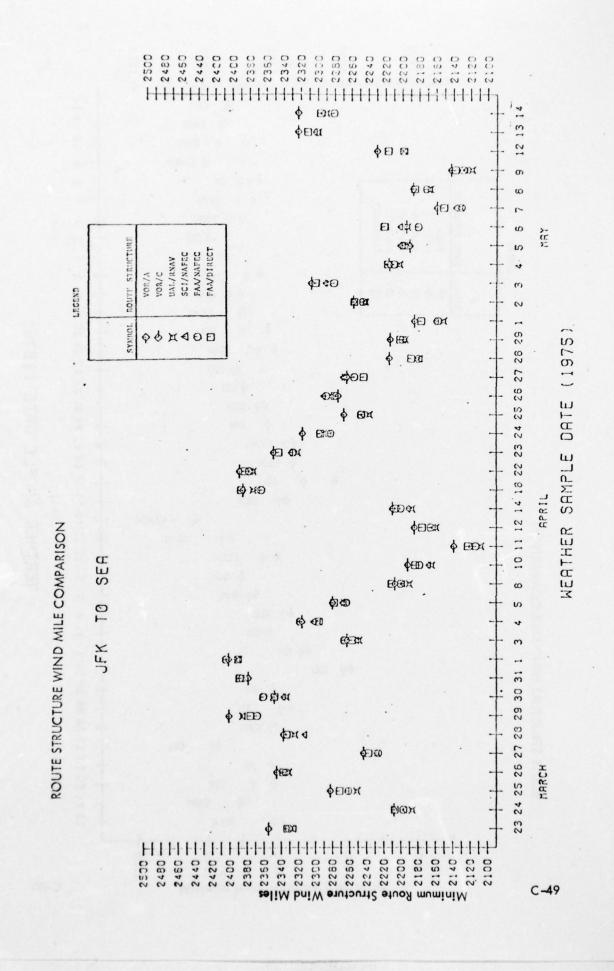


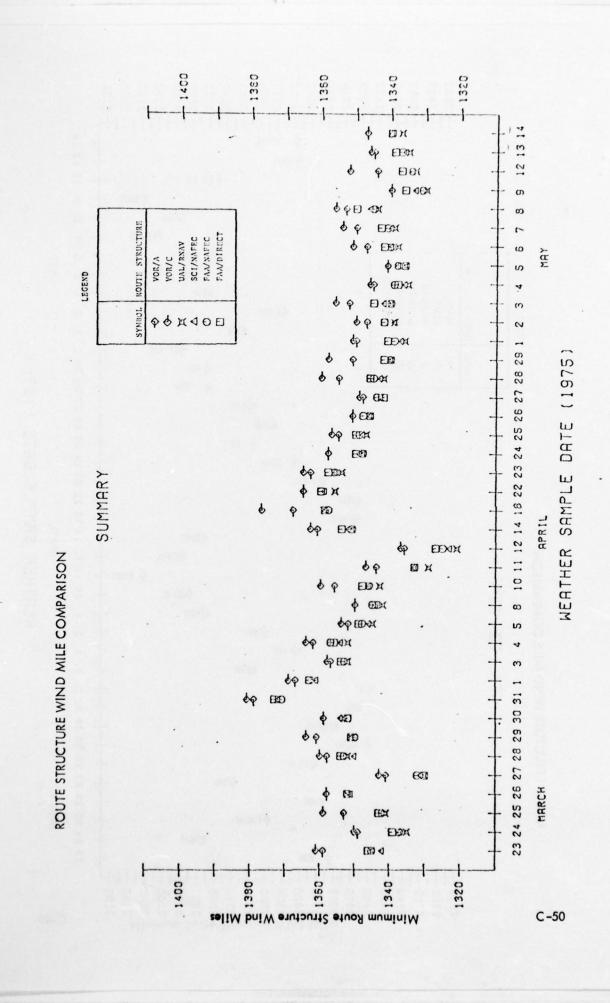












APPENDIX D

Development of Regression Model For Low Altitude Route Length Benefits

Implementation of a regression analysis, in its common form, simply involves the least squares fitting of a multidimensional plane to sets of independent variables. When only one independent variable exists, the end result is simply the fitting of a least squares straight line to estimate the linear relationship between the dependent and independent variables. In multivariable cases, the line is replaced by a linear model or hyperplane. The basic model is as follows:

$$Y - c + b_1X_1 + b_2X_2 + ... + b_mX_m$$

The variables χ_1,\ldots,χ_m are referred to as independent variables, while Y is the dependent variable. The relationship between the dependent variable and the independent variable is assumed to be as given above and the estimation of the constant c and the coefficients b₁,...,b_m is the objective of the regression analysis. The solution to the regression problem is the set of values \hat{c} , $\hat{b}_1,\ldots\hat{b}_m$ which minimizes the error sum of squares over a given sample. The error sum of squares is defined by

$$E = \sum_{i=1}^{n} (y_i - \hat{y}_i)^2$$

where

$$\hat{y}_i = \hat{c} + \hat{b}_1 X_{1i} + \hat{b}_{2i} + ... + \hat{b}_m X_{mi}$$

and the i subscript denotes the ith sample.

The solution of the coefficient values follows readily from matrix theory. The reader is referred to Reference 64 for any of the relevant details. Regression analysis need by no means be confined to the consideration of linear models. The most expedient way to address monlinear models is to include additional independent variables which are nonlinear combinations of the original independent variables. A quadratic equation can be fitted simply by including a second set of independent variables which are the squares of the first set.

In general, many regression models can be used to characterize a given set of data and some means to establish their relative merit is necessary. A common and favored approach is to compare the regression error sum of squares to the sum of squares of the deviations from the mean. By definition, the sums of squares of the dependent variables about their mean is the minimum over the sums of the squares about any other constant. Assuming that one is concerned with minimizing the expected square of this estimation error, the comparison of the two error sums of squares, therefore appropriately measures the ability of the regression model to reduce the error. The following definitions are useful:

Let

$$E_{B} = \sum_{i=1}^{n} (y_{i} - \overline{y})^{2}$$

$$E_{A} = \sum_{i=1}^{n} (y_{i} - \hat{y}_{i})^{2}$$

What is referred to as the "total regression coefficient" can now be written as:

$$R = \sqrt{\frac{E_B - E_A}{E_B}}$$

This measures the proportion of the variance, or error sums of squares, which is eliminated through the use of the regression model. The term, R, is also algebraically identical (except for the sign) to the correlation coefficient between the dependent variable and the resulting regression estimates.

Finally, if there are as many variables as samples, the error sums of squares can invariably be reduced to zero. This occurs in the same manner that a line can always be fit through two points, a plane through three points, etc. A total regression coefficient of units would result, which should obviously not be construed to imply perfect validity of the regression model. In a similar manner, the inclusion of an additional variable will result in an error sum of squares never greater than the previous amount. Thus, the addition of a set of random numbers as another independent variable produces a better regression fit and it is the responsibility of the analyst to understand that the regression model has in fact been degraded rather than improved. Fortunately, there are a variety of statistical tests designed to distinguish between a random improvement due to a statistically justifiable correlation. The technique applied in this study is referred to as a "stepwise selection procedure". This procedure essentially accepts or rejects independent variables from the model depending upon whether or not they reduce the error sums of squares by an amount significantly greater than that which would be expected by randomness alone. The specific significance level used on a given run must be stipulated by the analyst.

The twelve independent variables described in Section 3.3., in combination with the route length results of the design effort, was the fundamental data base used in this regression analysis. It is apparent that there are a great many potential regression models which can be derived from these data. In order to converge upon a satisfactory model in an expeditious and cost-effective manner, the regression attempts were designed to achieve the following;

- Determine the most suitable dependent variable and the independent variables which contribute most significantly to the regression fit.
- 2) Determine the improvement in the regression fit of nonlinear variations of the significant variables.
- 3) Compare the resulting models and select that which is the most appropriate.

Within the first set of regression attempts, three dependent variables were analyzed; the RNAV Absolute Mileage Benefit, the Percent Benefit and the Flight Mile Benefit. Many regression fits were made for each dependent variable. The runs of particular importance were those wherein all independent variables were forced into the model and those where very high levels of significance were used to select the variables. Forcing all variables into the model provides the best possible linear regression fit, while the stepwise approach provides the regression fit which can be statistically justified. It was found that the level of significance of the statistical tests did not play a major role in the model results. Variables which were correlated to the dependent variable at all were highly correlated. Statistical tests using lower levels of significance did not often result in the inclusion of additional dependent variables. Unless otherwise stated, the significance levels used were .05/.10. This implies that no variable was entered unless its improvement to the model was significant at the 95% level and included variables were rejected whenever the inclusion of more important variables caused their incremental improvement to drop below the 90% significance level.

The results of this first set of runs are summarized by a subset of runs, specifically I through 9 of Table D.I. From these results, it was apparent that the dependent variable, RNAV Absolute Benefit, would provide the best results. The Flight Mile Savings were virtually useless, which basically indicates a lack of correlation between the exchange rate and either the suitability of the VOR route or the RNAV benefit. The Percent Benefit was eliminated from further consideration since its results were poorer than those produced when using the Absolute Benefit, even at lower significance levels.

The stepwise procedure for the Absolute RNAV Benefit resulted in the inclusion of only three variables; the airports in the terminal areas, the airports overflown and the great circle distance. At no significance level down to and including .40/.50 was any other variable entered.

The second set of regression attempts were designed to determine whether or not certain nonlinear variations would provide better results. The variables previously included were the only ones upon which nonlinear variations were used. Specifically, the squares and square roots of each of these variables were added to the data set, in separate runs. In each case, a marked improvement was obtained culminating in Runs 10 and 11 of Table D.1, both of which produced a regression coefficient of .49.

				_												_
Total Regression	Coefficient			75.	0	12.	х.	.45	₽.	.48	3 .	94.	.49	. 49	64.	
Signi- ficance	Level			4	01./50.	.10/.20	.40/.50	u.	01./50.		.05/.10	.10/.20	01./50.	01./50.	.05/.10	
orts Area	Greater No.														0	
Dep. From Airports Within Terminal Area	Lesser No.										ola Maria				0	
Dep. 1 Within	Total			•	0	0	1	-	0	1	0	0	0	0		
13	iter	$A_{\rm L}^2$													0	
Number of Airports Within Terminal Area	Greater No.	A _L													-	
ring	Total Lesser No.	A A As													0	_
of A	No.	As													-	
1.5		4								i e i		131		-		
ith	ota	A A2) (a	٦	110		
		4		_	0	7	-	-	٦	-	7	-	7	0		
AP Pair	Exchange			1	0	1	1	1	0	1	0	0	0	0	0	
Intersections	Fits.			٦	0	0	0	-	0	-	0	0	0	0	0	
Interse	AP Pairs			•	0	0	0	1	0	1	0	0	0	0	0	
Great Circle	Stance C2 16				9 11						UIT			-		
Great	Sta G2												0		0	
80	0 0			-	0	0	0	-	0	-	-	-	-	0	-	
Airports Over- flown		O Dep. From	500	-	0	0	0		0	7	0	0	0	0		
ts		\$							1111-					0		
i odr		0,5		_									0		0	_
Ai	No.	0		-	0	0	0	-	-	-	0	-	-	-		
Dep. From	Both Ter-			•	0	0	1	1	0	1	0	0	0	0	0	
Dep.			,		ti.	u.	ш.	۵	۵.	4	4	4	ч	<	<	
Run 75				-	7	2	•	S	9	1	60	6	10	==	13	

Significance level of F implies that the variables were forced into the model; significant testing was not done. Zero entries imply that these variables were available for selection but were not included at the stated significance level. One entry implies that those variables were utilized in the resulting regression

equations.
Dependent Variables: F = flight mile benefit; P = percent benefit; A = absolute
 route mile benefit

The last regression model which was explored evolved by treating the airports (and departures) of each terminal area as separate characteristics rather than using only their respective sums. These data were ordered so that the area with the fewer airports was listed as one independent variable; the area with the greater number listed as the other. This allowed the regression model to make a distinction between whether or not the airports were evenly distributed between the two terminal areas. The results were highly useful in that they produced a regression coefficient of .49 (at a .05/.10 significance level) and as shown by Run 12 of Table D.1, which included the following variables:

- 1) The number of airports in the less dense* terminal area
- 2) The number of airports in the more dense area
- 3) The number of airports overflown
- 4) The great circle distance

Also of interest was the fact that the nonlinear variations which previously provided substantial improvements no longer did so. Even though available, they were not entered into the model at any significance level.

This model (Run 12) was favored over the two nonlinear models previously mentioned (Runs 10 and 11), even though the regression coefficients were the same. The linearity of the latter model provided more intuitive appeal. Equal results were obtained at equal levels of significance, thereby precluding further mathematical or statistical means as a method of defining the preferred regression model.

Thus, the final regression model (12), was selected for subsequent use in estimating, on a national scale, the potential RNAV low altitude benefits. Its parameter values are as follows:

$$\hat{R}_B = C + B_1 \cdot A_S + B_2 \cdot A_L + B_3 \cdot \theta + B_4 \cdot G$$

where

 A_S , A_L : number of airports in the terminal areas, 25 nm radius $(A_S \le A_L)$

 number of airports overflown (within 25 mi of the great circle arc)

G : great circle distance

 \hat{R}_{R} : estimated RNAV absolute mileage benefit

^{*}Less dense implying fewer airports

and C = 4.058 $B_1 = -1.455$ $B_2 = -0.285$ $B_3 = -0.428$ $B_4 = 0.031$

An assessment of this model reveals that the "average" RNAV route in the California design produced a route length benefit of 3.96 nm. The regression results imply that RNAV produces a constant 4.06 nm benefit component which, being constant over all routes, would be attributed to an improved RNAV terminal/ enroute interface capability. RNAV also produces a maximum route length benefit component of 3%, relative to a "worst case" VOR route (barring extremes), of the route length (great circle distance). This RNAV benefit is reduced as when the actual VOR routes are shorter than what was referred to as worst case. Specifically, when many airports exist in either of the terminal areas and/or when many airports are overflown, the model indicates that better than average VOR routes tend to exist. Initially, these airport-oriented characteristics were included so that degradation of the RNAV routes could be explained within the model. In retrospect, the predominant factor is the proliferation of VOR stations, and consequently the opportunity to design better than average VOR routes in areas where many terminals exist. Degradation of the RNAV routes is only a secondary effect which occurs only when the VOR structure is so complex that RNAV routes must be designed coincident with the VOR. The model produces large RNAV benefits in desolate areas and small benefits in high density areas, which is consistent not only with the results of the California design but also with what is intuitively expected elsewhere.

While the regression results are reasonable and interpretable, the total regression coefficient was lower than desired. In order to estimate an ultimate upper bound on the regression coefficient, and thereby judge the success of this effort, a final regression attempt was made wherein the actual VOR route lengths were used as an additional independent variable. The only unexplainable variation remaining for the regression model to address was, therefore, the RNAV route lengths. The resulting total regression coefficient of .86, lower than expected, can therefore be considered an upper bound. Computation of the VOR route length for all 8,000 airport pairs was logically not a practical approach. Since no set of independent variables would ever be expected to provide a perfect estimate of the VOR route lengths, the inclusion of more sophisticated variables would probably not have resulted a regression coefficient greater than about 0.7. The achieved regression coefficient of .49 is therefore not too far below that which could be attained even if computational limitations did not exist. Further, the four pertinent variables were included at a very high significance level.

The relationship between the RNAV benefits and the airport pair characteristics used with the model are statistically justified for routes within California. There is no reason to believe that the general validity of the model is not adequate when applied to other regions of the country.

APPENDIX E

SLANT RANGE ERROR FLIGHT TEST PROGRAM

Introduction

This appendix presents the test plan and resulting aircraft track plots (Figures E.5 through E.13) of the RNAV slant range error effect flight test program. The objective of the slant range error flight test is to determine what effects uncompensated slant range error has on achieved flight paths and airspace requirements in terminal area operations. In order to make this determination, four profiles at two altitudes have been designated for test purposes. The primary criteria defining these paths were to locate the point of closest approach to the VORTAC station either on a line of 45° elevation or 60° elevation with respect to the horizon, and to make the flight paths long enough to allow stabilization of the aircraft on the nominal flight path before entering the region where the slant range effect is felt. The altitudes selected are 8000 feet and 12000 feet. The lower level was selected because previous experience (Baseline Flight Test) [7] indicates that the effect is unnoticeable at lower altitudes. The higher altitude was selected as being a reasonable boundary on terminal area operations, particularly for GA aircraft which would typically have lower-capability RNAV equipment.

Test Plan

Because of convenience, the Miami ARTS III TRACON was selected for the flight test site. There are two VORTACs in the Miami area, Miami and Biscayne Bay. Miami VORTAC is surrounded by busy airspace, and so was discarded. Biscayne Bay, several miles South of the airport, is relatively free of traffic. An East-West course was selected since very little interfering traffic exists at the selected altitudes. The flight test profiles are defined in Table E.1, and depicted graphically in the attached navigation charts (Figures E.1 through E.4).

In order to bound the expected slant range effect, two subject pilots were selected to fly the profiles. The first was not advised of the purpose of the test, but was instructed to fly as he normally would. The second subject was aware of the purpose of the test and followed the RNAV guidance signal as closely as possible within reason considering normal maneuver restrictions and passenger comfort limitations. In this manner, the slant range effect to be expected given any typical pilot was included between the bounds defined by the two subject pilots. Each pilot flew each of the four profiles twice in order to provide adequate data.

The navigation signals were instrumented using a strip chart recorder in much the same manner as in the Denver and Miami flight tests ("Initial RNAV Operational Test Plan, Miami Area", June 12, 1974, Champlain Technology, Inc.). In addition to cross track deviation, distance to waypoint, time and transponder IDENT signal, three additional signals (DME range, TO/FROM flag and CDI Valid flag) were recorded. In particular, the DME signal was important since this was a slant range error test. The CDI flag was used for determining the extent of the zone of confusion around the VORTAC.

Data recovery and processing was conducted in the same manner as in the earlier flight test programs. Modifications to the merge and statistical analysis programs were required in order to properly process the DME measurement.

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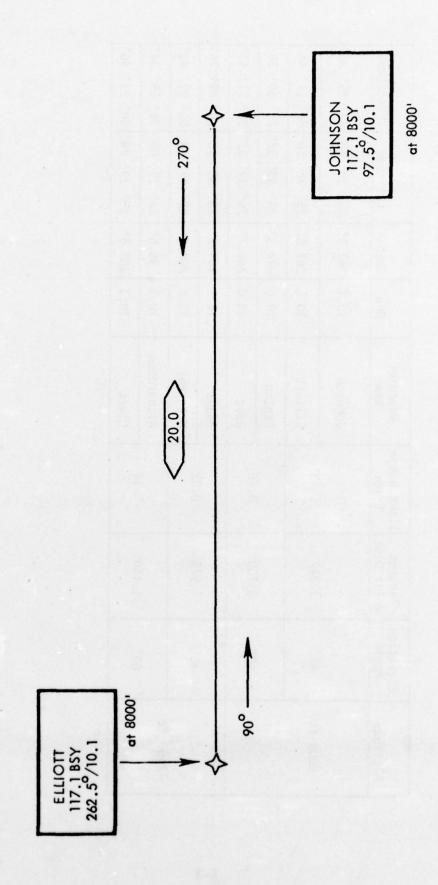
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Table E.1 Flight Profile Data

	Elevation	Tangent	Slant Range	Wavpoint					-		
Altitude	Angle	Point Dist	Error	Name	Rho	Theta	Lat	Latitude	_	Longi tude	rude
. 000	910	1.6.1	i.	Johnson	10.1	97.5° 25° 38' 58" 79° 59' 36"	25°	38, 28	79	. 59	36"
31 0000	ç	1.31/	0.33	Elliott	10.1	10.1 262.5° 25° 38' 58" 80° 21' 44"	25°	38 58	8	. 21	44"
	000	031.0	0	Pancho	10.0	10.0 94.3° 25° 39' 32" 79° 59' 38"	25°	39' 32	179	. 59	38"
	6	0.760	0.76	Bolz	10.0	10.0 265.7° 25° 39' 32" 80° 21' 41"	25°	39' 32	80	21,	41"
	45°	1 975		Adams	10.2	10.2 101.2° 25° 38' 18" 79° 59' 36"	25°	38, 18	. 13	. 29	36"
130001	2	6.6.	70.05	McConkey	10.2	10.2 258.8°	25°	25° 18' 18" 80° 21' 43"	80	, 21,	43"
11 00071	ê0°	1 140	77	Richardson	10.1	96.5° 25° 39' 08" 79° 59' 34"	25° 3	80 , 68	. 79	. 69	34"
	3	2	<u>.</u>	Clark	10.1	10.1 263.5° 25° 39' 08" 80° 21' 46"	25° 3	80 ,68	80	, 12 .	46"

Figure E. 1 TERMINAL RNAV PROCEDURE TEST



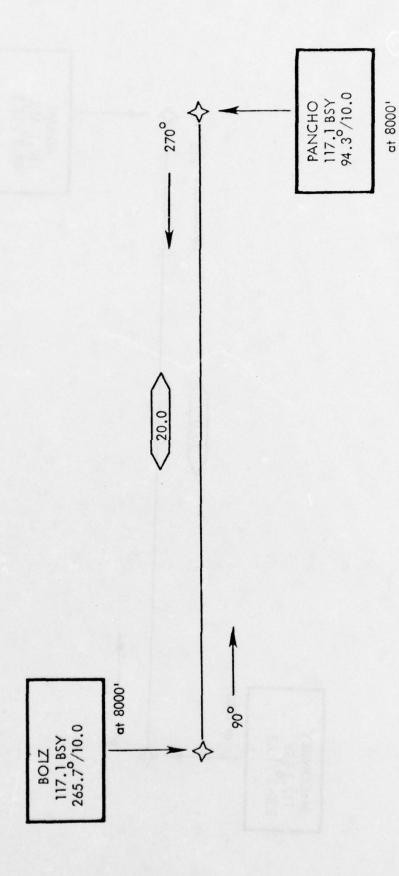


Figure E.3 TERMINAL RNAV PROCEDURE TEST

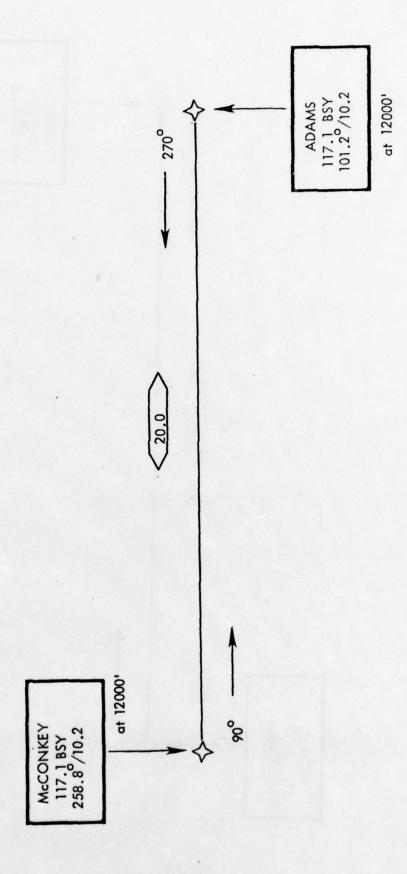
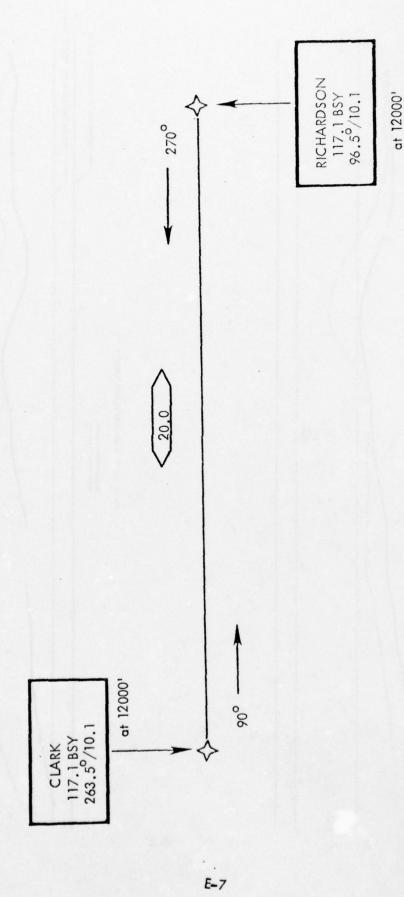


Figure E.4

TERMINAL RNAV PROCEDURE TEST



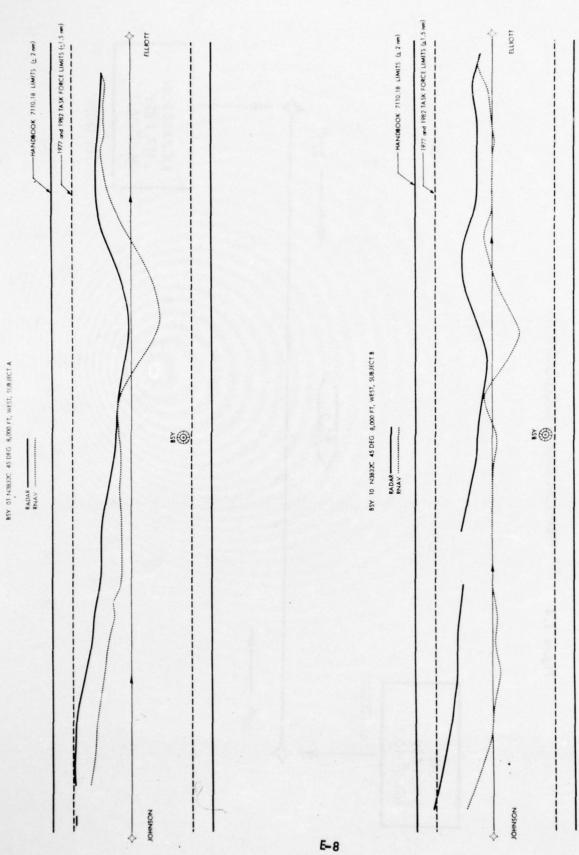


Figure E .5 SLANT RANGE ERROR FLIGHT TEST RESULTS

1977 and 1982 TASK FORCE LIMITS (£1.5 mm) 85Y 02 N3832C 45 DEG 8,000 FT, EAST, SUBJECT A (C)

FLLIOTT

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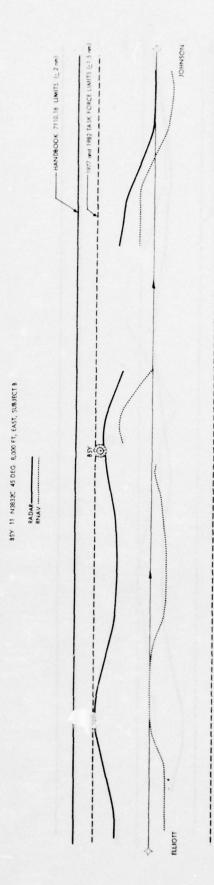
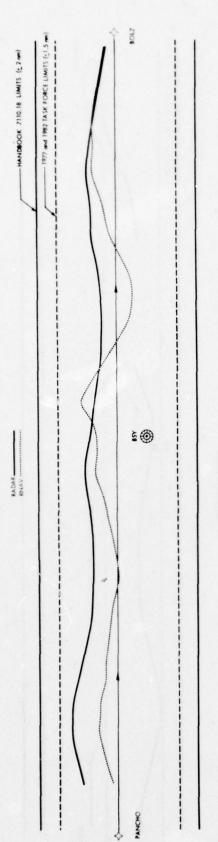
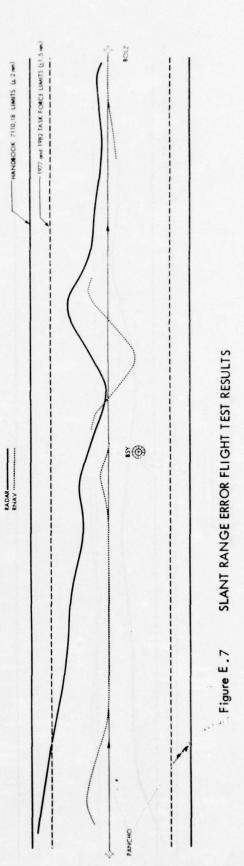


Figure E.6 SLANT RANGE ERROR FLIGHT TEST RESULTS

894 03 N383XC 60 DEG 8,000 FT, WEST, SUBJECT A



85Y 12 N3832C 60 DEG 8,000 FT, WEST, SUBJECT 8



SLANT RANGE ERROR FLIGHT TEST RESULTS

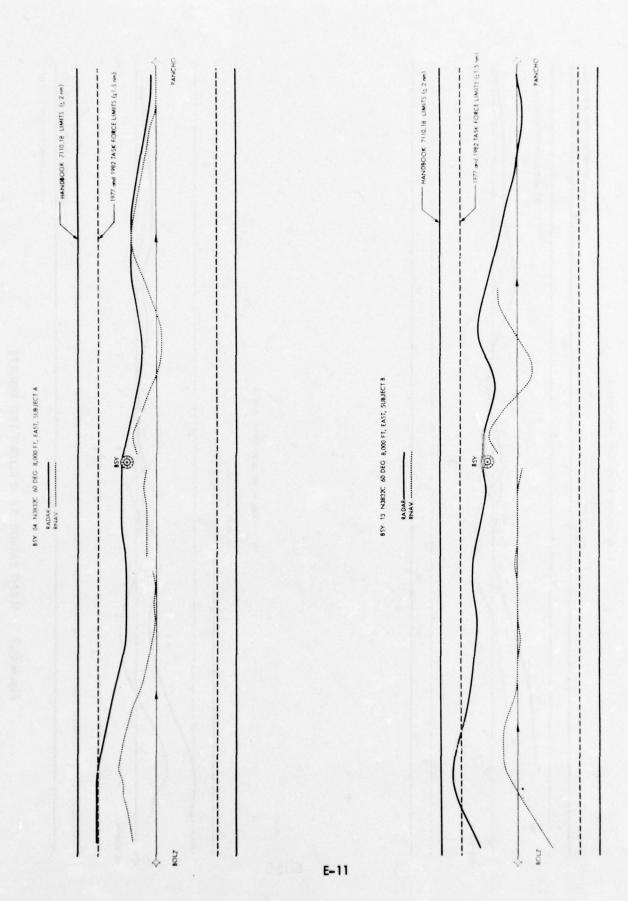


Figure E.8 SLANT RANGE ERROR FLIGHT TEST RESULTS

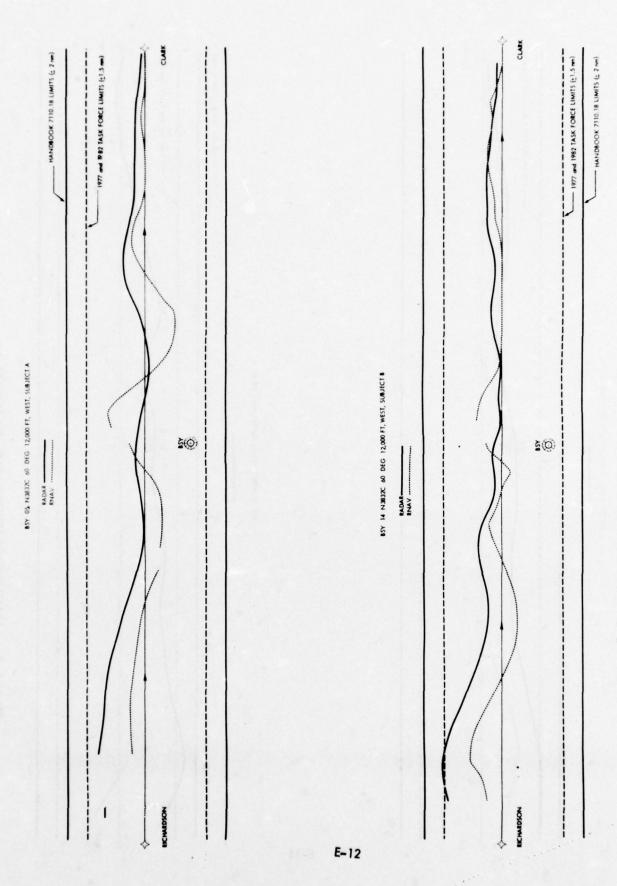


Figure E .9 SLANT RANGE ERROR FLIGHT TEST RESULTS

RICHARDSON 1977 and 1982 TASK FORCE LIMITS (£1.5 mm) HANDBOOK 7110,18 LIMITS (± 2 nm) 85Y 06 N3832C 60 DEG 12,000 FT, EAST, SUBJECT A RADAR RNAV 3.0 CLARK

RICHARDSON -- 1977 and 1982 TASK FORCE LIMITS (£1.5 nm) 85Y 15 N3832C 60 DEG 12,000 FT, EAST, SUBJECT 8 RADAR CLARK

Figure E.10 SLANT RANGE ERROR FLIGHT TEST RESULTS

- HANDBOOK 7110.18 LIMITS (£ 2 mm)

-1977 and 1982 TASK FORCE LIMITS (±1.5 mm) -HANDBOOK 7110.18 LIMITS (±2 m) 85Y 07 N3832C 45 DEG 12,000 FT, WEST, SUBJECT A RNAV

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85Y 16 N3832C 45 DEG 12,000 FT, WEST, SUBJECT 8 RNAV

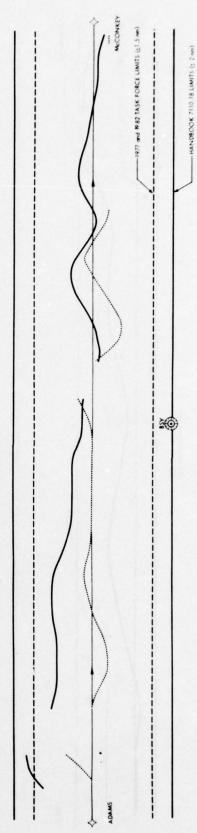
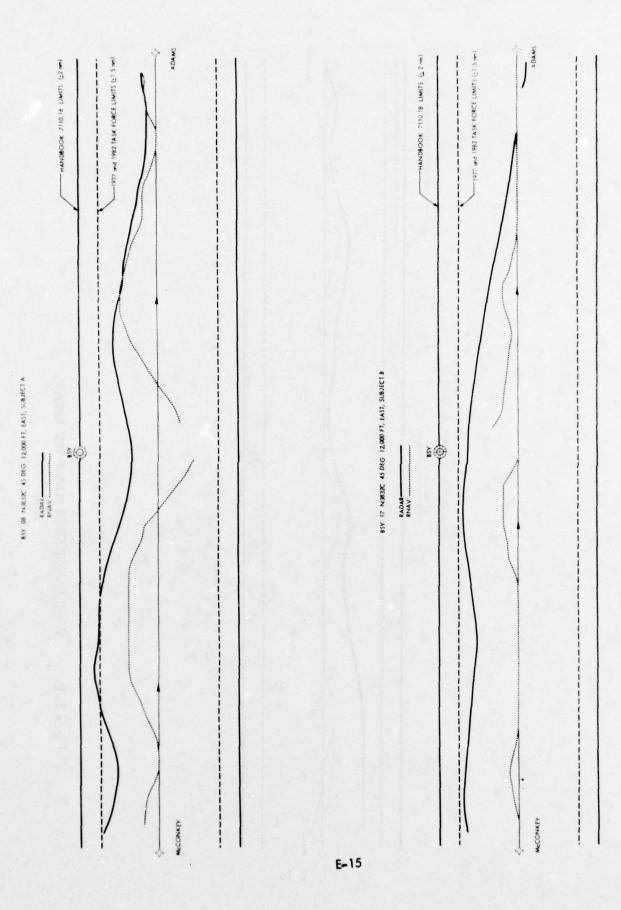
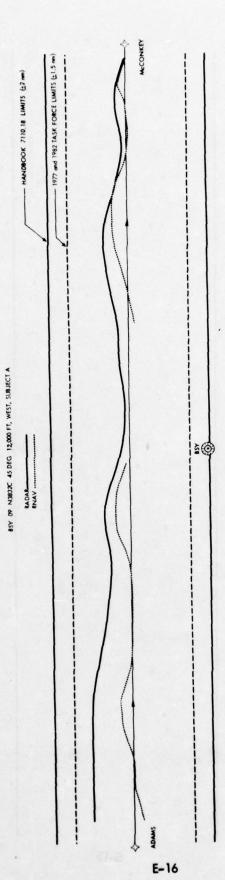


Figure E. 11 SLANT RANGE ERROR FLIGHT TEST RESULTS



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Figure E.12 SLANT RANGE ERROR FLIGHT TEST RESULTS



3 SLANT RANGE ERROR FLIGHT TEST RESULTS

Figure E.13

Appendix F

High Altitude VORTAC Requirements

Implementation Options and

Cost Sensitivities

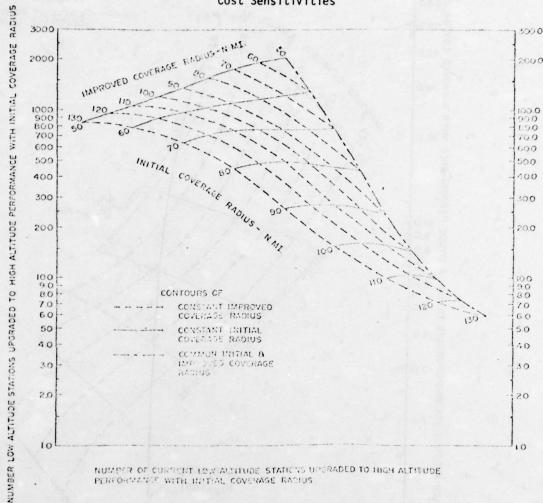


Figure F.1 Ground NAVAID System Implementation Cost
To Provide Full CONUS Coverage

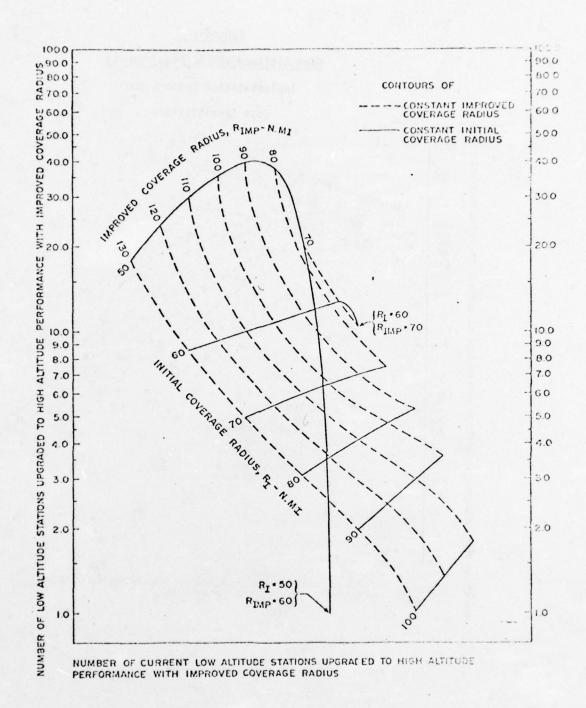
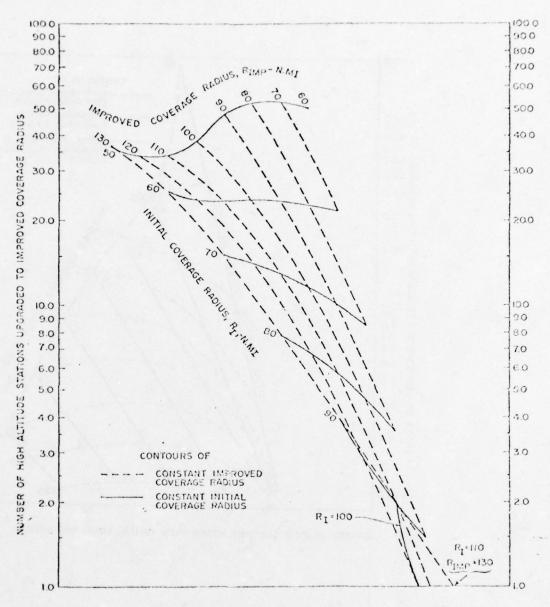
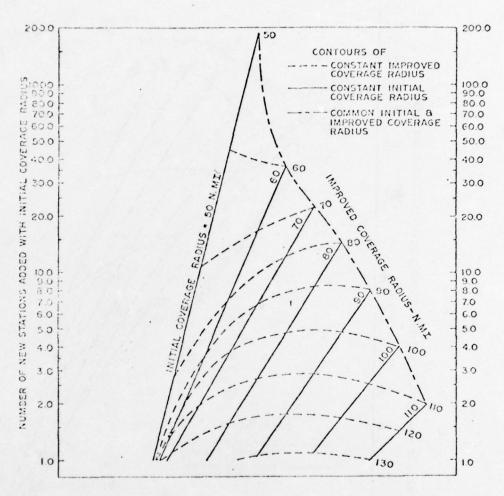


Figure F.2 Ground NAVAID System Implementation Costs
To Provide Full CONUS Coverage



NUMBER OF CURRENT HIGH ALTITUDE STATIONS UPGRADED TO IMPROVED COVERAGE RADIUS

Figure F.3 Ground NAVAID System Implementation Costs
To Provide Full CONUS Coverage



NUMBER OF NEW STATIONS ADDED WITH INITIAL COVERAGE RADIUS

Figure F.4 Ground NAVAID System Implementation Costs
To Provide Full CONUS Coverage

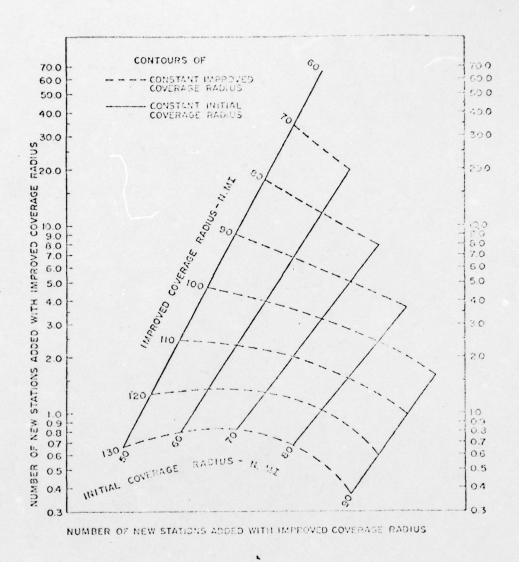


Figure F.5 Ground NAVAID System Implementation Cost
To Provide Full CONUS Coverage

APPENDIX G

ENROUTE RNAV BENEFITS ANALYSIS (Six Airlines)

This appendix consists of listings of RNAV benefits available to each of six airlines, based on their 1 February 1976 schedule and operating frequency. Tables G.1 through G.6 give the time and fuel benefit for each airport pair in the current airline structure. Tables G.7 through G.12 list 2D and 3D annual and per aircraft benefits in fuel, time, and dollars for each aircraft type. Tables G.13 through G.18 list projected 4D benefits.

TABLE G.1 NATIONAL AIRLINES ENROUTE RNAV BENEFITS ANALYSIS

	DST A/P	FLT	RANGE NMI	RNAV BENE	A/C	F	CRUISE ALT	TIME BENE	FUEL BENE
ROS	JFK	415	76.0	17.00	725	7	23880.	2.20	53.83
902	DEA	437	76.0	17.00	725	7	23880.	2.20	53.83
CHS	DCA	420	294.0	3.00	727	7	35000.	.39	6.87
UCV	CHS	421	249.0	-1.00	727	7	35000.	13	-2.29
DCA	CHS	423	289.0		727	7	35000.	13	-2.29
HCA	JAX	73	100.0		725	7	35000.	13	17.50
DCA	JFK	410	100.0	-1.00	727		25800.	13	-2.79
DCA	JFK	426	100.0		727	7	25800.	13	-2.79
DCA	JFK	496	100.0	-1.00	727	7	25800.	13	-2.79
DCA	MIA	101	718.0	5.00	727	7	35000.	.65	11.45
	MIA	109	718.0 718.0	5.00	121	7	35000.	.65	11.45
	FLL	123	868.0	7.00	725		35000.	.92	17.50
C M.D	XAL	121	621.0		725		35000.	.40	7.50
EAR	HIA	603	856.0	7.00	010	7	41000.	.92	29.20
FLL	EWR	124	874.0	7.00	725	7	35000.	.92	17.50
FLL	JFK	14	864.0		725	7	35000.	.66	12.50
JAY	DCA	605	462.0		727	7	35000. 35000.	.65	16.03
JAX	DCA	+28	452.0	7.00	721		35000.	. 41	16.03
JAX	EWM	70	626.0	4.00	725	7	35000.	.53	10.00
IAX	MIA	175	203.0	0.00	727	7	32360.	0.00	0.00
JAX	MIA	411	203.0	0.00	725	7	32360.	0.00	0.00
	605	128	100.0		725	7	26280.	6.37	150.04
IE W	DCA	403	103.0	2.00	725		26040.	.25	5.05
150	DCA	407	103.0	2.00	121	7	26040.	. 25	5.54
JEK	DCA	421	103.0	2.00	727 725	7	26040.	.25	6.05
JFK	DCA	493	103.0		725	7	25040.	.25	6.05
JEK	FLL	?	888.0	7.00	727	7	35000.	.91	16.03
JFK	FLL	439	626.0	7.00	725	7	35000.	. 42	17.50
JFK	MIA	81	874.0	0.00			41000.	0.00	0.00
Jex	FIA	97	874.0	0.00	725	7	35000.	0.00	0.00
JFK	MIA	601	759.0	0.00	727	7	41000. 35000.	.78	0.00
IFK	TOA	433	789.0	5.00	727	7	35000.	.78	13.74
	TPA	437	789.0	6.00		7	35000.	.79	15.00
LAS	LAX	21	293.0	2.00	725	7	35000.	0.00	0.00
LAS	SFU	27	293.0	0.00	010	7	36120.	0.00	0.00
	HIA	10	103.0	3.00	010	7	26040.	2.91	95.95
	HIA	50	1937.0			7	41000.	2.91	95.95
	MIA	54	1937.0	21.00	010	1	41000.	2.91	95.95
LAX	HIA	69	1937.0		010	7	41000.	1.17	95.95
LAX	TPA	36	-6.0	0.00	010	7	12600.	0.00	0.00
	MIA	91	865.0	7.00	125	7	35000.	.92	17.50
	HIA	611		7.00	727	7	35000.	.92	17.50
LGA	169	93	803.0	7.00	010	7	41000.	.89	29.20
	DCA	102	747.0	6.00	727	7	35000.	.78	13.74
	DCA	108	741.0	6.00	727	7	35000.	.78	13.74
#1A	EWH	8	884.0	7.00	010	7	41000.	.89	29.20
	ENH	126	889.0	1.00	725		35000.	.92	17.50
HIA	JFK	4	841.0	8.00	125	7	35000.	1.06	20.00
	JFK	600	861.0				41000.	1.01	33.37
	LAX	41	1937.0				41000.	1.65	33.37
MIA	LAX	43	1937.0	13.30	010	7	41000.	1.65	54.23
	LAA	53	1937.0	13.00			35000.	1.65	16.03
	LGA	90	892.0	1.00			35000.	.92	17.50
HIA	LGA	608	842.0	7.00	725	7	35000.	.42	17.50
AIM		25	509.0	2.00	725		35000. 41606.	.26	8.34
AIP	MSY	45	509.0	2.00	010	7	41000.	.25	A.34
HIA	SFU		5500.0	26.00	010	7	41000.	3.29	108.40
	TPA	21	2200.0	-4.00	010	7	25000.	3.29	108.46
	TPA	23	90.0	-4.00	010	7	25000.	50	-22.87

ORG	DST	FLT	RANGE	RNAV	A/C	F	CRUISE	TIME	FUEL
A/P	A/P	NO	NMI	BENE			ALT	BENE	BENE
-	IPA	475	90.0	-4.00	725	,	25000.	51	-12.49
MSY	LAA	31	1354.0	13.00	010	7	41000.	1.65	54.23
MSY	LAK	37	1354.0	13.00	D10	7	41000.	1.65	54.23
HSY	LAA	125	1354.0	13.00	010	7	41000.	1.65	54.23
MSY	HIA	26	505.0	7.00	725	7	35000.	.92	17.50
MSY	HIA	45	505.0	7.00	010	7	41000.	.89	29.20
MST	HIA	186	505.0	7.00	125	7	35000.	.92	17.50
MSY	TPA	22	354.0	0.00	010	7	38248.	0.00	0.00
MSY	TPA	34	354.0	0.00	010	7	38248.	0.00	0.00
MSY	TPA	64	354.0	0.00	D10	7	38248.	0.00	0.00
MSY	TPA	162	354.0	0.00	725	7	35000.	0.00	0.00
ORF	JFK	463	150.0	-5.00	127	7	30067.	64	-12.50
PBI	JFK	184	877.0	6.00	725	7	35000.	1.06	20.00
PAI	JFK	188	8/7.0	4.00	725	7	35000.	1.06	20.00
Pal	LGA	94	823.0	7.00	010	7	41000.	.89	29.20
SFO	LAS	24	277.0	24.00	725	7	35000.	3.43	72.51
SFO	LAS	44	275.0	27.00	010	7	35520.	3.65	13H.06
TPA	JFK	64	925.0	8.00	010	7	41000.	1.01	33.37
TPA	JFK	436	924.0	4.00	727	7	35000.	1.04	18.32
TPA	MIA	10	103.0	10.00	010	7	26040.	2.21	101.44
TPA	HIA	55	103.0	18.00	010	7	26040.	2.21	101.44
TPA	HIA	32	103.0	10.00	725	7	26040.	2.29	54.43
TPA	HIA	36	163.0	15.00	010	7	25040.	2.21	101.44
TPA	HIA	44	103.0	18.00	010	7	26040.	15.5	101.44
TPA	HIA	403	103.0	18.00	125	7	26040.	2.29	54.43
TPA	MIA	437	103.0	18.00	127	7	26040.	2.27	49.84
400000	HSY	23	353.0	0.00	010	7	38216.	0.00	0.00
TPA	MSY	37	353.0	0.00	010	7	38216.	0.00	0.00
	MSY	125	353.0	0.00	010	7	38216.	0.00	0.00
			3,3,0	00			300100		3.00

TABLE G.2 EASTERN AIRLINES ENROUTE RNAV BENEFITS ANALYSIS

ORG	DST	FLT	RANGE	RNAV	A/C	F	CRUISE	TIME	FUEL	ORG	DST	FLT	RANGE	RNAV	A/C	F	CRUISE	TIME	FUEL
A/P	A/P	NO	NMI	BENE			ALT	BENE	BENE	A/P		NO	NMI	BENE	A/C		ALT	BLNE	BENE
ATL	HAI	124	14H 0	1- 20	nos	7	34480.	1.89	23.45	. 71	080	454	626 0	-2.00	1.10	7	40400.	25	-9.07
ATL		132					344HQ.	1.89	23.45	ATL		161	389.0	-34.01	725	7	35000.		-57.51
AIL		134					34480.	1.89	23.45	ATL		245					35000.		-89.29
ATL		136					344HD.	1.89	23.45	ATL		329					35000.		-97.51
ATL		424					35000.	1.03	32.05	ATL		114						1.70	21.53
ATL		434	388.0	14.00	125	6	35000.	1.85	35.00	ATL	PHL	120	506.0	13.00	727	7	35000.	1.70	29.76
ATL		620	98.0				27640.	.13	3.06	ATL		122					35000.	1.76	21.58
ATL		524	98.0				25540.	.13	3.06	ATL		124					41300.	1.76	58.03
ATL		670	98.0	1.00	725	7	25640.	.13	3.06	ATL.	PHL	486	500.0	13.00	725	1	15000.	1.72	32.50
ATL		115	770.0				27640. 35000.	.13	1.95	ATL		320					35000.	.05	11.45
ATL		128	770.0				41000.	.78	26.78	ATL		988					35000.	.65	11.45
111		144	770.0	0.00	727	6	35000.	.78	13.74	AIL	SOF	252	196.0	-3.00	095	7	31667.	40	-5.30
ATL		534	770.0				35000.	.78	13.74	ATL		254					31667.	40	-5.30
ATL		330	535.0 535.0				35000.	.52	9.16	ATL		250					31667.	40	-5.30
ATL	BUF	962	535.0				41000.	.49	17.85	ATL	SOF	712					31667.	40	-5.30
ATL		355	116.0				27080.	1.01	21.43	ATL		98					30440.	3.59	139.76
AIL		344	116.0				29040.	1.05	15.01	ATL		270					35000.	3.40	72.51
ATL	CLT	360	116.0				27080.	1.02	23.41	ATL		274					34200.	3.40	48.82
ATL		476	115.0				27040.	1.02	23.41	ATL		276					34200.	3.70	48.62
ATL	CLT	130	350.0				27080.	1.02	23.41	ATL		289		29.00			35000.	3.78	27.50
ATL		145	356.0				35000.	.40	7.50	ATL		544					35000.	1.45	27.50
ATL		340	388.0	3.00	725	7	35000.	.40	7.50	ATL	TPA	365		11.00				1.45	27.50
ATL	DCA	135	388.0				35000.	.40	7.50	ATL		485					35000.	1.45	27.50
ATL		451	555.0				35000.	.13	1.57	ATL		127					35000.	1.45	27.50
ATL		523	555.0	1.00	727	7	35000.	.13	2.29	HAL		131	387.0	9.00	095	7	34470.	1.21	15.08
ATL		573	555.0				35000.	.14	1.66	BAL		147	387.0				34470.	1.21	15.08
ATL		106	555.0				35000. 35000.	.13	9.16	HAL		657	387.u 357.0	9.00			34470.	1.21	15.08
	E **	110	585.0				35000.	.52	9.16	BAL		424					34840.	2.22	39.07
ATL		112	585.0				35000.	.53	10.00	HAL		434					34840.	45.3	42.66
ATL		511	585.0				41000.	.49	17.85	HOL		593		23.00			31587	3.06	40.69
ATL	550	305	178.0	0.00	095	7	31347.	0.00	0.00	BUL		181		0.00				0.00	0.00
ATL		368	178.0				31027.	0.00	0.00	HHM		224		30.00				4.05	49.84
ATL		580	178.0				31347.	0.00	0.00	BNA		399	100.0				27800.	.52	7.75
ATL		86					29320.	76	-15.16	ANA		659	100.0				25800.	.51	12.19
AIL		182					29320.	77	-16.56	AVA		671	100.0				27800.	.52	7.75
ATL		533					29320.	76	-15.16	AVE		258	258.0	4.00			25800. 34720.	.51	12.19
ATL		578	145.0	-6.00	045	7	30240.	79	-10.94	AVA		894	258.0				33180.	.79	9.71
ATL		677					59350.	77	-15.56	405		129	748.0	5.00	725	7	35000.	.66	12.50
ATL	JEK	108	691.0				35000.	.52	9.16	HO5		145	748.0				35000.	.65	11.45
ATL			1542.0					2.22	39.92	305		533	748.0				35000. 35000.	.65	8.30
ATL			1542.0					2.22	38.92	805		537	740.0	5.00	152	6	35000.	.66	12.50
ATL			1642.0					2.22	38.92	305		537	748.0				35000.	.06	12.50
ATL		80	567.0	4.00	727	7	35000.	.52	9.16	40S		199	244.0				34160.	0.00	0.00
ATL	LGA	100	567.0	4.00	095	7	35000.	.54	6.64	905	DCA	393	244.0	0.00	727	7	34160.	0.00	0.00
ATL		432	567.0				35000.	.52	9.16	HO5		509	244.0				34160.	0.00	0.00
ATL		544	567.0				35000.	.52	9.16	H35		491	70.0				23880.	0.00	53.83
ATL	LGA.	572		4.00	725	7	35000.	.53	10.00	405	JFK	887	76.0	17.00	727	7	. 08HES	2.17	49.30
ATL		660					32397.	1.57	20.89	935			1030.0	0.00			35000.	0.00	0.00
ATL		980	217.0	12.00	727	7	33040.	1.55	28.70	HOS			1030.0				35000. 35000.	0.00	0.00
	MEN		217.0					1.57	31.34	ROS	MIA	419	1030.0	0.00	727	7	35000.		0.00
ATL		97					35000.	5.35	93.87	405		809					31933.	4.25	186.14
ATL		219					35000.	5.35	93.87	BUF	ATL	345					35000.	.53	10.00
ATL		387	470.0	41.00	095	7	35000.	5.54	68.05	AUF	ATL	961	551.0	4.00	113	7	41000.	.49	17.85
ATL	MIA	393	470.0	41.00	727	7	35000.	5.35	43.87		DHL	125					31720.	4.89	92.77
ATL		147					35000.	5.54	68.05		JFK 040	354					34500.	1.35	16.63
ATL		149					35000.	.52	9.10	CLT	020	214					34880.	1.35	16.63
ATL	MSY	661	200.0	4.60	725	7	J5000.	.53	10.00		CHO	218	428.0	10.00	095	7	34880.	1.35	16.63
ATL		429					35000.	.52	9.16	CLT		374					34880.	1.35	16.63
ATL		743					34840.	26	-3.14	CLT	246	540					35000.	1.61	27.47
ATL		242	424.0	-2.00	725	7	35000.	26	-5.00	CLT	OH!	596	312.0	12.00	121	7	35000.	1.57	27.47
ATL	OHO	244					35000.	26	-5.00	CLT		008					35000.	1.58	30.00
ATL		244					35000.	26	-4.58			310	240.0	10.00	095	7	33000.	1.34	17.19
ATL		248					34840.		-3.14		•								

TABLE G.2

EASTERN AIRLINES
ENROUTE RNAV BENEFITS ANALYSIS
Continued

ORG DST	FLT	RANGE			F CRUISE		FUEL		DST		RANGE	RNAV	A/C	F CRU	ISE	TIME	FUEL
A/P A/P	NO 348	NM1	BENE		7 33000.	1.34	BENE 17.19	A/P	A/P	NO 1HO	NMI 90.0	BENE	095	ALT 7 270	10	BENE 52	-7.92
CLT PIT	476				7 34000.	1.31	25.58	FLL		342		-4.00	-			52	-7.92
DCA 805	372				7 34840.	2.22	39.07	FLL	TPA	500		-4.00				52	-7.92
DCA HOS	398				7 34840.	2.22	39.07		LGA	362		11.00				1.48	18.70
DCA BOS	866				7 33210. 7 33210.	2.28	29.12	650		366 580		11.00				1.48	18.70
DCA HOS	878				7 33210.	2.28	29.12	650		206		0.00				0.00	0.00
DCA CLT	375				7 32093.	1.00	21.01	650		363	434.0	0.00.	095	7 349	40.	0.00	0.00
DCA CLT	385				7 32093. 7 32520.	1.60	28.95	IAD		558		-3.50				38	-8.39
DCA CLT	657				7 32520.	1.57	31.62	IAD		554 553		16.00				2.09	-3.05 36.63
DCA JAX	199	465.0	7.00	727	7 35000.	.91	16.03	JAX		364	148.0	2.00		7 2942		.26	5.51
DCA JAK	869	465.0			7 35000.	.95	11.62	JAX		368	148.0	5.00		7 2947		.26	5.51
DCA LGA					5 26040. 6 26040.	39	-6.10	JAX		63.0	148.0			7 2942		.26	5.51
DCA LGA					7 26040.	39	-6.10	JAX		630	148.0			2 244		.25	5.05
DCA LGA					7 26040.	39	-6.10	JAX	ATL.	676	148.0	2.00	095	7 3032	20.	.26	3.64
DCA LGA					7 26040.	39	-6.10	JAX		682	148.0			7 2942		.26	5.51
DCA LGA					7 26040.	39	-6.10	JAX.		866	462.0			7 3500		.95	11.62
DCA LGA					7 26040.	39	-6.10	JAX		156	634.0			7 3500		.68	8.30
DCA LGA					7 26040.	39	-6.10	JAX		397	203.0			7 3236		0.00	0.00
DCA LGA					7 26040.	39	-6.10	JFK		103		18.00				2.38	45.00
DCA LGA					7 26040.	39	-6.10	JFK		445		18.00				2.43	29.87
DCA LGA					7 26040.	39	-6.10	JFK		491		18.00				2.38	45.00
DCA LGA					6 26040.	39	-6.10		805	494		50.00				6.30	137.40
OCA LGA	175	718.0			7 26040. 7 35000.	39	-6.10 A.30	JFK		477	888.0			7 4100		.87	31.25 17.50
PCA MIA	177	718.0			7 35000.	.68	8.30	JFK		743	888.0			7 3500		.95	11.62
DCA HIA	195	718.0			7 35000.	.65	11.45	JFK		555	104.0			7 2612		.38	8.29
OCA MIA	197	718.0			7 35000.	.68	8.30	JFK		9	874.0			7 4100		0.00	0.00
DCA SOF	370	320.0			7 35000. 7 33800.	.94	8.30	JFK	MIA	23	874.0			7 3500		0.00	0.00
DCA SOF	509	320.0			7 35000.	.92	17.50	JFK		27	874.0			7 4100		0.00	0.00
nca STL	503	544.0	4.00	095	7 35000.	.54	6.64	JFK	MIA	401	874.0	0.00	LIO	7 4100	0.	0.00	0.00
OFW ATL	114	563.0			7 35000.	.54	6.64	JFK		405	874.0			7 3500		0.00	0.00
DEM ATL	150	563.0			7 35000.	.52	9.16	JFK		69	952.0			7 3500		1.04	18.32
DEW ATL	536	563.0	4.00	095	7 35000.	.54	6.64	JFK		161	739.0			7 3500		.79	9.40
DEW ATL	666	563.0			7 35000.	.52	9.16	JFK		167	789.0			7 3500		. 78	13.74
OTW ORD	341				1 25860. 7 24600.	50	-11.13	LGA		425	789.0			7 3500		.81	9.96
DI# PIT	349				7 26800.	13	-1.99	LGA		87	560.0			1 3500		.52	9.16
DIM PIT	739	85.0			7 24600.	13	-2.87	LGA		101	560.0	4.00	727	7 3500	0.00	.52	9.16
DIN TPA	335	-0.0			7 18200.	0.00	0.00	LGA		115	560.0			7 3500		.54	6.64
DIN TPA	639	-0.0	0.00		7 18200.	0.00	0.00	LGA		543	560.0			7 3500		.53	6.64
FIR ATL	105	569.0			7 35000.	.53	10.00	LGA		541	654.0			7 3500		.68	8.30
FAR ATL	135	569.4			7 35000.	.53	6.27	LGA		351		16.00				2.09	36.63
ENR ATL	353	377.0			7 41000. 7 35000.	.49	17.85	LGA		357 357	387.0	16.00	725	6 3500	00.	2.11	40.00
ENP CLT	359	377.0			7 34370.	.01	10.07	LGA		397	387.0	16.00	727	7 3500	0.	2.09	36.63
FHR CLT	383	377.0	6.00	095	7 34370.	.61	10.07	LGA	CLT	545	387.0	16.00	095	7 3447	70.	2.16	25.81
END FLL	407	368.0			7 35000.	.92	17.50			1401	105.0			5 2820		.52	7.67
EAR FLL	745	868.0			7 35000.	.91	16.03			1411	105.0			7 2820		.52	7.67
EWR FLL	759	868.0	7.00	727	7 35000.	.91	16.03	LGA	DCA	1431	105.0	4.00	045	7 2820	0.	.52	7.67
E 49 1AU	559	60.0	0.00	727	.00555 7	0.00	0.00			1441	105.0			7 2820		.52	7.67
ENR MIA	5	H56.0	7.00		7 35000. 5 35000.	.92	16.03			1451	105.0			7 2820		.52	7.67
EAR HIA	5	856.0	7.00		2 35000.	.95	11.62	LGA	DCA	1481	105.0			7 2820		.52	7.67
ENR HIA	7	A56.0			7 35000.	.92	17.50	LGA	DCA	1491	105.0			7 2820		.52	7.67
END TPA	403	A56.0			7 35000.	.91	16.03			1511	105.0			7 2820		.52	7.67
FAR TPA	165	811.0			7 41000.	.95	31.25			1521	105.0			7 2820		.52	7.67
FLL BOS		1029.0	9.00	727	7 35000.	1.17	20.61	LGA	DCA	1531	105.0	4.00	095	6 2820	0.	.52	7.67
FLL 905		1029.0			7 35000.	1.17	20.61			1541	105.0			7 2820		.52	7.67
FLL EWR	742	874.0			7 35000. 7 35000.	.91	17.50	LGA		153	627.0			7 3500		1.22	17.50
FLL EWH	746	874.0			7 35000.	.95	11.62	LGA		ii	865.0			1 3500		.92	17.50
FLL ENH	758	874.0	7.00	727	7 35000.	. 41	16.03	LGA	MIA	11	865.0	7.00	727	1 3500	.00	.91	16.03
FLL JFK	412	864.0	5.00	727	7 41000.	.62	22.32	LGA		17	865.0			6 3500		.87	16.03
FLL JFK	750	864.0	5.00	045	7 35000.	.65	8.30	LGA		51	865.0			7 4100		.87	31.25
FLL UND	468	951.0	8.00	727	7 35000.	1.04	18.32	LGA		415	865.0	7.00	L10	7 4100	.00	.87	31.25
FLL 090	790	957.0			7 35000.	1.04	14.32	LGA		193	803.0	7.00	725	7 3500	00.	.92	17.50
FLL PHL	795	853.0			7 35000.	.91	16.03	LGA		661	201.0	7.00	042	7 319	.0.	13	-1.76
FLL PHL	876	853.0			7 35000.	.91	16.03	MEM		665	201.0	-1.00	DC9	7 319	50.	13	-1.66

TABLE G.2

EASTERN AIRLINES
ENROUTE RNAV BENEFITS ANALYSIS
Continued

ORG DST	FLT	RANGE NMI	RNAV BENE	A/C	F	CRUISE ALT	TIME BENE	FUEL BENE	ORG A/P		FLT	RANGE NM)	RNAV BENE	A/C	F	CRUISE ALT	TIME BENE	FUEL BENE
MEM ATL	981					32253.	13	-2.65	090		791	960.0				35000.	1.04	18.32
MEM ATL	252					35000.	3.11	39.17	040		795	960.0				18200.	0.00	0.00
MIA ATL	452					35000.	3.00	52.66	040		209	-0.0				18200.	0.00	0.00
MIA ATL	564					35000.	3.04	57.50	090		75	953.0				35000.	1.04	18.32
MIA ATL	564					35000. 35000.	3.04	57.50	080	MIA	433	953.0				35000.	1.06	20.00
MIA ATL	616					35000.	3.04	57.50		HIA	433	953.0				35000.	1.04	18.32
MIA ATL	678					35000.	3.00	52.66	090	MIA	993	953.0				35000.	1.04	14.32
MIA BOL	172	977.0				35000.	1.19	15.00	020	PDU	203					35000. 35000.	6.49	79.67
MIA BOS		1081.0	9.00	727	7	35000.	1.17	20.61		TPA	237	804.0				35000.	.78	13.74
MIA BOS		1081.0				35000.	1.19	22.50	080	TPA	259	904.0				35000.	. 78	13.74
MIA CHH	302	814.0				35000. 35000.	.91	16.03	090	ATI	105	391.0	0.00			35000.	0.00	0.00
MIA DCA	159	747.0	6.00	727	5	35000.	.78	13.74	164		142	391.0	0.00	725	7	35000.	0.00	0.00
MIA DCA	158	747.0				35000.	.81	9.96	PHI		242	341.0				35000.	0.00	0.00
MIA DCA	190	747.0				35000.	.78	9.96	169		632	391.0				35000.	1.06	20.00
MIA DCA	192	747.0	6.00	725	7	35000.	.79	15.00	941	JFK	146	877.0	H.00	110	7	41000.	.49	35.71
MIA DEN	976	747.0				35000. 35000.	.81	25.19	IE9 IE9		440 PR.	877.0 823.0				35000.	1.04	18.32
WIA OTW	422	924.0				35000.	1.04	18.32	PHL		121	500.0				35000.	1.17	20.61
MIA DIM	952	424.0	8.00	L10	7	41000.	.99	35.71	PHL	ATL	123	500.0	9.00	727	7	35000.	1.17	20.61
MIA EWH	6	889.0				35000. 35000.	.92	17.50	PHL		125	500.0				35000.	1.17	20.61
MIA EWR	9	889.0				35000.	.92	17.50	PHL		449	500.0	9.00			41000.	1.11	49.17
HIA ENH	402	887.0				35000.	.92	17.50	PHL		561	500.0				35000.	1.19	22.50
MIA JEK	14	881.0				35000.	1.04	18.32	PHL		556					32627.	5.55	103.53
MIA JEK	22	881.0				35000.	1.09	13.28	PHL		373					33890.	2.82	35-55
MIA JEK	25	481.0				41000.	. 49	35.71	PHL		379					33390.	2.82	35.55
MIA JFK	16	892.0				41000.	.87	35.71	PHL		579					33890.	2.82	35.55
VIA LSA	20	0.568	7.00	725	6	35000.	.92	17.50	PHL		35					41000.	.87	31.25
MIA LGA	20	892.0				35000.	.91	16.03	PHL		37	416.0				35000.	.91	16.03
MIA LGA	24 28	892.0				35000.	.91	16.03	PHL	MIA	-11	H10.0				35000. 35000.	.91	16.03
MIA LGA	414	0.568	7.00	L10	7	41000.	.87	31.25	PHL	MSY	575	-0.0	0.00	727	7	12600.	0.00	0.00
MIA MSY	495	509.0				41000.	.87	31.25	PIT		327					35000.	2.56	47.50
MIA MSY	907	509.0				35000.	.26	3.32	PIT		989		19.00				2.51	47.50
USO AIP	72	959.0	8.00	727	7	35000.	1.04	18.32	PIT		311	235.0	3.00	725	7	33800.	.39	7.71
WIA ORD	78 430	959.0				35000. 35000.	1.04	18.32	PIT		317	235.0				33800.	.39	7.05
MIA PHL	34	820.0				35000.	.91	16.03	PIT		349	235.0				32867.	.40	5.17
MIA PHL	36	620.0				35000.	.91	16.03	PIT		309					35000.	2.22	38.92
MIA PHL	956	820.0				41000.	.92	31.25	DIT		483	809.0				35000.	.79	15.00
MIA PIT	300	811.0	6.00	725	7	35000.	.79	15.00	PIT		483	809.0	6.00	095	5	35000.	.81	9.96
MIA PIT	482	811.0				35000.	.81	9.96	RDU		351	223.0				33320.	.26	5.19
MIA STL	522	861.0				35000. 35000.	.95	11.62	ROU		361	553.0				33320.	.26	5.19
HIA TPA	614	90.0	-4.00	095	7	27000.	52	-7.92	ROU		393	223.0				33320.	.26	4.76
MIA TPA	787					27000. 35000.	3.38	-7.92 41.49	HDU		549	223.0				33320.	.26	5.19
MKE ATL	784	518.0	25.00	727	7	35000.	3.26	57.24	200	ATL	585	223.0	2.00	727	5	33320.	.26	4.76
MSY ATL	103	288.0				35000.	0.00	0.00	RDU		204	489.0				35000.	0.00	0.00
MSY ATL	428 565	584.0				35000. 33480.	0.00	0.00	HIC		208 562	146.0				2932U.	0.00	0.00
MSY ATL	634	288.0	0.00	725	7	35000.	0.00	0.00	RIC	LGA	898	146.0	0.00	095	7	30240.	0.00	0.00
MSY ATL	780	0.885				33480. 35000.	0.00	0.00		ATL	261		12.00			32013.	1.56	21.05
MSY MIA	521	505.0				35000.	1.04	16.63	SOF	ATL	269		12.00				1.56	19.87
MSY MIA	906	505.0				35000.	.91	16.03	SOF		+35					32013.	1.60	21.05
ONC DEN	980	365.0	20.00	727	7	35000. 34950.	2.61	45.79	SDF		254					32013.	1.60	21.05
ORD ATL	234	435.0	17.00	095	7	34950.	2.30	28.24	SDF	DCA	508	321.0	12.00	725	7	35000.	1.58	30.00
OPD ATL	245	435.0	17.00	727	7	35000.	2.22	38.92	STL		97					35000.	.13	65.5
ORD ATL	247					34950.	2.25	26.66	STL		271	334.0				37608.	.12	1.69
ORD ATL	957	435.0	17.00	L10	7	40840.	2.10	76.26	STL	ATL	273	334.0	1.00	095	7	33940.	.13	1.69
ORD BAM	225	409.0	3.00	009	7	34690.	.40	4.73	STL		699	334.0				35000.	.13	2.50
OPD BNA	891					34560. 33140.	.13	2.31	STL		510	532.0	4.00	095	7	35000.	.54	6.64
DED CLT	512	443.0	20.00	095	7	35000.	2.70	33.19	TPA	ATL	280	301.0	35.00	095	7	33610.	4.70	59.54
ORD CLT	217					35000.	2.70	33.19	TPA		476 48H		35.00			35000.	4.62	87.51
040 CF1	803					35000. 26040.	.13	2.77	TDA	ATL	546	301.0	35.00	725	7	35000.	4.62	87.51
OPD FLL	469					35000.	1.08	13.28	TPA		572					35000.	4.62	87.51
									, , ,	WIL	624	301.0	35.00	125	,	32000.	4.62	87.51

TABLE G.2 EASTERN AIRLINES ENROUTE RNAV BENEFITS ANALYSIS Continued

		DST A/P	NO	RANGE NMI	RNAV BENE	A/C	F	ALT	BENE	BENE		DST A/P	FLT NO	RANGE NMI	RNAV BENE	A/C	F CRUISE ALT	TIME	FULL BENE
1	-	ATL	476	301.0	35.00	725	7	35000.	4.62	87.51									
1	PA	ATL	488	301.0	35.00	121	7	35000.	4.57	80.13									
1	DA	ATL	546	301.0	15.00	125	7	35000.	4.62	87.51									
1	DA	ATL	572	301.0	35.00	725	7	35000.	4.62	87.51									
1	PA	ATL	624	301.0	35.00	725	7	35000.	4.67	87.51									
T	DA	CLE	316	728.0	5.00	095	7	35000.	.68	A.30									
1	PA	DT	340	797.0	7.00	045	7	35000.	.95	11.62									
1	DA	DI#	344	797.0	7.00	095	7	35000.	.95	11.62									
1	PA	DIO	464	797.0	7.00	095	7	35000.	. 45	11.62									
		EWR	168	916.0				41000.	.74	26.78									
1	PA	FLL	180					26040.	5.59	54.43									
1	DA	FLL	553	103.0	13.00	095	7	24040.	2.36	34.67									
1	PA	FLL	339	103.0	18.00	095	7	28040.	2.36	34.67									
		FLL	485					26040.	5.59	54.43									
1	PA	FLL	529	103.0				28040.	2.31	32.73									
1	PA	JFK	160	928.0	8.00	727	7	35000.	1.04	18.32									
1	PA	JFK	165	0.650	8.00	727	7	35000.	1.04	18.32									
1	PA	JFK	426	928.0	8.00	727	7	35000.	1.04	18.32									
1	PA	MIA	127	103.0	18.00	121	7	20040.	2.27	49.84									
1	PA	MIA	437	103.0	18.00	095	7	28040.	2.36	34.67									
1	PA	MIA	645					26040.	5.59	54.43									
		MIA	645					26040.	5.59	54.43									
		ORU	530	793.0				35000.	.78	13.74									
		OPD	595	793.0				35000.	.79	9.40									
1	PA	ORD	466	793.0	6.00	727	7	35000.	.78	13.74									

TABLE G.3 DELTA AIRLINES ENROUTE RNAV BENEFITS ANALYSIS

	FLT NO	RANGE NMI	RNAV BENE	A/C		CRUISE ALT	TIME BENE	FUEL BENE			DST A/P	FLT	RANGE NMI	RNAV BENE	A/C	F	CRUISE ALT	TIME BENE	FUEL BENE
ATL PAL	500	348.0	14.00	095	7	34480.	1.89	23.45			MSY	924	288.0				35920.	.51	17.01
	610	388.0	14.00	095	7	34480.	1.89	23.45			ORD	957					35920.	26	-5.02
	629					34480.	1.89	23.45			ORD	138					35000.	26	-5.00
	684	388.0	14.00	095	7	34480.	1.89	23.45			ORD	192					35000.	26	-5.00
	790					25080.	1.89	43.61			080	234					35000.	26	-5.00
ATL CAE	309	91.0	14.00	725	7	25080.	1.78	43.61		ATL	080	1138	424.0	-2.00	LIO	7	40488.	25	-9.07
ATL CAE	320					25080.	1.78	43.61			040	249					40488. 35000.	25	-9.07 -97.51
	637					27080.	1.03	27.65		ATL	PHI	267	389.0-	-39.00	725	7	35000.	-5.15	-97.51
	787					27080.	1.83	27.65			PHI	737					35000.		-97.51 -65.32
	907	116.0				25080.	1.73	23.41		ATL		280					35000.	1.72	32.50
ATL CLT	520	116.0	8.00	095	7	29040.	1.05	15.01			PHL	316					35000.	1.72	32.50
	620	116.0				27080.	1.05	15.01		ATL	PHL	318					35000.	1.72	32.50
ATL CLT 1		110.0				27080.	1.02	23.41		ATL	PHL	522	506.0	13.00	095	7	35000.	1.76	21.58
ATL CLT 1		116.0				29040.	1.05	15.01			SAV	233					27080.	2.30	58.03
	208	388.0				35000. 35000.	.40	7.50			SAV	305	116.0	18.00	725	7	27050.	2.30	52.66
	210	388.0	3.00	725	7	35000.	.40	7.50			SAV	331					27080.	2.30	52.66
	214	388.0				35000. 35000.	.40	7.50			SAV	345 657					27080.	2.30	33.77
	288	380.0				35000.	.40	7.50		ATL	SAV	802	115.0	18.00	DBF	7	27080.	2.23	76.28
	302	388.0				35000.	.40	7.50		ATL		248					31667.	39	-8.00
ATL UF	15	555.0				34480.	.13	6.89		ATL		483					31667.	39	-8.00
ATL OF	21	555.0				37000.	.13	6.89		ATL		595					31667.	40	-5.30
	783	555.0 555.0				35000. 41000.	.14	1.66		ATL		632	190.0				31667.	40	-5.30 -5.30
ATL OFW	158	555.0	1.00	DBF	7	41000.	.13	4.20		ATL	SFO	453	1800.0	20.00	085	7	41000.	2.58	91.85
ATL OF .	919	555.0				41000.	.13	4.59			TPA	1191	1500.0				35000.	1.45	89.27 27.50
ATL OF # 1		555.0				41000.	.12	4.46			TPA	287					35000.	1.45	27.50
ATL OF 1		555.0				41000.	.13	4.20			TPA	355					35000.	1.45	27.50
	148	555.0				35000. 35000.	.13	2.50			TPA	991					35000.	1.41	46.65
ATL DIN 1		555.0	1.00	L10	7	41000.	.12	4.46				1139					35480.	1.35	56.05
ATL DIA 1	116	555.0				41000.	.12	10.00				1147					35480.	1.35	56.05
ATL EAR	384	585.0	4.00	725	7	35000.	.53	10.00		BAL	ATL	315	387.0	9.00	725	7	35000.	1.19	22.50
	135	585.0				29320.	77	18.37			ATL	505	387.0				34470.	1.21	15.08
	289					29320.	77	-16.56		HAL		699	387.0				34470.	1.21	15.08
	343					24320.	17	-16.56			ATL	705	387.0				34470.	1.21	15.08
ATL JAX 1	000					29320.	77	-16.56			805	790					33210.	2.28	29.12
ATE JAK 1	095	144.0	-5.00	L10	7	29320.	72	-33.34			JAN	303	107.0	12.00	725	7	26350.	1.53	35.92
ATL JAX 1	516		4.00				72	6.64			JAN	397					26360.	1.53	35.92
ATL JFK	424	691.0	4.00	DRS	7	41000.	.52	18.37		PHF	JAN	1227	107.0	12.00	727	7	26350.	1.51	32.89
ATL JFK 1			4.00				2.19	17.85			040	542	258.0				34940.	4.05	10.28
ATL LAX 1							2.10	75.88		AVE	(120)	569	258.0				33180.	.80	10.28
ATL LAX 1							2.10	75.88			ORD	668	258.0				33180.	.80	10.28
ATL LGA	120	567.0				35000.	.53	10.00		105		760					34120.	13	-2.55
ATL LGA	128	557.0	4.00	725	7	35000.	.53	10.00		305	BAL	711	243.0	-1.00	095	6	33030.	13	-1.72
	136	567.0				35000. 35000.	.53	10.00			DCA	215	244.0				34160.	0.00	0.00
ATL LGA	200	567.0	4.00	725	7	35000.	.53	10.00		105	DCA	275	244.0	0.00	725	6	34160.	0.00	0.00
	515	567.0				35000. 33080.	1.57	31.34			DCA	303	244.0				34160.	0.00	0.00
	211		12.00				1.57	31.34			DCA	323	244.0				34160.	0.00	0.00
	307					330A0.	1.57	31.34		205			244.0					0.00	0.00
	450					33080. 33080.	1.57	31.34			JFK	175					23880.	2.20	53.83
ATL MEM	617	217.0	12.00	095	6	32387.	1.60	20.89		305	MIA	129	1030.0	0.00	725	7	35000.	0.00	0.00
ATL MEM 1	690	217.0	12.00	D95	7	32387. 33080.	1.60	62.60	, and	105	MIA	375	1030.0	0.00	725	7	35000.	0.00	0.00
ATL MIA	149	470.0	41.00	725	7	35000.	5.41	102.51		105	PHL	219	195.0	35.00	725	7	31933.	4.55	93.02
	335					35000.	5.41	102.51			PHL	261					31933.	4.56	93.02
	989	470.0	41.00	DHS	7	41000.	5.30	188.29	c	SCE	PHL	491	195.0	35.00	725	6	31933.	4.55	93.02
ATL MIA	993	470.0	41.00	DAS	7	41000.	5.30	188.29			PHL	577					31800.	4.65	61.65
ATL MIA 1		470.0	41.00	L10	7	41000.	5.07	183.01	(LT	JFK	326	340.0	25.00	725	7	35000.	3.30	2.55
ATL MST	119	288.0	4.00	725	7	35000.	.53	10.00	(LT	JFK	520	390.0	25.00	095	7	34500.	3.37	41.87
	325	288.0				35000. 35000.	.53	10.00		YAC		339					35000.	1.06	20.00
ATL MSY	327	288.0	4.00	725	7	35000.	.53	10.00		YAC	ATL	773	382.0	8.00	095	7	34420.	1.08	13.42
	917	298.0	4.00	095	7	33480. 35920.	.54	17.01		MY	ATL	785	395.0	8.00	095	7	34420.	1.08	13.42
4.6 1.3					1														

TABLE G.3 DELTA AIRLINES ENROUTE RNAV BENEFITS ANALYSIS Continued

ORG DST FLT A/P A/P NO	RANGE NM1	RNAV	A/C F	CRUISE	TIME	FUEL	ORG A/P		FLT	RANGE NMI	RNAV	A/C	F CRUISE	TIME	FUEL
DAY ATL 789	102 0		001 7	34420.	1 00	13.42				2.00	4 00	006	7 33080.	.80	10.30
				39144.	1.08	37.77	IND		633	248.0	6.00	095	7 33080.	.80	10.30
DCA 805 208	261.0 1	17.00	725 7	34840.	2.24	42.66	JAN	HEH	517	75.0	-1.00	095	7 26400.	13	-2.01
DCA 805 210				34840.	2.24	42.66	JAN		558	75.0	-1.00	095	7 26400.	13	-2.01
DCA 805 214				34840.	2.24	42.66	MAL		709				7 26400. 7 26400.	13	-2.01
DCA 805 230				34840.	2.24	42.65			303	101.0			7 25680.	0.00	0.00
				34840.	2.24	42.66	MAL		629	101.0			7 27880.	0.00	0.00
DCA 805 302 DCA 805 775				34840.	2.28	42.66	MAL		729	101.0	0.00	727	7 27880. 7 25880.	0.00	0.00
				37000.	.50	27.58	JAX		412	148.0			7 29427.	.26	5.51
DFW ATL 126	563.0	4.00	725 7	35000.	.53	10.00	JAX		448	148.0			7 29427.	.26	5.51
DEN ATL 704				35000.	.54	6.64	JAX		936	148.0			7 29427.	.25	8.84
OFW ATL 820				41000.	.51	16.81	JAX		1019	148.0			7 29427.	.26	11.09
OFW ATL 985				41000.	.52	18.37			1117	148.0	2.00	L10	7 29427.	.24	11.09
DEM ATE 1110				41000.	.49	17.85			1196	148.0			7 29427.	.24	11.09
DFW ATL 1116	563.0			41000.	.49	17.85	JAX		1235	203.0			7 32360. 7 35000.	2.38	45.00
DFW LAX 21	999.0			37000.	0.00	0.00	JFK		389	572.0	18.00	725	7 35000.	2.38	45.00
				41000.	0.00	0.00	JFK		959				7 41000.	2.33	82.66
OFW LAX 1125	999.0			41000.	0.00	0.00	JFK		254 175	106.0			7 26280. 7 35000.	6.37	17.50
DF# LAX 1227				35000.	0.00	0.00	JFK		369	0.858			7 35000.	.92	17.50
DEW LAX 1885	999.0	0.00	DAF 7	41000.	0.00	0.00	JFK		1071	988.0	7.00	L10	7 41000.	.87	31.25
				33620.	54	-6.90			1081	888.0			7 41000.	.87	31.25
DEM MSY 814				36480.	54	-6.80	JFK		193	874.0			7 35000. 7 35000.	0.00	0.00
DFW MSY 898	302.0 -	-4.00	DBF 7	36480.	51	-15.62	JFK	MIA	479	874.0	0.00	725	7 35000.	0.00	0.00
OFW MSY 920 OFW MSY 928				36480.	52	-17.07	JFK		121	952.0			7 35000.	1.06	20.00
				36480.	52	-17.07 -17.07	JFK		177	789.0			7 35000. 7 35000.	1.06	20.00
DEM MSY 1080	302.0 -	-4.00	L10 7	36480.	49	-20.18	JFK		495	789.0	6.00	725	7 35000.	.79	15.00
				35000.	.66	12.50	LAS						5 41000.	1.29	45.92
				41000.	.64	21.01	LAS		911	293.0			5 36120. 7 41000.	0.00	0.00
OF# SFO 819 1	177.0	0.00	DBF 7	41000.	0.00	0.00	LAX	DFW	922	991.0			7 41000.	0.00	0.00
				41000.	0.00	0.00	LAX		1080	991.0			7 41000.	0.00	0.00
DFW SFU 1019 1				41000.	0.00	0.00	LAX		1110	991.0			7 41000.	0.00	0.00
	449.0			35000.	0.00	0.00	LAX		1228	991.0			7 35000.	0.00	0.00
				41000.	0.00	0.00	LAX			1386.0			7 41000.	1.81	64.29
				41000.	0.00	0.00	LAX						2 41000.	2.16	71.97
				27890.	52	-7.74	LAX		983				2 32147.	2.16	71.97
				27880.	52	-7.74	LGA	ATL	123	560.0	4.00	725	7 35000.	.53	10.00
DTW 50F 745	-0.0			12600.	0.00	8.87	LGA		201	560.0			7 35000. 6 35000.	.53	10.00
PTW TPA 1185	-0.0			12600.	0.00	0.00	LGA		223	560.0			7 35000.	.53	10.00
				35000.	.53	10.00	LGA		397	560.0			7 35000.	.53	10.00
				35000. 35000.	.54	6.64	LGA		732	560.0			7 35000.	.54	6.64
				35000.	.54	6.64	LGA		629 327	387.0			7 35000. 7 35000.	2.11	40.00
				41000.	.52	18.37	LGA		525				6 34470.	2.16	26.81
				35000.	.79	15.00	LGA		721				1 34470.	2.16	17.50
EAR FLL 189				35000.	.92	17.50	LGA	HIA	565	865.0			7 35000.	. 45	11.62
	868.0	7.00	725 7	35000.	.92	17.50	MOW	STL	595	137.0	14.00	045	6 29880.	1.85	25.74
	029.0			35000.	.91	16.03	MEM		247				7 32253.	13	-2.65
	029.0			35000.	1.19	22.50	MEM		363				7 32253. 7 32253.	13	-2.65
FLL 805 1168 1	629.0	4.00	L10 7	41000.	4.11	40.17	MEM	ATL	475	201.0	-1.00	725	7 32253.	13	-2.65
	874.0			35000.	.92	17.50	HEH		651				7 31960.	13	-1.76
	874.0			35000.	.92	17.50	MEM		666				7 31960. 7 31960.	13	-1.76 -1.76
FLL JFK 254	854.0	5.00	725 7	35000.	.66	12.50			1139				7 32253.		-5.29
		5.60	725 7	35000.	.65	12.50	MEM	BHM	324	96.0	0.00	725	7 25480.	0.00	0.00
				35000.	.66	12.50	MEM		671 390	243.0			7 27480.	0.00	0.00
FLL OPD 388	957.0			35000.	1.06	20.00	MEM		558	243.0			7 33030.	0.00	0.00
FLL OPU 452	957.0			35000.	1.06	20.00	MEN	INU	762	243.0	0.00	095	7 33030.	0.00	0.00
				35000. 35000.	1.06	17.50	HEH		768	243.0			7 33030. 7 26400.	13	-2.01
FLL PHL 782	653.0	7.00	095 7	35000.	.95	11.62	MEM	JAN	669	75.0	-1.00	D95	7 26400.	13	-2.01
FLL TPA 1140				25000.	48	-24.47	MEN		741	75.0	-1.00	095	7 26400.	13	-2.01
	434.0	2.00	095 7	34940.	0.00	0.00	MEH		763				7 26400.	13	-2.01
140 ATL 107	390.0 1	2.00	08F 5	39400.	1.54	49.05	MEM		295	214.0	-4.00	725	7 32947.	52	-10.47
	248.0				.79	15.24	MEH	MSY	615	214.0	-4.00	095	6 32307.	53	-6.97
140 MEM 615	248.0	6.00	042 6	33080.	.60	10.30	MEH	MSY	1133	214.0	-4.00	L10	7 32947.	49	-20.90

TABLE G.3
DELTA AIRLINES
ENROUTE RNAV BENEFITS ANALYSIS
Continued

1.0 1.0			DST A/P	FLT NO	RANGE NMI	RNAV BENE	A/C	F	CRU1SE ALT	TIME BENE	FUEL BENE	ORG A/P	DST A/P	FLT NO	RANGE NMI	RNAV BENE	A/C	F	CRUISE ALT	TIME	FUEL BENE
147 080 782 322.0 1.00 087 7 31800. 133 1.69 000 844 567 244.0 1.00 087 7 31800. 131 1.72 41 080 78 1.00 087 7 31800. 131 1.72 41 080 7 1.00 087 7 31800. 131 1.72 41 080 7 1.00 087 7 31800. 131 1.74 41 080 7 1.00 087 7 31800. 131 1.74 41 080 7 1.00 087 7 31800. 131 1.74 41 080 7 1.00 087 7 31800. 131 1.74 41 080 7 1.00 087 7 31800. 131 1.74 41 080 7 1.00 087 7 31800. 131 1.74 41 080 7 1.00 087 7 31800. 127 1.75 9 0.00 087 1.00 087 7 31800.																					
## 900 766 322.0 1.00 097 33860 . 133 1.09	-																				
## STL 671 136.0 4.00 057 2946 32 7.39 ## STL 671 136.0 4.00 057 2946 32 7.39 ## STL 671 136.0 4.00 057 2946 32 7.39 ## STL 671 136.0 4.00 057 2946 32 7.39 ## STL 671 136.0 4.00 057 2946 32 7.39 ## STL 671 136.0 4.00 057 2946 32 7.39 ## STL 671 136.0 4.00 057 2946 32 7.39 ## STL 671 136.0 4.00 057 2946 32 7.39 ## STL 671 136.0 4.00 057 2946 32 7.39 ## STL 671 136.0 4.00 057 2946 32 7.39 ## STL 671 136.0 4.00 057 2946 32 7.39 ## STL 671 136.0 4.00 057 2946 32 7.39 ## STL 671 136.0 4.00 057 2946 32 7.39 ## STL 671 136.0 4.00 057 2946 32 7.39 ## STL 671 136.0 4.00 057 2946 32 7.39 ## STL 671 136.0 4.00 057 2946 32 7.39 ## STL 671 136.0 4.00 057 2946 32 7.39 ## STL 671 136.0 4.00 057 2946 32 7.39 ## STL 671 136.0 4.00 057 3946 32 7.39 ## STL 671 136.0 4.00 057 3946 32 7.39 ## STL 672 136.0 4.00																					
New York 233 161.0 2.00 054 7 306.0 .27 3.59 0.20 01.0 1.00 0.05 2.600.0 1.04 2.00 1.00 2.00	M.E	4 (020																		
### STL 67 136.0 4.00 055 7 2948553 7.39																					
## 511 567 132.0 4.00 095 7 29660. 523 7.39																					
## 516 57 134.0										.53											
### 51, 670 132.0, 4.00 DVS 7 2964053 7.39 ### 57 160 132.0, 4.00 DVS 7 2964053 7.39 ### 67 132.0, 4.00 DVS 7 396053 10.00 ### 67 132.0, 4.00 DVS 7 396053 10.00 ### 67 132.0, 4.00 DVS 7 3960130 ### 67 132.0, 4.00	145		STL							.53											
Company Comp																					
New New No. 17 181.0 0.00 0.05 7.0900 1.10 1.17 1.17 0.00 1.17 1.17 1.10 1.17																					
NOM MSY 517 161.0 9.00 095 730840. 1.10 16.17 000 MET 763 123.00.0 055 7 1383055 -6-7.8																					
HIA ART HE ARE - 93.0 23.00 725 7 35000. 3.00 57.55 000 HE HIBS 322.000.0 LID 7 3725629 - 19.5 HIA ART - 94.0 453.0 23.00 085 7 41000. 2.97 105.03 000 HIA 351 933.0 8.00 725 7 35000. 1.06 62.00 WIE ARE - 93.0 23.00 085 7 41000. 2.95 102.65 000 HIA 351 933.0 8.00 725 7 35000. 1.06 62.00 WIE ARE - 93.0 23.00 085 7 41000. 2.95 102.65 000 HIA 351 933.0 8.00 725 7 35000. 1.06 62.00 WIE ARE - 93.0 23.00 085 7 41000. 2.95 102.65 000 HIA 351 933.0 8.00 725 7 35000. 1.08 22.00 080 HIA 351 933.0 8.00 725 7 35000. 1.08 22.00 080 HIA 351 933.0 8.00 107 7 41000. 2.95 102.65 000 HIA 135 933.0 8.00 LID 7 41000. 2.95 102.65 000 HIA 135 933.0 8.00 LID 7 41000. 2.95 102.65 000 HIA 135 933.0 8.00 LID 7 41000. 2.95 102.65 000 HIA 135 933.0 8.00 LID 7 41000. 2.95 102.65 000 HIA 135 933.0 8.00 LID 7 41000. 2.95 102.65 000 HIA 135 933.0 8.00 LID 7 41000. 2.95 102.65 000 HIA 135 933.0 8.00 LID 7 41000. 2.95 102.65 000 HIA 135 933.0 8.00 LID 7 41000. 2.95 102.65 000 HIA 135 933.0 8.00 LID 7 41000. 2.95 102.65 000 HIA 135 933.0 8.00 LID 7 41000. 2.95 102.65 000 HIA 135 933.0 8.00 LID 7 41000. 2.95 102.65 000 HIA 135 933.0 8.00 LID 7 41000. 2.95 102.65 000 HIA 135 933.0 8.00 LID 7 41000. 2.95 102.65 000 HIA 135 933.0 8.00 LID 7 41000. 2.95 102.65 000 HIA 135 933.0 8.00 LID 7 41000. 2.95 102.65 000 HIA 135 933.0 8.00 LID 7 41000. 2.95 102.65 000 HIA 135 933.0 8.00 LID 7 41000. 2.95 102.65 000 HIA 135 933.0 8.00 LID 7 41000. 2.95 102.65 000 HIA 135 933.0 8.00 HIA 135 933.0 HIA 135 933.						9.00	095	7	30840.	1.19	16.17	ORD	MEH		323.0	-4.00	095	7	33830.	54	
141 ATL 456 453.0 23.00 25.7 35000 3.04 57.50 300 MIA 315 93.00 40.57 735000 1.06 62.00																					
MIA ATL 1990 453,0 23.00 1985 7 41000 2.47 195.63 197.14 431 953.0 8.00 725 7 35000 1.06 20.06 198.44 198.56 453.0 23.00 11.07 41905 23.55 102.66 102.66 198.44 1315 953.0 8.00 102 7 41905 23.55 102.66 102.																					
### ART 1199																					
MIA 805 A76 1061.0 9.00 725 A15000. 1.19 22.50 980 200 533 507.0 98.00 095 73500. 6.49 77.07 MIA 805 A76 1061.0 9.00 725 735000. 1.19 22.50 980 200 533 507.0 98.00 12.00 725 728880. 1.59 31.50 MIA 805 476 1061.0 9.00 725 73500. 1.19 22.50 980 511 137 135.0 12.00 725 728880. 1.59 31.50 MIA 807 35 92.00 980 511 137 135.0 12.00 725 728880. 1.59 31.50 MIA 807 35 92.00 980 511 137 135.0 12.00 725 728880. 1.59 31.50 MIA 807 35 92.00 980 511 137 135.0 12.00 725 728880. 1.59 31.50 MIA 807 35 92.00 980 511 539 130.0 12.00 725 728800. 1.59 22.00 980 511 539 130.0 12.00 985 728800. 1.59 22.00 MIA 807 35 980 980 780 780 980 980 980 980 980 980 980 980 980 9																					
MIA 805 276 1081.0 9.00 725 6 35000. 1.19 22.50 000 500 503 507.0 9.0.00 005 7 35000. 6.90 79.57 MIA 805 487 1081.0 9.00 725 7 35000. 1.19 22.50 000 511 137 130.0 12.00 725 7 28860. 1.54 32.50 MIA 805 488 1081.0 9.00 725 7 35000. 1.00 725 7 35000. 0.00 725 7 35000. 1.00 725 7 35000																					
Mix 805 478 1081.0 ".00 725 7 35000. 1.07 22.50 0RD STL 163 13.6.0 12.00 725 7 26860. 1.54 33.58 Mix 805 48 1081.0 ".00 625 7 35000. 1.05 11.02 0RD STL 157 116.0 12.00 725 7 26860. 1.54 33.58 Mix 614 755 814.0 7.00 725 7 35000. 1.05 62.00 0RD STL 557 115.0 12.00 057 6 28860. 1.53 22.03 Mix 814 814 92 924.0 6.00 725 7 35000. 1.05 62.00 0RD STL 557 116.0 12.00 057 6 28860. 1.58 22.03 Mix 814 814 352 889.0 7.00 725 7 35000. 1.06 62.00 0RD STL 651 110.0 12.00 057 6 28860. 1.58 22.03 Mix 814 814 378 889.0 7.00 725 7 35000. 1.06 62.00 0RD STL 651 110.0 12.00 057 6 28860. 1.58 22.03 Mix 814 814 814 814 814 814 814 814 814 814																					
MIA DIA 359 924.0 6.00 725 7 35000. 1.06 20.00 020 5TL 557 115.0 12.00 056 29840. 1.58 22.09 MIA DIA 492 924.0 6.00 725 7 35000. 1.06 20.00 020 5TL 651 136.0 12.00 056 29840. 1.58 22.09 MIA DIA 492 924.0 6.00 725 7 35000. 1.06 20.00 020 5TL 651 136.0 12.00 056 29840. 1.58 22.09 MIA JIK 470 RAIL 0 8.00 725 7 35000. 1.06 20.00 020 5TL 651 136.0 12.00 056 29840. 1.58 22.09 MIA JIK 470 RAIL 0 8.00 725 7 35000. 1.06 20.00 020 5TL 650 136.0 12.00 056 7 28840. 1.58 22.09 MIA JIK 470 RAIL 0 8.00 725 7 35000. 1.06 20.00 020 5TL 650 136.0 12.00 057 7 28840. 1.58 22.09 MIA JIK 478 RAIL 0 8.00 725 7 35000. 1.06 20.00 020 5TL 650 136.0 12.00 057 7 28840. 1.58 22.09 MIA JIK 478 RAIL 0 8.00 725 7 35000. 1.06 20.00 020 5TL 650 136.0 12.00 057 7 28840. 1.58 22.09 MIA JIK 478 RAIL 0 8.00 725 7 35000. 1.06 20.00 020 5TL 751 136.0 12.00 057 7 28840. 1.58 22.09 MIA QUI 122 959.0 8.00 725 7 35000. 1.06 20.00 020 5TL 751 136.0 12.00 057 7 28840. 1.58 22.09 MIA DIA 122 959.0 8.00 725 7 35000. 1.06 20.00 020 5TL 751 136.0 12.00 057 7 28840. 1.58 22.09 MIA DIA 122 959.0 8.00 725 7 35000. 1.06 20.00 020 5TL 751 136.0 12.00 057 7 3500079 15.00 MIA DIA 122 959.0 8.00 725 7 35000. 1.07 12.00 020 7 28840. 1.58 22.09 MIA DIA 122 959.0 8.00 725 7 35000. 1.07 12.00 020 7 28840. 1.58 22.09 MIA DIA 122 959.0 8.00 725 7 35000. 1.07 12.00 020 7 28840. 1.58 22.09 MIA DIA 122 959.0 8.00 725 7 35000. 1.07 12.00 020 7 28840. 1.58 22.00 020 020 020 020 020 020 020 020 0																				1.54	
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	031	0 1	ATL	749	435.0	17.00	095	7	34950.			SFO	ATL	1096	1775.0	25.00	LIO	7	41000.	3.09	111.59
	0.51) 1	ATL	1151	435.0	17.00	L10	7	40940.	2.10	76.26	SFO	ATL	1126	1770.0	25.00	C10	7	41000.	3.09	111.59

TABLE G.3
DELTA AIRLINES
ENROUTE RNAV BENEFITS ANALYSIS
Continued

FUEL BENE

	DST A/P	FLT NO	RANGE NMI	RNAV BENE	A/C	F	CRUISE ALT	TIME	FUEL BENE		DST A/P	FLT NO	RANGE NMI	RNAV BENE	A/C F	CRUISE	
SOF		730					30160.		-21.92								
SOF		754					10160.		-21.92								
SFO			1776.0					3.09	111.59								
SFO			1207.0					1.54	50.42								
SFO			1207.0					1.54	50.42								
SFO			1207.0					1.55	55.11								
SFO			1207.0					1.48	53.56								
SFO		986					35520.	3.73	123.02								
SFO		972	182.0				31240.	.38	12.89								
SFO		984	182.0				31240.	.38	12.89								
SHV	JAN	188	102.0	1.00	725	7	25900.	.13	3.03								
SHY	JAN	392	102.0	1.00	725	7	25950.	.13	3.03								
SHV	JAN	524	102.0	1.00	095	7	27960.	.13	1.93								
SHY	JAN	724	102.0	1.00	D95	7	27960.	.13	1.93								
SHY	MAL	1558	102.0	1.00	727	7	25460.	.13	2.78								
SIL		565					59500.	26	-3.74								
STL		770					59500.	26	-3.74								
SIL		157					59360.	13	-5.85								
SIL		595					296A0.	13	-1.85								
STL		651					5APH0.	13	-1.85								
SIL		009					29640.	13	-1.85								
STL		761					54500.	13	-3.74								
STL		570					24500.	26	-3.74								
STL		617					54500.	26	-3.74								
	040	630					29200.	26	-3.74								
STL		652					24200.	26	-3.74								
SIL		500					29200.	26	-3.74								
SIL	040	070	120.0	-2.00	D95	7	29200.	26	-3.74								
STL	020	640	120.0	-2.00	095	7	24500.	26	-3.74								
STL	090	778	120.0	-2.00	095	7	29200.	26	-3.74								
SIL		746					29200.	26	-3.74								
IPA		450					15000.	4.62	87.51								
124		596					35000.	4.62	87.51								
TPA		952					36440.	4.51	149.33								
		1044					36440.	4.32	176.67								
	ATL						35440.	4.32	176.67								
	ATL	1146					35440.	4.32	176.67								
TPA		140	797.0				35000.	4.32	17.50								
TPA		490	797.0				41000.	.90	32.15								
	FLL	1185					26040.	2.16	108.54								
TPA		250	928.0				35000.	1.06	20.00								
TPA		284	928.0				35000.	1.06	20.00								
TPA		287					26040.	2.29	54.43								
TPA		145					28040.	2.36	34.67								
TPA		753					28040.	2.36	34.67								
TPA		256	793.0				35000.	.79	15.00.								
IPA	040	282	793.0	0.00	725	7	35000.	.79	15.00								
TPA	URD	352	793.0	6.00	725	7	35000.	.79	15.00								

TABLE G.4 AMERICAN AIRLINES ENROUTE RNAY BENEFITS ANALYSIS

ORG DST A/P A/P	FLT NO	RANGE NMI	RNAV BENE	A/C F	CRUISE	TIME BENE	FUEL BENE		DST A/P		RANGE NMI	RNAV BENE	A/C	F	CRUISE ALT	TIME	FUEL BENE
HAL OF	371	1006.0	8.00	707	41000.	1.02	30.66	OCA	090	411					35000.	5.49	96.15
RAL DEN		1006.0			7 41000.	1.02	30.66	DCA		137					35000.	5.48	96.16
HAL DE		1006.0			7 41000.	1.02	30.66	DCA		305					35000.	5.48	96.16
HAL DE	211	1006.0			41000.	1.02	30.66	DCA		437	455.0	42.00	727	7	35000.	5.48	96.16
BOL ORD		1006.0			7 35000. 7 35000.	1.04	18.32	DEM		504					41000. 35000.	2.17	38.92
HOL ORD	193	589.0	-2.00	727	35000.	26	-4.58	DFW	HAL	324	980.0	17.00	707	7	41000.	2.17	65.15
ANA DCA					35000.	2.22	-4.58 38.92	DFW		262		17.00				2.17	38.92
ANA DCA	326	413.0	17.00	727	35000.	2.22	38.92	DFW		324					41000.	2.17	65.15
BVA DCA		413.0 87.0			7 35000.	2.22	38.92	DEW		274					35000.	2.87	50.37
RVA MEM					24760.	0.00	0.00	DFW		280					35000.	2.87	50.37
HVA MEN		87.0			24760.	0.00	0.00	DFW		270	480.0	00.55	727	6	35000.	2.87	50.37
HOS HAL	35	87.0			34120.	13	-3.51	DEM		130	835.0				35000.	.89	26.82
BOS BAL	371	243.0	-1.00	707	7 34120.	13	-3.51	DFW		333	396.0	4.00	727	7	35000.	.52	9.16
BUS DCA	429	244.0			34160.	0.00	0.00	DEM		85	396.0	4.00	727	7	35000.	.52	9.15
HOS DCA	275	244.0			34160.	0.00	0.00	DEM		141	396.0				39592.	.52	14.95
905 DCA	651	244.0			34160.	0.00	0.00	DF#		325	396.0	4.00	727	7	35000.	.52	9.16
HOS DCA	305	244.0			34160.	0.00	0.00	DEM		286	955.0				41000.	1.02	30.66
HOS DCA	453	244.0			34160.	0.00	0.00	DFW		378	955.0	8.00	010	7	41000.	1.01	33.37
HOS DEN	607	244.0			41000.	1.53	0.00	DFW			1185.0					1.28	38.32
305 JFK	117				23880.	2.12	70.69	DFW		27	1186.0				35000.	0.00	0.00
HOS JEK	95				23880.	2.12	70.69	nF#	LAA	371	499.0	0.00	707	7	41000.	0.00	0.00
HOS ORD	273				41000.	51	112.63	DFW		239	999.0				35000.	0.00	0.00
905 040	445	660.0	-4.00	707 7	41000.	51	-15.33	DFW		511	999.0				41000.	0.00	0.00
905 080	135				41000.	51	-15.33	DEW		439	999.0				41000.	0.00	0.00
POS OPD	157				41000.	52	-16.69	DEN		3439	999.0				41000.	0.00	0.00
HUF DIN	53				27960.	.37	11.39	DEM	LAX	95	999.0	0.00	707	7	41000.	0.00	0.00
PUF DRU	181	314.0			35000.	0.00	0.00	OFW		2095	999.0				35000.	6.13	0.00
BUE 020	355	314.0	0.00	727 7	35000.	0.00	0.00	DFW		672	1158.0	47.00	727	5	35000.	6.13	107.61
CLE DEM	131	839.0			36960.	0.00	0.00	DEM			1158.0					6.13	107.61
CLE LAX					41000.	2.30	68.98	DFW			1158.0					6.13	107.61
CLE LAX					41000.	2.30	68.98	DFW		376					31987.	2.32	43.77
CLE LGA	360				34720.	91	-16.13	DFW		122		18.00				2.32	43.77
CLE LGA	130	258.0	-7.00	727 7	34720.	91	-16.13	DFW	LIT	177	196.0	18.00	727	7	31987.	5.35	43.77
CLE LGA	475				34720.	1.28	-16.13 36.35	DEM		180		19.00			35000.	2.48	43.50
CLE STL	133	345.0	10.00	707 7	37960.	1.28	36.35	DFW	MEM	398	304.0	19.00	727	7	35000.	2.48	43.50
CLE STL	385				35000.	1.30	22.90	DFW		240		19.00				2.48	43.50
CHE STL	209				35000.	1.30	22.90	DEW		572		-1.00				13	-2.93
CHH LGA	615				35000.	1.30	55.90	DFW		340		-1.00				13	-2.93
DCA BNA	605	318.0			35000.	1.30	2.29	DFW		306		-1.00				13	-2.93
DCA BNA	369	392.0	1.00	727 7	35000.	.13	2.29	DFW	080	210	600.0	5.00	707	7	41000.	.64	19.16
DCA BOS	511	392.0			35000.	2.22	39.07	DFW		332	600.0				35000.	.65	11.45
DCA BOS	535	261.0	17.00	727 7	34840.	5.55	39.07	DFW		442	600.0				41000.	.64	19.15
DCA BOS					34840.	2.22	39.07	DFW		590	500.0				35000.	.65	11.45
DCA BOS	350				34840.	2.22	39.07	DFW	ORD	315	600.0				35000. 35000.	.65	11.45
DCA BOS	352	261.0	17.00	727 7	34840.	2.22	39.07	DFW	040	402	600.0	5.00	727	7	35000.	.65	11.45
DCA 805					34840.	2.22	39.07	DEM		158	713.0				41000.	.64	19.16
DCA LGA	446	60.0	-3.00	727 5	23080.	39	-A.79	DFW		652	713.0	5.00	727	7	35000.	.65	11.45
DCA LGA					23080.	39	-8.79 -8.79	DFW			127.0					0.00	30.37
DCA LGA					23080.	39	-8.79	DFW		487	1177.0	0.00	707	7	41000.	0.00	0.00
DCA LGA	505	66.0	-3.00	727 6	23080.	39	-8.79	DFW	SFO	603	1177.0	0.00	127	7	35000.	0.00	0.00
DCA LGA					23080.	39	-8.79 -8.79	DFW			1177.0				35000.	0.00	0.00
TCA LGA	550	56.0	-3.00	727 6	23080.	39	-8.79	DFW	SFO	267	1177.0	0.00	707	7	41000.	0.00	0.00
DCA LGA		66.0	-3.00	727 6	23080.	39	-8.79	DEW		358	400.0				35000.	1.17	33.74
DCA MEM					35000.	.26	4.58	DFW		560	400.0				35000.	1.15	20.61
DCA OPD	563	455.0	42.00	727 7	35000.	5.48	96.16	DIM	608	218	469.0	0.00	727	7	3500V·	0.00	0.00
DCA OPD					35000.	5.48	96.16	DIW		374	469.0				41000. 35000.	0.00	0.00
DCA ORD					35000.	5.48	96.16	DTW			469.0				41000.	0.00	0.00

TABLE G.4 AMERICAN AIRLINES ENROUTE RNAV BENEFITS ANALYSIS Continued

ORG D		FLT	RANGE NMI	RNAV BENE	A/C	F	CRUISE	TIME BENE	FUEL BENE			DST A/P	FLT NO	RANGE NMI	RNAV BENE	A/C	F	CRUISE ALT	TIME BENE	FUEL BENE
DIW 8		92					30227.	5.47	160.29			JFK								-110.30
DIA L		604	1563.0				41000.	1.53	45.99			JFK						41000.		-61.31 -66.75
DIAL		41	1663.0	12.00	010	7	41000.	1.52	50.06	L	AX	OKC	382	936.0					.91	16.03
DIM F		380	351.0				35000.	.52	9.16			OND	178	1412.0				41000.	1.77	26.42 58.40
DIM L	GA	512	351.0	4.00	727	7	35000.	.52	9.16	L	AX	UNU	194	1412.0	14.00	010	7	41000.	1.77	54.40
DIM L		435	351.0				35000. 35000.	.52	9.16			080		1412.0					1.79	32.05
DIN L		530	351.0	4.00	727	7	35000.	.52	9.16			ORD		1412.0					1.79	53.65
DIM L		552	351.0				35000. 35000.	.52	9.16			PHL		1995.0					1.83	53.65
DTW L		618	351.0				35000.	.52	9.16	L	AX	PHA	630	208.0				32627.	0.00	0.00
DIN L		650	351.0				35000.	.52	9.16			PHA	54	208.0				32627.	0.00	0.00
DIN L	UPU	107	351.0				35000.	49	9.16			PHA	422	208.0				32627.	0.00	0.00
DIN C		159	101.0	-4.00	010	7	25880.	49	-22.59			STL		1304.0					1.57	27.47
DIM C		105					25880.	49	-16.05 -16.05			STL		1304.0					1.57	45.99
DIM	USO	505	101.0	-4.00	727	7	25880.	50	-11.13	L	AX	TUS	288	288.0	0.00	707	6	35920.	0.00	0.00
DIN C		553					25880.	49	-16.05			BUF	529					29320.	38	-7.58 -7.58
OTE C		435					25880.	50	-11.13		GA	RUF	355	146.0	-3.00	727	7	29320.	38	-7.58
DIM S			1736.0					2.30	68.98			BUF	531					29320.	38	-7.58 -7.58
FLP L		427	1736.0 508.0	0.00	727	7	35000.	0.30	0.00			BUF	419					29320.	38	-7.58
FLP S	TAT	532	342.0	0.00	727	7	35000.	0.00	0.00	L	GA	CLE	363	270.0	10.00	727	5	35000.	1.30	22.90
ELP T		69	145.0				29267.	0.00	0.00			CLE	239					35000.	1.30	22.90
ELP T	rus	325	145.0	v.00	727	7	29257.	0.00	0.00	Ĺ	GA	CLE	35	270.0	10.00	727	7	35000.	1.30	22.90
ENR L	*	317	60.0				41000.	0.00	0.00			DAY	417	105.0				35000.	1.70	29.76
F. Q C	SPU	449					41000.	2.17	65.15			DCA	440	105.0				26200.	.50	11.02
ERR C		513					41000.	2.15	70.92			DCA	463	105.0				26200.	.50	11.02
ENR C	050	591					41000.	2.17	65.15			DCA	369	105.0				26200.	.50	11.02
140 8	105	112	264.0	19.00	707	7	34960.	2.41	67.01	L	GA	UCA	173	105.0	4.00	727	7	26200.	.50	11.02
IAD E		378					34960. 25720.	37	-16.98			DCA	511	105.0				26200.	.50	11.02
IAD E		110					25720.	37	-12.10			DTW	539	338.0	1.00	727	5	35000.	.13	2.29
IAD		114					25800.	12	-4.02			DTW	519 593	338.0				35000.	.13	2.29
JEK H		10	1947.0				26280.	6.12	280.83			DTW	379	338.0				35000.	.13	2.29
JEK 6		118	106.0	50.00	707	7	26280.	6.13	198.11			DTW	155	338.0				35000.	.13	2.29
JEK B	325 0F#	81					26280.	6.13	38.32			DTW	505	338.0				35000.	.13	2.29
JFK D	DF#	95	1127.0	10.00	707	7	41000.	1.28	38.32	L	GA	DIA	591	338.0	1.00	727	6	35000.	.13	2.29
JFK I			2040.0				26120.	0.00	0.00			DTW	595 455	338.0				35000.	.13	2.29
JFK L			2040.0				37000.	0.00	0.00			DIW	447	338.0	1.00	727	7	35000.	.13	2.29
JFK L			2040.0				41000. 41000.	0.00	0.00			ORO	201					35000.	1.96	34.34
75 × 5			1731.0					2.43	72.81			080	235					35000. 35000.	1.96	34.34
JF × P			1781.0					2.43	72.81	L	GA	040	303	532.0	15.00	727	7	35000.	1.96	34.34
JEK S			2041.0					3.07	79.26			040	221					35000.	1.96	34.34
JFK S	FU	54	2165.0	20.00	707	7	41000.	3.32	99.63		GA	OPD	443	532.0	15.00	727	7	35000.	1.96	34.34
JFK S			2165.0					3.29	108.46			040	633					35000. 35000.	1.96	34.34
JFK S		151					41000.	.64	19.16			040	315					35000.	1.96	34.34
JFK S		37	0.180				41000.	3.29	20.86			090	391					35000.	1.96	34.34
LAX			1684.0					2.30	68.98			080	377					35000.	1.96	34.34
LAX C			1684.0					2.30	68.98	L	GA	SCH	537	121.0	7.00	727	6	27480.	.88	18.55
LAX D		155					35000.	1.17	20.61			ROC	337	121.0				27480.	.88	18.55
LAX D	Fw	324	991.0	4.00	707	7	41000.	1.15	34.49	L	GA	HOC	301	121.0	7.00	727	7	27480.	.88	18.55
LAX D		159	991.0				41000.	1.15	34.49			HOC	403					27480.	1.95	18.55
LAX (F .	314	991.0	9.00	707	7	41000.	1.15	34.49			SOF	247					35000.	1.96	34.34
LAX D		37A	991.0				41000.	1.14	37.54	L	SA	STL	533	675.0	5.00	727	7	35000.	.65	11.45
LAX D		330	991.0				41000.	1.15	34.49			STL	606	87.0				35000.	.65	8.60
LAX D		68	1619.0	15.00	707	7	41000.	1.92	57.48	**	EM	BNA	529	87.0	3.00	707	7	24760.	.37	12.39
LAX E			513.0					0.00	0.00			BOA	264	597.0				35000.	4.57	80.13
LAX E	-	A	2032.0	24.00	707	6	41000.	3.07	91.97			DCA	398		35.00	727	6	35000.	4.57	80.13
LAX I							41000.		91.97			OFW	495	295.0				35000.	1.17	20.61
LAX J							41000.		-66.75			DFW	283					35000.	1.17	20.61

TABLE G.4 AMERICAN AIRLINES ENROUTE RNAV BENEFITS ANALYSIS Continued

	DST A/P	FLT	RANGE NMI	RNAV BENE	A/C	F	CRUISE ALT	TIME BENE	FUEL BENE		DST A/P	FLT	RANGE NMI	RNAV		F	CRUISE	TIME	FUEL BENE
	OF A	173	295.0				35000.	1.17	20.61	ORD	SFO	265	1513.0	16.00	610	7	41000.	2.03	66.75
	LAX	353						1.70	29.76	090			1513.0					2.03	66.75
	LGA	488					35000.	.78	13.74	080			1513.0					2.05	61.31
MEM	LGA	122	757.0	6.00	727	7	35000.	.78	13.74	080	SLC		1007.0				35000.	1.17	20.61
	LAA	147					41000.	~. 38	-11.50	040			1007.0				35060.	1.17	20.61
	HOL	119					41000.	2.30	08.98	090		609	136.0				35000. 28680.	1.17	30.75
	BOL	310	598.0	18.00	727	7	35000.	2.35	41.21	CPC	SIL	493					23680.	1.52	30.75
	HOS	525					41000.	2.30	68.98	090		483					35000.	3.65	64.11
	805	300					41000.		-57.48	090	TUL	119					35000.	3.65	64.11
	805	550	655.0	-15.00	707	7	41000.	-1.92	-57.48	090	TUL	437	440.0	28.00	727	7	35000.	3.65	64.11
	805	404					41000.		-57.48 -57.48	040			1175.0					1.43	25.19
	BUF	254					35000.		89.29		TUS		11/5.0					1.41	42.15
	BUF	558					35000.	5.09	89.29	ORD			1175.0					1.43	25.19
	BUF	316					35000.	5.09	89.29	PHL		63	2026.0				35000.	2.94	11.45
	DCA	312					35000.	3.26	57.24	PHX		498	716.0				41000.	.64	19.10
	DCA	572					35000.	3.26	57.24	РНХ		622	715.0				35000.	.65	11.45
050	DC4	116					35000.	3.26	57.24	PHX		2346	716.0				35000.	.65	11.45
090		352					35000.	3.26	57.24	PHX			716.0					2.17	65.15
	DCA	638					J5000.	3.26	57.24	РНХ		118	1799.0	17.00	707	7	41000.	2.17	65.15
050	DCA	490					35000.	3.26	57.24	PHX		533	1799.0				32627.	.38	70.92
060	DCA	334	444.0	25.00	727	7	35000.	3.26	57.24	PHX	LAX	291	208.0				32627.	.37	14.73
	DFA	333					35000.	2.48	43.50	PMX		283	208.0				32627.	.39	7.23
	DF#	599					41000. 35000.	2.43	43.50	PHX			1163.0					1.39	45.89
090	CFA	141	0.559	19.00	727	7	35000.	2.48	43.50	онх			1163.0					1.41	42.15
	DFW	633					35000.	2.48	43.50	РНХ			1163.0					1.39	45.89
	DFA	301					35000.	2.48		2HA		366	192.0				31773.	0.00	0.00 4.58
	DFW	627	6.22.0	14.00	727	6	35000.	2.48	43.50	PHX	STL	134	1012.0	2.00	727	7	35000.	.26	4.58
	DIM	611	103.0				26040.	.12	72.91	ROC		384 450	116.0				27080.	0.00	0.00
	DT	374	103.0				26040.	.13	2.77	80C		538	110.0				27080.	0.00	0.00
	DTW	284	103.0				25040.	.13	2.77	ROC	LGA	26	116.0	0.00	727	7	27080.	0.00	0.00
	DIM	630	103.0				26040.	.12	2.77	HOC		367 667	366.0				35000. 35000.	.39	6.87
	01.	104	103.0				26040.	.12	5.64	ROC		301	366.0				35000.	.39	6.87
	DTW	54	103.0				26040.	.12	3.99	SAN		126	-0.0	0.00	707	7	12600.	0.00	0.00
	DTW	202	103.0				26040.	.12	2.77	SAN			1440.0					1.96	57.48
	EWH	196					41000.	38	-11.50	SAN			1440.0					1.96	34.34
	EWH	418					41000.	3A	-11.50	SAN		464	192.0				31773.	0.00	0.00
	EWH	234					41000. 35000.	39	-12.51	SAT		554	125.0	5.00	727	7	35000.	.85	11.43
	LAX	181	1404.0	10.00	010	7	41000.	1.27	41.72	SJF	LGA	298	508.0	24.00	727	7	35000.	3.13	54.45
	LAX		1409.0					1.27	41.72		HEM	250					35000.	2.19	54.95
	LAX		1409.0					1.28	38.32		OF #		1207.0					1.57	27.47
080	LAX	435	1409.0	10.00	727	7	35000.	1.30	22.90	SFO			1207.0					1.52	50.06
	LGA	236	519.0				35000.	.13	2.29	SFO	OF #		1207.0					1.57	50.06
	LGA	305	519.0	1.00	727	5	35000.	.13	2.29	SFO	DFW	84	1207.0	12.00	707	7	41000.	1.53	45.99
	LGA	446	519.0				35000.	.13	2.29	SFO			1207.0					1.57	27.47
	LGA	359	519.0				35000.	.13	5.29	SFO SFO	JEK		2129.0					3.29	108.46
	LGA	304	519.0	1.00	727	6	35000.	.13	5.29	SFO		56	2124.0	26.00	101	6	41000.	3.35	. 99.63
	LGA	33A	519.0	1.00	727	7	35000.	.13	2.29	SFO			1498.0					2.05	61.31
	LGA	166	519.0				35000.	.13	2.29	SFO			1498.0					2.03	66.75
040	LGA	154	519.0	1.00	727	7	35000.	.13	2.29	SFO	PHX	366	490.0	32.00	727	7	35000.	4.17	73.27
	LGA	344	519.0	1.00	727	6 7	35000.	.13	2.29	SFO		134					35000.	4.17	73.27
	PHA		1204.0					1.39	45.89	550		246					41000.	4.05	133.49
040	PHX	291	1204.0	11.00	D10	7	41000.	1.39	45.89	SLC	090	208	991.0	5.00	727	7	35000.	.65	11.45
	PHX	157	1204.0	11.00	010	7	41000	1.43	25.19 45.89	SLC		538	991.0				35000.	.65	11.45
	30C	368	386.0	23.00	727	7	35000.	3.00	52.66	SLC		686	991.0	5.00	727	1	35000.	.05	11.45
040	SOC	482	386.0	23.00	707	7	39272.	2.94	85.56	STL	CLE	412	358.0	20.00	707	7	38376.	2.56	73.24
	ROC	402					35000.	3.00	52.66	STL		366					35000.	2.61	45.79
	SAN		1400.0	3.00	727	7	35000.	.39	6.87	STL	DFW	4.75	309.0	1.00	707	7	39688.	.13	3.75
	SAN		1400.0	3.00	707	7	41000.	.38	6.87	STL		349	399.0				39688.	.13	3.75
	SAN		1400.0				35000.	.39	11.50	STL		491	344.0				35000.	.13	2.29
	-	-																	

TABLE G.4 AMERICAN AIRLINES ENROUTE RNAV BENEFITS ANALYSIS Continued

		DST A/P	FLT NO	RANGE NM1	RNAV BENE	A/C	F	CRUISE	TIME	FUEL BENE	ORG A/P		RANGE NMI	RNAV BENE	CRUISE	TIME BENE	FUEL
5	TL	JFK	152	703.0	13.00	707	1	41000.	1.66	49.82							
5	TL	LAX	241	1273.0	7.00	727	7	35000.	.91	16.03							
S	TL	LAX	353	1273.0	7.00	707	7	41000.	.89	26.82							
S	TL	LAX	37	1273.0	7.00	010	7	41000.	.89	29.20							
5	TL	LGA	106	692.0	2.00	727	7	35000.	.26	4.58							
S	TL	LGA	598	642.0				35000.	.26	4.58							
5	TL	040	35A	120.0	-2.00	727	7	27400.	25	-5.31							
S	TL	050	26	120.0	-2.00	727	7	27400.	25	-5.31							
5	TL	040	200					27400.	25	-5.31							
		AHG		1010.0				41000.	. 26	7.66							
		PHX		1010.0				35000.	.26	4.58							
		TUL	233					330HU.	0.00	0.00							
		TUL	473	217.0				33080.	0.00	0.00							
		UTH	67	241.0				34040.	.38	10.52							
		050	277	432.0				35000.	.39	6.87							
		050	197	432.0				40744.	.38	12.61							
		030	503	432.0				35000.	.39	6.87							
		DF	672	122.0				27560.	0.00	0.00							
		DFW	467	155.0				27560.	0.00	0.00							
		090	446					35000.	1.30	22.90							
		050	338					35000.	1.30	55.90							
		040	490					35000.	1.30	55.40							
		040	179					40296.	1.28	37.86							
		SIL	345	218.0				33120.	.26	4.78							
1	UL	STL	484	218.0				33120.	.25	6.97							
		ELP	336	144.0				29450.	.62	18.43							
		ELP	288	149.0				29480.	.65	18.43							
		050		1378.0					1.41	42.15							
		CFO		1378.0					1.43	25.19							
T	US	050		1378.0					1.43	25.19							
1	US	ORU	236	1378.0	11.00	727	7	35000.	1.43	25.19							

TABLE G.5 UNITED AIRLINES ENROUTE RNAV BENEFITS ANALYSIS

ORG DST A/P A/P		RANGE NMI	RNAV BENE		F	CRUISE ALT	TIME BENE	FUEL BENE	ORG A/P		FLT	RANGE NMI	RNAV BENE	A/C	F	CRUISE ALT	TIME BENE	FUEL BENE
ARE PIT	765	140.0		727	7	29000.	.64	12.69	CLE	805	732	404.0	-1.00	737	7	34640.	14	-1.59
ARE PIT	935 478	140.0	5.00			29000.	.56	6.30	CLE		642					31987. 31987.	4.82	58.29
ATL BUF	836	535.0				35000.	.52	9.16	CLE		665	202.0	35.00	737	1	31987.	4.82	58.29
ATL HUF	835	535.0	4.00	727	1	35000.	.52	9.16	CLE	DCA	722	202.0	35.00	737	7	31987.	4.82	58.29
ATL CLE	396					35000.	2.51	47.50	CLE		905					31947.	4.62	7.72
ATL CLE	476					35000.	2.51	47.50	CLE		434					34760.	.26	4.60
ATL CLE	888					35000.	2.51	47.50	CLE		572	254.0	2.00	727	7	347h0.	.26	4.60
ATL CLE	11115					35000.	77	43.50 -16.56	CLE		134	259.0				34700.	.26	3.25
ATL LGA	368					35000.	.52	9.16	CLE		850	259.0				34760.	.26	5.03
ATL MIA	575					35000.	5.35	93.87			1154	259.0				34760.	.26	5.03
ATL PSI	763					35000.		-89.29	CLE		182	267.0				35080.	1.13	43.03
ATL PBI	763	309.0	-39.00	737	1	34490.	-5.43	-61.99	CLE	JFK	804	267.0	9.00	737	7	33270.		14.62
ATL PIT	384					34360.	.70	7.97	CLE							33270.	2.30	14.62
ATL PIT	440					35000. 35000.	.65	11.45	CLE			1668.0				41000.	2.32	75.63
ATL PIT	470	376.0	5.00	727	7	35000.	.65	11.45	CLE		326	256.0	-7.00	737	7	13180.	47	-11.39
ATL PIT	672					34360.	.70	7.97	CLE		386					33180. 33180.	97	-11.39
ATL 1PA	475					33370.	1.52	52.39	CLE		394					33180.		-11.39
HAL CLE	697	188.0	26.00	727	7	31560.	3.34	63.62	CLE	LGA	712	258.0	-7.00	737	7	33100.	97	-11.39
HAL CLE	717					31560.	3.28	43.50	CLE		840					34720.	91	-16.13
HAL DIW	805					35000.	2.48	43.50	CLE		918					331HU.	97	-11.39
RAL LAA		1924.0	21.00	DCA	7	4100U.	2.70	88.23	CLE	LGA	915	258.0	-7.00	737	6	33180.	97	-11.39
HAL ORD	307					35000.	2.83	50.37	CLE		1232					32413.	88	15.42
HAL ONU	741	436.0	22.00	727	7	35000.	2.87	50.37	CLE		701					32413.	.51	15.42
HAL ORD	793					35000.	2.87	50.37	CLE		101					31400.	1.01	31.36
BOL ORD	103					35000.	25	50.37 -8.34	CLE			185.0					.99	40.13
BOL ORU	121					41000.	26	-9.18	CLE			185.0				31400.	.99	40.13
HOL OHD	135					41000.	26	-9.18 -4.58	CLE			185.0				31400.	.98	57.28
BDL ORD	343					35000.	26	-4.58	CLE		1401	185.0				31400.	1.01	31.36
POL ORD	789	589.0	-2.00	085	7	41000.	26	-9.18	CLE	PHL	364	237.0	7.00	725	7	33880.	.92	17.95
HOI PDA	587 761	216.0				33040.	.92	18.30	CLE		406	237.0				32920.	.97	16.44
HOI SLC	248	164.0				30280.	0.00	0.00	CLE		940	237.0				33880.	.42	17.95
HOI SEC	278	164.0	0.00	727	7	30280.	0.00	0.00	CHH	DCA	668	179.0	4.00	737	5	31373.	.55	6.74
BOI SEC	374 868	164.0				30280.	0.00	0.00	CHH		66A					31373.	.55	6.74
ROS CLE	301	418.0	10.00	DC8	7	40296.	2.06	66.42	DCA		898					30493.	2.61	54.68
HOS CLE	317					35000.	2.09	36.63	DCA		235					31613.	3.57	69.48
HOS CLE	769					34780.	2.23	25.30 36.63	DCA		357					31560.	3.38	59.48
BOS ORD	121	660.0	-4.00	280	1	41000.	51	-18.37	DCA	CLE	455	185.0	20.00	737	7	31613.	3.57	43.62
ROS ORD	223					41000.	51	-16.81	DCA		779 285					31613.	3.57	44.85
AOS ORD	237					41000.	51	-16.69	DCA		847					32573.	3.48	65.09
805 030	471					41000.	51	-18.37			1285					32120.	3.72	44.85
AOS ORD	489	2299.0				41000.	3.58	128.59	DCA		493					32091.	2.89	34.90
AUF ATL	473	551.0	4.00	737	7	350UO.	.56	6.30	DCA	DAY	727	200.0	21.00	737	1	32093.	2.89	34.40
AUF ATL	833	551.0				35000.	.52	9.16	DCA		221					35000.		43.50
HUF ATL	139	314.0				35000.	0.00	0.00	DCA		669					33260.	.78	13.74
AUF GAD	659	314.0	0.00	737	7	33740.	0.00	0.00	DCA	080	207	455.0	42.00	727	7	35000.	5.48	96.16
AUF ORD	H31	314.0				35000.	4.89	92.77	DCA		247					35000.		105.01
CAK ORD	620					32627.	6.76	80.56	DCA							35000.		105.01
CAK OHD	481					33440.	0.35	116.24	DCA	ORU	299	455.0	42.00	125	7	35000.	5.54	105.01
CAK ORD						32627.		80.56		OHU	327					35000.	5.48	96.16
CLE ATL	207	414.0	00.55	725	7	35000.	2.90	55.00	OCA	0.0	667	455.0	42.00	727	1	35000.	5.48	96.16
CLE ATL							2.87	50.37	DCA		644					27240.		0.00
CLE ATL							3.06	94.15	DCA	PIT	787					27240.		0.00
CLE ATL	941	414.0	22.00	727	7	35000.	2.87	50.37	OCA	TYS	593	287.0	2.00	737	7	33470.	85.	3.24
CLE ATL		414.0	22.00	727	7	35000.	2.87	50.37	DCA			267.0					.26	4.58
CLE HAL		165.0	1.00	727	7	30493.	.13	3.98	DEN							35000. 35000.		-12.50
CLE BOL	740	348.0	10.00	727	7	35000.	1.30	22.90	DEN	DTw	23R	913.0	4.00	010	7	41000.	.51	16.69
CLE HOL						35000.	1.30	22.90	DEN			913.0				35000.	.52	25.19
CLE BOS							13	-2.29								35000.		25.19
CLE HOS	584	404.0	-1.00	727	7	35000.	13	-5.59	DEN	IAO	935	1214.0	11.00	085	7	41000.	1.40	50.52

TABLE G.5 UNITED AIRLINES ENROUTE RNAV BENEFITS ANALYSIS Continued

ORG DST A/P A/P	FLT NO	RANGE NMI	RNAV BENE	A/C	F	CRUISE ALT	TIME BENE	FUEL BENE		DST A/P	FLT NO	RANGE NMI	RNAV BENE	A/C	F	CRUISE ALT	TIME BENE	FUEL BENE
DEN JFK	160	1323.0	6.00	Des	7	41000.	.77	27.55	EMR		747	268.0				33280.	1.11	12.99
DEN JEK		1323.0				41000.	.76	25.03	EWR	CLE	1153	301.0				35000.	1.06	20.00
DEN JEK		1323.0				41000.	.77	27.55		650	657	301.0				33610.	.55	6.46
DEN LAS	249	451.0				41000.	.63	20.85		650	781	301.0				35000.	.52	9.16
DEN LAS	305	451.0				41000.	.64	21.01		LAX	11	2034.0				41000.	0.00	0.00
DEN LAS	753 855	451.0				41000.	.65	21.01		ORD	147					41000.	2.15	70.92
DEN LAS	465	451.0				35000.	.65	11.45	EWR		201	531.0	17.00	727	7	35000.	2.22	38.92
DEN LAX	193	630.0				37000.	.38	20.68		ORU	225					41000.	2.15	70.92
DEN LAA	701	636.0				41000.	.38	12.51	EWR	040	543					41000.	2.24	70.92
DEN LAA	726	630.0				35000.	.39	6.87	EWR	ORU	1199					41000.	2.19	71.42
DEN LAK	799	636.0				35000.	.39	6.87	EWR			2158.0					3.45	124.00
DEN CHA	208	338.0				35000. 35000.	.39	6.87	GEG		529	584.0				35000.	.52	9.16
DEN OMA	340	338.0				35000.	.65	11.45	GEG		809					35000.	.52	9.16
DEN OMA	444	338.0				35000.	.65	11.45	650		298					33640.	1.66	19.36
DEN OTA	130	336.0				41000.	.66	16.81	650 650		780					33640.	1.57	19.36
DEN VIV	236	688.0				41000.	.51	16.69	650		78H					33640.	1.66	19.36
DEN ORU	240	688.0				35000.	.52	9.16	IAD			1220.0					1.53	55.11
DEN OPD	282	688.0				41000.	.51	18.37	IAD			1220.0					1.57	50.42
DEN OND	492	56H.0				41000.	.51	16.09		LAX		1947.0					2.78	91.78
DEN 040	086	640.0				41000.	.51	16.81		LAX		1947.0					2.81	101.03
DEN ORD	916	688.0				35000. 35000.	.56	6.30	IAD	040	673					35000.	5.48	96.16 66.15
DEN ORD		688.0				35000.	.52	9.16	IAD	030	673	455.0	42.00	737	1	35000.	5.86	66.15
DEN DAD		688.0				41000.	.51	16.69	IAD			2159.0					2.94	105.63
DEN ORD		688.0				35000.	.52	9.16	JAX		396	148.0					2.94	5.51
DEN POA	167	776.0				41000.	.64	27.96	JFK	305	644	106.0	50.00	085	7	25280.	6.13	237.41
DEN POA	743	776.0				41000.	.64	21.01	JFK		773					33330.	1.80	21.09
DEN FOX	177	776.0				35000. 41000.	.65	25.21			1233	273.0					1.64	62.01
DEN SEA	177	808.0				41000.	.77	25.21	JFK	DEN		1406.0					1.65	54.23
DEN SEA	379	720.0				41000.	.76	25.03		DEN		1406.0					1.67	54.62
DEN SFO	393	726.0				35000.	.52	9.16	JFK			1888.0					2.70	88.23
DEN SFO	647	726.0	4.00	725	7	35000.	.53	10.00	JFK			2040.0				37000.	0.00	0.00
DEN SFO	869	726.0				35000. 35000.	.52	9.16	JFK		119	558.0				41000.	.51	16.81
DEN SFO	463	726.0				35000.	.56	10.00		040	159	550.0				41000.	.51	16.81
DEN SEO		726.0				35000.	.52	9.16	JFK		229	558.0				41000.	.51	18.37
DEN SEU	151	245.0				41000. 3420U.	.51	7.70		SEA	239	2011.0				35000.	2.91	95.95
DEN SLC	173	245.0				34200.	.25	7.70	JFK	SFO	25	2165.0	20.00	010	7	41000.	3.29	108.45
DEN SEC	761	245.0				34200.	.26	4.66		SFU		2165.0					3.29	108.46
DEN SLC	793	245.0				34200. 35000.	1.17	20.61	LAS		238	458.0				35000.	.76	25.03
DSM 090	252	161.0				30120.	0.00	0.00	LAS		468	458.0				35000.	.78	13.74
D2W 0KD	272	161.0				30120.	0.00	0.00	LAS		686	458.0				41000.	.77	25.21
05M 0PU	412	161.0				30120.	0.00	0.00	LAS	DEN	962	458.0				35000.	.77	25.21
DSM ORU	662	161.0	0.00	725	7	30120.	0.00	0.00	LAS	DEN	1148	450.0	6.00	727	7	35000.	.78	13.74
DIM BAL	284	161.0				30120.	0.00	0.00	LAS		229	1852.0				20120.	1.29	9.54
DIW BAL	765	248.0				34320. 34320.	0.00	0.00	LAS		345	104.0				26120.	.25	11.26
OTW DCA	442	285.0	37.00	737	7	33450.	5.12	59.90	LAS	LAX	571	104.0	2.00	DCA	5	26120.	.25	8.73
DIN DEN	752					33450. 41000.	5.12	59.90	LAS		857	104.0				20120.	.25	5.52
DIN DEN	311					35000.	66	-12.50	LAS			1233.0					1.54	50.42
DTW LAA		1663.0	12.00	010	7	41000.	1.52	50.06	LAS	080	214	1233.0	12.00	747	7	37000.	1.50	82.73
DIM OND	203	101.0				25880. 25880.	51	-12.16	LAS			1233.0					1.52	50.06
OTW CHO	251	101.0	-4.00	737	7	27880.	54	-7.34	LAX			1407.0					2.83	92.43
090 WTO	707					27880.	54	-7.34	LAX	CLE	7-	:684.0	10.00	DAS	7	41000.	2.30	82.66
DIM PHL	460					35000.	1.43	25.19	LAX			63/.0					.88	17.50
DIN PHL	014	350.0	11.00	UCB	7	37160.	1.42	43.21	LAX			63/.0					.89	29.20
DTW PHL	668					37160.	1.42	43.21	LAA	DEN	574	637.0	7.00	127	7	35000.	.91	16.03
DT# SFU		1736.0					2.28	-5.75 75.09	LAX			1156.0					1.90	62.57
EWR CLE	187	200.0	6.00	DCB	4	35120.	1.02	30.97	LAX			0.5605						25.011
EMP CLE	305					35120.	1.02	30.97	LAA			1913.0						100.12
EWH CLE	403	268.0	8.00	737	7	33240.	1.11	12.99	LAX			2032.0						-66.75
ENR CLE			8.00	727	7	35000.	1.04	18.32	LAX	JFK	A	2032.0-	10.00	747	7	37000.	-2.00	-110.30
EWR CLE	457	200.0	0.00	008	,	33120.	1.02	30.97	LAX	LAS	150	103.0	3.00	008	5	26040.	.37	13.13

TABLE G.5 UNITED AIRLINES ENROUTE RNAV BENEFITS ANALYSIS Continued

	RANGE RNAV NMI BENE		ALT	BENE	BENE		DST A/P		RANGE NMI	BENE	A/C I	ALT	BENE	BENE
		010 7		.37	16.91		000		190.0			31667.	0.00	0.00
		727 7		.38	8.31		040		190.0			31667.	0.00	0.00
		DCA 7		.37	13.13		DEN		3+0.0			35000.	.65	11.45
		725 7		1.19	22.50		DEN	047	340.0	5.00	725	35000.	.66	12.50
		127 1		1.17	20.61		DEN		340.0			35000.	.05	11.45
	412.0 14.00			1.75	96.52		DEN		340.0			35000.	.65	11.45
LAX 040 104 1	412.0 14.00	010 7	41000.	1.77	5A.40		DEN		340.0			37800.	.64	21.72
	412.0 14.00			1.77	58.40 58.82		UHU		264.0			34960.	.26	5.01
	412.0 14.00			1.75	96.52		ORU		264.0			34960.	.26	5.01
CAX FOX 202	660.0 5.00	725 7	35000.	.66	12.50		040		264.0			34960.	. 45	4.45
		727 7		.64	21.01		ORD		264.0			34960.	.26	5.01
		727 7		.65	11.45		ORD		264.0			34960.	.26	3.25
LAX PDA 598	660.U 5.00	727 7	35000.	.05	11.45	AMO	ORD	980	264.0	2.00	737	33240.	.28	3.25
	761.0 6.00			1.40	58.82		HAL					35000.	3.78	66.40
		727 7		.78	13.74		BAL					35000.	3.78	66.40
	761.0 6.00	727 7	35000.	.78	13.74		BAL	834	447.0	29.00	727	35000.	3.78	66.40
		727 7		.78	13.74		HAL					35000.	3.78	56.40
		D10 7		.76	25.03		BOL					41000.	2.28	75.09
		085 7		.77	27.55		BUL					41000.		75.63
	199.0 17.00			2.13	71.97		HUL					41600.		82.66
	199.0 17.60			2.13	71.97		805					41000.		-63.02
	199.0 17.00			2.19	41.24		805					41000.		-63.02
	199.0 17.00			2.34	28.36		905					41000-		-63.02
	199.0 17.00			2.19	41.24		BUF					7 41000.		62.43
LAX 5FO 508	199.0 17.00			2.34	28.36		BUF					7 34040.	5.42	62.43
	199.0 17.00			2.19	41.24		BUF					7 34090.		62.43
	199.0 17.00			2.19	41.24		CAK					7 32733.		73.83
LAX SFO 526	199.0 17.00	737 7	31907.	2.34	28.36	090	CAK	724	230.0	45.00	737	7 32733.	6.21	73.83
	199.0 17.00			2.15	65.84		CAK	876				33600.	5.83	106.37
	199.0 17.00			2.19	28.36		CLE					31187.	25	-8.60
	149.0 17.00			2.19	41.24	080	CLE	218	161.0	-6.00	741	311+7.	24	-14.38
	199.0 17.00			2.11	84.18		CLE					31187.	25	-10.07
	199.0 17.00			2.22	45.04		CLE					31187.	26	-4.92
	199.0 17.00			2.09	119.94		CLE	316	101.0	-2.00	010	31187.	25	-10.07
	560.0 4.00			.52	9.16		CLE					31187.	26	-5.37 -5.37
	270.0 10.00			1.26	47.76							7 31187.	25	-10.07
LGA CLE 411	270.0 10.00	727 7	35000.	1.30	22.90		DCA					7 35000.	3.26	57.24
	270.0 16.00			1.38	16.23		DCA					7 35000.	3.26	62.50
	270.0 10.00			1.38	16.23		DCA					35000.	3.49	39.38
LGA CLE HOS .	270.0 10.00	737 7	33300.	1.38	16.23		CCA					35000.	3.49	39.38
	270.0 10.00			1.38	16.23		DCA					7 35000.	3.30	62.50
	270.0 10.00			1.38	16.23		DCA					35000.	3.30	62.50
LGA 0PD 903	532.0 15.00	727 7	35000.	1.96	34.34		DCA					35000.	3.49	39.38
	532.0 15.00 532.0 15.00			1.98	37.50		DEN		683.0			7 41000.	.53	16.69
	532.0 15.00			1.96	34.34	ORD	DEN	237	683.0	4.00	D10 7	7 41000.	.51	16.69
LGA UPD 929	532.0 15.00	727 7	35000.	1.96	34.34		DEN	263	683.0			7 35000.	.53	10.00
	379.0 13.00			1.70	29.16		DEN		683.0			7 41000.	.51	18.37
	453.0 23.00			3.00	29.76	090	DEN	1407	683.0	4.00	727	7 35000.	.52	9.16
MIA PIT 26H	411.6 5.00	727 7	35000.	. 78	13.74				683.0			7 41000.	.51	10.09
#IA PIT 578	811.0 6.00			.76	25.03			1417	683.0	4.00	727	7 35000.	.56	9.16
	90.0 -4.00			50	13.74	040	DEN	4007	683.0	4.00	DC8	41000.	.51	16.61
MKE CLE 306	207.0 8.00	727 7 3	32573.	1.03	19.29			4009				7 41000.	.51	9.16
	207.0 8.00 733.0 6.00	727 7 3		.76	25.03				159.0			7 35000.	.52	5.47
MKE DEN 243		DC8 7 4		.77	25.21	ORD	DSM	497	159.0	2.00	725	7 30013.	.26	5.47
MKE DEN 701	733.0 6.00	DC8 7 4	41000.	.77	25.21		DSM		159.0			30013.	.26	5.01
		727 7 3		.77	27.55		DSM		159.0			7 30760.	.26	3.42
MSP 0WD 270		727 5		0.00	0.00	ORD	DSM	835	159.0	2.00	725	7 30013.	.26	5.47
45P 0RD 464	190.0 0.00	727 7 3	31667.	0.00	0.00		DSM		159.0			7 30013.	.26	3.02
MSP 020 484		727 7 3			0.00		DTW		103.0	1.00	737	7 28040.		1.83
MSP 0RD 716		727 7 3	31667.	0.00	0.00	0.30	DTM		103.0			7 25040.	-14	1.83

TABLE G.5 UNITED AIRLINES ENROUTE RNAY BENEFITS ANALYSIS Continued

ORG DST FLT RANGE							TIME	FUEL
A/P A/P NO NMI	BENE ALT	BENE BENE			BENE	ALT	BENE	BENE
040 DT# 704 103.0		.13 2.77	POX HO				.25	4.79
020 07# 882 103.0 020 6## 136 516.0	1.00 727 7 26040.	38 -13.78				7 35000.	.52	9.16
080 ENN 142 516.0	-3.00 010 7 41000.	38 -12.51	POX LA	x 487 639.0	4.00 DC8	7 41000.	.51	16.81
	-3.00 DC8 7 41000.	39 -12.60 38 -12.51					.52	9.16
040 ENH 442 510.0	-3.00 010 7 41000.	38 -12.51	AD XON	K 369 399.0	21.00 727	7 35000.	2.74	48.08
040 ENR 1198 516.0	-3.00 DC8 7 41000. 0.00 DC8 2 41000.	0.00 0.00			21.00 727	7 35000.	2.74	48.08 52.50
040 JFK 120 514.0		0.00 0.00			-7.00 D13		90	-29.20
020 JFK 685 518.0	0.00 DC8 6 41000.	0.00 0.00	POX CR	0 157 1415.0	-7.00 OHS	7 41000.	89	-32.15
080 JFK 666 518.0		0.00 0.00			21.00 725		2.77	33.32
ORD LAS 229 1255.0	12.00 De5 7 41000.	1.53 55.11	POX SF		21.00 UCB		2.70	86.27
	12.00 010 7 41000.	1.52 50.06 1.50 82.73	POX SF	0 523 399.0	21.00 737	7 34590.	2.92	33.32
	12.00 727 7 35000. 10.00 DC6 7 41000.	1.57 27.47			21.00 737 53.00 727		6.85	33.32
OND LAA 103 1464.0	10.00 010 7 41000.	1.27 41.72	PHL CL	E 363 230.0	-1.00 737	7 32733.	14	-1.64
	10.00 747 7 37000.	1.25 68.94			-1.00 725		13	-2.58
ORD LAX 115 1409.0	10.00 010 7 41000.	1.27 41.72			-1.00 DC8		1.85	-3.83 35.00
040 LG4 900 519.0 040 LG4 904 519.0	1.00 725 7 35000.	.13 2.29 .13 2.50	PHL DT	w 571 326.0	14.00 DC8	5 37352.	1.80	55.19
040 LGA 910 519.0 040 LGA 914 519.0		.13 2.29	PHL DI		14.00 DCB		1.80	55.19
080 LGA 922 519.0	1.00 727 7 35000.	.13 2.29	PHL LA	A 99 2025.0	23.00 008	7 41000.	2.96	96.63
	23.00 727 7 32840.	2.97 55.23		0 133 509.0	16.00 010	7 41000.	2.03	73.48
	23.00 727 1 32840.	2.97 55.23 3.01 60.31			16.00 725		2.11	40.00
040 MSP 661 212.0	23.00 727 7 32840.	2.97 55.23	PHL UR	0 841 564.0	10.00 727	7 35000.	2.09	36.63
	23.00 727 7 32840.	2.97 55.23			25.00 DHS 3.00 727		.38	7.77
	16.00 DAS 7 41000. 16.00 727 7 35000.	2.04 73.48			19.00 737		2.64	30.21
	-4.00 725 7 34920.	53 -10.02	PIT AT	L 589 388.0	19.00 737	7 34480.	2.64	-30.21
	-4.00 725 7 34920. -4.00 DRS 7 34920.	53 -10.02 51 -16.90			19.00 737		.95	12.77
040 UMA 471 263.0	-4.00 DAS 7 34920.	51 -16.90	PIT DC	A 638 104.0	7.00 737	7 25120.	.95	12.77
	-4.00 725 7 34920. -4.00 727 7 34920.	53 -10.02		A 960 104.0			.88	19.33
	-4.00 725 7 34920. 12.00 DC8 7 41000.	53 -10.02 1.54 50.42			17.00 DBS		2.17	78.07
ORD PDX 159 1423.0	12.00 DC8 7 41000.	1.54 50.42	PIT MI	A 315 809.0	6.00 727	/ 35000.	.78	13.74
	28.00 010 7 41000.	3.58 128.59	PIIMI	0.60P 4211 V	6.00 010	7 41000.	.76	25.03
	28.00 DCA 7 41000. 28.00 725 7 35000.	3.60 117.64			6.00 727		1.24	13.74
OHD PHL 786 526.0	28.00 010 7 41000.	3.54 116.81	PIT OR	0 143 267.0	10.00 085	7 35080.	1.27	42.30
	-2.00 747 7 34960. -2.00 DBS 7 34960.	25 -13.83 25 -A.45			10.00 727		1.30	42.30
080 PIT 404 264.0	-2.00 725 7 34960.	26 -5.01	PIT OR		10.00 737		1.38	22.90
090 PIT 776 264.0	-2.00 727 7 34960. -2.00 D8S 7 34960.	26 -4.58 25 -8.45	RNO LA	x 223 259.0	13.00 DCA	7 34760.	1.66	50.21
	23.00 727 7 35000.	3.00 52.66			42.00 727			126.67
040 SAN 225 1400.0	3.00 DIO 7 41000.	.38 12.51	RNO SF		42.00 727		5.29	116.00
	3.00 727 7 35000. 14.00 085 7 41000.	1.19 64.29	ROC OR	785 366.0	3.00 737	7 34260.	.42	4.79
ORD SEA 147 1415.0	14.00 DIO 7 41000. 14.00 DIO 7 41000.	1.77 58.40 1.77 58.40			3.00 727 15.00 DC8		1.93	63.02
OND SEA 157 1415.0	14.00 DIO 7 41000.	1.77 58.40	SAN OR	210 1440.0	15.00 DCH	7 41000.	1.93	63.02
	16.00 DBS 6 41000.	2.04 73.48	SAN SF	0.HPS 28H.0	0.00 727	7 35000.	0.00	0.00
040 SFO 123 1513.0	16.00 010 7 41000.	2.03 66.75	SEA DE		6.00 DCH		.76	25.03
ORD SFO 135 1513.0	15.00 DAS 7 41000.	2.00 110.30	SEA LA	337 705.0	0.00 010	7 41000.	0.00	0.00
040 SLC 247 1007.0	9.00 725 7 35000.	1.19 22.50	SEA LA	353 765.0	0.00 T25	7 35000.	0.00	0.00
OHD SLC 375 1067.0	9.00 725 7 35000.	1.19 22.50	SEA LA		0.00 725	7 35000.	0.00	0.00
040 SLC 489 1007.0	9.00 0CA 7 410no. 9.00 727 7 35000.	1.16 37.81	SEA LA	383 765.0	0.00 727	7 35000.	0.00	0.00
040 SLC 4075 1007.0	9.00 727 7 35000. 0.00 727 6 35000.	1.17 20.61	SEA ORI		4.00 DHS		.51	18.37
PHI ATL 440 371.0	0.00 727 1 35000.	0.00 0.00	SEA ORI	150 1404.0	4.00 DAS		.51	18.37
PSI ATL 600 341.0	0.00 737 7 34510.	0.00 0.00	SEA ORI	134 1404.0	4.00 010	1 41000.	.51	16.69

TABLE G.5 UNITED AIRLINES ENROUTE RNAY BENEFITS ANALYSIS Continued

ORG DST A/P A/P	FLT	RANGE NMI	RNAV BENE	A/C	F	CRUISE ALT	TIME BENE	FUEL BENE
SEA DAD	154	1404.0	4.00	010	6	41000.	.51	16.69
SEA SFO	245	50/.0	19.00	727	7	15000.	2.48	43.50
SEA SFO	257	507.0	14.00	121	5	35000.	2.48	43.50
SEA SFO	275	507.0	19.00	121	7	35000.	2.48	43.50
SEA SFO	339	507.0	14.00	121	7	35000.	2.48	43.50
SEA SEO	352	507.0	19.00	121	6	35000.	2.48	43.50
SEA SFO	345	507.0	19.00	151	7	35000.	2.4A	43.50
SEA SEO	840	507.0	19.00	727	6	15000.	2.4R	43.50
SEA SFO	11/7	501.0	19.00	727	7	35000.	2.48	43.50
SFO BOS	256	2275.0	-2.00	727	7	35000.	3.10	133.18
SFO DEN	262	726.0	-2.00	DC8	7	41000.	26	-8.40
SFO DEN	282	726.0	-2.00	010	7	41000.	25	-8.34
SFO DEN	730	726.0	-2.00	725	7	35000.	26	-5.00
SFO DEN	4000	726.0	-2.00	DCB	7	41000.	26	-4.58
SFO ENH	34	2320.0	26.00	085	7	41000.	3.32	419.40
SFO GEG	504 756	545.0	4.00	727	7	35000.	.52	9.16
SFO TAD	50	2010.0	7.00	727	7	41000.	.52	9.16
SFO IAD	56	2010.0	7.00	OHS	7	41000.	.89	32.15
SFO JFK	22	2129.0	26.00	016	7	41000.	3.29	108.46
SFO JFK	64	2129.0	26.00	010	7	31240.	3.29	108.46
SFO LAX	283	182.0	3.00	725	7	31240.	.39	8.05
SFO LAK	352	182.0	3.00	727	6	31240.	.39	7.38
SFO LAA	352	182.0	3.00	727	1	31240.	.39	7.38
SFO LAK	507	152.0	3.00	777	7	31240.	.39	7.38
SFO LAK	519	142.0	3.00	737	7	31453.	.41	5.05
SFO LAX	523	182.0	3.00	737	7	31453.	.41	5.05
SFO LAX	525 527	182.0	3.00	280	7	31240.	.37	12.89
SFO LAX	529	182.0	3.00	727	7	31240.	.39	7.38
SFO LAX	533	182.0	3.00	737	7	31453.	.41	5.05
SFO LAK	010	182.0	3.00	727	7	31240.	.39	7.38
SFO LAX	700	182.0	3.00	727	7	31240.	.39	7.38
SFO LAX	782	182.0	3.00	727	7	31240.	.39	7.38
SFO LAX	896	182.0	3.00	727	6	31240.	.39	7.38
SFO LAX	1181	182.0	3.00	010	7	31240.	.37	15.09
SFO LAX	1185	182.0	3.00	747	7	31240.	.37	21.55
SFO LAA	1189	182.0	3.00	727	7	31240.	.39	7.38
SF0 040	126	1448.0	16.00	747	7	37000.	2.00	110.30
SFO ORD	129	1498.0	16.00	010	7	41000.	2.03	73.48
SFO OPD	136	1498.0	16.00	Das	7	41000.	2.04	73.45
SFO PUX	462	380.0	1.00	121	7	35000.	.13	2.29
SFO POX	554	380.0	1.00	737	7	34400.	.14	1.59
SFO POX	774	380.0	1.00	727	7	35000.	.13	2.29
SFO POX	878	350.0	1.00	137	7	34400.	.14	1.59
SFO SEA	242	492.0	3.00	727	7	35000.	.39	6.87
SFO SEA	388	492.0	3.00	727	7	41000. 35000.	.39	6.87
SFO SEA	420	442.0	3.00	727	7	35000.	.39	6.87
SFO SEA	950	442.0	3.00	727	7	35000.	.39	6.87
SFO SEA	707	492.0	3.00	727	7	35000.	.39	6.87
SFO SEA	1176	492.0	3.00	727	7	35000.	.39	6.87
SFO SLC	502	425.0	9.00	727	7	35000.	1.17	20.61
SFO SLC	704	425.0	9.00	727	7	35000.	1.17	20.61
SFO SLC	946	425.0	9.00	127	7	34850.	1.25	20.61
SLC HOI	277	164.0	-2.00	725		30280.	26	-5.44
SLC BCI	375	164.0	-2.00	725			26	-5.44
SLC BUI	761					30280.	26	-5.44
SLC DEN	166		0.00	727	7	34200.	0.00	0.00
SLC DEN	178	245.0	0.00	DC8	7	34200.	0.00	0.00
SLC DEN	812	245.0	0.00	727	7	34200.	0.00	0.00
SLC DEN	946	245.0	0.00	727	7	34200.	0.00	0.00
SLC ORU	220	991.0	5.00	DCa	7	41000.	.64	21.01
SLC OND	274	991.0				41000.	.65	21.01
SLC ORU	374	991.0				41000.	.05	21.01

	DST A/P	FLT NO	RANGE NMI	RNAV BENE			CRUISE ALT	TIME BENE	FUEL BENE
SLC	040	1412	991.0	5.00	727	7	35000.	.65	11.45
SLC	SFO	137	425.0	13.00	727	7	35000.	1.70	29.76
SLC	SFO	329	425.0	13.00	DCB	7	40520.	1.67	54.17
TPA	ATL	416	301.0	35.00	725	7	35000.	4.62	87.51
TPA	ATL	479	301.0	35.00	737	7	33610.	4.05	56.51
TPA	ATL	750	301.0	35.00	737	7	33610.	4.85	56.51
TPA	CLE	182	724.0	5.00	010	7	41000.	.63	20.86
TOA	CLE	580	728.0	5.00	725	7	35000.	.66	12.50
TYS	DCA	550	294.0	14.00	727	7	35000.	1.63	32.05
TYS	DCA	610	294.0	14.00	727	7	35000-	1.83	32.05

TABLE G.6 TRANS WORLD AIRLINES ENROUTE RNAV BENEFITS ANALYSIS

ORG DST	FLT	RANGE NMI	RNAV BENE	A/C I	F CRUISE ALT	TIME BENE	FUEL	ORG A/P	DST A/P	FLT	RANGE NMI	RNAV BENE	A/C	F CRUIS	E TIME BENE	
ARD LAX	217	485.0	-1.00	725	7 35000.	40	-7.50	DIW	511	903	321.0	26.00	727	1 35000	. 3.39	59.53
AHO LAX	231	485.0	-3.00	725	7 35000.	40	-7.50	OTW	STL	903	327.0	26.00	727	2 35000	. 3.39	59.53
ABO LAX	315				1 35000. 6 35000.	40	-7.50 -7.50	DIM		403	2034.0			7 41000		
ABO PHX	163				5 32040.	26	-4.86	EAR	090	109				7 41000		
ABO PHX	303				7 32040.	26	-5.31	END		193				7 41000		
ATL STL	529				7 34200.	3.82	46.09	ENR		305	193.0	4.00	009	7 35000	. 2.22	
ATL STL	537	360.0	29.00	DC9	7 34200.	3.82	46.09	EWR	PIT	305	193.0	4.00	009	1 31747	52	5.66
HAL ORD	243				7 40872.	2.81	84.12	ENR		401	193.0			7 31827		14.20
BAL URD	381				7 35000.	2.87	50.37	ENR		527	193.0			7 31827		
40F 050	175				7 41000.	25	-8.37	ENR		575	193.0			7 31747		6.66
ROL ORD	403				7 35000.	2.04	77.19	EMR		239	675.0			7 41000		7.83
905 JFK	911				7 23880.	2.04		[40	DEN	203	1220.0	15.00	151	7 35000	. 1.57	27.47
HOS LAX					7 41000.	3.34	120.52	IAD		900				2 25800		
ROS LAX	117				7 41000.	3.34	120.52	IAD						7 41000		
405 04U	195				7 41000.	49	-17.85	IAD						7 41000		92.05
405 090	265				7 35000. 7 35000.	53	-10.00	140						7 41000		92.05
405 090	811				4 41000.	49	-16.74	IAD	SFO	67	2154.0	23.00	833	7 41000	. 2.83	96.23
905 090	811				41000.	49	-16.74	ICT		346	409.0			6 35000		0.00
HOS PIT	173	353.0	25.00	727	6 35000. 7 35000.	3.26	57.24	107		376	409.0			7 35000		0.00
905 P11	551	353.0	25.00	004	7 34130.	3.29	39.78	ICT		348	409.0			6 35000		0.00
HOS STU	107	881.0			7 41000.	3.46	25.82	140		139	109.0			7 26520		0.00
HOS STL	137	891.0	7.00	707	7 41000.	.89	26.82	IND	STL	181	109.0	0.00	727	6 26520	. 0.00	0.00
GLE JFK	700	861.0			7 35000.	. 10	16.03	IND		305	109.0			7 25520		0.00
CLE STL	199	345.0			7 34050.	1.19	15.94	INO		531	109.0			1 26520		0.00
CLE STL	577	345.0	10.00	DC9	1 34050.	1.32	15.94	JFK		44				7 26280		216.30
CLE STL	577				6 34050. 7 37960.	1,32	39.69	JFK		703				7 35320		50.19
CHH DCA	294	179.0	4.00	725	7 31080.	.52	10.76	JFK	DEN	155	1406.0	13.00	837	7 41000	. 1.60	54.39
CHH DCA	426	179.0			7 31080.	.51	9.86	JEK						7 41000		54.39
CHH DCA	112				7 31080.	1.30	10.76	JFK		903	338.0			1 35000		2.29
CAH FEV	124	318.0	10.00	727	7 35000.	1.30	22.90	JFK		903	338.0			2 35000		2.29
DCA CMH	538				7 35000.	3.53	71.08	JFK		903	333.0			7 26120		13.04
OCA CHH	441	207.0	27.00	725	32573.	3.53	71.08	JFK	LAS	149	1858.0	21.00	L10	7 41000	. 2.60	93.74
DCA CMH	531				32573.	3.53	71.08	JFK			2040.0			7 41000		0.00
DEA DAY	373				32520.	2.74	55.33	JFK	LAX	9	2040.0	0.00	83J	7 41000	. 0.00	0.00
DCA OPD	183				7 35000.	5.48	95.16	JEK		11	558.0			7 41000		15.33
DC4 0RU	217				7 35000.	5.54	96.16	JFK		405	558.0			7 41000		15.33
OCA ORD	377	455.0	42.00	725	7 35000.	5.54	105.01	JFK			1781.0	19.00	707	7 41000	. 2.43	72.81
DC4 DRD	423				7 35000.	5.54	105.01	JFK						7 41000		79.50
OCA ORD	449				7 35000.	5.54	105.01	JFK						7 41000		104.78
DCA STL	407	544.0			7 35000.	.53	10.00	JFK						7 41060		12.50
DCA STL	431	544.0			7 35000.	.53	9.16	LAS		306	485.0			7 35000		7.50
nca STL	461	544.0	4.00	727	7 35000.	.52	9.16	LAS	450	440	445.0	3.00	725	7 35000	40	7.50
DEN STL	551	1214.0			7 35000.	1.43	25.19	LAS						7 41000		11.50
DEN JEK	132	1323.0	0.00	83J	7 41000.	.74	25.10	LAS	LAX	417	104.0	2.00	725	7 26120	25	6.03
DEN JEK		1323.0			7 41000.	.74	25.10	LAS		102	1233.0	12.00	727	7 41000	1.48	53.56
DEN OPD		1323.0			7 41000.	.74	15.33	LAS						7 35000		30.00
DEN URD	278	688.0	4.00	707	7 41000.	.51	15.33	L45	ORD	780	1233.0	12.00	L10	7 41000	1.48	53.56
DEN ORD	292	688.0			7 35000. 1 35000.	.52	9.16	LAS		192	140.0	9.00		1 55000		36.49
DEN ORD	545	688.0	4.00	727	5 35000.	.52	9.16	LAS	PHA	464	140.0	9.00	725	1 29000	. 1.16	24.95
DEN SEO	366	688.0			7 35000. 1 35000.	.52	9.16	LAS		910	140.0			6 29000		33.42
DEN SFO	173	726.0			35000.	.52	9.16	LAS	SFO	333	293.0	0.00	725	7 35000	. 0.00	0.00
DEN SFO	185	726.0	4.00	83J	7 41000-	.49	16.74	LAS						7 36120		179.24
DEN STE	108	726.0			7 41000.	.49	19.16	LAX		6	5035.0	24.00	707	7 41000	3.07	91.97
DEN STL	430	599.0	5.00	725	7 35000.	. 56	12.50	LAX	CAL	18	1913.0	24.00	707	7 41000	. 3.07	91.97
DEN STE	456				7 41000.	.64	19.16	LAX						7 41000		-61.31
nT" STL	167	327.0	26.00	707	7 37384.	3.35	93.54	LAX	JFK	702	2032.0	-15.00	837	7 41000	1.97	-66.94
OTW STL					7 37384.	3.35	93.54	LAX								-110.30
1114 516	414	361.0	20.00	101	31384.	3.35	93.54	LAX	J. K	704	203210			2 -1000		

TABLE G.6 TRANS WORLD AIRLINES ENROUTE RNAV BENEFITS ANALYSIS Continued

ORG DST	FLT NO	RANGE NMI	RNAV BENE	A/C		RUISE	TIME BENE	FUEL BENE	ORG A/P		FLT	RANGE NMI	RNAV	A/C		RUISE	TIME BENE	FUEL BENE
LAX JFK		2032.0-						-71.42		DCA	358		25.00				3.09	111.59
LAX LAS	306	103.0	3.00				.38	9.07		DCA	376		25.00				3.30	62.50
LAX OKC	535	936.0				1000.	.89	28.82		DEN	439	683.0	25.00			35000.	3.26	9.15
LAX CKC	232	936.0	7.00	707	7 4		.89	26.82	090	DEN	121	683.0	4.00	727	6 3	35000.	.52	9.16
LAX OPD		1412.0					1.72	58.58		DEN	193 387	683.0				1000.	.51	9.16
LAX 020		1412.0					1.79	53.65	040	DEN	415	683.0	4.00	747	7 4	.1000.	.51	15.33
LAX ORD		1412.0					1.79	62.49	ORD		130	516.0	4.00				37	10.00
LAX PHL		1995.0	14.00	LIQ	7 4	1000.	1.73	62.49	050	EWH	384	516.0	-3.00	707	7 4	.1000.	18	-11.50
LAX PHX	16	208.0	0.00				0.00	0.00	040		217	516.0	5.00				40	12.50
LAX DHA	190	0.805	0.00	707	7 3	2627.	0.00	0.00	0.50	ICT	315	414.0	5.00	725	1 3	15000.	.66	12.50
LAX PHA	53		17.00				2.13	60.05	090		315	419.0				5000.	.66	12.50
LAX SFO	61	199.0	17.00	707	7 3	2147.	2.13	60.05	000	JFK	800	519.0	0.00	707	4 4	1000.	0.00	0.00
LAX SFO	471		17.00				2.13	60.05	040		880	518.0				1000.	0.00	0.00
LAX SFO	761		17.00				2.07	90.08	050		195	1255.0	12.00	L10	7 4	1000.	1.48	53.56
LAX SEC	411	199.0	17.00				2.05	65.57	080			1255.0					1.58	30.00
LAX STL	12	1304.0	12.00	707	7 4	1000.	1.53	45.99	080	LAS	711	1255.0	12.00	L10	7 4	1000.	1.48	53.56
LAX STL	16	1304.0	12.00	707	7 4	1000.	1.53	30.00	080			1409.0					1.28	38.32
LAX STL	448	1364.0	12.00	LIO	7 4	1000.	1.48	53.56	040	LAX	21	1404.0	10.00	707	7 4	1000.	1.28	38.32
LAX TUS	106	7HH.0	13.00				1.72	32.50	080			1409.0					1.28	38.32
LGA DAY	411	385.0	13.00	DC9	7 3	4450.	1.71	20.57	040	LGA	306	519.0	1.00	725	7 3	5000.	.13	2.50
LGA DAY	465		13.00				1.70	29.76	040		310	519.0				5000.	.13	2.50
LGA ONU	303	532.0	15.00	725	7 3	5000.	1.98	37.50	ORD	LGA	318	519.0	1.00	727	7 3	15000.	.13	2.29
LGA CHU	315		15.00				1.98	37.50	040		322	519.0				5000.	.13	5.59
LGA UPU	323		15.00				1.98	37.50	000	LGA	330	519.0	1.09	725	7 3	5000.	.13	2.50
LGA ORD	329		15.00				1.96	34.34	040		334	519.0				5000.	.13	5.59
LGA ORU	339		15.00				1.96	34.34	080	LGA	342	519.0	1.00	727	7 3	5000.	.13	2.29
LGA ORD	341		15.00				1.96	34.34	080		346	519.0				5000.	.13	2.50
LGA ORD	343		15.00				1.98	37.50	0.50	LGA	354	519.0	1.00	725	7 3	5000.	.13	2.50
LGA ORD	351		15.00				1.98	37.50 37.50	0.40			1531.0					2.09	36.63
LGA OPD	359		15.00				1.98	37.50	090	PHL	50	526.0	28.00	833	7 4	1000.	3.45	117.15
LGA PIT	115		15.00				1.56	19.89	050		110		28.00				3.46	124.98
LGA PIT	225	201.0	12.00	727	7 3	2253.	1.55	29.07	0.50	PHL	238	526.0	28.00	707	7 4	1000.	3.58	107.30
LGA PIT	251		12.00				1.56	19.89	080		278	1204.0	28.00				1.36	49.10
LGA PIT	515	201.0	12.00	DC9	7 3	1960.	1.56	19.89	090	PHX	235	1204.0	11.00	727	7 3	5000.	1.43	25.19
LGA PIT	393		12.00				1.96	19.89	040			1204.0					1.43	49.10
LGA STL	147	675.0	5.00	125	7 3	5000.	.66	12.50	090	PHX	435	1204.0	11.00	727	7 3	5000.	1.43	25.19
LGA STL	171	675.0	5.00	DCY	7 3	5000.	.66	12.50	050		36		-5.00				25	-7.05
LGA STL	495	675.0	5.00	727	7 3	5000.	.65	11.45	020	PII	104	264.0	-2.00	707	7 3	4960.	25	-7.05
MIA STL	493	851.0	7.00				.92	29.29	080		270		-2.00				26	-4.58 -7.05
MIA TPA	485	90.0	-4.00	725	7 2	5000.	51	-12.49	080	PIT	230	264.0	-2.00	727	7 3	4960.	26	-4.58
OKC LAX	107		-4.00				51	-12.49	ORD		262		-2.00				26	-4.5H
OKC LAX	197	925.0	-3.00	707	1 4	1000.	38	-11.50	040		593	264.0	-2.00	121	6 3	4950.	26	-4.5H
ORD ARD	243		7.00				38	-11.50	040			1513.0					2.05	61.31
ORD AND		0.588	7.00	725	7 3	5000.	.45	17.50	040	550	135	1513.0	10.00	101	6 4	1000.	2.05	01.31
DED VEG	373		7.00				.92	17.50	040			1513.0					1.97	71.42
OPD BAL	24	447.0	29.00	707	7 4	1000.	3.71	111.13	090	TUS	323	1175.0	11.00	725	7 3	5000.	1.45	27.50
OND HAL	92 350		29.00				3.71	111.13	250			1175.0					1.43	25.19
OPD BUL	82	598.0	18.00	831	7 4	1000.	2.21	75.31	ORD	TUS	355	1175.0	11.00	725	6 3	5000.	1.45	27.50
090 80L	262		18.00				-1.86	45.00	PHL		27	509.0					2.85	61.31
090 605	105	656.0-	-15.00	725	7 3	5000.	-1.98	-37.50	PHL	020	151	504.0	16.00	727	1 3	5000.	2.09	36.63
040 805	202						-1.85	-62.76	PHL		121	509.0	16.00				2.09	61.31
090 905	429	556.0-	-15.00	707	7 4	1000.	-1.92	-57.48	PHL	040	501	509.0	16.00	L10	7 4	1000.	1.98	71.42
ORD DCA	168	444.0	25.00	725	7 3	5000.	3.30	62.50	PHL		711	159.0	8.00				1.98	13.59
040 004	. 76		23.00		. ,		2.50		200		-							

TABLE G.6 TRANS WORLD AIRLINES ENROUTE RNAV BENEFITS ANALYSIS Continued

ORG DST	FLT	RANGE NMI	RNAV BENE	A/C	CRUISE ALT	TIME	FUEL BENE		DST A/P	FLT	RANGE NMI	RNAV		F CRUIS	E TIME BENE	FUEL BENE
PHL PIT	539	159.0	8.00	DC9	7 30760.	1.04	13.59	SFO	LAX	760	182.0	3.00	LIO	7 31240	36	16.14
PHL PIT	565	159.0			7 30760.	1.04	13.59		050					7 41000		66.94
PHL PIT	755	159.0			7 30013.	.96	21.86		ORD					7 41000		66.94
DHL SFO			25.00	707	7 41000.	3.20	95.80	SFO	OPD	770	1498.0	16.00	L10	7 41000	. 1.98	71.42
DHE STE	101	621.0			7 35000.	.40	7.50		PHX	94				7 41000		122.63
PHX ABQ	150	200.0			7 32200.	0.00	0.00		PHA	296				7 35000		13.27
DHX AHU	170	200.0	0.00	727	7 32200.	0.00	0.00	SFO	PHA	358	490.0	32.00	L10	7 41000	. 3.96	142.84
DHX JFK					7 41000.	2.09	71.13	SFO			1409.0			7 35000 7 41000		12.50
DHE JEK					1 41000.	2.09	71.13		STL					7 35000		11.45
PHX LAS	73	140.0			1 29000.	.64	12.69		ATL	528	334.0			7 33940		1.60
PHX LAS	191	140.0			7 29000.	.60	12.69	STL		562	334.0			7 33940.		1.60
PHX LAX	17	208.0	3.00	707	7 32627.	.38	10.51	STL	CLE	554	358.0	20.00	DC9	1 34180	. 2.63	31.80
PHX LAX	207	208.0			7 32627.	.36	7.23	STL		554				6 34180. 7 34180.		31.80
PHX LAX	533	208.0			6 32627.	.39	7.89	STL		700				7 35000		50.00
DHX LAX	533	208.0			1 32627.	.39	7.89	STL		80	216.0	4.00	725	7 33040	52	10.45
DHX 020					7 41000.	1.36	25.19	STL		542	216.0			7 32360. 7 33040.		9.57
PHX ORU	354	1163.0	11.00	725	7 35000.	1.45	27.50	STL		374	532.0			7 35000.	52	9.16
PHX STL					7 41000.	1.36	49.10	STL		430	532.0			7 35000.		10.00
PHA STL		1012.0			4 41000.	.26	7.66	STL		450	532.0			7 35000. 7 35000.		9.16
PHE STL	876	1012.0	2.00	707	3 41000.	.26	7.66	STL	DCA	482	532.0	4.00	727	7 35000.	52	9.16
P11 805	244				7 35000.	3.13	54.95	STL		401	599.0			7 41000.		0.00
P11 905	550				7 34320.	3.16	38.06	STL		457	599.0			7 41000.		0.00
SIT EWN	76				7 32733.	3.65	101.46	STL		561		0.00	727	7 35000	0.00	0.00
PIT ENH	158				7 32733.	3.75	69.75	STL		210				7 37160. 7 37160.		96.74
PIT ENH	264	210.0	29.00	727	7 32733.	3.75	69.75	STL		565				7 37160		96.74
PIT EWN	518				.00526 7	3.78	47.84	STL		832				4 37160		96.74
PIT LGA	214	185.0			7 31533.	.52	6.69	STL	IND	832	122.0			3 37160. 7 27560.		34.45
PIT LGA	246	145.0	4.00	DC9	7 31533.	.52	6.69	STL	IND	254	122.0	9.00	725	7 27560	1.15	25.99
PIT LGA	255	185.0			7 31533.	.52	9.81		IND	434	155.0			7 27560		25.99 34.45
PIT LGA	564	185.0			7 31400.	.51	9.81		IND	454	122.0	9.00	DC9	7 29280	. 1.16	15.85
PIT LGA	570	185.0			7 31533.	.52	6.69		IND	456	122.0			7 27560	1.11	34.45
PIT CHU	572	267.0			7 31533. 7 35080.	1.23	51.16		LAX		1273.0			7 41000		26.82
DIL ODD	135	267.0	10.00	707	1 35080.	1.27	35.30		LAX	269	1273.0	7.00	725	7 35000	92	17.50
PIT ORD	225				7 35000. 7 35000.	1.30	22.90		LGA	56	1273.0			7 35000		3.13
P11 090	261	267.0	10.00	727	7 35000.	1.30	22.90		LGA	182	692.0			7 35000		5.00
PIT ORD	271				7 35080.	1.27	35.30		LGA	266	692.0			7 35000		5.00
PIT PHL	516				7 35080.	1.27	35.30		PHL	436	692.0			7 41000		11.50
PIT PHL	520	144.0	-5.00	DC9	7 30160.	65	-8.62	STL	PHL	55A	629.0	3.00	725	7 35000	40	7.50
PIT PHL	522				7 29213. 7 30160.	65	-12.66		PHX		1010.0			1 35000. 6 35000		4.58
PIT PHL	545				7 30160.	65	-8.62		PHX		1010.0			7 35000		4.58
PIT PHL	548				7 30160.	65	-8.62	STL		76	407.0			7 39944		30.11
SOF LGA	756 396				7 29213.	3.17	-20.21	STL		876	407.0	8.00	707	7 34670.	1.06	30.11
SOF STL	259	136.0	2.00	125	7 28680.	.26	5.60	STL	PIT	876	407.0	8.00	707	3 39944.	. 1.05	30.11
SOF STE	421	136.0			7 28680.	.25	7.48	STL	SOF	136	136.0			7 28680		11.22
SFO BOS					7 41000.	3.59	129.45	STL		444	136.0			7 28680		11.22
SFO DEN	188	726.0	-2.00	727	7 35000.	26	-4.58	STL	550			14.00	707	7 41000	. 1.79	53.65
SED DEN	252 810				7 35000.	26	-4.58 -8.37		SFO					7 41000		53.65 35.00
SFO DEN	810	726.0	-2.00	B37	3 41000.	25	-8.37	STL	TUL	107	217.0	0.00	707	7 33080.	. 0.00	0.00
SFO TAU		2010.0			7 41000.	.86	29.29		TUL	187				1 33080 6 33080		0.00
SEC JEK	44	2129.0	26.00	837	7 41000.	3.20	108.78		TUL	495	217.0	0.00	727	7 33080	. 0.00	0.00
SFO JFK					7 41000.	3.22	116.06	TPA	MIA	478	103.0	18.00	725	7 25040	. 2.29	54.43
SFU LAS					7 41000.	3.22	102.65		STL	232				7 26040 6 33120		6.97
SFO LAS	440	278.0	24.00	725	7 35000.	3.83	72.51	TUL	STL	435	218.0	5.00	727	7 33120	26	4.78
SFO LAK	76	182.0			7 31240.	.37	10.75		STL					7 33120.		27.50
SFO LAX	90	182.0	3.00	707	7 31240.	.37	10.75		OND	334	1378.0	11.00	727	7 35000	. 1.43	25.19
SFC LAX	92	182.0			7 31240.	.37	10.75		020					7 35000		25.19
SFO LAX	535	182.0			31240.	.37	10.75									
SFO LAK	305				7 31240.	.39	8.05									

TABLE G.7 RNAV Benefits over VOR for 2/1/76 schedule and route structure

NAL

		ANNUAL	IAL				JG DE	PER AIRCRAFT		
Aircraft	Ranafit	Fuel	Time	\$ Ra	Range	# of	Fuel	Time	\$ Ra	Range
lype	n l l l l l l l l l l l l l l l l l l l	(gal)	(min)	Low	High	Aircraft	(ga1)	(min)	Low	High
	2D Enroute	145K	X7	67K	115к		11.2K	609	5,409	9.190
121	2D TMA	651K	34K	306K	521K	13	50.1K	2603	23,508	39,932
	3D TMA	214K	10%	94K	160K		16.4K	752	7,138	12,114
	Total	1,010K	51K	467K	796K		77.7K	3964	36.055	61.236
	2D Enroute	876K	40K	400K	670K		35.0K	1616	16.082	26.936
725	20 TMA	1,494K	72K	705K	1,180K	30	59.8K	2865	28.097	47.068
	3D TMA	496K	21K	216K	362K	67	19.8K	841	8,652	14,490
	Total	2,866K	133K	1,321K	2,212K		114.6K	5322	52.831	88.494
		1,049K	29K	437K	814K		70.2K	1876	28.793	53.562
010	20 TMA	1,100K	34K	490K	918K	15	73.3K	2264	32,664	61.157
	3D TMA	221K	6K	91K	170K		14.8K	423	6.301	11 750
	Total	2,370K	X69	1,018K	1,902K		159.3K	4563	67 758	176 477
							1000	2000	06/1/0	1000

TABLE G.8 RNAV Benefits over VO? for 2/1/76 Schedule and Route structure

EAL

		ANNUAL	JAL					PE	PER AIRCRAFT		
Aircraft	Bonofit	Fuel	Time	\$ Range	nge	# of		Fuel	Time	\$ Ra	Range
Type		(ga1)	(min)	Low	High	Aircraft	raft	(ga1)	(min)	Low	High
	2D Enroute	891К	18K	273K	X/89			23.1K	607	9,202	23.248
L10	20 TMA	1498K	41K	610K	1552K	30		49.9K	1382	20,475	52,146
	3D TMA	227K	6K	91K	230K			7.6K	198	3,015	7,606
	Total	2416K	65K	974K	2569K		L	80.6K	2187	32,692	83,000
	2D Enroute	2.20M	122K	1.24M	2.02M			29.4K	1631	16,544	26,975
727	2D TMA	3.10M	162K	1.68M	2.73M	75		41.3K	2161	22,344	36,449
	3D TMA	W68.	41K	.44M	.72M			11.9K	544	968'5	9,627
	Total	6.19M	325K	3.36M	5.47M			82.6K	4336	44,784	73,051
	2D Enroute	1.39M	70K	647M	647M 1 114M			35.5K	1800	16,566	28,570
725	2D TMA	2.36M	120K	1 103M	MC06 L			60.5K	3066	28,213	48,657
	3D TMA	.67M	28K	.279M	.480M	39	-	17.2K	729	7,215	12,414
	Total	4.42M	218K	2.029M 3.496M	3.496M		<u> </u>	113.2K	5835	51,994	89,641
	2D Enroute	147K	12K	77K	138K			18.3K	1471	9.427	16.922
600	2D TMA	322K	25K	162K	291K	α		40.2K	3169	20,440	36,672
	3D TMA	179K	11K	78K	139K)		22.4K	1429	986,6	17,819
	Total	648K	481.	317K	568K			80.9K	6909	39,853	71,413
	2D Enroute	1.97M	157K	1.15M	1.90M			27.1K	2157	15,692	26,094
260	2D TMA	3.37M	229K	1.76M	2.92M			46.2K	3136	24,048	39,993
	3D TMA	1.44M	89K	.71M	1.17M	73		19.7K	1213	9,631	16,018
	Total	6.78M	475K	3.62M	5.99M			92.9K	9059	49,371	82,105
				-			-		-	-	-

TABLE G.9 RNAV Benefits over VOR for 2/1/76 Schedule and Route Structure

DAL

		ANNUAL	JAL			-		PER	R AIRCRAFT		
Aircraft	1,3	Fuel	Time	\$ Ra	Range	#	Jf	Fuel	Time	\$ Ra	Range
Type	Benetit	(gal)	(min)	Low	High	Air	Aircraft	(ga1)	(min)	LOW	High
	2D Enroute	321K	У6 .	121K	194K			15.3K	461	5,859	9,427
D8F	2D TMA	425K	15K	178K	285K		21	20.2K	705	8,383	13,451
	3D TMA	173K	5K	65K	104K			8.2K	234	3,052	4,913
	Total	919K	29K	364K	583K			43.7K	1400	17,294	27,791
	2D Enroute	143K	3.75K	60K	161K			47.3K	859	16,564	41,468
747	2D TMA	213K	4.0K	76K	191K		3	70.9K	1458	26,443	67,610
	3D TMA	22K	.3K	7K	16K		,	7.2K	112	2,347	5,720
	Total	378K	8.1K	143K	250K			125.4K	2429	45,354	114,798
	2D Enroute	1.28M	34K	527K	1.34M			71.0K	1882	29,184	73,894
L10	2D TMA	1.49M	44K	652K	1.68M		0.	82.9K	2419	36,007	92,481
	3D TMA	.30M	8K	124K	.31M		2	16.4K	430	6,700	16,939
	Total	3.07M	86K	1303K	3.33M			170.3K	731	71,891	183,314
	2D Enroute	858K	24K	330K	585K			66.0K	1885	25,581	45,316
DBS	2D TMA	938K	30K	387K	687K			72.1K	2269	29,466	52,337
	3D TMA	280K	7K	101K	179K		13	21.5K	562	7,954	14,056
	Total	2074K	61K	818K	1451K			159.6K	4716	63,001	111,709
	2D Enroute	2.89M	145K	1.31M	2.40M			38.0K	1907	17,200	31,580
727	2D TMA	4.45M	226K	2.03M	3.73M		1 9/	58.6K	2974	26,707	49,055
	3D TMA	1.45M	62K	.60M	1.09M		2	19.1K	815	7,866	14,355
	Total	8.79M	433K	3.94M	7.22M			115.7K	9699	51,773	94,990
	2D Enroute	1.26M	97K	598K	1.09M			20.3K	1562	9,636	17,484
600	2D TMA	2.19M	147K	958K	1.73M		62	35.2K	2377	15,447	27,912
	3D TMA	1.60M	99K	668K	1.20M		!	25.8K	1590	10,739	19,347
	Total	5.05M	343K	2224K	4.02M			81.3K	5529	35,822	64,743

TABLE G.10 RNAV Benefits over VOR for 2/1/76 Schedule and Route Structure

AAL

		ANNUAL	IAL			-		DE	PER AIRCRAFT		
Aircraft	Beaufit	Fuel	Time	\$ Range	nge	#	# of	Fuel	Time	\$ Range	nge
Type	benerit	(gal)	(min)	Low	High	A	Aircraft	(ga1)	(min)	Low	High
	2D Enroute	2.47M	82K	1059K	1.98M			28.1K	933	12,060	22,511
707	2D TMA	2.73M	100K	1241K	2.33M		88	31.0K	1135	14,092	26,420
	3D TMA	. 96м	30K	398K	.74M			10.9M	341	4,519	8,413
	Total	6.16M	212K	2698K	5.05M	_		70.0K	2409	30,671	57,344
	2D Enroute	3.99M	227K	1995K	3.55M			37.6K	2135	18,802	33,496
727	2D TMA	5.47M	263K	2466K	4.38M		106	51.6K	2477	23,242	41,241
	3D TMA	1.35M	62K	592K	1.05M	_	3	12.7K	583	5,570	9,873
	Total	10.81M	552K	5053K	8.98M			101.9K	5195	47,614	84,610
	2D Enroute	1.03M	30K	475K	м16.			41.0K	1195	19,130	36,697
010	2D TMA	1.22M	36K	574K	1.10M			48.7K	1449	23,016	44,194
	3D TMA	.21M	6K	97K	. 19М		25	8.2K	237	3,807	7,299
	Total	2.46M	72K	1146K	2.20M			97.9K	2881	45,953	88,190
	2D Enroute	04M	-1.0K	-30K	M. 1057M			-4.0K	-73	-2,417	-4,497
747	2D TMA	.14M	3.0K	95K	.178M		01	14.2K	569	8,810	16,410
	3D TMA	W10.	.1K	4K	.008M			.6К	10	341	632
	Total	М11.	2.1K	869K	.129M			10.8K	206	6,734	12,545

TABLE G.11 RNAV Benefits over VOR for 2/1/76 Schedule and Route Structure

NAL

		ANNUAL	UAL					PER AIRCRAFT		
Aircraft	114000	Fuel		\$ Ra	\$ Range	# of		Time	\$ Ra	Range
Type	peneric	(ga1)	(min)	LOW	High	Aircraft		(min)	LOW	High
	2D Enroute	4.25M	236K	2.17M	2.68M		34.8K	1934	17,731	30,154
727	2D TMA	3.16M	156K	1.49M	2.54M	122	25.9K	1282	12,269	20,846
	3D TMA	1.37M	63K	.62M	1.05M		11.3K	515	2,090	8,643
	Total	8.78M	455K	4.28M	7.27M		72.0K	3731	35,090	59,643
	20 Enroute	1.10M	57K	533K	945K		39.4K	2041	19.100	33.829
725	2D TMA	. 95M	42K	418K	738K		34.0K	1515	15.033	26.537
	3D TMA	.48M	20K	204K	360K	28	17.1K	725	7,343	12,948
	Total	2.53M	119K	1155K	2043K		90.5K	4281	41, 476	73,314
	2D Enroute	1.42M	120K	948K	1.56M		22.1K	1874		24,277
737	2D TMA	1.29M	95K	780K	1.28M		20.2K	1481	12,176	20,006
	3D TMA	.84M	56K	474K	.78M	64	13.2K	883	7,468	12,275
	Total	3.55M	271K	2202K	3.62M		55.5K	4238	34,429	56,558
	2D Enroute	1.31M	38K	622K	1.22M		35.3K	1048	16,856	32,221
010	2D TMA	1.61M	48K	711K	1.52M	37	43.6K	1290	20,791	40,966
	3D TMA	.33M	9K	150K	. 29м	5	8.8K	177	3,360	6,466
	Total	3.25M	95K	1543K	3.03M		87.7K	2515	41,007	79,653
	2D Enroute	1.61M	50K	622K	1.12M		30.2K	944	11,673	21,154
800	2D TMA	1.66M	55K	662K	1.20M		31.3K	1045	12,531	22,766
	3D TMA	.46M	13K	169K	.30M	53	8.7K	247	3,196	5,772
	Total	3.73M	118K	1453K	2.62M		70.2K	2236	27,400	49,692
	2D Enroute	1.37M	26K	406K	785K		34.9K	645	10,312	19,913
088	2D TMA	M66.	31K	371K	750K		25.4K	786	9,471	19,109
	3D TMA	.44M	13K	160K	322K	39	11.3K	333	4,110	8,257
	Total	2.80M	70K	937K	1857K		71.6K	1764	23,893	47,279
	2D Enroute	.38M	7K	142K	274K		21.4K	381	8,037	15,511
747	2D TMA	.51M	10K	202K	392K		28.2K	562	11,258	21,882
		. 06М	1K	22K	42K	18	3.4K	53	1,195	2,286
	Total	.95M	18K	366K	K 19K		53.0K	966	20,490	39,679

TABLE G.12 RNAV Benefits over VOR for 2/1/76 Schedule and Route Structure

TWA

		ANNUAL	AL			-		JE DE	PER AIRCRAFT		
Aircraft	113000		1	\$ Range	nge	#	of	Fuel	Time	\$ Rai	Range
Type	benerit	(ga1)	(min)	Low	High	A	Aircraft	(gal)	(min)	Low	High
01.1	2D Enroute	1095K	29.5K	464K	1003K		30	36.5K	1002	15,613	33,864
	30 TMA	828K 151K	22K 4K	348K	752K		3	5.0K	132	2,092	4,521
	Total	2074K	55.5K	875K	1892K			69.1K	1872	29,340	63,553
	2D Enroute	976K	55K	519K	878K			27.9K	1575	14,813	25,065
727	2D TMA	1329K	61K	618K	1045K		35	38.0K	1752	17,732	29,971
	3D TMA	380K	17K	174K	294K			10.8K	496	5,027	8,498
	Total	2685K	133K	1311K	2217K	_		76.7K	3823	37,572	63,534
	2D Enroute	1558K	81K	Х969	1356K			40.0K	2061	17,839	34,722
725	2D TMA	1686K	73K	681K	1311K		30	43.2K	1870	17,438	33,592
	3D TMA	554K	24K	224K	431K		33	14.2K	603	5,672	10,913
	Total	3798K	178K	1601K	3098K			97.4K	4534	40,949	79,227
	2D Enroute	484K	39K	323K	564K			26.9K	2181	18,037	31,479
60G	2D TMA	716K	48K	419K	730K			39.8K	2692	23,450	40,834
	3D TMA	250K	16K	142K	247K		8	13.9K	885	7,856	13,669
	Total	1450K	103K	884K	1541K	_1		80.6K	5758	49,343	85,982
	2D Enroute	3.06M	98K	1.20M	2.14M			30.6K	066	11,974	21,454
707	2D TMA	2.88M	99K	1.17M	2.09M		100	28.8K	994	11,678	20,966
	3D TMA	1.01M	30K	.38M	.68M			10.1K	302	3,791	6,775
	Total	6.95M	227K	2.75M	4.91M			69.5K	2286	27,443	49,195
	2D Enroute	25K	.5K	9K	23K			.25K	5	92	234
747	2D TMA	67K	1.4K	25K	65K			6.72K	140	2,523	6,466
	3D TMA	6K	.1K	2K	5K		10	.62K	10	205	504
	Total	98K	2.0K	36K	93K			7.59K	155	2,820	7,204
									-	-	-

4D RNAV PROJECTED BENEFITS

NAL

-		-			
	inge	High	29,008	36,016	88,473
	\$ Range	Low	17,045 29,008	21,490	46,913 88,473
PER AIRCRAFT	Time	(min.)	2077	2400	3467
PER A	Fuel	(ga1)	30.5K	38.7K	95.0K
	# 0f	Aircraft	13	25	15
	inge	High	377K	900K	1327K
	\$ Range	Low	222K	537K	734K
	Time	(min.)	27K	60K	52K
ANNUAL	Fuel	(gal)	396K	968K	1425K
	Aircraft	Type	727	725	010

4D RNAV PROJECTED BENEFITS

EAL

	-	T					
	nge	High	47,497	39,504	191,751	59,442	38,077
	\$ Range	Low	18,067	24,260	29,935	32,952	22,901
PER AIRCRAFT	Time	(min.)	1333	2533	3487	5500	3301
PER A	Fuel	(gal)	38,8K	37.7K	56.8K	55.8K	35.2K
	# of	Aircraft	30	75	39	8	. 73

	Range	High	1425K	2963K	2018K	476K	2780K
	\$ RA	LOW	542K	1820K	1167K	264K	1672K
JAL	Time	(min.)	40K	190K	136K	44K	241K
ANNUAL	Fuel	(gal)	1163K	2825K	2217K	441K	2571K
	Aircraft	Type	710	727	725	600	260

TABLE G.15

DAL 4D RNAV PROJECTED BENEFITS

		1	7			1	T
	Range	33,910	147,044	175,189	88,085	511,112	39,824
	\$ Ra	21,212	55,938	66,636	49,306	33,036	21,822
PER AIRCRAFT	Time (min)	1952	3333	4778	4154	3961	3790
PER A	Fuel (gal)	44.8K	137.0K	138.8K	106.2K	64.0K	40.5K
	# Of Aircraft	21	т	18	13	92	. 29
			1				
	Range	712K	441K	3153K	1145K	4644K	2469K
	\$ Ra	445K	168K	1199K	641K	2511K	1353K
١,	Time (min)	41K	10K	86K	54K	301K	235K
ANNUA	Fuel (nal)	941K	411K	2499K	1381K	4863K	2512K
	Aircraft	DBF	747	710	D8S	727	600

AAL

4D RNAV PROJECTED BENEFITS

	Range	High	50,449	53,968	87,927	17,480
	\$ Ra	Low	26,624	30,177	45,313	9,333
PER AIRCRAFT	Time	(min.)	2352	3632	3080	300
PER A	Fuel	(gal)	49.9K	54.0K	83.5K	13.1K
	# of	Aircraft	88	106	25	10
-						
	\$ Range	High	4440K	5721K	2198K	175K
	\$ Ra	Low	2343K 4440K	3199K	1133K	93K
	Time	(min.)	207K	385K	77K	3K
ANNUAL	Fuel	(gal)	4391K	5722K	2087K	131K
	Aircraft	Type	707	727	010	747

TABLE G.17

UAL 4D RNAV PROJECTED BENEFITS

,										1
	Range	High	40,847	58,224	35,446	76,567	46,604	45,544	59,672	
	\$ R3	Low	23,988	32.755	21,599	38,400	25,415	22,123	30,253	
PER AIRCRAFT	Time	(min.)	2787	3714	2844	2568	2358	2051	1991	
PER A	Fuel	(gal)	41.4K	60.5K	28.9K	70.2K	54.7K	51.9K	66.2K	
	# 0f	Aircraft	122	28	64	37	53	. 68	18	
	- L									
	\$ Range	High	4983K	1630K	2268K	2833K	2470K	1776K	1074K	
	\$ Ra	Low	2926К	917K	1382K	1421K	1347K	863K	545K	
UAL	Time	(min.)	340K	104K	182K	95K	125K	80K	30K	
ANNUAL	Fuel	(gal)	\$051K	1696К	1849K	2598K	2900K	2023K	1192K	
	Aircraft	Type	. 727	725	737	010	800	085	747	

TABLE G.18

TWA 4D RNAV PROJECTED BENEFITS

		_						1
	Range	High	56,830	40,823	46,528	30,602	39,011	8,457
	\$ Ra	Low	25,644	24,105	23,669	17,494	21,539	3,137
PER AIRCRAFT	Time	(min.)	1800	2714	2923	2222	2060	200
PER A	Fuel	(ga1)	52.1K	40.2K	47.6K	22.5K	44.9K	7.0K
	# of	Aircraft	30	35	39	18	100	. 01
	Range	High	1705K	1429K	1815K	551K	3901K	85K
		Low	769К	844K	923K	315K	2154K	31K
IUAL	Time	(min.)	54K	95K	114K	40K	206K	2K
ANNUAL	Fuel	(gal)	1564K	1406K	1856К	405K	4489K	70K
	Aircraft	Type	710	727	725	600	707	747

APPENDIX H

FIELD CONTROLLER APPRAISAL OF THE USE OF RNAV/VNAV

IN TERMINAL AREA ATC OPERATIONS

H.1 INTRODUCTION

Considerable interest has been expressed by various groups in the involvement of journeymen air traffic controllers, drawn from field facilities, in this simulation in which RNAV/VNAV operations are introduced into a high density terminal ATC environment. The purpose of the introduction of field controllers in the training, exploratory, and data collection phases of the simulation served four major purposes: (1) to draw upon our current field experience in the refinement of the environment/procedures to be simulated, (2) to solicit our comments and reactions to the use of RNAV/VNAV in terminal area operations, (3) to derive quantitative simulation results from data collection runs in which we participated as test subjects, and (4) to provide us with some degree of familiarity with RNAV/VNAV operations through simulations.

Through the cooperation of the Air Traffic Service, Washington, D.C., and the regions and facilities represented, we were detailed to NAFEC to participate in the simulation. While NAFEC pool controllers also participated in the simulation, due to the high degree of interest expressed in our participation, the following appraisal represents our opinions only. The presentation of our appraisal, independent of any comments or opinion expressed by the NAFEC pool controllers, is provided to be responsive to this interest and does not imply any prejudicial judgment between the value to NAFEC pool controllers and our opinions.

H.2 BACKGROUND

We started the training phase in the operational use of RNAV/VNAV at NAFEC on June 16, 1975. The purpose of the training phase was to familiarize the digitial simulation facility (DSF) target generator operators (DSF pilots), general aviation trainer (GAT) pilots, and the air traffic controllers with all required aspects of the simulation including geography, equipment and procedures. The training phase was scheduled to be followed by an exploratory phase during which the initial procedures, geography, etc. could be modified and refined prior to the data collection phase which was scheduled to start no later than August 4. Due to the numerous problems in the shakedown of the simulation equipment, the planned training and exploratory phase were frequently disrupted by various simulation system performance problems and failures and data collection was delayed until August 18, 1975. The need for extensive shakedown runs impacted severely on controller training and the exploratory phases of the simulation. The failure of the DSF targets to react in a predictable manner to ATC clearances (due to problems in the DSF not associated with "real world" RNAV/VNAV performance) worked to the detriment of an early understanding and efficient use of RNAV/VNAV functions in the control of air traffic. There may be a residual effect of this uncertainty as to compliance with ATC clearances and anomalous performance of the simulated targets which will impact

the objective results from the data runs. However, the opinions upon which the following of the data collection period and are expressions of our subjective analysis of the operational application of RNAV/VNAV to terminal air traffic control.

H.3 TERMINAL AREA RNAV ROUTE STRUCTURE DESIGN (2D)

It is our opinion that the RNAV (2D) structure design originally planned for simulation required some modification to provide a higher degree of flexibility for the controller, if such modifications did not adversely impact route miles, altitude restrictions, etc. to an undue extent on the system user. A modified design was developed which appeared to satisfy this requirement. The major difference between this design, which was developed during the exploratory period for use in data collection runs, and the original design was in the area immediately to the east of JFK. The original design located the departure routes serving departures to the northwest, north and northeast parallel to, and inside the downwind leg. The new design, which is discussed in more detail in the NAFEC simulation report, placed the departure routes outside the downwind leg. This change appeared to have no adverse impact on the system user. The modification was made based on the following operational consideration:

There was a need for radar vectoring airspace, so that we could compare a 100 percent radar vectored operation with a 100 percent RNAV/VNAV operation. We would have been unable to compare these operations if we had to vector aircraft along the RNAV track.

The original design was made by Champlain Technology Industries, and there were no provisions made for radar vectored aircraft in their design.

The arrival route was moved inside the departures to give us more flexibility. We wanted to have the ability to shortcut traffic to runway 22L, from the downwind leg. This gave the final controller a true dump zone.

The new design had more waypoints. Two of the new waypoints were positioned closer to the outer marker. This enabled the final controller to switch any arrival to either runway by the use of RNAV.

H.4 TERMINAL AREA VNAV ROUTE STRUCTURE DESIGN (3D)

The original design planned for simulation allowed for the use of "stacked routes" for arrival/departure traffic. (The term "stacked routes" is used here to describe two or more routes having common or near common horizontal paths which are separated vertically based on VNAV (3D) separation criteria.) It was envisioned that a unique application of VNAV arrival routes would result through the use of two-segment approaches which were to be included in certain parts of the simulation tests. However, when it was learned that the FAA did not support the use of two-segment approaches, this application was no longer considered viable. Therefore a renewed and major emphasis was placed on determination of other potential uses for VNAV and its unique capabilities

as they might relate to both terminal airspace design and ATC operational use of VNAV as a control tool.

In order to clarify the unique use of VNAV in combination with the two-segment approach concept, and why this combination appeared to offer some potential advantage in the use of stacked routes, the following illustration (Figure H.1) is provided. As shown, using a two-segment approach to runway 22R and a single-segment approach to runway 22L, traffic from the east could fly stacked routes with the aircraft on the higher route intercepting the localizer for runway 22R at a higher altitude and executing a two-segment approach. The aircraft on the lower stacked routes would intercept the localizer for runway 22L at a lower altitude and vertical separation could be provided between the two aircraft until both were established on their respective localizers. Since two-segment approaches were dropped from the simulation tests it is not known whether this combination would provide any operational advantage or not. However, two-segment approaches, when used as illustrated, did appear to provide a means for "unstacking" stacked routes.

Our efforts to develop discrete VNAV routes was not limited to stacked routes but was an extension of the previous work done by Champlain Technology Industries in their terminal route design activities and other analyses by SRDS and NAFEC prior to and during simulation planning. While the capabilities of VNAV were recognized by us as potentially beneficial to the ATC system user, it was the consensus that the only unique, potentially advantageous, property of VNAV related to terminal airspace is the capability to define the vertical dimension of a path through space as though the path were described with an infinite number of altitude checkpoints.

A number of applications of VNAV to route structure design were considered. During these studies we were advised not to consider either the original or modified route structures as a constraint to the development of routes discrete to VNAV operations. We were in effect given complete freedom to invent any route that potentially would exploit the use of VNAV. The effort was aimed at defining a route or series of routes that, by their nature, could be used exclusively by VNAV equipped flights rather than routes that could be used by both RNAV and VNAV equipped traffic. This approach was taken to identify any airspace design application based on the unique capabilities of VNAV.

As a result of this effort, no VNAV-only charted route structure or individual routes were developed. It was our opinion that no need or advantage could be found in airspace design for discrete VNAV-only routes. It was concluded that a good terminal route structure would accommodate both RNAV and VNAV traffic.

H.5 RNAV ATC APPLICATION

The following represents our opinions as to the advantages, disadvantages and limitations to the use of RNAV in terminal ATC operations. These opinions presuppose that certain conditions relative to avionics equipment and pilot performance are met.

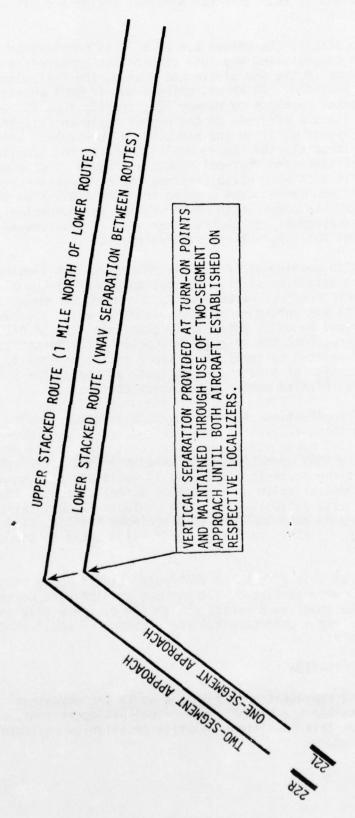


Figure H.1 EXAMPLE OF STACKED ROUTE/TWO-SEGMENT

APPROACH APPLICATION

- Conditions (1) RNAV turn anticipation (both automatic and manual) will be performed in such a manner that the approximate ground track can be anticipated by the controller. This assumes that both automatic and manual turn anticipation procedures be standarized to minimize the requirement for radar vectoring to compensate for turns that deviate from the expected ground path.
- (2) All RNAV flights will be capable of flying at least a ten-mile parallel offset.
- (3) All turns to and from offsets will be accomplished using a common departure angle from the parent/offset route unless otherwise specified by the controller.
- (4) All RNAV equipment will permit assignment of "direct to" waypoint clearances and compliance with such instructions will result in the aircraft flying a direct path to the assigned waypoint upon completion of any required turn.
- (5) Offsets may be cancelled prior to the time the aircraft achieves the assigned parallel offset distance from the parent track.
- (6) Flights on a "direct to" clearance can be assigned an offset parallel to the direct flight path.
 - (7) All RNAV functions simulated will be available.
- (8) Charted SIDs and STARs with altitude restrictions are published and that such SIDs and STARs are so designed as to provide flexibility for spacing and sequencing of traffic equivalent to that required in a radar vector operational environment.

While condition (6) was not met by the DSF targets, condition (6) is believed to be realistic and available in some, if not all, RNAV systems, and our appraisal of the use of RNAV in terminal operations assumes that all of the preceding conditions would be met. This appraisal, based on both experience in the NAFEC simulation and in current field facility terminal air traffic control, is organized by specific areas of potential ATC impact and summarized in a general appraisal statement.

 $\frac{\text{Controller Radio Communications}}{\text{in radio communications for the feeder controllers in a 100 percent RNAV/VNAV operation.}}$

There would be a greater reduction of radio communications in a 100 percent RNAV/VNAV departure operation. However, there was minor reduction on the final control positions.

The reason for the reduced communications is that each SID departure or STAR arrival has a predetermined route to fly with all the altitude restrictions on it. The controller need only monitor the flight and make occasional RNAV maneuvers to accommodate overtaking or merging traffic situations.

The Role of RNAV Maneuvers vs. Phase of Flight - RNAV Limitations:
Because of differences in navigational error and turn anticipation, separation standards in critical areas such as base leg or turns to final, can, and often do, diminish separation to less than prescribed minimums. Whereas radar vectors, being more precise when employed properly, can be benefically substituted in these same areas to provide more exact required separation.

Impact of Mixes of RNAV/Non-RNAV Operations - Both the departure and feeder controllers found no appreciable differences between mixed traffic situations. It was just as easy to assign a heading off a fix or off the runway, as it was to issue an RNAV maneuver. However, the final controller's workload increases if he incorporates RNAV instructions for the RNAV aircraft and vectors to the non-RNAV aircraft.

Impact of Mixes of VNAV/RNAV Operations - No differences noted.

System Capacity - RNAV will not affect traffic capacity in the terminal area, in that it is possible to run a three-mile final with RNAV or with radar vectors. Present day standards require three-mile separation and this can be accomplished with or without RNAV.

General Appraisal Statement - It is our opinion that RNAV/VNAV procedures may well be applied in the terminal area to provide a safe, orderly and expeditious flow of air traffic. We feel that RNAV routes with altitude restrictions to which VNAV usage can be applied, as pilots may desire, should be established at as many busy terminal areas as may be deemed beneficial by FAA and user groups. We feel that these routes should initially co-exist with established airspace allocations to the maximum extent possible to insure little or no adverse impact on present day operations. We also feel strongly that radar vector procedures should be employed at the discretion of the controller in critical areas where RNAV/VNAV may not be as precise as radar. We believe that RNAV/VNAV will be beneficial to the user in that properly established routes can and will reduce flying miles and time. It will be beneficial to the user and more particularly to the controller under all traffic densities, in that the controller will normally have to provide fewer control instructions, subsequently allowing him to perform duties which may include handling more aircraft per sector, combining sectors or portions of sectors, and freeing him to provide both essential and additional services at a reasonable level.

It is felt that RNAV/VNAV could work well in a high density terminal area. RNAV STARs should be made for the entire route of flight including the final approach. RNAV/VNAV could be used to set up straight-in approaches to satellite airports that have no navigational aids.

VNAV ATC APPLICATION

The following represents our opinions as to the advantages, disadvantages and limitations to the use of VNAV in terminal operations.

During the pre-data collection and data collection periods of the simulation, little or no operational use was made of the functions peculiar to

VNAV. While those functions common to both RNAV and VNAV were used to a major degree, there were no occasions found for the use of VNAV as a control tool. In addition, it was our opinion that while VNAV capabilities have the potential for providing advantages to the user in the manner in which climbs and descents can be accomplished, these potential advantages do not require the establishment of exclusively VNAV routes. Such advantages are available in a well structured RNAV terminal route system.

When VNAV vertical separation is being applied between aircraft on crossing courses, vertical separation criteria are predicated upon mathematical curves, which increase separation requirements proportionately with any change of the course angle convergence or divergence, and any increase of degree of vertical path angle. These VNAV separation standards can only increase the present day minimums which dictate one thousand feet vertical separation between IFR aircraft and which can more efficiently and effectively be applied through step-up or step-down procedures in use today. Also due to the complex nature of the mathematical curve, a controller could very rarely move an aircraft laterally from an established track and still insure separation from a crossing course. Impromptu courses would be out of the question, as altitude separation requirements could not possibly be computed by the controller.

When VNAV vertical separation is being applied between aircraft in a parallel climb or descent on the same lateral track, separation criteria in accordance with the Vertical Separation Requirements curves is increased over criteria which can be applied through the use of today's step-up or step-down procedures. If an aircraft is moved laterally from the main track, separation from another aircraft, which had previously been separated by the minimum criteria, either above or below, immediately ceases to exist due to the proportionate vertical separation increase caused by course angle divergence in the mathematical curve. Impromptu courses would again be out of the question, as controllers could not compute descent angles or altitude separation requirements.

VNAV could be used as a useful tool to pilots as a more economical means of climb or descent.

RNAV/VNAV could be used to set up straight-in approaches to satellite airports that have no navigational aids. It could also be used to set up an artifical glide path to aid in VOR approaches, which would possibly lower minimums.

The foregoing appraisal represents the expressed opinions of the following named field controllers who participated in the NAFEC RNAV/VNAV simulation and is based on our experience in the simulation and our judgments as current field facility air traffic control specialists.

[Signed]

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